



NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

REFERENCE

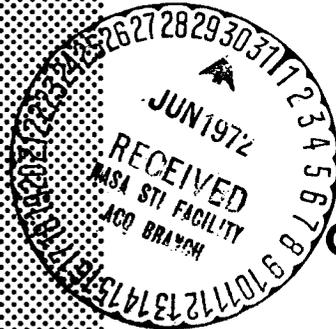
SKYLAB FLIGHT PLAN

SL-1/SL-2, SL-3, AND SL-4
(APRIL 30, 1973 SL-1 LAUNCH)

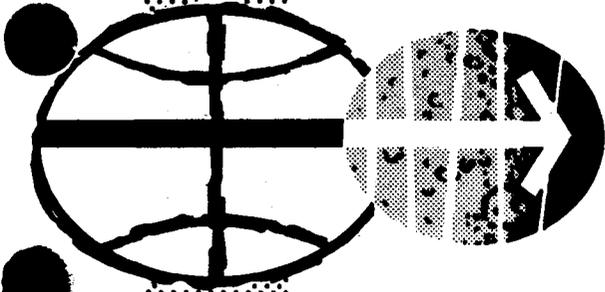
(NASA-TM-X-68459) REFERENCE SKYLAB FLIGHT
PLAN, SL-1/SL-2, SL-3, AND SL-4 (APRIL 30,
1973 SL-1 LAUNCH) C. Hassler, et al (NASA)
1 May 1972 312 p

N72-73394

Unclas
00/99 32497



PREPARED BY
FLIGHT PLANNING BRANCH
CREW PROCEDURES DIVISION



MANNED SPACECRAFT CENTER
HOUSTON, TEXAS
MAY 1, 1972

Rpt-67599

FLAG SHEET

SKYLAB REFERENCE FLIGHT PLAN

This flag sheet identifies changes to the mission requirements that affect the flight plan.

1. Scheduling of blood sampling (M110) and urine sampling (M071) has been revised by CCB action as follows: (not incorporated in this issue).
 - a) Blood samples are taken as follows:
SL-2: day 2,4,12, and 27.
SL-3 and 4: day 2,4,12,19,29,37,47, and 55.
 - b) Full urine samples are required on the first seven (7) and last seven (7) days of each mission.
 - c) Full urine samples are required on day before, day of, and day after blood sampling.
 - d) Half (1/2) urine samples shall not be taken on consecutive days.
 - e) If two (2) UCTA's per crewman are used on the first day, half (1/2) samples shall be taken from each UCTA.
 - f) Half (1/2) samples of urine should be taken on the following days:
SL-2: day 8,10,14,16,18, and 20.
SL-3 and 4: day 8,10,14,16,21,23,31,33,35,43,45, and 49.
2. A period of 30 minutes is set aside each day for each crewman for personal exercise except when the crewman is subject for M509, EVA, or M171. (incorporated in this issue)
3. T025, "Coronagraph Contamination Measurement", is included although the requirements are in the process of being redefined. In this issue, 5 viewing periods are scheduled for SL-1/SL-2 and 10 viewing periods are scheduled for SL-3. SL-4 viewing periods are TBD.
4. The Skylab Educational Program requires video participation by the crew. The exact nature of this program has not yet been defined. For planning purposes, one crew man-hour per week has been allocated to meet the educational TV requirement. (incorporated in this issue)
5. A number of investigators selected from a national contest of Skylab Student Projects will be performed. Until specific investigations are assigned 1 1/2 crew man-hours per week have been allocated for flight planning purposes. (incorporated in this issue)

6. The ozone photography portion of S063 "UV Airglow Horizon Photography" is scheduled during 2 of the 15 Z-LV passes on SL-1/SL-2. Due to recent changes in the stowage list, this experiment will be deleted from the next SL-1/SL-2 timeline.
7. Operational radiation measurements have been included pending approval of the DTO defining the required measurements.
8. T027/S073 "Contamination Measurement and Gegenschein/Zodiacal Light" has been scheduled against the proposed changes to the experiment observing programs.
9. The schedule for M509 "Astronaut Maneuvering Equipment" in the SL-3 timeline will be modified in the June 1 update to facilitate atmosphere management considerations.

NOTE
Refer to page 1-9 and 1-10 for a listing of
experiment constraints which were not met in
this issue of the SL-1/SL-2 and SL-3 timelines.

CONTENTS

	<u>Page</u>
Introduction	ii
Abbreviations	iii
 <u>SECTION 1 - FLIGHT PLAN NOTES</u>	
Mission Description	1-1
Crew Scheduling	1-16
Activation/Deactivation	1-25
Communications and Instrumentation	1-28
Guidance, Navigation, and Control	1-37
Environmental Control	1-47
Electrical Power	1-49
Systems Housekeeping	1-51
Habitability	1-56
Venting	1-58
Experiments	1-68
Photography	1-116
Stowage	1-117
Updates	1-120
 <u>SECTION 2 - SUMMARY TIMELINES</u>	
SL-1/2 Summary Timeline	2-1
SL-3 Summary Timeline	2-30
 <u>SECTION 3 - DETAILED TIMELINE - SL-1/2 AND SL-3</u>	
SL-1 Activation	3-1
SL-1/2 Rendezvous	3-3
SL-1/2 Activation	3-8
SL-1/2 Deactivation and Entry	3-11
SL-3 Rendezvous	3-23
SL-3 Activation	3-26
SL-3 Decativation and Entry	3-32
 <u>SECTION 4 - EXPERIMENT AND TEST OBJECTIVES</u>	
Experiment Data Sheets	4-1

INTRODUCTION

This flight plan schedules the Skylab mission operations and crew activities to fulfill, when possible, the objectives defined in the Mission Requirements Document, 1-MRD-001D, volume 1, dated November 1, 1971. Detailed timelines are presented for the SL-1/2 and SL-3 activation and deactivation sequence while summary timelines are presented for SL-1/2 and SL-3.

The trajectory parameters used for this flight plan are for the April 30, 1973, Workshop Launch and were obtained from the Skylab Reference Trajectory to be published by the MSC Mission Planning and Analysis Division.

NOMENCLATURE

AA	Aerosol Analyzer
ACT	Activation
AE	Activity Element
AGO	Santiago
AL	Airlock
ALSA	Astronaut Life Support Assembly
AM	Airlock Module
AMRV	Astronaut Maneuvering Research Vehicle
AMU	Astronaut Maneuvering Unit
AOS	Acquisition of Signal
APCS	Attitude and Pointing Control System
ASC	Ascension
ASMU	Automatically Stabilized Maneuvering Unit
A-SOL SAL	Anti-Solar Scientific Airlock
ATM	Apollo Telescope Mount
ATMDC	Apollo Telescope Mount Digital Computer
ATT	Attitude
AZ	Azimuth
B(Beta)	Minimum angle between the Earth-Sun Line and the Vehicle Orbital Plane
B&W	Black and White
BDA	Bermuda
BET	Best Estimate of Trajectory
BMAG	Body-Mounted Attitude Gyro
BMMD	Body Mass Measurement Device
BP	Biopack
BPMS	Blood Pressure Measuring System
BPSS	Biopack Subsystem
CB	Circuit Breaker
CCU	Crew Communication Umbilical
CCW	Counter Clockwise
C&D	Control and Display
CDR	Commander
CM	Command Module
CMN	Crewman
CMC	Command Module Computer
CMD	Command
CMG	Control Moment Gyro
OMGS	Control Moment Gyro Subsystem
COAS	Crewman Optical Alignment Sight
CRO	Carnarvon
CS	Crew Station
CSM	Command and Service Module
CST	Central Standard Time
CW	Clockwise
C&W	Caution & Warning
CYI	Grand Canary Island

D	Dump
DA	Deployment Assembly
DAC	Data Acquisition Camera
DAP	Digital Auto Pilot
DAS	Digital Address System
DCS	Digital Command System
DOD	Department of Defense
DOY	Day of year
DSE	Data Storage Equipment
D/T	Delayed Time
DTO	Detailed Test Objective
EA	Experiment Assembly
ECS	Environmental Control System
ED	Educational
EMOD	Erasable Memory Dump
EOH	Experiment Operations Handbook
EPCS	Experiment Pointing Control Subsystem
EPS	Electrical Power System
ER	Earth Resources
ERD	Experiment Requirements Document
EREP	Earth Resources Experiment Package
ERG	Ergometer
ESS	Experiment Support System
ETC	Earth Terrain Camera
EVA	Extravehicular Activity
EXP	Experiment
EXT	Exterior
FAC	Facility
FAS	Fixed Airlock Shroud
FCMU	Foot Controlled Maneuvering Unit
FMS	Force Measuring System
FMSC	Film Magazine Stowage Container
FMU	Force Measuring Unit
FO	Functional Objective
FOV	Field of View
FWD	Forward
G&N	Guidance and Control
GDC	Gyro Display Coupler
GDS	Goldstone
GET	Ground Elapsed Time
GG	Gravity Gradient
GMT	Greenwich Mean Time
GND	Ground
GSE	Ground Support Equipment
GSFC	Goddard Space Flight Center
GWM	Guam
HAO	High Altitude Observatory
HAW	Hawaii
HBR	High Bit Rate

HCO	Harvard College Observatory
HD	Highly Desirable
HHMU	Hand Held Maneuvering Unit
HOU	Houston
HPN	Heavy Primary Nuclei
HQ	Headquarters
HRS	Hours
HSK	Honeysuckle Creek
HSS	Habitability Support System
IG	Inner Gimbal
IMC	Image Motion Control
IMD	Inhibit Momentum Dump
IMU	Inertial Measuring Unit
INIT	Initiate
IPS	Inches Per Second
IR	Infrared
IRS	Inertial Reference System
ISS	Inertial Subsystem
IU	Instrument Unit
IVA	Intervehicular Activity
KSC	Kennedy Space Center
LBNP	Lower Body Negative Pressure
LBR	Low Bit Rate
LCG	Liquid Cooled Garment
LES	Launch Escape System
LIMS	Limb Motion Sensor
LO	Lift-off
LOS	Loss of Signal
LV	Launch Vehicle
LVMS	Leg Volume Measuring System
M	Mandatory
MA	Metabolic Activity
MAG	Magazine
MD	Mission Day
MAD	Madrid
MCC	Mission Control Center
MCC-H	Mission Control Center - Houston
MCSS	Microscopic Camera Subsystem
MDA	Multiple Docking Adapter
MD/CSU	Motor Drive/Cassette Support Unit
MIL	Merritt Island
MIN	Minute(s)
MOL	Molecular
MPF	Multispectral Photographic Facility
MRD	Mission Requirements Document
MSC	Manned Spacecraft Center

MSFC	Marshall Space Flight Center
MSFEB	Manned Space Flight Experiment Board
MSFN	Manned Space Flight Network
MSS	Motion Sickness Susceptibility
MU	Measuring Unit
N2	Nitrogen
N/A	Not Applicable
NAV	Navigation
NC1	Phase Adjust Maneuver
NCC	Corrective Combination Maneuver
NEG	Negative
NM	Nautical Mile
NOAA	National Oceanographic and Atmospheric Administration
NPV	Non-Propulsive Vent
NRL	Naval Research Laboratory
NSR	Coelliptic Maneuver
O	Observer
O2	Oxygen
OA	Orbital Assembly
OBS	Operational Biomedical Harness
ODB	Operational Data Book
OG	Outer Gimbal
OGI	Occulogyral Illusion
OMSF	Office of Manned Space Flight
OPS	Operations
OTG	Otolith Test Goggles
OWS	Orbital Workshop
PCM	Pulse Code Modulation
PCU	Pressure Control Unit
PERF	Performance
PGA	Pressure Garment Assembly
PGU	Propulsion Gas Umbilical
PH	Personal Hygiene
PLN	Planning
PLT	Pilot
PMT	Photomultiplier Tube
PNL	Panel
POTS	Precision Optical Tracking System
PRF	Pulse Repetition Frequency
PROG	Program
PS	Payload Shroud
PSS	Propellant Supply Subsystem
PT	Physical Training
PWR	Power
QTY	Quantity

RAD	Radiation
RCDR	Recorder
RCS	Reaction Control System
REF	Refrigerator
REV	Revolution
RLS	Rotating Litter Chair
RPM	Revolutions Per Minute
R&R	Rest and Recreation
RSS or RS	Refrigeration Sybssystem
RT	Real Time
S	Subject
SAA	South Atlantic Anomaly
SAL	Scientific Airlock
SC	Spacecraft
SCT	Scanning Telescope
S/HK	Systems Housekeeping
SHT	Shutter
SI	Solar Inertial
SIA	Speaker Intercom Assembly
SL	Skylab
SLOH	Skylab Operations Handbook
SM	Service Module
SMMD	Specimen Mass Measurement Device
SMRD	Spin Motor Rotation Detector
SOL-SAL	Solar Scientific Airlock
SOP	Secondary Oxygen Pack
SPAN	Solar Particle Alert Network
SPEC	Specification(s)
sps	samples per second
SPS	Service Propulsion System
SPT	Scientist Pilot
S-SAL	Solar Scientific Airlock
ST	Stow
STS	Structural Transition Section
STU	Student
SU	Set Up
SWS	Saturn Workshop
SXT	Sextant
SYS	System
TACS	Thruster Attitude Control Subsystem
TBD	To Be Determined
TBS	To Be Supplied
TCS	Thermal Control System
TEX	Texas
TGT	Target
TM	Telemetry
TPI	Terminal Phase Initiation
TPF	Terminal Phase Final
TPM	Terminal Phase Midcourse
TPTR	Teleprinter
TV	Television

UHF	Ultra-High Frequency
UMB	Umbilical
UNATTD	Unattended
USB	Unified S-Band
UV	Ultraviolet
VAN	Vanguard
VCG	Vectrocardiogram
VHF	Very High Frequency
VVOH	Vacuum Valve Operating Handle
VX	Voice
VTS	Viewfinder Tracking System
W/O	Without
WMC	Waste Management Compartment
WMS	Waste Management System
WR & WRDM XFER	Wardroom Transfer
XFER	Transfer
X-IOP/Z	Solar Inertial Attitude
X-REA	X-ray Event Analyzer
X-RT	X-ray Telescope
X-PONDER	Transponder
XUV	Extreme Ultraviolet
Z-LV(E)	Earth Pointing Attitude
Z-LV(R)	Rendezvous Pointing Attitude

SECTION 1
FLIGHT PLAN NOTES

Flight Plan Notes Mission Description

1. The Flight Plan Notes have been included to provide:
 - . Information relative to system hardware operations that may or may not be called out in the timeline.
 - . Groundrules for scheduling experiments and other activities.
 - . Miscellaneous information pertinent to an understanding of the timeline.
 - . Experiment scheduling accomplishments.
 - . Constraints violated in scheduling.
2. Table 1-1 presents Skylab mission oriented parameters and event times.
3. Figures 1-1 and 1-2 illustrate the traces of window centerlines on the celestial sphere for the duration of the Skylab missions for the CSM and OA, respectively (SL-4, TBS). Docked solar inertial attitude is used. The CSM traces are from nominal eye positions to the center of the window, while the OA traces represent the normal through the center of the window.
4. Figure 1-3 presents the beta angle history and sunlight time for Skylab.
5. Table 1-2 presents known constraints violated in scheduling experiments for SL-2, SL-3 and SL-4(TBS).
6. Table 1-3 presents the experiments with functional objectives allocated per day for the SL-2, SL-3 and SL-4 (TBS) missions.
7. Figure 1-4 is a man-hour allocation summary for SL-2, SL-3 and SL-4 (TBS).

Flight Plan Notes
Mission Description

PARAMETER	MISSION			
	SL-1	SL-2	SL-3	SL-4
CSM Number	--	116	117	188
Launch Vehicle	2 Stage Saturn V	Saturn IB	Saturn IB	Saturn IB
Launch	30 Apr 73	1 May 73	30 July 73	31 Oct 73
Day of Year (DOY)	120	121	211	304
(GMT)	17 ^h 30 ^m 00 ^s	16 ^h 59 ^m 33.6 ^s	04 ^h 43 ^m 57.9 ^s	14 ^h 56 ^m 55.1 ^s
Insertion ha/hp/nm	236.9/235.8	120.0/80.9	120.0/80.9	120.0/80.9
Inclination	50.0	50.0	50.0	50.0
CSM/OWS Docking (GET)	--	08 ^h 00 ^m 00 ^s	08 ^h 40 ^m 00 ^s	08 ^h 40 ^m 00 ^s
Orbital Elements after Docking				
ha/hp, nm	234.6/234.4	234.6/234.4	235.7/233.4	237.0/230.2
Period, min	93.3	93.3	93.3	93.3
Trim #1		122 15:20:13	211 17:00:33	TBD
Trim #2 (DOY) (GMT)		129 12:08:42.4	229 12:25:06.3	318 15:32:46
Trim #3 (DOY) (GMT)		144 12:02:00.6	249 12:50:36.5	338 15:46:41
Trim #4 (DOY) (GMT)			263 12:39:00.0	
Activation Complete				
SWS (DOY) (GMT)		122 21:00	212 07:00	TBD
ATM (DOY) (GMT)		124 01:30	213 04:24	TBD
EVA (DOY) (GMT)		146 10:00	213 22:00	TBD
			237 14:45	TBD
			264 11:00	TBD

Table 1-1. Skylab Mission Parameters (Sheet 1 of 2)

Flight Plan Notes
Mission Description

PARAMETER	MISSION			
	SL-1	SL-2	SL-3	SL-4
Mission Duration	--	28	56	56
Experiment Days	--	21	45	TBD
Storage Period, Days	--	--	61	34
SWS Deactivation (DOY) Begin (GMT)	--	148 07:30	266 11:00	TBD
CSM/SWS (DOY) Undocking (GMT)		149 08:01	267 08:38	360 TBD
Splashdown (DOY) Time (GMT) Latitude (DEG) Longitude (DEG)		149 13:01:43 30°N2 60°W2	267 13:35:57 30°W2 60°W2	360 22:45 ² 20°N2 165°W2
2. Estimated; data not available at time of publication				

Table 1-1. Skylab Mission Parameters (Sheet 2 of 2)

Flight Plan Notes Mission Description

- 1. Center CM Hatch
 - 2. Left CM Side Window
 - 3. Left CM Dock Window
 - 4. Right CM Hatch
 - 5. Right CM Side Window
 - 6. Right CM Dock Window
- (all views from nominal eye position to center of window)

© 0418 GMT, 1 May 1973
 Δ 0614 GMT, 29 May 1973

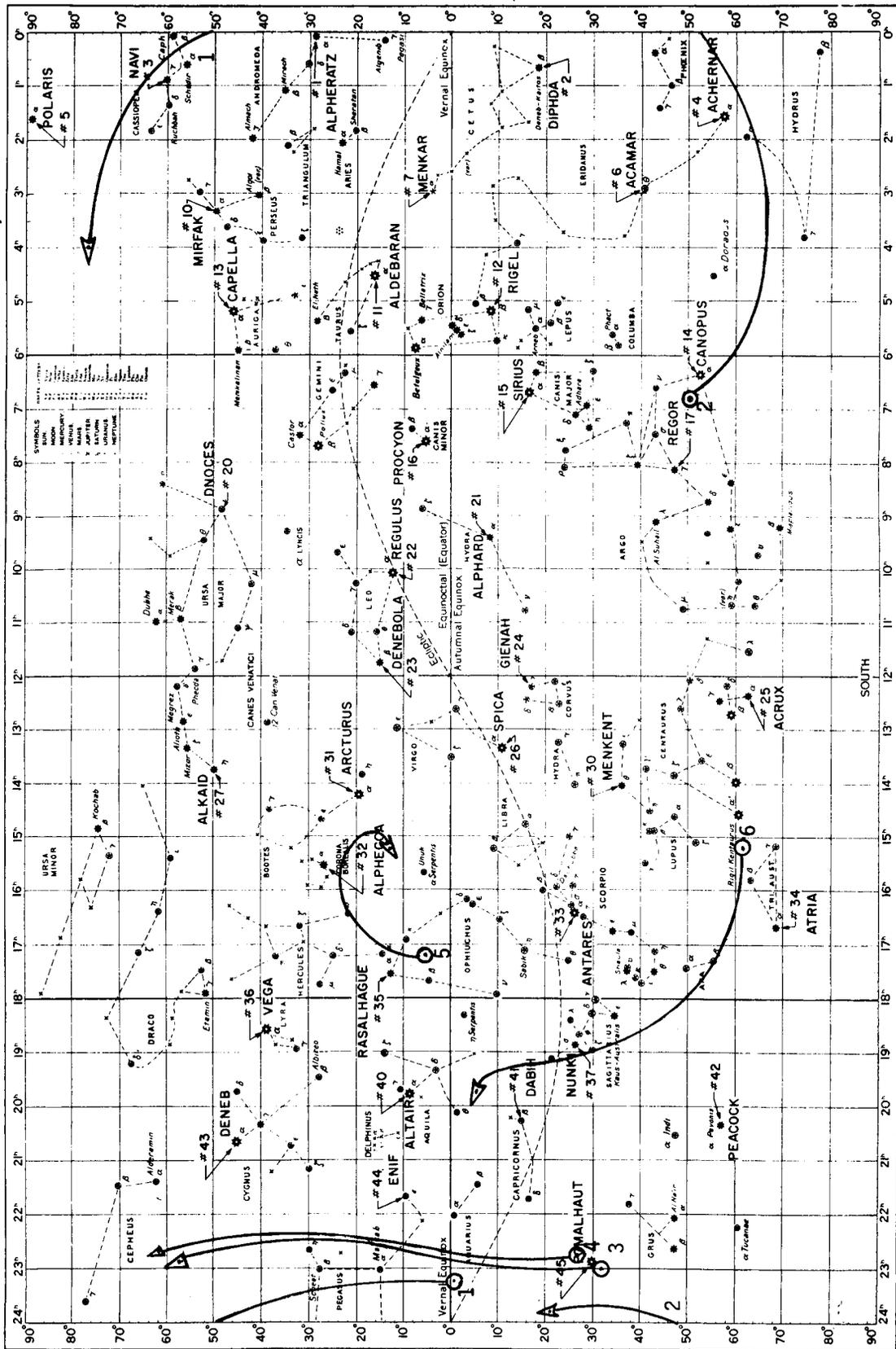


Figure 1-1a. CM Window Traces for SL-2

Flight Plan Notes
Mission Description

© 0418 GMT, 1 May 1973
 ▲ 0614 GMT, 29 May 1973
 (Normal through center of window)

- 7. Structural transition Sect. No. 1
- 8. Structural transition Sect. No. 2
- 9. Structural transition Sect. No. 3
- 10. Structural transition Sect. No. 4
- 11. Solar Airlock
- 12. Anti-solar Airlock
- 13. Multiple Docking Adapter
- 14. Ward Room

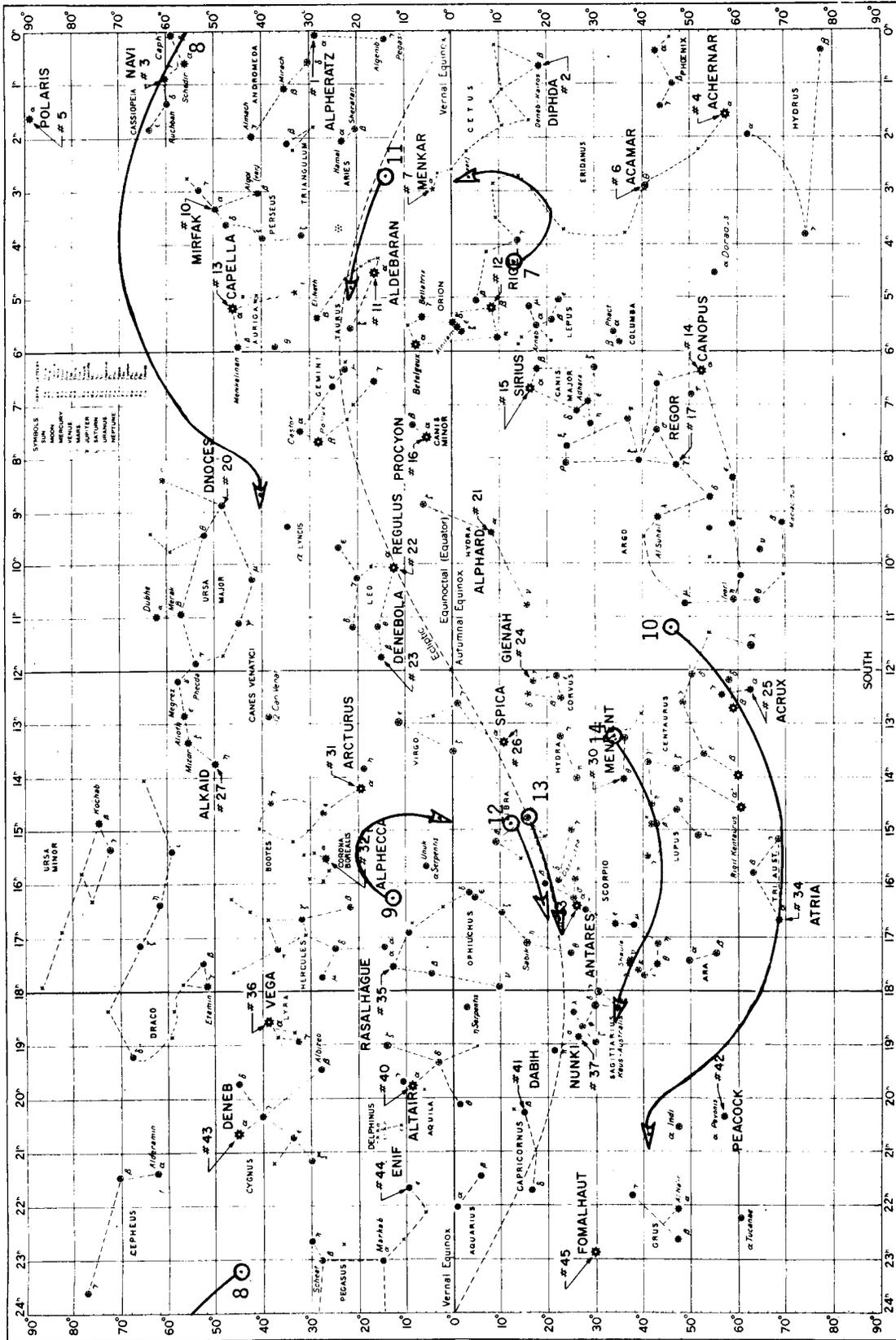


Figure 1-2a. Saturn Workshop Window Traces for SL-2

Flight Plan Notes
Mission Description

- ☉ July 30, 1973
- + Sept. 21, 1973
- (Normal through center of window)
- 7. Structural Transition Section #1
- 8. Structural Transition Section #2
- 9. Structural Transition Section #3
- 10. Structural Transition Section #4
- 11. Solar Airlock
- 12. Anti-Solar Airlock
- 13. Multiple Docking Adapter
- 14. Ward Room

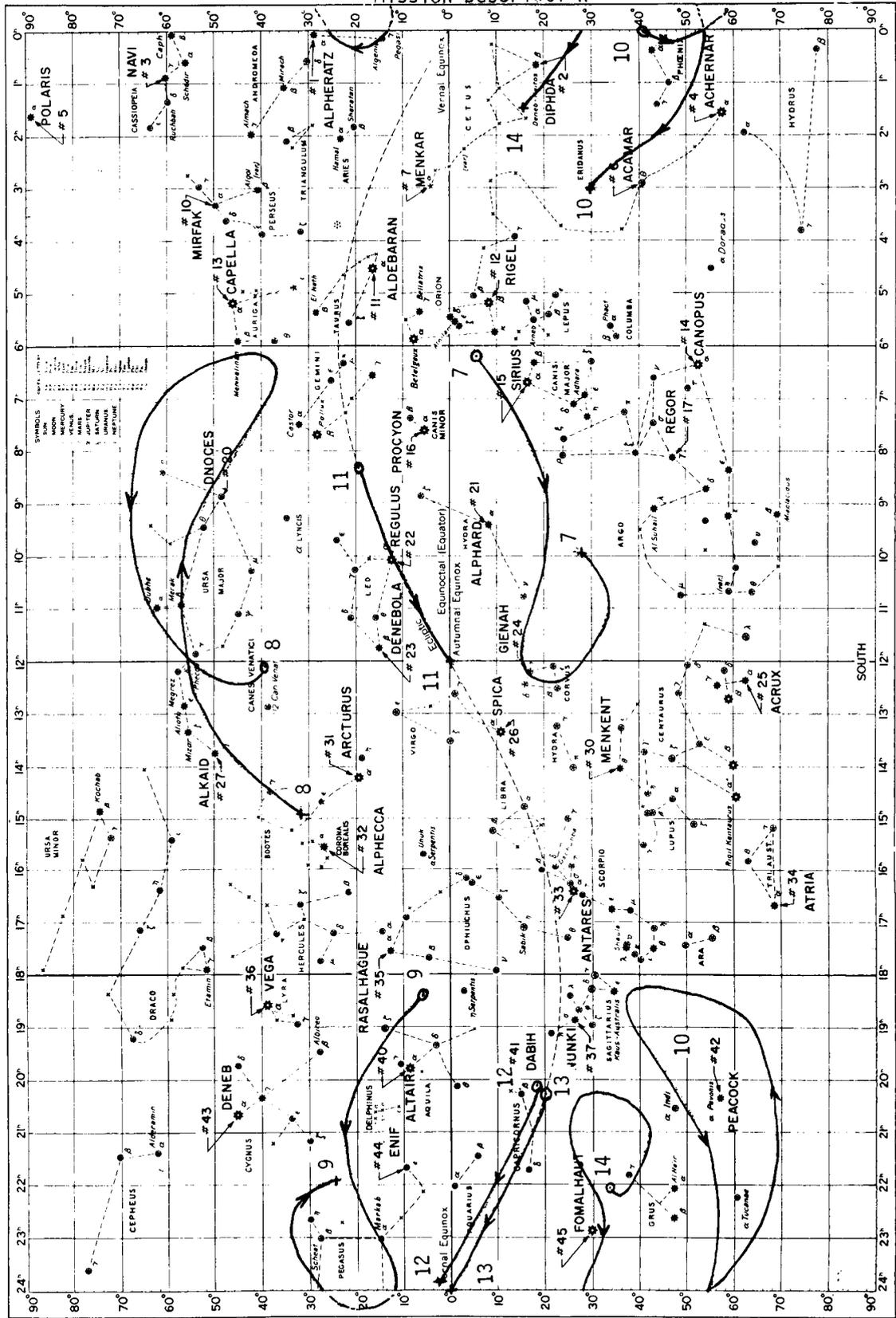


Figure 1-2b. Saturn Workshop Window Traces for SL-3

Flight Plan Notes
Mission Description

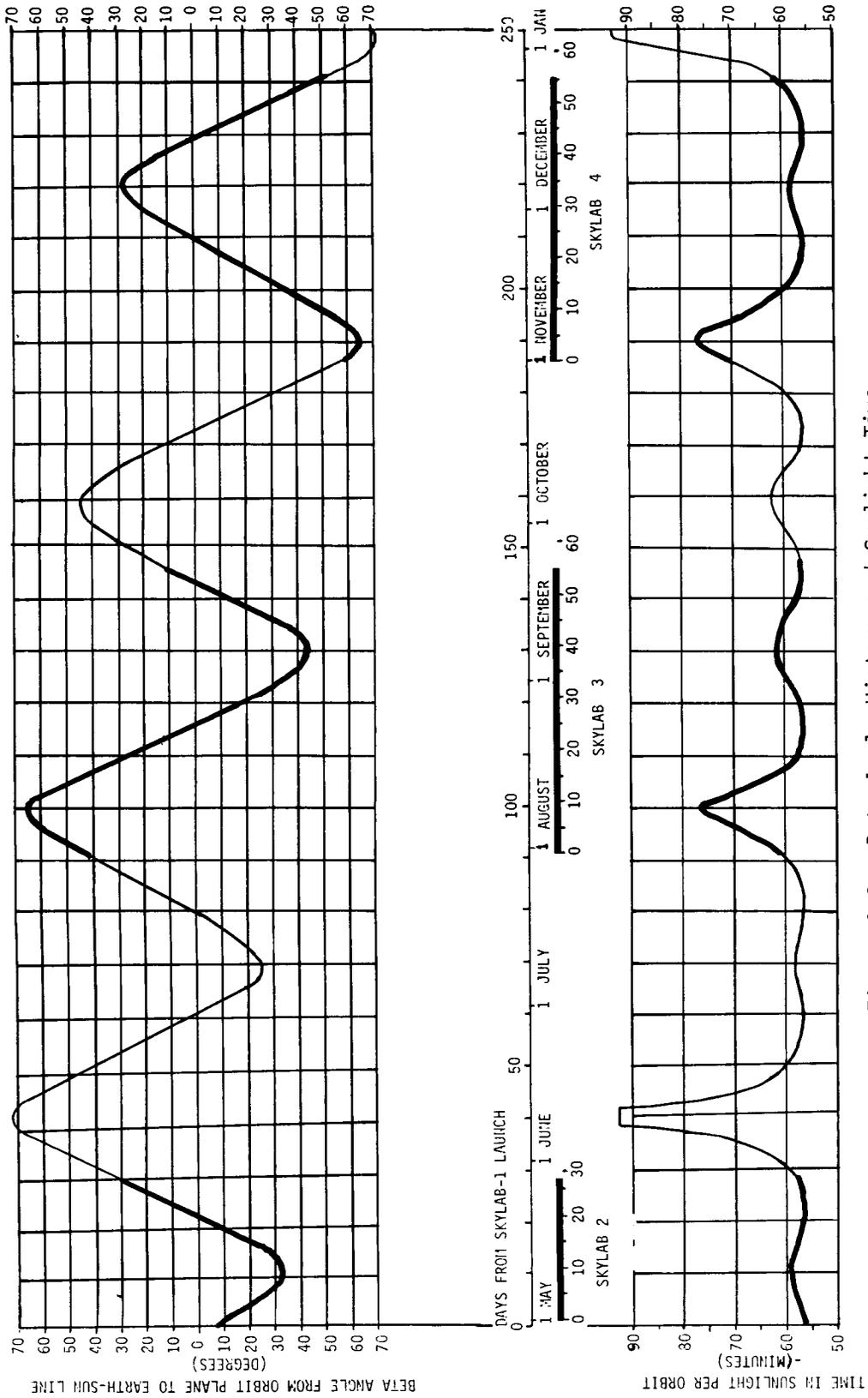


Figure 1-3. Beta Angle History and Sunlight Time

Flight Plan Notes
Mission Description

Day of Year	CMN	Activity	Scheduling Violation	Reason
123	PLT	M092/M171	Wrong time of day	First available time for experiment performance
124	CDR	RAD DTO	Between PLN AND R&R	SAA crossing availability
125	CDR	RAD DTO	After R&R	SAA crossing availability
125	PLT	M092/M171	Wrong Day	EREP pass on day 6
130	PLT	M092/M093	Wrong Day	EREP pass on day 9
132	ALL	Lunch	Greater than 1 hour from normal time	EREP pass and M092/171 in before lunch period
133	PLT	M092/M171	Wrong Day	EREP pass on day 12
135	ALL	Sleep	Early wake up	Early EREP pass
137	PLT	S149	Experiment retracted on Off-Duty-Day	Nine hour warm-up required prior to stow
139	PLT	M092/M171	Wrong Day	EREP pass on day 18
144	ALL	Sleep	Early wake up	Reentry simulation
144 145 146 147 148 149	ALL	Meals, M071 Sleep, Etc.	Early schedule for remainder of mission	Synchronize sleep periods with reentry events and EVA

Table 1-2a. Scheduling Constraints Not Met for SL-2.

Flight Plan Notes
Mission Description

Day of Year	CMN	Activity	Scheduling Violation	Reason
211 212 213	ALL	Sleep, eat, PH, M071, PLN, R&R, etc	Non-standard days and times	Synchronize sleep periods with docking, activation and EVA
212	CDR	M092/M171	Wrong day and time	First available time for performance
225	CDR PLT	Morning PH	None scheduled	M509 performed entire morning
229	ALL	Morning eat, PH, M071	Non-standard sequence	Scheduled around trim burn
235	SPT PLT	M092/M171	Wrong day	Scheduled around EREP pass
237	ALL	Sleep, eat, M071, PLN, R&R, etc	Non-standard day	Scheduled around EREP pass and EVA
238	ALL	Sleep	Early sleep period	Scheduled around EREP pass
239	ALL	Eat, PLN, R&R	Non-standard sequence	Scheduled around EREP pass
243	ALL	PLN, R&R	Non-standard sequence	Scheduled around EREP pass
247	ALL	P50, P51, P52	Scheduled during dinner and PLN	Preparation for re-entry sim.
249	PLT	M092/M171	Wrong day	Scheduled around EREP pass
249	ALL	Morning eat, PH, M071	Non-standard sequence	Scheduled around trim burn
249	ALL	Evening eat, PLN, R&R	Non-standard times	Scheduled around EREP pass
252	PLT	M092/M093	Wrong day	Scheduled around EREP pass

Table 1-2b. Scheduling Constraints Not Met for SL-3

(Sheet 1 of 2)

Flight Plan Notes
Mission Description

Day of Year	CMN	Activity	Scheduling Violation	Reason
258	CDR SPT	Evening eat, PLN, R&R	Non-standard times	Scheduled around EREP pass and M092/M093
259	ALL	Sleep	Crew wakes early	Scheduled around EREP pass
261	ALL	Sleep	Crew wakes early	Scheduled around EREP pass
261	ALL	Evening eat, PLN, PH, M071, R&R	Non-standard times	Scheduled around M092/M171 and early sleep
262	ALL	Sleep	Scheduled early	Scheduled around re-entry sim.
262	ALL	Evening eat, PH, PLN, R&R, M071	Non-standard times	Scheduled around early sleep
263 264 265 266 267	ALL	Sleep, eat, PH, M071, PLN, R&R, etc	Non-standard days and times	Synchronize sleep periods with EVA, undocking and deorbit events
264	SPT	M092/M093	Wrong day	Scheduled around EVA and EREP pass

Table 1-2b. Scheduling Constraints Not Met for SL-3

(Sheet 2 of 2)

Flight Plan Notes
Mission Description

		DAY OF YEAR																												
ACTIVITY	CMN	121	122	123	124	125	126	127	128	129	130	131	132	133	134	135	136	137	138	139	140	141	142	143	144	145	146	147	148	149
Activation	ALL	X	X																											
Off Duty	ALL							X									X								X					
EVA	CDR/SPT																										X			
M071/73	ANY	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X
M074	SPT			1												1													1	
M092/93	CDR									1						1						1								
	SPT									1						1						1								
	PLT										1					1						1								
M092/171	CDR		1			1						1							1							1		1		
	SPT		1			1						1							1					1						
	PLT		1		1								1						1					1			1			
M131	CDR					2									2					2			1							
	SPT			1	2			1					1	2			1			2		1				1				
	PLT			1	2			1					1	2			1			2		1				1				
M133	SPT			1	1	1		1			1		1	1		1				1		1	1	1	1	1	1			
M151	ANY	6	1,2,2	5,1,3	1,6						3,5	1	1,2	2			5						1,6			4				
M172	SPT		1													1												1		
ATM	SPT/ANY			X	X	X		X	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X			
EREP	CDR/PLT			X	X	X		X	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X			
S190B	SPT							1,1	1	1	1	1	1	1		1				1						1				
D008	CDR																					1,2						1,2		
D024	CDR/SPT																										X			
M110	ALL		1	1							1																	1		
M479																SL-4														
M487	PLT/ANY							1								1												1		
M509	CDR/PLT	X		1							2						3									4				
M516	PLT/ANY	1,3												2													5			
M551	PLT/CDR							1	1				1																	
M552	PLT/CDR															1	1													
M553	PLT/CDR												1																	
M554	PLT/CDR																1													
M555	PLT/CDR																	1						1						
S009	CDR/ANY		1																									1		
S015	CDR	2	2	2	1,2	2	2	2	2	2	1																			
S019	PLT/CDR		1,2		3,4,5,6				7,8,9													10	1,1,2							
S020																SL-3														
S063	SPT														1						1									
S073/T027	PLT/ANY		1,2								5	3	4	1-4			7,8,1	7,4,9	3	7	1,2,3	10	2							
S149	PLT/ANY															1	--	1										2		
S183	PLT/CDR										1,2,3										4,5,6		7,8,9,10				1,1,2			
T002	PLT															SL-3														
T003	PLT/ANY	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1
T013																SL-3														
T020																SL-3														
T025	PLT/CDR	X	1												1							1	1							
T027	PLT/ANY					1						1																		
RAD DTO		6	1,1	1	1	1					6											6							6	
MICRO DTO													1,2														1,3			
STUDENT							1		1	1,1			1							1,1		1	1			1	1			
ED T V								1	1		1	1								1		1	1	1						
Reentry Sim																									X					
Deactiv.																												X	X	X
TRIM BURN		X								X																X				

X=EXP. PERFORMANCE
ARABIC NUMERAL=FO PERFORMANCE

Table 1-3a. Experiment Allocation Matrix (SL-2), FO Accomplishment

Flight Plan Notes
Mission Description

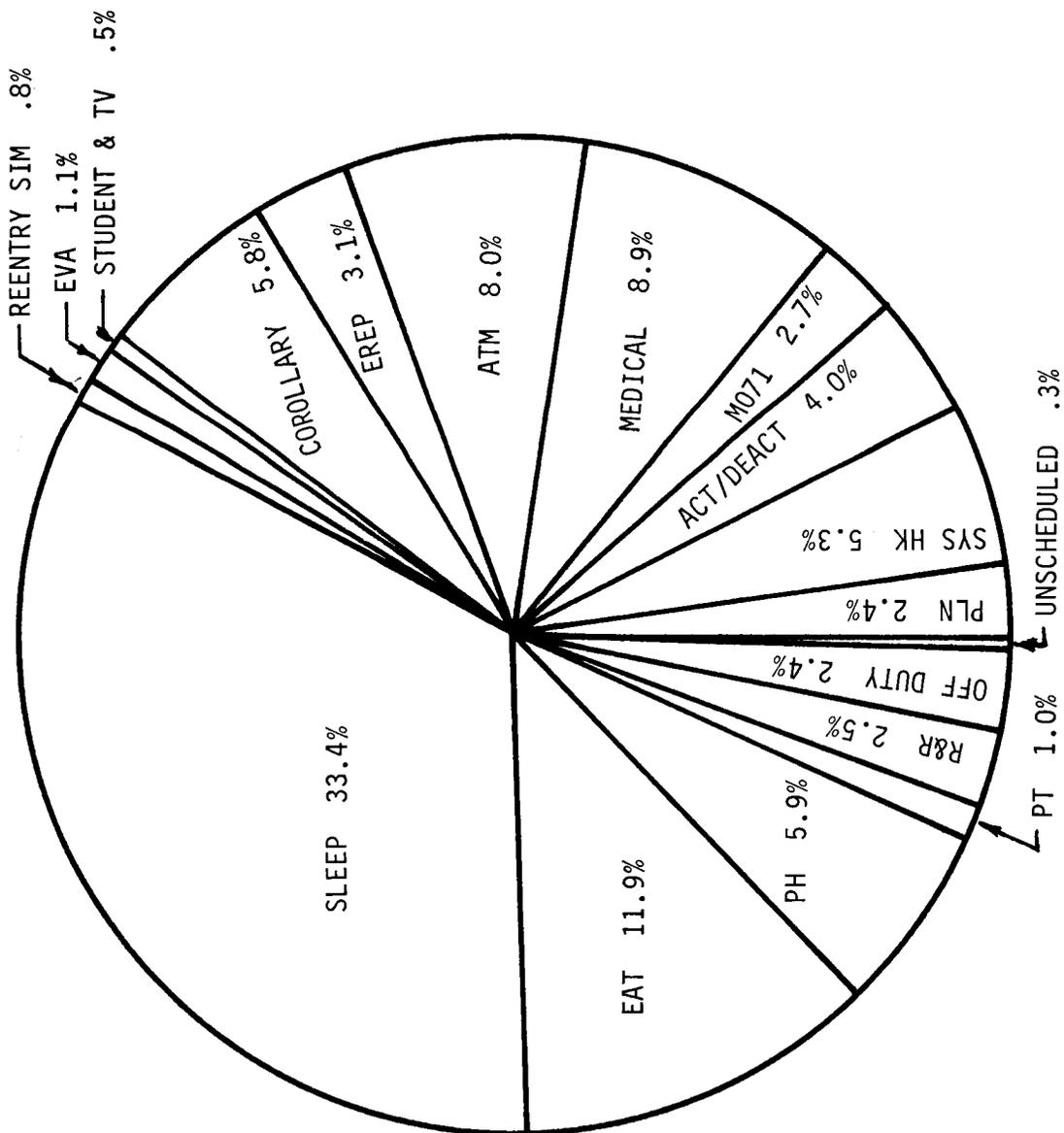


Table 1-4a. Skylab Man-Hour Allocation SL-2

Flight Plan Notes
Mission Description

TBS

Figure 1-4b. Manhour Allocation Summary (SL-3)

Flight Plan Notes
Crew Scheduling

A. Crew Scheduling

1. Crew designations for SL-2, SL-3, and SL-4 are as follows:

Mission	Designation	Prime	Backup
2	CDR	Conrad	Schweickart
	SPT	Kerwin	Musgrave
	PLT	Weitz	McCandless
3	CDR	Bean	Brand
	SPT	Garriott	Lind
	PLT	Lousma	Lenoir
4	CDR	Carr	Brand
	SPT	Gibson	Lind
	PLT	Pogue	Lenoir

2. The nominal CM couch positions are:

Activity	Left	Center	Right
Launch thru docking	CDR	SPT	PLT
Undocking thru splashdown	CDR	SPT	PLT

3. The crew/garment configurations are delineated as follows:

Activity	Pressurized Hard Suit	Suited (Soft Suit)	Partial Suit W/O Helmets and Gloves	Shirt Sleeves
Launch		All		
Docking				All
AM/MDA Activation				All*
OWS Activation				All*
Experiment Ops				All
EVA Ops	2 CMN	1 CMN		
M509-3/IVA	1 CMN			2 CMN
Deactivation				All
Undocking & separation		All		
Entry				All
* w/masks				

Flight Plan Notes
Crew Scheduling

4. The operational biomedical harness will be worn as follows:

Activity	Crewmen
Launch	A11
Rendezvous	A11
Activation	TBD
Orbital Ops	TBD
EVA	2 CMN
Deactivation	TBD
Undocking & Separation	ALL
Deorbit & Entry	A11

The OBS will not be used for daily monitoring but may be used if changes in crew condition are indicated by experimental biomedical data. The OBS is required for subject during M509/IVA hard suit operations.

The operational biomedical measurements include electrocardiogram, respiration rate, heart rate, body temperature, and subject identification.

5. A typical crew day is shown in Figure A-1.

- . All crewmen are scheduled for 8 hours of sleep each day.
- . The crew duty day is between 6 AM and 10 PM CST or CDT.
- . Simultaneous 1 hour eat periods (7 AM, 12 noon and 6 PM Houston time) except for 1 crewman at ATM for noon meal (does not eat at console).
- . Each crewman is scheduled 1.5 hours per day for personal hygiene (30, 15, 15, 30 minute blocks).
- . Four to four and one-half hours each day are scheduled for systems housekeeping.
- . Mission planning and R&R periods prior to crew day off may be used for experiment performance.
- . Day off every 7 days (\pm TBD).
- . Each crewman is allowed 30 minutes per day for exercise. Normally, this is scheduled with 15 minutes of PH just prior to dinner.

Flight Plan Notes
Crew Scheduling

Two crewmen are scheduled for simultaneous mission planning and off-duty periods (1 hour each) at the conclusion of each crew day. The third crewman will delete mission planning and off-duty periods each day to increase ATM viewing time.

6. Crew days off will be scheduled to avoid impacting EREP passes of primary interest and/or the scheduling constraints levied against the major biomedical experiments.
7. No crew activities scheduled on day off except system housekeeping, flares (real time), M071, re-entry simulations, debriefings, and crew actions required for passive experiments (S009, S015).
8. Every third day is a medical day, with all three crewmen performing one series of M092/M093 or M092/M171. If this schedule conflicts with an EREP pass of primary interest, the medical series will be moved 24 hours (before or after the EREP pass).
9. Each crewman is assumed to be proficient in specific system operations and mission assigned experiments. Specialties as scheduled in the timelines are delineated in Table A-1.
10. Extravehicular activity is scheduled for a maximum of 3 hours from egress start to ingress completion. One EVA is scheduled on SL-2 DOY 146 for D024 and ATM. Three ATM EVA's are scheduled on SL-3, DOY 213, 237, 264, and two EVA's are scheduled on SL-4 (TBS).
 - . Two crewmen will be fully suited for each EVA. The third crewman (softsuited) will be located forward of the airlock and will perform monitoring systems housekeeping as required.
 - . Scheduling of EVA while in the South Atlantic Anomaly will be held to a minimum.
 - . Although not a presently identified requirement, scheduling of EVA over the continental United States was attempted to achieve maximum network coverage.
11. One crewman will weigh all food residue after each meal.
12. Orbital Assembly (OA) crew station panel accessibility is shown in Figure A-2. Estimated crew translation times between crew stations are shown in Table A-2.
13. Crew status reports will be voiced to MCC each morning.

Flight Plan Notes
Crew Scheduling

14. The ATM audible flare alarm in the OWS may be inhibited during sleep with concurrence of the ground.
15. Re-entry simulations are scheduled for DOY 144 of SL-2, DOY 247 and 262 of SL-3, and DOY TBS of SL-4 for crew review of detailed entry procedures.
16. At the end of each day, a crew status report will be made. This report will include the status of consumables, unscheduled maintenance, housekeeping checks, anomalies, and stowage.
17. ATM crew changes during sunlight require 5 minutes for change of crew briefings.
18. Whenever possible on SL-2, the SPT will be the observer of biomed experiment activities. On SL-3 and SL-4, this responsibility is equally assigned to all three crewmen.

Flight Plan Notes
Crew Scheduling

Experiment	SL-2			SL-3			SL-4		
	C	S	P	C	S	P	C	S	P
Flight Planning	P			P			P		
AM/MDA/OWS Sys			P			P			P
ATM Exps	P	P	P	P	P	P	P	P	P
ATM Sys		P			P			P	
Med Sys		P			P			P	
M074 (Ca1)		P			P			P	
M133		P			P				
M172 (Ca1)		P			P			P	
M151	P	P	P	P	P	P	P	P	P
D008		P							
D024		P			P				
M479							B		P
M487	B	B	P	B	B	P	B	B	P
M509	B		P	P		B	P		B
M551-M555	B		P						
M516	B	B	P	B	B	P	B	B	P
S009	P	B	B						
S015	P								
S019	B		P	B		P			
S020				B		P	B		P
S063			P			P			P
S071/72						P			
S073/T027	B	B	P	B	B	P	B	B	P
S149	B	B	P	B	B	P	B	B	P
S183	B		P	B		P			
EREP C&D	P		P	P		P	P		P
EREP V/TS	P		P	P		P	P		P
S190B			P			P			P
T002						P			P
T003	B	B	P	B	B	P	B	B	P
T013				B	B	P			
T020				P		B	P		B
T025	B		P	B		P	B		P
T027	B	B	P						

P- Prime; B-Backup; C-CDR; S-SPT; P-Plt

Table A-1a. Crewmen Speciality
Assignment

Flight Plan Notes
Crew Scheduling

CREWMAN POSITION DESIGNATIONS FOR EVA			
CREWMAN	SL-2 EVA 1 NOMINAL ALTERNATE	SL-3 EVA 1 NOMINAL ALTERNATE	SL-3 EVA 2 NOMINAL ALTERNATE
CDR	EV1 EV2	EV1/EV2	EV1
SPT	EV2	EV1	EV2
PLT	EV1	EV2	EV1/EV2
CREWMAN	SL-3 EVA 3 NOMINAL ALTERNATE	SL-4 EVA 1 NOMINAL ALTERNATE	SL-4 EVA 2 NOMINAL ALTERNATE
CDR	EV2	EV1/EV2	EV2
SPT	EV1/EV2	EV2	EV1 EV2
PLT	EV1	EV1	EV1
<p>EV1 - Crewman who egresses first and performs operations at the FAS Work Station (VF)</p> <p>EV2 - Crewman who egresses second, transfers along handrails and performs operations at the Center Work Station (VC) and Sun-end Work Station (VS)</p>			

Table A-1b. Crewmen Specialty Assignment

Flight Plan Notes Crew Scheduling

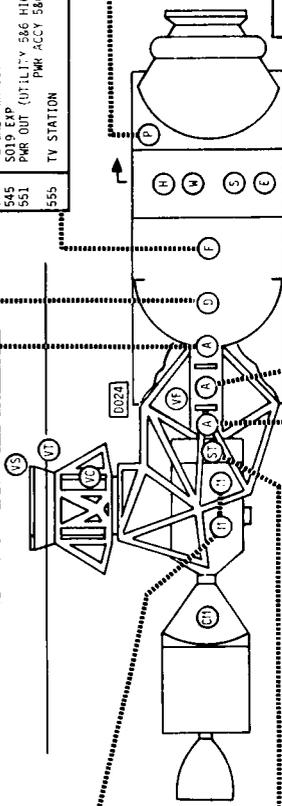
ORBITAL ASSEMBLY CREW STATIONS		PANEL NUMBERS	STORAGE NUMBERS
CM	-COMMAND MODULE	300 SERIES	M100 SERIES
M	-MDA: FMD & AFT COMPT	200 SERIES	S7200 SERIES
ST	-STS	300 SERIES	A300 SERIES
A	-A11: FMO, LOCK & AFT COMPT	-	-
VF	-EVA EXT TO BAY	-	-
DO2A	-EVA EXT TO BAY	-	-
VC	-A11: SUN END TRFR	400 SERIES	I400
VT	-A11: SUN END MS	500 SERIES	E600
VS	-A11: SUN END MS	500 SERIES	F500
E	-A11: SUN END MS	500 SERIES	S900
F	-A11: SUN END MS	500 SERIES	W700
S	-A11: SUN END MS	500 SERIES	H800
M	-A11: SUN END MS	500 SERIES	1000 SERIES
H	-A11: SUN END MS	-	-
P	-A11: SUN END MS	-	-

NO.	PANEL NAME
100	AXIAL HATCH
101	INT LIGHT PNL
102	VENT PNL
103	UTILITY OUT 2
104	M512 EXP
105	S130 EXP
106	WINDOW COVER LATCH
107	WINDOW COVER LATCH
108	S191 EXP 'D & C PNL
109	EREP PNL
110	S009 EXP
111	UTILITY OUT 1
112	M512 HBR VENT VLV
113	M512 BLND VENT VLV
114	HIGH PMR ACCY OUT 1
115	WINDOW HTP PNL
116	RADIAL HATCH
117	WINDOW HTP PNL
118	FIRE SENSOR CTRL PNL
120	DAS
121	RMB1 (UPPER)
128	RMB1 (LOWER)
129	ATM DAC PNL
130	S140 PNL
131	TV STATION
132	UTILITY OUT 3
133	TV STATION
134	UTILITY OUT 3 (MOL SV B)
135	EREP COOL VLV
136	S192 EXP
139	HIGH PMR ACCY OUT 2
140	UTILITY OUT 4
160	ATH-EXT CNST ROLL

NO.	PANEL NAME
200	CB PNL EGS/IBC
201	CB PNL EPS
202	CB PNL
203	CTRL PNL ECS
204	CTRL PNL IAC
205	CTRL PNL IAC
206	CTRL PNL EPS, CBM
207	CTRL PNL LGT, CBM
208	CTRL PNL PROT SPECT
209	CTRL PNL TELEPRINT
210	PEG CTRL SCH
211	BUS DISTR. SCH
212	GHT CLOCK
213	UTILITY OUT 1
214	UTILITY OUT 2
215	CTRL PNL CUSSTATE
216	MOL SV B VENT VLV
217	MOL SV A VENT VLV
218	UTILITY OUT 3 (MOL SV B)
219	BED O ₂ VLV
220	MOL SV A VENT VLV
221	UTILITY OUT 4 (MOL SV A)
222	SYS 1 LCG RES PRESS VLV
223	SYS 2 LCG RES PRESS VLV
224	SYS 2 LCG RES PRESS VLV
225	O ₂ M ₂ CTRL SYS PNL
226	MOL SV A VLV CTRL
227	MOL SV B VLV CTRL
228	MOL SV A ABS DES VLV
229	MOL SV B ABS DES VLV
230	MOL SV A HT EXCH CTRL PNL
231	AM CLNT FLOW CABIN HK VLVs
232	MOL SV B HT EXCH CTRL PNL
233	MOL SV A AIRFLOW VLV PNL
234	ATH-EXT CNST ROLL
235	ATH-EXT CNST ROLL
236	FIRE SENSOR CTRL PNL
237	FIRE SENSOR CTRL PNL
238	MOL SV B AIRFLOW VLV PNL
239	MOL SV A AIRFLOW VLV PNL
241	S15 WINDOW CRANK -Y
242	S15 WINDOW CRANK -Z
243	S15 WINDOW CRANK -Y
244	S15 WINDOW CRANK -Z

NO.	PANEL NAME
400	OMS HATCH, PRESS EQ VLV
401	SIA
402	PHR OUT (UTILITY 182 HIGH PMR ACCY 182)
403	SIA
404	SOLENOID VENT PORT
405	PNEUMATIC VENT PORT

NO.	PANEL NAME
390	CTRL PNL OMS LGT, TCS
391	CTRL PNL OMS LGT, TCS
392	FIRE SENSOR CTRL PNL



NO.	PANEL NAME
300	FMD COMPT PRESS RELIEF VLV

NO.	PANEL NAME
311	FMD HATCH PRESS EQ VLV
312	FMD HATCH HANDLE
313	LOCK PRESS RELIEF VLV
314	EVA SUPPORT CTRL PNL
315	LOCK DEPRESS VLV
316	EVA HATCH
317	TV STATION
321	EXTENDABLE BOOT PNL
323	EVA PNL 2
325	AFT HATCH, PRESS EQ VLV
326	AFT HATCH HANDLE

NO.	PANEL NAME
800	CTRL PNL WRC
801	SIA
803	PHR OUT UTILITY 172 HIGH PMR ACCY 182
818	CB PNL WASTE PROCESSOR
822	M074 EXP
825	WASTE COLLECTION MODULE
831	WATER DUMP VLV

NO.	PANEL NAME
500	WATER SYS PRESS PNL
512	-Z SAL PMR OUT
517	SIA
520	-Z SAL SIA
521	PHR OUT (UTILITY 182 HIGH PMR ACCY 182)
522	T020 EXP
523	M509 EXP
524	T027 EXP
527	M172 ERROROR CTRL PNL
530	FIRE SENSOR CTRL PNL
531	PHR OUT (UTILITY 348 HIGH PMR ACCY 344)
534	T013 EXP
540	-Z SAL SIA
542	EXP RCORR CTRL PNL
543	-Z SAL
546	PHR OUT
548	S013 EXP
551	PHR OUT (UTILITY 586 HIGH PMR ACCY 586)
555	TV STATION

NO.	PANEL NAME
1001	REFR PUMP PKG
	VAC VENT VLV

NO.	PANEL NAME
600	SIA
601	TTY OUT 182
602	FLARE ALERT PNL
603	GHT LOG
611	CB PNL REFRIG
612	CB PNL HT: PMR & UTIL OUT, TV
613	CB PNL HSS TCS
614	CB PNL AT OMS PHR FEEDERS
616	CB PNL LGT, REFRIG, CBM
617	CB PNL LGT, REFRIG, CBM
618	FIRE SENSOR CTRL PNL
619	FIRE SENSOR CTRL PNL
620	M171 LSS H ₂ SUPPLY PNL
621	M092 EXP
622	METABOLIC ANALYZER
623	ESS
624	M171 VAC VENT VLV
625	M171 VAC VENT VLV
626	M507 EXP
627	UTILITY OUT 384
628	UTILITY OUT 384
629	M002 VAC VENT VLV
630	CTRL PNL WARDROB/MHC LGTS
631	PHR OUT (UTILITY 586, HIGH PMR ACCY 182)
632	M131 EXP
633	FIRE SENSOR CTRL PNL
634	TRASH AIRLOCK
638	FIRE SENSOR CTRL PNL
639	FIRE SENSOR CTRL PNL
640	WATER BELT DUSTRETER
642	TV STATION

NO.	PANEL NAME
901	SIA
902	SIA
903	SIA
908	M133 EXP

NO.	PANEL NAME
700	CTRL PNL H ₂ O DUMP/ WINDOW HTR
701	SIA
702	S063 UV SYHC
703	M074 EXP
705	FOOD PREP TABLE
706	H ₂ O DUMP VLV
707	FOOD TRAY 1
708	FOOD TRAY 2
709	FOOD TRAY 3
711	UTILITY OUT WARDROB

Figure A-2. Crew Station Panel Accessibility

Flight Plan Notes
Crew Scheduling

	H	W	S	E	F	D	A	ST	M
Command Module	2.8	2.8	2.8	2.6	2.3	1.8	1.2	.8	.6
Multiple Docking Adapter	2.2	2.2	2.2	2.0	1.7	1.2	.6	.2	
Structural Transition Section	2.0	2.0	2.0	1.8	1.5	1.0	.4		
Airlock Module	1.6	1.6	1.6	1.4	1.1	.6			
Forward Dome	1.0	1.0	1.0	.8	.5				
Forward Compartment	.5	.5	.5	.3					
Experiment Compartment	.2	.2	.2						
Sleep Compartment	.2	.2	.2						
Wardroom	.2								

NOTE:

Times are estimated and noted in minutes. These are unencumbered times, i.e. not suited and not carrying equipment. The time estimate applies for direct translation from Crew Station to Crew Station (Fig A-2) in either direction.

Table A-2. Translation Times (Estimated)

Flight Plan Notes
Activation/Deactivation

- B. Activation/Deactivation (times are Julian Days and GMT - hours and min.)
1. SL-2 CSM Deactivation to quiescent mode is scheduled from 122^d01^h00^m to 122^d16^h07^m and for SL-3 from 211^d13^h25^m to 211^d16^h50^m.
 - a) After fuel cell shutdown (approximately 13 days until cryo depletion) residual H₂ will vent to space and O₂ will be vented to cluster.
 - b) Detailed procedures for CSM systems deactivation to quiescent state are in CSM/SLOH Vol. 2, Sec. 4.5.
 2. CSM quiescent mode configuration and power requirements are TBD pending redefinition.
 3. SL-2 CSM Activation from quiescent mode is initiated at 149^d03^h20^m and for SL-3 at 267^d03^h30^m.
 - a) Detailed procedures for CSM systems activation from quiescent state are in CSM/SLOH Vol. 2, Sec. 4.5.
 4. SWS Activation is scheduled as follows:
SL-2: 122^d13^h00^m to 122^d17^h00^m and 122^d18^h30^m to 122^d21^h00^m.
SL-3: 211^d15^h05^m to 211^d17^h05^m and 212^d02^h50^m to 212^d07^h05^m.
SL-4: TBS
 - a) OWS entry and activation is scheduled with the crew in shirtsleeves. If required, masks will be donned for initial entry into the MDA and OWS.
 - b) Detailed SWS activation procedures are in MDA/AM/OWS SLOH Vol. 2.
 - c) Detailed ATM activation procedures are in ATM SLOH Vol. 2, Sec. 4.
Initial systems activation is performed by IU or ground command and verified by telemetry monitoring.
 5. SWS Deactivation is scheduled as follows:
SL-2: 147^d14^h55^m to 147^d18^h20^m, 148^d07^h30^m to 148^d11^h00^m,
148^d12^h00^m to 148^d18^h00^m, and 149^d03^h50^m to 149^d08^h03^m.
SL-3: 266^d15^h11^m to 266^d14^h00^m, 266^d15^h00^m to 266^d17^h20^m,
266^d19^h30^m to 266^d20^h30^m, 267^d04^h00^m to 267^d07^h00^m, and
267^d07^h30^m to 267^d08^h15^m.
SL-4: TBS

Flight Plan Notes
Activation/Deactivation

- a) Detailed SWS deactivation procedures are in MDA/AM/SLOH Vol.2.
- b) Detailed ATM deactivation procedures are in ATM SLOH Vol. 2,
Sec. 4.

*ATM C&D console will be enabled for ground command.

Flight Plan Notes
Communications and Instrumentation

C. Communications and Instrumentation

1. CSM

- a) CSM high bit rate TM with the power amplifier turned off, will be the normal mode of operation.
- b) S071/S072 experiment memory will be dumped by ground command via the CSM real time TM link.
- c) The CSM tape recorder (DSE) will not normally be used to record routine systems data, with the exception of the periods from launch through activation and deorbit through re-entry.
- d) The DSE is required for D008 operations (initial 2 weeks at approximately 6 to 7 hours/day and for astronaut operations) and electron spectrometer.
- e) All normal cluster voice is via the CSM communications system. The CSM is an integral part of the intercom system and will also be used for downlinking R/T voice on S-band (VHF backup). If VHF is required, antenna switching must be performed by the crew.
- f) Selection of CSM omni antennas is normally under ground control via CSM or AM command systems. Crew selection of antennas is available as backup.
- g) TV ground rules and schedules are presented in note C.2.g, SWS Communications and Instrumentation.

2. SWS

a) Intercom

- Both parallel channels may be used simultaneously and either A or B can be recorded, but only the selected one can be recorded at a time unless the CSM audio stations are bussed together or the AM to CSM call or crew alert command is initiated.
- Experiment recording may be effected by crew at recorder control panels 204, 542, and 617 in the STS, OWS forward compartment and OWS experiment compartment. Voice recording can be initiated at AM tunnel Aft lock compartment panel 316 or at any of 13 speaker intercom boxes which are located as follows:

Flight Plan Notes
Communications and Instrumentation

Location	Crew Station	Panel No.	Location	Crew Station	Panel No.
MDA Forward Compartment	M	102	Wardroom	W	702
MDA Forward Compartment	M	116	WMC	H	801
MDA Aft Compartment	M	131	Sleep Compartment	S	901
OWS Dome	D	401	Sleep Compartment	S	902
OWS Forward Compartment	F	520	Sleep Compartment	S	903
OWS Forward Compartment	F	540			
Experiment Compartment	E	600			
Experiment Compartment	E	627			

- The same voice channel is routed to all AM recorders.
- Selection of channel for recording is made at the AM instrumentation panel.
- The volume control on each SIA is bypassed by the C&WS caution time. The warning tone is applied directly to the SIA speaker except in the sleep mode. The sleep mode is overridden by AM crew alert or crew call.

b) Teleprinter

- Ground command functions will be given priority over teleprinter messages on the data uplink.
- The teleprinter will be controlled by ground command.
- All ground stations have teleprinter uplink capability.
- Detailed information concerning teleprinter message characteristics and specific update formats are found in SKYLAB FLIGHT PLAN UPDATE MESSAGES, MSC, September 27, 1971.

c) STDN (Spacecraft Tracking and Data Network) ground stations considered for this Skylab flight plan are Ascension, Bermuda, Carnarvon, Goldstone, Grand Canary, Guam, Hawaii, Huneysuckle Creek, Madrid, Merritt Island, Texas, Newfoundland for launch support, and the Vanguard off the coast of South America.

d) Cluster Data Systems General Information

Table C-1 presents AM and ATM experiment data recording requirements.

Flight Plan Notes
Communications and Instrumentation

- Table C-2 presents a summary of which systems and functions are controlled by each cluster command system. Also indicated is whether the crew or ground has primary control over the function for normal mission operations.
- Table C-3 presents a summary of the cluster data links by function and frequency for each module data system. Where a function for one module is routed to another module is also indicated (i.e., D/T voice for all cluster modules is down-linked via the AM).
- Table C-4 presents a summary of tape recorder characteristics for each cluster data system, EREP and M133. This information will be used as a baseline for determining recorder utilization to be scheduled in the detailed timeline.

e) Airlock Module Data System

- The systems housekeeping recorder will normally be operated by ground command with crew as backup - as will the transmitter operations. The two experiment data recorders will be operated in the record mode by the crew. Dump of these recorders will be by ground command with crew backup.
- The AM tape recorders are replaceable in flight.
- AM systems housekeeping will normally be recorded 2 to 4 hours per day during the manned phases and nominally dumped once per revolution. During the first 2 weeks of SL-2 and from deactivation to end of the manned period AM recording will be continuous with dump normally once per revolution.
- AM recording should be limited to two hours to allow dump within a nominal site contact.
- The data recorder is prime for voice recording of operational and experiment voice data and is under crew control for voice recording. Voice recording requiring time tags necessitates the simultaneous dump of the PCM track and the voice track to allow time correlation.
- Antenna selection is normally made by ground command with crew backup.

f) ATM Data System

- ATM recorders may not record or dump simultaneously, however, one recorder may dump while the other records.

Flight Plan Notes
Communications and Instrumentation

- The normal operation is for one ATM recorder to be used as the prime recorder, with the second as a backup in case the first fails; However the secondary records will be used one rev per day to ensure reliability.
- All ATM recorder dumps, regardless of length of data recorded, will require 5 minutes.
- The ATM recorder record and dump functions are normally controlled by ground command, but may be controlled by the crew. If no record or stop command is sent, the recorder will automatically return to the record mode 6 minutes following the dump command.
- As no ATM voice recording capability exists, when ATM operations require voice recording, the AM tape recorder must be used.
- Cluster operations requiring recording of attitude information necessitate ATM recording.

g) Television

- Color TV from the cluster (ATM monitor TV is black & white) will be transmitted via the CSM from one of 6 cluster TV connection point, located as follows:
 - MDA, AM, OWS forward dome, OWS forward compartment, OWS experiment compartment, CSM.
- The TV camera is an Apollo color camera with a 30ft. cable.
- EVA coverage is via TV mounted on S073/T027 boom.
- Video recorder capability exists onboard the vehicle to record up to 30 minutes of color TV. The recorder has a 1 to 1 record/dump ratio with dump capability nominally controlled by ground command.
- TV transmission, other than crew setup of camera, and selection of source, is effected by ground control.
- TV coverage of rendezvous and docking, formation flight (undocking), experiment operations, workshop tour, systems housekeeping, EVA, and ATM are scheduled. Additionally educational TV programs are scheduled for recovery via recorded and RT TV.
- R/T TV to the MCC is available only from MIL and GDS. The capability to record TV exists at all ground stations.
- ATM TV is scheduled for one GDS or MIL pass each daylight period when pass is concurrent with ATM operation. Each will consist of cycling of the C&D monitor switch through H -1, H -2, XUV monitor, and WLC positions. Video from passes other than over GDS or MIL are not scheduled and are left to R/T decision.

Flight Plan Notes
Communications and Instrumentation

- h) The ground rules that will be used for scheduling ATM and AM recorder utilization in the detailed timeline are as follows:
- The ATM recorder and the AM data recorder will be run 2 to 4 hours per day for systems housekeeping and as required for experiment support and will be dumped as required (nominally once per revolution).
 - The ATM recorder should be dumped every revolution or as required and not exceed 90 minutes of recording time as recorder capability is limited to 90 minutes.
 - The AM data recorder should be dumped every revolution and not exceed two hours of recording time in order that nominal site contact time is not exceeded.
 - A minimum acceptable site pass for ATM data dump is considered to be 6 minutes above 3 degrees elevation, because the ATM recorder requires 5 minutes for dump and 1 minute for configuration.
 - An additional one minute above AM recorder dump time is required at each site pass for configuration and reconfiguration of the AM recorders.
 - The ATM recorder and AM data recorder are normally under ground command for both record and dump.
 - The AM experiment recorders are normally under crew control for recording and ground command for dump.
 - Each recorder's recording and dump cycle will be shown on the detailed timeline. If only systems data is required, no indication will be shown for each recorder. Where the recording is required for specific experiments, the experiment number is so indicated. Recorder dumps will be indicated by a "D" in the proper recorder column. The block selected for a dump indicates the entire dump opportunity (site pass) while the number in the block indicates the time (in minutes) required to dump the data recorded on that particular recorder. No time will be indicated for the ATM recorder as all ATM recorder dumps require 5 minutes.
 - CSM and MDA Spectrometer recordings will be shown on the detailed timeline.

Flight Plan Notes
Communications and Instrumentation

- i) EREP sensor data will be recorded on the EREP tape recorder. All recording will be 7-1/2 ips except when S192 is required - then recording will be at 60 ips. All EREP recorder utilization is by crew control. Further EREP tape recorder utilization ground rules are TBD.
 - Voice recording in support of EREP is by utilization of the AM recorders. Attitude data recording requires the ATM recorder.
 - EREP sensor data is recorded for return of EREP data tapes at the end of each mission.
- j) M133 data is recorded on one of two available recorders reserved for M133 (one parameter requires AM recorder support). Tapes will be returned for post-mission evaluation.
- k) Clocks located within the orbital assembly module are as follows:
 - DDU - STS
 - DDU - OWS experiment compartment (ceiling)
 - GMT Clock - MDA
 - Digital Clock (elapsed time or time remaining) - STS
 - 4 Portable Timers (time remaining)

GMT clocks are resettable to zero by DCS command or manually by crew.

Flight Plan Notes
Communications and Instrumentation

Experiment	Data Recorder	Exp 1 Recorder	Exp 2 Recorder	ATM Recorder
M092	D+V	D+(V)	D	D
M093	D+V	D+(V)	D	D
M131	D+V	D+(V)		D
M133*	D			
M171	D+V	D+(V)	D	D
M509	D+V	D+(V)		D
M509 (Suited)	D+V	D+(V)	D	D
M512	D+V			D
S020	D+V			D
S063	D+V			D
S073/T027	D+V	D+(V)	D	D
S183	D+V	D+(V)		D
T002	D+V			D
T013	D+V	D+(V)		D (RT)
T020	D+V			D
T027 (Sample Array)	D			
ATM ***	D+V			D
EREP**	D+V			D

D = Data Record V = Voice Record (V) = Backup Voice

* M133 has own recorder for detailed data.

** EREP has own scientific data recorder.

*** Onboard TV recorder capability exists for recording of ATM or operational video.

Other experiments not listed may require either log entries or voice annotation requiring data recorder.

Table C-1. Experiment Recording Requirements

Flight Plan Notes
Communications and Instrumentation

DATA SYSTEM	DOWNLINK	UPLINK	FREQUENCY (MHZ)
CSM (USB)	R/T TM R/T VX		2287.5 (All cluster R/T VX via CSM)
	D/T TM D/T VX R/T TV D/T TV		2272.5 (Primary D/T VX via AM)
		CMD VX	2106.4 (All cluster R/T VX via CSM)
CSM (VHF)	VX	VX	296.8/259.7
AM (UHF)		CMD (Including Teleprinter)	450
AM (VHF)	R/T TM D/T TM D/T VX		230.4 and 235.0 and 246.3 (All cluster D/T VX normally via AM)
ATM (UHF)		CMD	450
ATM (VHF)	R/T TM D/T TM		231.9 and 237.0

Table C-3. Skylab Data Link Summary

Flight Plan Notes
Communications and Instrumentation

RECORDER CHARACTERISTIC	DATA SYSTEM				TV	EREP	M133
	CSM		AM	ATM			
No. of Recorders	1		3/4 *	2	1	1/1 *	2
No. of Channels/Recorder	5-PCM/1-VX/8-Spare		1-PCM/1-VX	1-PCM	1	28	7
Record Speed (ips)	LBR (1)	HBR (2)	1-7/8	4	12	7.5/60(4)	-
Dump Speed (ips)	3.75	15	41-1/4	72	12	N/A	N/A
Dump Ratio	120	15	22:1	18:1	1:1	N/A	N/A
Record Data Rate (KBPS)	32:1	1:1	5.12(3)	4	--	N/A	N/A
Dump Data Rate (KBPS)	1.6	51.2	112.64(3)	72	--	N/A	N/A
Maximum Record Time	51.2	30 min.	4 hours	90 min	30	192 min/ 24 min	100 hrs
Maximum Dump Time	120 min.	30 min.	10.9 min	5 min	30	N/A	N/A
Data Content	Operational & Exp D008 TM and CSM voice, if required		MDA/AM/OWS Opr. TM, Exp. TM and Cluster VX	ATM Opr. and Exp. TM (Including Cluster Pointing)	TV	EREP Sensor scientific & support data	M133 Sensor Data

* Spares

(1) Low Bit Rate

(2) High Bit Rate

(3) Experiments M509, T013 require record and dump data rates of 5.76 and 126.72 KBPS, respectively.

(4) All EREP recording is at 7.5 ips except when S192 is used - then 60 ips speed is required.

Table C-4. Skylab Recorder Characteristic Summary

Flight Plan Notes
Guidance, Navigation and Control

D. Guidance, Navigation, and Control

1. CSM

- a) CSM G&N status during docked mode consists of the inertial and optics subsystem powered down and the CMC in standby.
- b) IMU is not aligned periodically during SKYLAB quiescent mode unless operationally required.
- c) Scheduled IMU realignments are shown in Table D-1.
- d) Star availability for CSM IMU realignment for the duration of SL-2 and SL-3 is presented in figures D-1 and D-2 respectively which shows the path of the gimballed optics axis (45 degree half-cone-angle field of view).
- e) Prior to TPI the SWS lights shall be tracked by COAS. The ground will turn acquisition lights off shortly after sunrise or set the AM timer to shut them off to prevent crew observation of the intense light at short range.

2. Orbital Assembly APCS

a) OA Primary Attitude Modes

- Solar Inertial (SI) - The solar inertial attitude is defined as the principal SWS X axis in the orbital plane with the Z axis coincident with the sun line. The +Z axis points directly toward the sun and at orbital noon the +X axis is in the direction of the velocity vector. Refer to Figure D-3, Skylab Dynamic Body Axis System for axes reference.
- Z-Local Vertical (Z-LV) - The Z-local vertical attitude is defined as the SWS geometric X axis in the orbital plane with the Z axis along the local vertical. The +X axis is in the direction of the velocity vector except for rendezvous when -X is in direction of velocity vector.

b) APCS Control Modes

- Standby Mode - Used during activation and deactivation of the APCS and when OA control by APCS is not required.
- Solar Inertial Mode - Maintained for CSM docking and undocking, nominal workshop operations and SWS storage.
- Solar Inertial (Experiment Pointing) Mode - Maintained for ATM experiment operation.

Flight Plan Notes
Guidance, Navigation, and Control

- Z-Local Vertical Mode - Maintained for CSM rendezvous and EREP pointing. Offset pointing while in Z-LV is possible in 0.1 degree increments up to 51.1 degrees about the previously selected Z-LV experiment pointing attitude reference.
- Attitude Hold (CMG) Mode will be used during post separation fly-around and the docked RCS trim burns.
- c) Primary control of the APCS is by crew selection of attitude modes (switch positions on ATM C&D console) under control of the ATMDC. The crew can override automatic control or input supplementary pointing commands via the digital address system (DAS). Ground command may control the attitude by input to the ATMDC via the digital command system (DCS).
- d) Ground command and crew access to the ATMDC is gained by crew selection of ENABLE position on CMPTR ENABLE/INHIBIT switch. Ground command access to ATMDC also requires crew selection of ENABLE position on GRD CMD ENABLE/INHIBIT switch.
- e) During crew input to the ATMDC via the DAS, ATM DCS commands are automatically inhibited.
- f) The ATMDC is scheduled for navigation and timing update daily, and prior to Z-LV maneuvers. The following navigation and timing parameters may be updated by ground uplink or crew input via DAS.
 - Timer A and B
 - Greenwich Meridian Celestial Longitude
 - Bias term for rotation about sun reference axis to SI axis
 - Z-LV maneuver time
 - Change to previous ascending node crossing time
 - Orbital ascending node
 - Mean anomaly of earth in solar orbit
 - Ascending node period
 - Orbital Inclination
 - Ratio of earth to orbital radius
 - Change to previous orbital midnight time
 - First order nodal regression rate
 - Time interval between orbital midnights
 - Bias CMG momentum components

Flight Plan Notes
Guidance, Navigation, and Control

- Orbital rate
- Minimum dump interval

All parameters are updated in each update. ATMDC updates are effected each morning and prior to each Z-LV maneuver.

- g) TACS will be used for momentum management when necessary.
- h) CMG desaturation maneuvers are inhibited during experiments S183, S019, M074 (SMMD calibration), BMMD operation, T027/S073 (Modes 1a, 1b, 1d, 1e, 2a, 3a, 3c), S063 (F02).
- i) For Z-LV maneuvers CMG desaturation may not be inhibited one rev prior to a Z-LV pass nor two revs following the pass. Figure D-4 represents the number of consecutive momentum dump inhibits which can be done without using TACS. Curve A resulted from a nominal case computer simulation of orbital dumps and inhibits for 16 days. Curve B is the conservative estimate of what is really required for momentum management and will be the basis for scheduling.
- j) Control system gain changes are scheduled after CSM docking to configure the APCS for the docked mode and prior to CSM undocking to reconfigure the APCS for the unmanned mode. These changes are effected by ground command to the ATMDC.
- k) A reference star will be tracked continuously or partly through the orbit on all manned orbits for which ATM and other selected experiments are scheduled. Reacquisition by the crew of one of the three specified target stars is required when star occultation occurs for TBD hours or during an experiment period requiring star tracker data. The three specified target stars are Canopus, Archnar, and Alpha Crux. The star reacquisition schedule is TBD. There are no requirements to operate the star tracker during the unmanned portions of the Skylab mission.

Flight Plan Notes
Guidance, Navigation and Control

RENDEZVOUS SEQUENCE		
APPROXIMATE GET (HRS)	ALIGNMENT TYPE	PURPOSE/NOTES
0.6	P52 OPT 3	CHECK G-SENSITIVE LAUNCH DRIFTS
IMMEDIATELY FOLLOWING FIRST ALIGNMENT	P52 OPT 2 (TALIGN) USE NORMAL COARSE ALIGN FOLLOWED BY SIGHTING ON TWO STARS	ESTABLISHES RENDEZVOUS REFSMMAT (LVLH AT TPI)
2.3	NC1	
3.4	P52 OPT 3	DRIFT CHECK
NOTE: IF PC IS RQD, PERFORM P52 OPT 3 DURING 1st NIGHT PERIOD AFTER NC1. THEN ENTER P38 SEQUENCE.		
4.6	NC2	
8.0	DOCKING	
8.5	P52 OPT 3 TWO STARS	PROVIDES ACCURATE IMU FOR THE ESTABLISHMENT OF THE IMU/ATM MATRIX IN P50 + ANOTHER DRIFT CHECK
8.7	P50 OPT 1 PARTIAL UPDATE USING SUN SENSOR INFORMATION	P50 SHOULD BE PERFORMED WITH VEHICLE UNDER SUN SENSOR CONTROL (I.E., NOT DURING CG DUMP OR IN DARKNESS) - GROUND WILL PROVIDE A TIME TO START IN THIS CASE - RELAY N23 TO GROUND
22.0 22.3	P52 OPT 3 TWO STARS RCS TRIM 1A (IF REQ'D)	AGAIN PROVIDES AN ACCURATE IMU FOR P50 + DRIFT CHECK AND FOR +X RCS BURNS (IF REQUIRED)
22.5	P50 OPT 2 TOTAL UPDATE USING SUN SENSOR + STAR TRACKER DATA	SUN SENSOR CONTROL RQD-- C&D PANEL MUST BE ACTIVATED TO PROVIDE STAR TRACKER DATA (2 CMN RQD) RELAY N23 TO GROUND

Table D-1. IMU Alignments
(Sheet 1 of 3 Sheets)

Flight Plan Notes
Guidance, Navigation and Control

RENDEZVOUS SEQUENCE (Cont'd)		
APPROXIMATE GET (HRS)	ALIGNMENT TYPE	PURPOSE/NOTES
22.7	P52 OPT 3, ATM SUN SENSOR + STAR	SAME RESTRICTIONS AS P52 OPT 2. THIS ALIGNMENT IS PERFORMED AS A FINAL VERIFICATION OF DOCKED ALIGNMENT CAPABILITY.
23.3	RCS TRIM 1b	(2 CMN REQ'D)
23.4	G&N POWER DOWN	N93 TORQUE ANGLES SHOULD BE APPROXIMATELY ZERO

ENTRY SIMULATIONS ARE SCHEDULED TO DUPLICATE
THE FOLLOWING ENTRY SEQUENCE ON THE NEXT PAGE.

Table D-1. IMU Alignments
(Sheet 2 of 3 Sheets)

Flight Plan Notes
Guidance, Navigation and Control

ENTRY SEQUENCE		
APPROXIMATE MINUS TIME TO SHAPING	ALIGNMENT TYPE	PURPOSE/NOTES
-17.6	P51 ATM SUN SENSOR AND STAR TRACKER	REQUIRES SUN SENSOR CONTROL AND C&D PANEL DATA
NOTE: IF G&N IS POWERED UP LATE (IN DARKNESS) CREW CAN DO P51 (TWO STARS).		
-17.0	P52 OPT 1 USE NORMAL COARSE ALIGN FOLLOWED BY SIGHTING ON TWO STARS	ALIGNMENT IS TO A PRELIMINARY ENTRY REFSMMAT - 0-180-0 AT SPS2TIG
-16.5	P50 OPT 2	REAFFIRM ATM/NB ALIGNMENT
-15.8	P52 OPT 3 (not done during Entry Sim)	DRIFT CHECK
G&N POWERDOWN		
-5.4	P51 USE ATM SUN SENSOR + STAR TRACKER DATA	SAME NOTES AS ALIGNMENT AT -19 HRS
-5.5	P52 OPT 1 NORMAL COARSE ALIGN FOLLOWED BY SIGHTINGS ON TWO STARS	ESTABLISHES FINAL ENTRY ORIENTATION: 0-180-0 AT SPS2TIG
-1.5	P52 OPT 3	DRIFT CHECK
-1.3	UNDOCKING	
	SHAPING BURN	ORBIT SHAPING
+1.3	P52 OPT 3	FINAL DEORBIT/ENTRY ALIGNMENT
+3.0	DEORBIT	

Table D-1. IMU Alignments
(Sheet 3 of 3 Sheets)

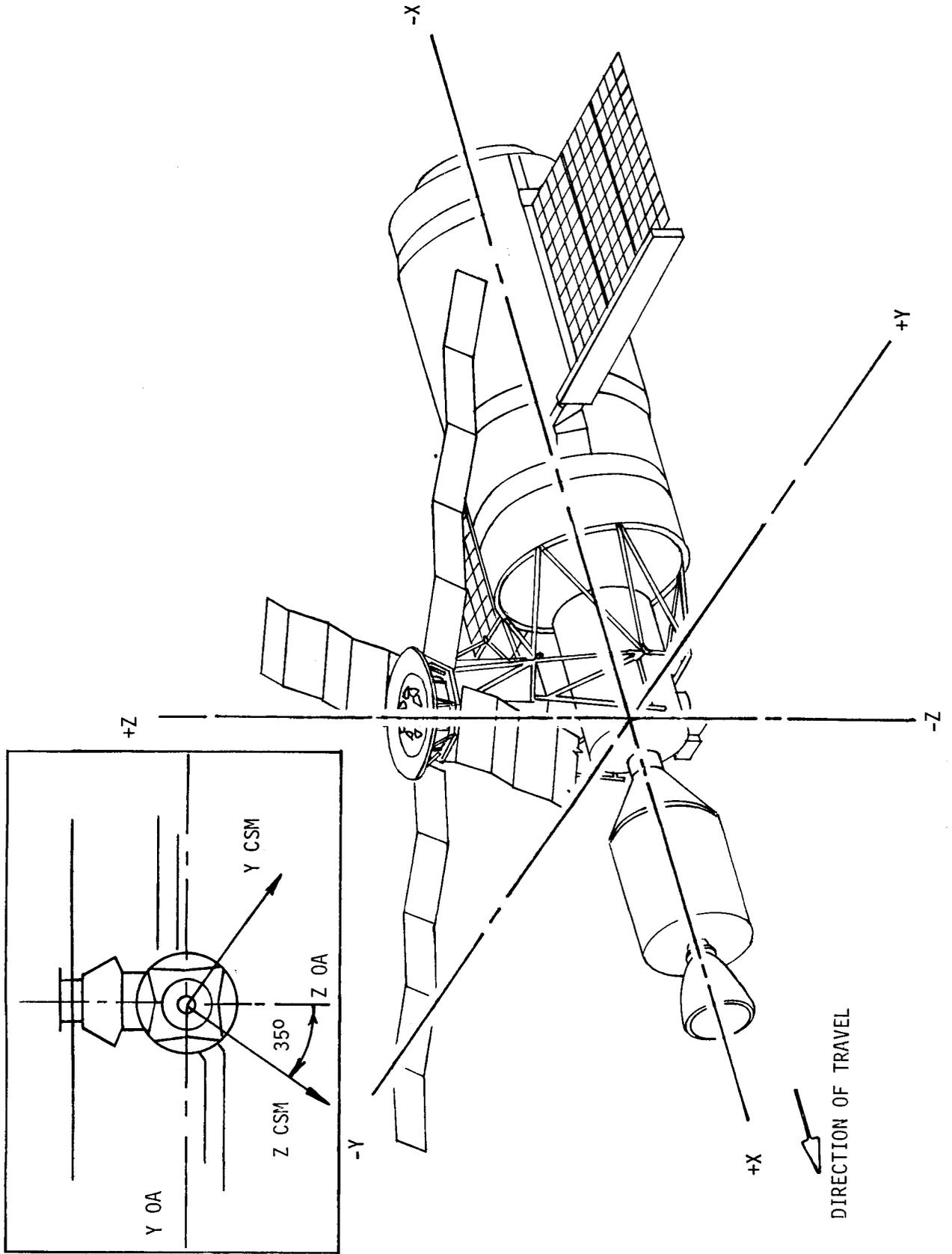


Figure D-3. Skylab Dynamic Body Axis System
(Inset - CSM docking attitude)

Flight Plan Notes
Guidance, Navigation and Control

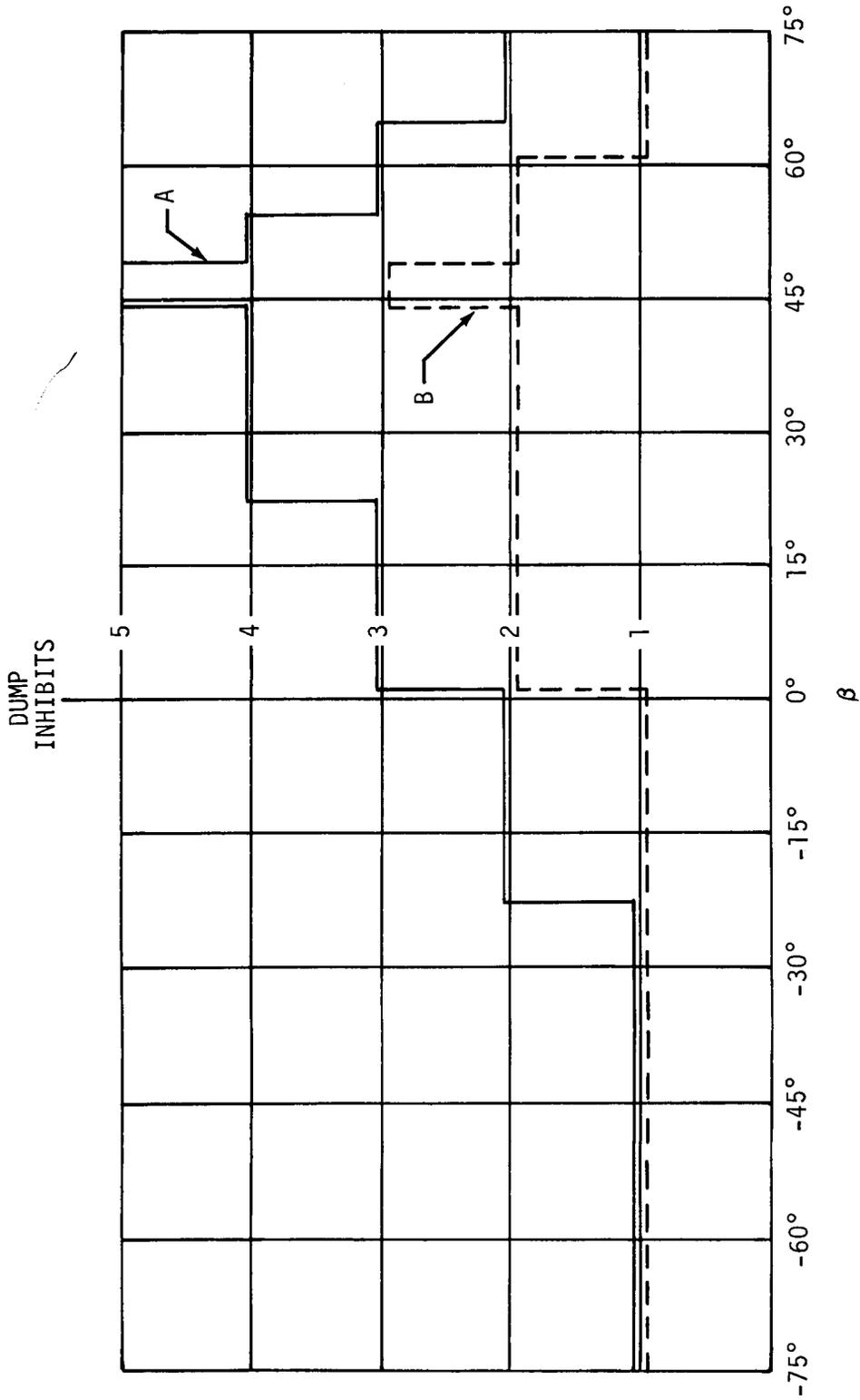


Figure D-4 MOMENTUM DUMP INHIBITS WITH NO TACS

Flight Plan Notes
Environmental Control

E. Environmental Control

1. CSM

- a) LiOH cartridge units are changed after 12 hours of use or if CM CO₂ partial pressure exceeds 5.5 mm Hg. Replacements are stored in MDA and deployed in the CM. The LiOH system is operated during CM use and during molecular sieve bakeout at deactivation. There are 10 units launched aboard the MDA, 10 launched with each of the SL-2 and SL-3 CSMs, and 6 in the SL-4 CSM. LiOH canister change schedule is as follows:

MISSION	DOY	GMT	COMMENTS
2	121	0300	Change one cannister
	122	1300	Remove both cannisters
	148	1650	Install both cannisters
	149	0500	Change one cannister
3	211	1520	Remove both cannisters
	266	1630	Install both cannisters
	267	0515	Change one cannister

- b) At lift-off the CSM cabin will contain 60 percent O₂ and 40 percent N₂.
- c) The CSM fuel cells remain on until H₂ cryogenic depletion (approximately 13 days).
- d) CSM ECS activation requires that the cluster dewpoint be raised to 50°F to prevent suit circuit heat exchanger dryout.
- e) The cluster total pressure must be raised to a point higher than the CSM regulation point by over-ride of O₂ pressure control system prior to CSM suit loop activation.
- f) During OWS activation the CSM suit loop will be deactivated within 30 minutes after opening the MDA hatch to prevent suit circuit heat exchanger dryout.

2. SWS

- a) All command system controlled cluster ECS functions, except the habitation vent valves and the waste tank vent valves, have a manual crew over-ride. Ground command controls ECS functions, except fans and heaters, during storage and activation phases and provides crew back-up during manned phases.
- b) Molecular sieve bakeout is performed approximately every 28 days of habitation.
- c) Prior to the molecular sieve bakeout at deactivation, the CSM CO₂ removal system must be activated. Bakeout requires 5 hours for ² each bed. After bakeout the beds are placed in the stowage configuration.

Flight Plan Notes
Environmental Control

- d) Prior to molecular sieve bakeout at the end of the first 28 day period of SL-3 and SL-4 the second molecular sieve must be activated. The primary molecular sieve beds are then baked out and placed in the stowed condition. The mission is continued on the second molecular sieve.
- e) Egress for EVA is provided in the AM lock compartment. The minimum time required to depressurize the AM lock compartment from 5.0 to 0.15 psia is 50 seconds and to pressurize the AM lock compartment from 0 to 4.95 psia is 25 seconds. Slower depressurization or pressurization is achievable by partially opening the appropriate equalization valve.
- f) The OWS refrigeration system must run continuously throughout the Skylab mission. Either pump or loop selection can be made by the crew. Loop selection only is provided to ground control.
- g) MSFN coverage is required during the following:
 - 1) Transfer of the AM Coolant Loop Inverters to Ground Command. MSFN coverage is required to reactivate the necessary pumps for coolant loop operation since they will be shut off by switching the inverters to command.
 - 2) Switching of the RSS pumps. Switching is accomplished by the crewmen but must be confirmed by the MSFN that pump switching did occur.
- h) Following rendezvous and docking, the PGAs will be dried and then placed inside PGA bags. The bags with PGAs will be stowed in the CM on orbit to minimize crew handling in the event of a fire in the OWS and subsequent emergency crew transfer into the CM. Before EVA, the crew will retrieve the PGA bags from the CM and transfer the bags to the OWS suit donning station. PGAs will be removed from the bags and donned for EVA. Following EVA, the PGAs will again be dried, placed inside the PGA bags, and stowed in the CM.

Flight Plan Notes
Electrical Power

F. Electrical Power

1. CSM

- a) The entry and post-landing batteries A and B will be turned on five minutes prior to an SPS burn and turned off immediately after completion of burn.
- b) Charging of the entry and post landing batteries A and B will begin at docking. Each battery will require approximately 10 hours of charging, for a total of 20 hours. Additional charging is TBD.
- c) The fuel cells will operate until H₂ cryo is depleted (any residual H₂ will be vented to space and excess O₂ vented to the cluster atmosphere or overboard). Cryo depletion is after approximately 13 days (14-18 days for SL-3 and SL-4) at which time the CSM EPS will be paralleled with the AM/ATM EPS.
- d) Twenty minutes prior to, during and 10 minutes after a fuel cell purge, the H₂ Purge Line Heater (PN3) must be turned on. O₂ is purged for 120 seconds per fuel cell and H₂ purged for 80 seconds. Purging frequency until fuel cell shutdown is once per day for O₂ and every other day for H₂.
- e) Inverter 1 feeds AC bus 1 and inverter 2 feeds AC bus 2 for boost through docking and during deorbit and entry. For docked operations, both inverters will be used.

2. AM/ATM

- a) Time of CSM-AM/ATM power transfer initiation for SL-2 is DOY 134, GMT 1340; SL-3 is DOY 224, GMT 1300. Time of power termination for SL-2 is DOY 149, GMT 0445; SL-3 is DOY 265, GMT 1200.
- b) During some EREP passes or other periods of high electrical power usage, it may be necessary to inhibit certain astronaut support equipment. This power conservation is required to balance power usage with average electrical power available and prevent exceeding the maximum depth of discharge for the AM and ATM batteries. Some loads that can be inhibited during periods of high power usage are given in Table F-1.

Flight Plan Notes
Electrical Power

Load	Maximum Potential Load If Not Inhibited (Watts)
<u>OWS</u>	
Waste Management Compt. Water Heater (1)	124
Wardroom Hot Water Heater (1)	124
Food Trays (3)	476
Duct Heaters (8)	1582
Waste Processors (3)	247
Light in Compartments Not Occupied	50
<u>MDA</u>	
Wall Heaters (4)	557
Radial Docking Port Heater (1)	17
<u>ATM</u>	
EPEA	65
Experiment Electronics	191

Table F-1. Candidates for Electrical Power Reduction

Flight Plan Notes
Systems Housekeeping

G. Systems Housekeeping

1. Periodic CSM housekeeping functions are scheduled in the timeline as follows:

Every 2 Days	Weekly Checks
Voltage checks Fuel cell checks Cryo system monitor Batt manifold pressure check & vent H ₂ O condensate check SPS monitor check RCS monitor check Total crew time req'd ≈ 20 minutes	Alt. prim. loop glycol pumps Top-off CM O ₂ supply (repress package) Circulate glycol in sec. loop & recheck in 1 hour Circulate glycol in prim. loop thru radiator #2 & suit heat xchgr & recheck in 1 hour O ₂ main regulator check Inspect & dry CM girth rings and CM window frames Place batt A&B on batt bus and check Total crew time req'd ≈ 45 minutes

2. Periodic SWS S/HK functions scheduled in the timeline are delineated in Table G-1. The functions are grouped into tasks which are common by function or frequency of performance. S/HK task identifiers are assigned to each task with the number indicating frequency of task performance in days and the letter indicating each separate task for the indicated frequency.

3. Checklists

a) Pre-sleep checklist

- 1) Stow any loose objects.
- 2) Configure two intercom boxes to the A-sleep position and one to channel A-ON with reduced volume.
- 3) Adjust ventilation diffusers as desired and heat control for crew comfort during sleep.

Flight Plan Notes
Systems Housekeeping

- 4) Configure bunks for sleep.
 - 5) Check all cameras (except ATM and those experiments operating) to insure that all film is in the film vaults. (Only 2 transport mechanisms can be stored at once.)
 - 6) Check status of experiments operating during sleep period.
 - 7) Inhibit audible ATM flare alarm (with concurrence of ground).
 - 8) Enable ATM ground command.
 - 9) Ensure adequate paper supply for teleprinter.
 - 10) Dump condensate system.
 - 11) Set up breakfast tray and configure necessary switches and timer.
 - 12) Set lights.
 - 13) Fuel cell purge as req'd.
 - 14) Take Mineral Supplements (W)
 - 15) Log H₂O Gun counter readings (W)
- b) Post-sleep checklist
- 1) Configure bunks for stowage position.
 - 2) Configure intercom for normal operation.
 - 3) Adjust ventilation diffusers and heat control for crew comfort.
 - 4) Check status of experiments operating during sleep period.
 - 5) Voice down crew status report.
 - 6) Enable ATM audible flare alarm.
 - 7) Check teleprinter for messages.

Flight Plan Notes
Systems Housekeeping

S/HK TASK I.D.	TASK	SCHEDULED INTERVAL (DAYS)	ESTIMATED CREW TIME (MIN)	CREW STA. & LOC.	REMARKS
1a	Dump condensate system	1	20	ST, H	Dump during presleep checklist.
	Reload teleprinter paper cartridge	1	5	ST	Reload during presleep checklist and due to TPTR paper-low indication.
	Dry urine drawers and cold plates	1	3-5 min/cmn	H	Perform during collection bag replacement (M071).
	Vacuum clean and dry any collected moisture	1	10	ALL	Vacuum loose particles as required.
	Resupply WR and WMC trash bags	1	5	W,H	
	Resupply experiment and sleep compartment trash bags	1	5	E,S	
1c	Dump trash airlock	1	15	E	Approximately 5 times daily, 3 minutes for each dump.
	ECS Periodic Verification	1	2	E	
	Resupply utility wipes	3	5	W,H,E,S	As required.
3a	Replace waste water bag	3	5	H	Replace as required
	Disinfect trash locker diaphragm and inner locker	3	10	W,H,E,S	Schedule with 1c resupply of trash bags.
	Disinfect WMC collection module	7	10	H	Disinfect WMC collection module during urine collection bag replacement. Cleaning may be required more often if odor develops or heavy urine spillage occurs. Schedule with 1b.
7a	Urine drawers Urine stowage areas Fecal collector seat Urine receiver stowage area Urine receiver bracket				

Table G-1. SWS Housekeeping Functions
(Sheet 1 of 3 Sheets)

Flight Plan Notes
Systems Housekeeping

S/HK TASK I.D.	TASK	SCHEDULED INTERVAL (DAYS)	ESTIMATED CREW TIME (MIN)	CREW STA. & LOC.	REMARKS
14a	(Continued)				
14b	Dry moisture from food chiller Replace sleep restraints (3), top blankets of sleep restraints and pillows	14 14	10 10 min/ cmn	W S	
28a	H ₂ O purification check	28	30	D	Check prior to connecting an unused tank to a H ₂ O system and prior to activation of the H ₂ O systems for SL-3 and SL-4
28b	Transfer H ₂ O source Replace MMC charcoal canister Replace fecal-urine collector filter Replace molecular sieve A and B charcoal canisters	28 28 28 28	10 15 10 15	D E H ST	
28c	Replace bottom blankets of sleep restraints	28	10	S	
28d	Bakeout molecular sieve(s) Replace outlet CO ₂ filters PP0 ₂ Sensors - Replace	28 28 28	15 10 10	ST ST ST	Done after bakeout.

Table G-1. SWS Housekeeping Functions
(Sheet 3 of 3 Sheets)

Flight Plan Notes
Habitability

H. Habitability

1. Unused food bags from the 12 man-day package in the CSM will be transferred to the OWS food pantry during activation. The bags will be used for days 1-5 or disposed if not used. 3 man-day left in CSM.
2. Every 7 days, a crewman is scheduled to remove a 7-day food supply from one of the eleven food storage containers (CS-F) and transfer the supply to the wardroom food pantry (CS-W).
3. Prior to each 7-day supply of the food pantry, leftovers in the pantry will be returned to the food lockers.
4. After each meal, a crewman is scheduled to place the food to be heated for the next meal in the food trays and set the timers. Frozen foods require 62 minutes to thaw and 70 additional minutes to heat to 150°F. Food at ambient temperature requires 30 minutes to heat to 150°F. Food tray timers can be set 15 minute increments up to a maximum of 12 hours.
5. Any food that remains after the completion of a meal must be weighed and documented prior to disposal.
6. The crewman must log the total amount of drinking water consumed during a 24-hour period. Deviations from the amount of water to be used for food reconstruction must also be logged.
7. Each defecation, vomitus deposit, and 120 ml sample of a crewmember's daily accumulated urine is processed for eventual return to Earth in the Command Module. Residual urine in its collection bag, after sample withdrawal at beginning of each day (normally between 11 and 12 GMT), is packaged in a trash bag and disposed of through the trash airlock.
8. The OWS trash airlock (CS-E) is scheduled for use 5 times per day. Those items deposited in the trash airlock are to be recorded on tape to be dumped to the ground. Trash dumps are part of normal systems housekeeping (See table G-1).
9. Visible moisture should be dried at any time, from any area or system, when noticed. Utility wipes may be used for the purpose of spot drying. Additional housekeeping functions are in table G-1.
10. Water volumes used to purge the water distribution system must be recorded.

Flight Plan Notes
Habitability

11. The first and last "eat" periods in the OWS are scheduled as follows:

MISSION	FIRST EAT (DOY,GMT)	LAST EAT (DOY,GMT)
SL-2	122,2200	148,1100
SL-3	212,0719	266,0945
SL-4	TBS	TBS

Flight Plan Notes
Venting

I. Venting

1. Figure I-1 shows cluster axes, window locations, and vent locations.
2. Table I-1 lists the windows shown in figure I-1 and gives angular locations for these windows. This gives a reference as to which windows may be affected by venting.
3. Table I-2 lists the vents shown in figure I-1 and gives angular locations for the vents as well as duration and frequency.
4. Ground rules for venting are as follows:
 - No direct urine dumps into the waste tank
 - 4680 ml urine per day sealed in bags
 - Five trash dumps per day
 - Selective trash dumps containing bacterial growth potential
 - Condensate Tank will be normally vented into the waste tank. Contingency overboard dump.
5. CSM RCS firings should be minimized.
 - No controlled venting, dumping or RCS firings 15 minutes prior to, or during, optical experiment operations (except 30 minutes prior to, and during S019).
 - All window, optical and attitude sensor covers in place and during controlled venting dumping, and hypergol firings.
 - All protective covers in place at least 15 minutes after conclusion of RCS or controlled venting/dumping operations.
6. Table I-3 shows scheduling criteria for vents.

Flight Plan Notes
Venting

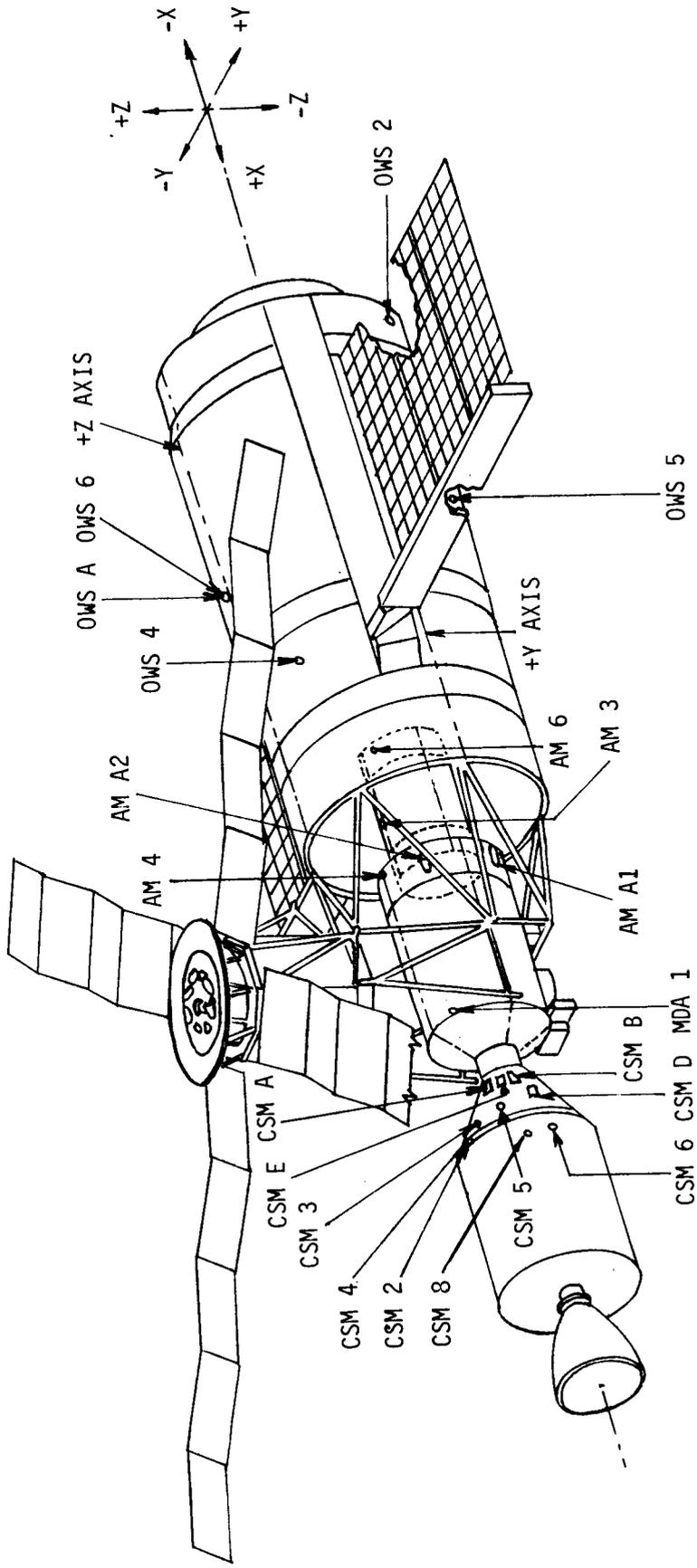


Figure I-1. OA Window and Vent Location
(Sheet 1 of 2 Sheets)

Flight Plan Notes
Venting

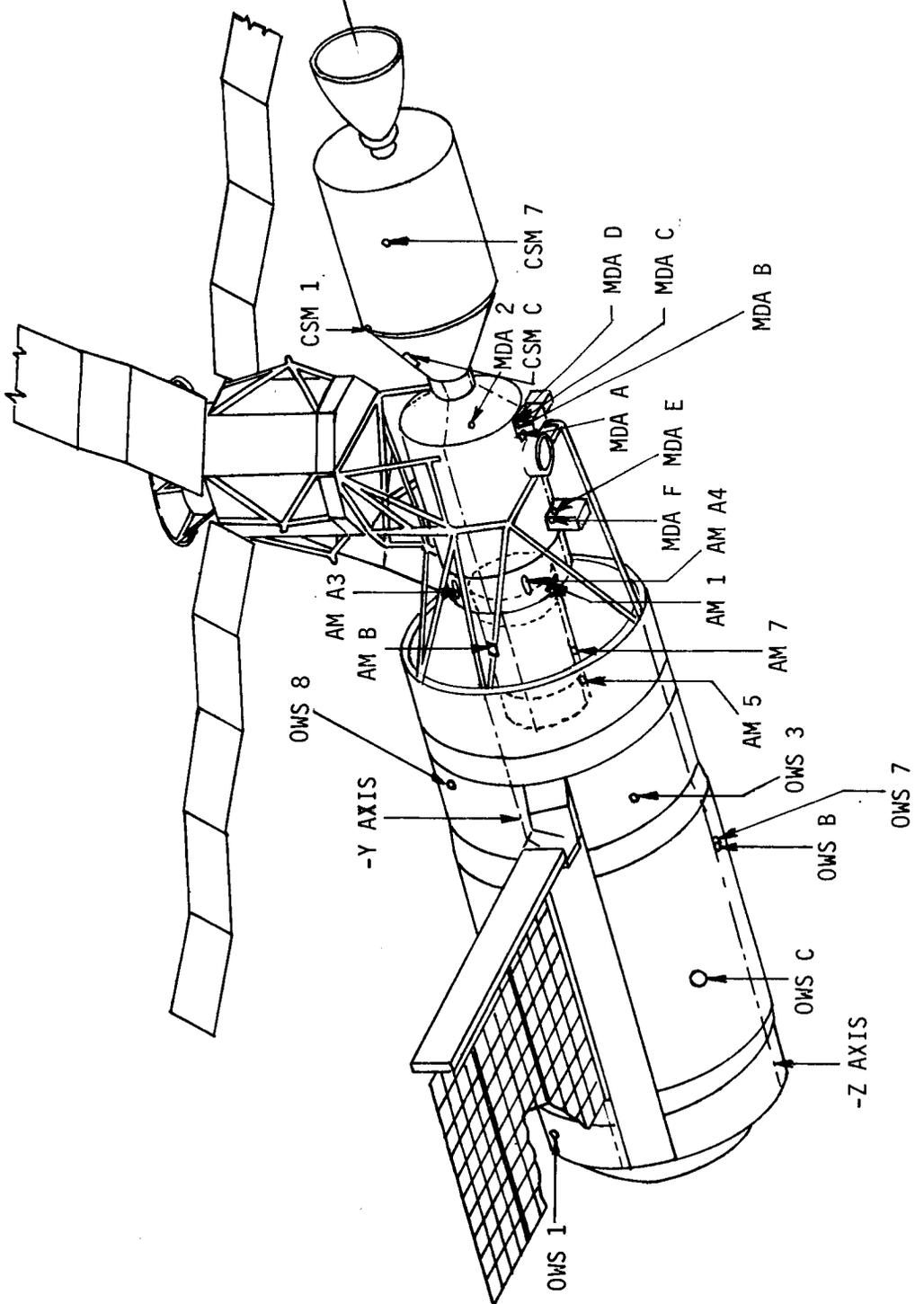
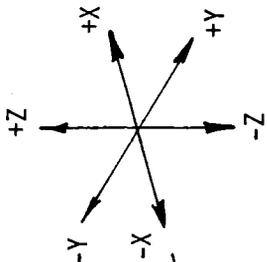


Figure I-1. OA Window and Vent Location
(Sheet 2 of 2 Sheets)

Flight Plan Notes
Venting

MODULE AND WINDOW NAME	WINDOW LOCATION
	ANGULAR (DYNAMIC AXIS)
<u>CSM</u>	
A. LEFT RENDEZVOUS (DOCKING)	3.3° OFF +Z TOWARD +Y
B. RIGHT RENDEZVOUS (DOCKING)	23.3° OFF +Y TOWARD +Z
C. LEFT SIDE VIEWING	25.5° OFF +Z TOWARD -Y
D. RIGHT SIDE VIEWING	5.5° OFF +Y TOWARD -Z
E. CENTER HATCH	35° OFF +Z TOWARD +Y
<u>MDA</u>	
A. S190 EXPERIMENT WINDOW	ON -Z AXIS
B. S191 VIEWFINDER/TRACKING PORT	12.7° OFF -Z TOWARD -Y
C. S191 CRYOGENIC UNIT PORT	2.9° OFF -Z TOWARD +Y
D. S191 SHORT WAVELENGTH PORT	13.7° OFF -Z TOWARD +Y
E. S192 SHORT WAVELENGTH PORT	0.3° OFF -Z TOWARD -Y
F. S192 THERMAL PORT	5.3° OFF -Z TOWARD -Y
<u>AM</u>	
A. AM VIEWING PORTS	
1)	37.87 OFF -Z TOWARD +Y
2)	37.87 OFF +Y TOWARD +Z
3)	37.87 OFF +Z TOWARD -Y
4)	37.87 OFF -Y TOWARD -Z
B. EVA HATCH WINDOW	45° OFF +Z TOWARD -Y
<u>OWS</u>	
A. SOLAR SCIENTIFIC AIRLOCK	0.15° OFF +Z TOWARD -Y
B. ANTI-SOLAR SCIENTIFIC AIRLOCK	1.88° OFF -Z TOWARD +Y
C. WARDROOM WINDOW	25.9° OFF -Z TOWARD -Y

Table I-1. OA WINDOW LOCATIONS

Flight Plan Notes
Venting

MODULE AND VENT NAME	CREW CONTROLLED	ANGULAR LOCATION	FREQUENCY	DURATION	SCHEDULING CRITERIA
CSM					
1. Steam	N/A	27.5° off +Z toward -Y	---	---	Contingency use only; CM cooled by OWS ECS.
2. Urine Dump	N/A	16° off +Z toward -Y	---	---	Contingency use only; CMN will use OWS waste facility.
3. Aux Urine Dump	N/A	25° off +Z toward +Y	---	---	Contingency use, back-up to above.
4. Waste Water Dump	N/A	6° off +Z toward -Y	---	---	Contingency use; fuel cell water will be stored in SM tank.
5. Air	Yes	34.5° off +Y toward +Z	once per mission	---	Exhausts docking tunnel at undocking.
6. Fuel Cell Hydrogen Purge	Yes	4° off +Y toward -Z	every day for 13 days	80 sec per fuel cell	Manual purge, frequency requirements approximate.
7. Fuel Cell Oxygen Purge	Yes	4° off -Y toward -Z	every other day for 13 days	120 sec per fuel cell	Manual purge, frequency requirements approximate.
8. Fuel Cell Hydrogen Cryogenic Dump	Yes	2.58° off +Y toward +Z	---	See NOTE	Will vent fuel cell cryogenics if fuel cells are shut down prior to cryo depletion. O ₂ cryogenics are vented to the cluster atmosphere or overboard. NOTE: Duration dependent upon cryo amount remaining.

Table I-2. Vent Location and Schedule
(Sheet 1 of 5 Sheets)

MODULE AND VENT NAME	CREW CONTROLLED	ANGULAR LOCATION	FREQUENCY	DURATION	SCHEDULING CRITERIA
MDA					
1. Vent Valves	No	45° off +Z toward +Y	once	9 min 32 sec	Opened 5 min before SL-1 liftoff - closed when MDA press reaches nominal .5 psia. No further planned usage, manned or unmanned.
2. Experiment	Yes	30° off -Y toward -Z	M551, 552, 553, 554, 555	See Note	Open during each performance M551-2:39, 3 times; M552-8:00; M553-0:44, 2 times; M554-6:08; M555-118:22. Most contamination from M551, M552, and M553.
AM					
1. Molecular Sieve (NPV) Bake out event (NPV)	No Yes	On -Z Axis	6/orbit 1/28 days	15 min 5 hrs for 1 bed (2 beds)	Cyclic during manned phases; no anticipated usage during unmanned portions.
2. Condensate Tank	Yes	---	1/day	6 min	Will normally be vented into the waste tank. May be vented earlier but not later than 24 hours. Contingency overboard vent.
3. EVA Depressurization	Yes	80° off +Z toward -Y	1/EVA	92 sec	Vents the Airlock prior to EVA
4. Nitrogen Regulator Relief	N/A	On +Z Axis	---	---	Contingency use only; active only if GO_2 supply system regulator fails open. ² Could occur with any N_2 overpressure.

Table I-2. Vent Location and Schedule
(Sheet 2 of 5 Sheets)

MODULE AND VENT NAME	CREW CONTROLLED	ANGULAR LOCATION	FREQUENCY	DURATION	SCHEDULING CRITERIA
AM (Cont)					
5. Aft Compartment Relief	N/A	10° off -Z toward +Y	---	---	Contingency use only; functions only if cabin pressure exceeds 5.5 psia
6. Airlock Compartment Relief	N/A	30° off +Z toward +Y	---	---	Contingency use only; functions only if cabin pressure exceeds 5.5 psia
7. Forward Compartment Relief	N/A	100° off -Z toward +Y	---	---	Contingency use only; functions only if cabin pressure exceeds 5.5 psia
OMS					
1. Waste Tank NPV-A	See 2.	180° off +Z toward -Y	See 2.	See 2.	See 2.
2. Waste Tank NPV-B	No	180° off -Z toward +Y	Once	(to 0.5 psia)	Initial blowdown to vacuum occurs prior to solar array deployment and CMG spinup.
Urine Dump	Yes		1/day	10 min	After sample withdraw, each bag (3) will be crimped to prevent urine leakage and placed inside trash bag and then dumped.
Trash Dump	Yes		5/day	10 min	Contains waste foods, clothes, towels, tissues

Table I-2. Vent Location and Schedule
(Sheet 3 of 5 Sheets)

Flight Plan Notes
Venting

MODULE AND VENT NAME	CREW CONTROLLED	ANGULAR LOCATION	FREQUENCY	DURATION	SCHEDULING CRITERIA
OWS (Cont)					
Waste Processor	Yes		1/day	≈10 hr	Dries fecal matter and vomitus. Very small amount of moisture vented.
Water Purge and drain (MMC and WR)	Yes		1/Mission	20 hr	Drain water system at end of SL-2,3,4. Purge water system with iodine rich water at beginning of SL-3 and SL-4. Drain water system of air at beginning of SL-2.
3. Habitation Area NPV-A	See 4.	25° off -Z toward -Y	See 4.	See 4.	See 4.
4. Habitation Area NPV-B	No	25° off +Z toward +Y	Once	30 min	Initial blowdown to 0.74 psia occurs prior to solar array deployment and CMG spinup. Final blowdown to 2.0 psia occurs after crew departure.
Initial Blowdown	No		1/Mission	4.5-6 hrs	
Final Blowdown					
5. Experiment Vacuum M171	Yes	40° off -Z toward +Y	M171-15	25 min	Vented during M092 or M171 performance; both vent under meteoroid shield; vented mass will come out the sal's, wardrobe window, each annular end of meteoroid shield
M092	Yes	31° off -Z toward +Y	M092-25	15 min	
6. Solar Scientific Airlock	Yes	0.14° off +Z toward -Y	1/installation in SAL and 1/removal	Less than 5 min	Vacuum source for: S020, S063, S073/T027, T027 and T025 (very small volume vented)
7. Anti-Solar Scientific Airlock	Yes	4.3° off -Z toward +Y	Same	Less than 5 min	Vacuum source for: S019, S063, S073/T027, S183, S149 (very small volume vented)

Table I-2. Vent Location and Schedule
(Sheet 4 of 5 Sheets)

MODULE AND VENT NAME	CREW CON- TROLLED	ANGULAR LOCATION	FREQUENCY	DURATION	SCHEDULING CRITERIA
8. Pneumatic Bottle Dump	No	43° off +Z toward -Y	once	Less than 4 hours	Blown down after Initiation of habitation area pressurization. No further planned usage.

Table I-2. Vent Location and Schedule
(Sheet 5 of 5 Sheets)

Flight Plan Notes
Venting

LOCATION	CREW CONTROLLED VENT	PREFERRED SCHEDULING CONSIDERATIONS
CSM	Fuel cell purges	Prior to sleep period after ATM
MDA	M551 M552 M553	Schedule operation at night and at least 30 minutes prior to EREP and SAL experiments
	M554 M555	Same as above although not as critical
	M479	Schedule operation at night and at least 30 minutes prior to EREP and SAL experiments
AM	Mole Sieve Bakeout	Schedule during sleep period to preclude venting during optical experiment operations
	Condensate tank dump	Prior to sleep period after ATM
OWS	Urine Dump	Schedule with morning trash dump
	Trash Dump	Morning (after breakfast), mid morning, after lunch, mid afternoon, after dinner. Schedule at least 30 minutes prior to EREP and SAL experiments
	Waste Processor	Schedule after breakfast (actually, this will occur as crew need arises. In such a case, schedule SAL experiments at least 30 minutes after processor is activated.
	M092 M171	Schedule at least 30 minutes prior to SAL experiments
		The ground should inhibit S149 and ATM (unmanned) during deactivation vents (WMC and WR water system drain) and unmanned vents (habitation area blowdown)
		T025 and S073/T027 modes 1c and 3d requires coordination with contamination events. S073/T027 modes other than 1a, 1d, 1e, 2a, 3a, 4b, and 5b are not affected by vents.

Table I-3. Scheduling Considerations For Vents

Flight Plan Notes
Experiments

J. Experiments

1. General

- a) Table J-1 is an experiment accomplishment matrix for SL-2, SL-3, and SL-4 (TBS). The matrix gives the experiments and functional objectives scheduled as well as those not scheduled for the missions. Experiments are arranged by Flight scheduling precedence numbers.
- b) Each experiment (except Medical, ATM and EREP) is assigned a Flight scheduling precedence number (100 to 450) for each mission. The precedence number is a measure of an experiment's value to the program on a given mission. The criteria for using the precedence number in scheduling experiments are as follows:
 - . Medical experiments, ATM and EREP take precedence above individual experiments and are considered first for Flight scheduling. With available time remaining, experiments are scheduled in order of preference.
 - . If a unique experiment requirement exists (e.g., Beta angle, new moon, etc.), the experiment is scheduled when that requirement is met.
 - . If a constraint does not permit an experiment to be scheduled (e.g. SAL availability), move through preferences until an experiment fits the available time block.
 - . Attempt to schedule film usage to prevent return of partial magazines.
 - . Attempt to fulfill total requirements as presented in the MISSION REQUIREMENTS document, I-MRD-001E, November 1, 1971.
- c) Table J-2 is a man hour/mission hour utilization summary for each mission (SL-4,TBS). Experiments and times are those scheduled in each individual timeline. ATM data taking time indicates daylight time and setup time for each data period between the 400 K.M. atmospheric constraint. In addition to the 100 hour minimum ATM data taking time for SL-2(220 hours for SL-3 and SL-4), additional time will be allowed for JOP 7 and JOP 13.
- d) Figure J-1 is a man hour utilization chart for SL-2, SL-3 and SL-4 (TBS). The chart shows man hours per day of year.
- e) Table J-3 is a scheduling consideration and constraint matrix for Skylab experiments. The table indicates only that some major constraint or requirement exists. The Experiment Test Objective in Section 4 of this document gives specifics on the constraint.

Flight Plan Notes
Experiments

- f) Figure J-2 gives Scientific Airlock usage times as scheduled for SL-2. Ground rules for scheduling of SAL experiments are as follows:
- . One performance of T025 (solar airlock) is required before any other SAL experiments are performed.
 - . Mode 1a and 0a (solar airlock) of S073/T027 must follow T025 but precede the performance of any other SAL experiments.
 - . Determine viewing opportunities of experiment objectives for desired mission (See 5-g, 5-h, ...of the Experiments section).
 - . Schedule experiment using Flight Scheduling Precedence logic.
 - . Minimize SAL operational (set up and stow) repetitions to conserve crew time.
 - . Attempt to meet all requirements per MRD.
- g) Figure J-3 shows the boundary for the South Atlantic Anomaly as defined in the Radiation DTO. The boundary defined by D008 is also shown.
- h) Scheduling criteria for the SAA as follows:
- . EVA exposure is minimized.
 - . The X-Ray Event Analyzer of S056 cannot operate in the SAA.
 - . The Flare Auto Switch of S054 should be inhibited when passing through the SAA.
 - . Film exposure to the SAA should be minimized.
 - . S020 and S063 performances should be minimized.
 - . D008 must operate in the SAA.
 - . Radiation DTO requires operation in the SAA.
- i) This Flight Plan has not kept account of film radiation accumulation, however, this will be done on future flight plans.
- j) Due to size of limitations on the summary timeline, time blocks less than 10 minutes cannot be plotted in the crewman columns. Affected experiment tasks are noted in the NOTES column.

Flight Plan Notes
Experiments

- k) The following table presents the times for a full moon for SL-2, SL-3, and SL-4:

Year	Month	Day	Hour (GMT)
1973	5	17	0654
1973	6	15	2042
1973	7	15	1224
1973	8	14	0512
1973	9	12	1942
1973	10	12	0700
1973	11	10	1636
1973	12	10	0136

2. Medical

- a) On SL-2, the PLT performs an additional M092/M171 on DOY 146 and the CDR performs an additional M092/M171 on DOY 147. On SL-3, the CDR and PLT perform an additional M092/M093 on DOY 263 and the SPT performs an additional M092/M093 on DOY 264.
- b) Criteria for meeting the 24-hour urine freezing requirement after launch are as follows:
- . Samples from individual urine collections will be frozen within 24 hours after urination. This 24-hour period is assumed to start at lift-off.
 - . The urine freezer in the OWS is chilled down prior to OWS activation.
 - . Urine may be processed only after WMC activation.
 - . One crewman will transfer all individual urine collections from the CM to OWS, obtain and identify samples, and store samples in the freezer.

Rendezvous of M=5 through M=8 are acceptable operationally: however, any delay past M=5 or delay in OWS entry will impact crew procedures, but the 24-hour requirement will not be violated. Urine will be placed in the freezer at the following times:

Mission	GET- HR:MIN
SL-2	23:15
SL-3	12:16
SL-4	TBS

Flight Plan Notes
Experiments

3. ATM

- a) Since specific solar behavior cannot be predicted prior to flight, blocks of time have been allocated throughout the timeline to meet anticipated ATM requirements, considering all other demands on the flights.
- b) No less than 15 minutes of ATM usable observing is scheduled in any one time block.
- c) The crewman performing ATM should be at the ATM console 75 percent of the ATM night periods.
- d) ATM operations will require realtime re-scheduling during the mission to accommodate actual solar activity as it occurs,
- e) In realtime scheduling, reasonable adjustments to the crew schedule (for all activities except sleep periods) may be made to accommodate realtime flare occurrences.
- f) S063-2 should be scheduled 1 to 3 days after a realtime flare.
- g) S149 should be scheduled 1 to 2 days after a realtime flare.
- h) ATM requires 5 daylight passes for checkout on SL-2 and 4 daylight passes for checkout on SL-3 and SL-4.
- i) A TBD rocket will be launched from White Sands on DOY 128 and DOY 134 of SL-2 (SL-3, TBS) for calibration purposes and must be correlated with ATM observation.
- j) Lagranian point L4 will be available for ATM observation on June 26, 0000 to 0800 GMT and L5 available for observation on July 15, 1700 to 2400 GMT.
- k) All ATM experiment power must be off (except TCS) during EVA in order to ensure safety for the crewman. However, prior to ingress, power will be turned on to insure nominal operation.
- l) Table J-4 presents a description of ATM Joint Observing Programs (JOPS) and Building Blocks (BB).
- m) Tables J-5 and J-6 present JOP/BB correlation and BB/Experiment correlation respectively.

Flight Plan Notes
Experiments

4. EREP

- a) EREP sensors (S190, S191, S192, S193, S194) are scheduled collectively as it is assumed they all can operate at some time during any EREP pass. The SPT crewman is shown performing "S190B" or "S063". For the first three EREP passes, he is shown as "EREP" which consists of assisting with the Z-LV maneuver, monitoring and checkout of equipment.
- b) S063-1 requires 120⁰ Z-LV passes for operation. EREP Z-LV times (it not 120⁰ passes) are increased to meet the S063 constraint.
- c) EREP selection logic for scheduling is as follows:
 - . For Solar Inertial passes, $-10^{\circ} \leq \beta \leq +10^{\circ}$.
 - . For 60⁰ passes, $-50^{\circ} \leq \beta \leq +50^{\circ}$. For 120⁰ passes $-30^{\circ} \leq \beta \leq +50^{\circ}$.
 - . For S190 F01, S192 F0's, 1, 2, 3, 5, 6, and S194 F01, the sun elevation angle must be greater than 20⁰ for the winter hemisphere and greater than 30⁰ for the summer hemisphere.
 - . Each scheduled pass should have at least one schedulable backup pass, which does not necessarily have to be another scheduled pass.
 - . Backup passes cover same ground track as prime passes due to repeatability of ground track every 71 revolutions.
 - . Crew available time is considered to be 1300-0300 GMT.
 - . "EAT" period may be moved ± 1 hour to accommodate a pass.
 - . M092/93 and M092/171 may be moved ± 1 day or ± 1 hour to accommodate a pass.
 - . S190B is not scheduled for the first three EREP passes to allow the crewmen familiarization with the other EREP sensor operations.
 - . Weather probability not taken into account.
 - . 15, 25, and 20 Z-LV passes scheduled for SL-2, SL-3, and SL-4 respectively.
 - . 3 and 2 Solar Inertial passes scheduled for SL-3 and SL-4 respectively.

Flight Plan Notes Experiments

- Passes selected from a shopping list supplied for each mission by MSC MPAD personnel. Prime passes are highest ranking.
 - Test sites and test conditions based on EREP Detailed Test Objective document, dated November 1970.
 - The OA must be in Z-LV at least 4 minutes prior to EREP data take.
- d) Due to activation, the earliest EREP may be scheduled is as follows:

Mission	DOY
SL-2	124
SL-3	212
SL-4	TBS

- e) Due to deactivation, the latest EREP may be scheduled is as follows:

Mission	DOY
SL-2	147
SL-3	265
SL-4	TBS

- f) Table J-7 gives parameters for those passes scheduled in the timeline for SL-2 and SL-3.
- g) Table J-8 shows EREP prime and backup revolutions available for SL-2 and SL-3. The DOY is for primes. A backup occurs 5 days later (71 revolutions). Data is from MPAD shopping list.
- h) Figure J-4 gives a pictorial representation of the EREP passes for SL-2 and SL-3. Traces are when data taking begins and ends.
- i) Figure J-5 shows typical EREP pointing sequences and Z-LV capabilities.

5. Individual

- a) M415 and S150 require no crew interface.
- b) S071/S072 (SL-3) require a crewmember only to remove electrical power from the experiments upon completion. Any other operations would be as a backup to ground commands. The ground schedules data dumps and commands the experiments during operation.

Flight Plan Notes
Experiments

- c) S149 will be set up and stowed by the crew (two ~~men~~, but will be commanded by the ground during operation. The crew will perform post-operation tasks and will be the backup for onboard activation. The crew must advise ~~the~~ ground when the experiment is ready for activation.
- d) S063-2 desires coordination with high altitude flights.
- e) S020 doors must be closed during orbital night in periods of contamination.
- f) All experiment film should be in the film vault at the end of each day. See Section 4 of this document for other film constraints.
- g) Programs are being developed to show viewing opportunities for the following experiments: S019, S063, S073/T027, S183, and T002. The program will show by mission day and GMT time the opportunities for viewing the experiment's particular area of interest. Figures J-6 and J-7 are examples of experiment viewing opportunities for S073/T027 mode 1-e (fixed position, celestial pole). The objective of this mode is to sight on the celestial poles, with simultaneous observations to the North Pole from a ground station. Constraints used to develop figure J-6 are as follows (each constraint is shown on the graphs):

- . North celestial pole within spacecraft field of view.
- . Experiment to be performed within one week of new moon.
- . Simultaneous observation by spacecraft and a ground station. Ground observations to be made from Maui Island, Hawaiian Islands; (a) during the darkness between astronomical twilights and, (b) when the Moon is not above the horizon.
- . Experiment to be conducted from the anti-solar airlock.
- . Spacecraft observations at times other than when spacecraft is in Earth's shadow are permitted.

Constraints used to develop figure J-7 for the South celestial pole are the same as above except for ground observations.

For the spacecraft, with location continually changing and instrument freedom restricted to a half-cone angle of 112.5 degrees, each celestial pole is visible for a 60.5 minute period of each orbit (i.e., it is not visible for a 32.8 minute period of each orbit).

Flight Plan Notes
Experiments

- h) The constraint for S063-2 (twilight airglow) is that the spacecraft be in such a position that the depression angle of the sun is between 26.5 degrees and 37.5 degrees. In general, there are two twilight periods per orbit; sunset twilight and sunrise twilight, each occurring during orbital darkness. Figure J-8 shows the case when the beta angle is so high that the twilight periods have merged into one long period. Figure J-9 shows the total time per orbit that depression angles are suitable for the experiment.
- i) The viewing times for experiment S019 are subject to the following constraints:
- . Nighttime on spacecraft
 - . Starfield within gimbal capability of mirror system
 - . Moon less than 50% illuminated or more than 90° from starfield

The twenty-six criterion A starfields (See MRD) were tested, using these constraints, for the period of SL-2. Figure J-10 shows that at the beginning of the period, there will be eight starfields viewable, and these will be viewable for the entire night period. On May 10, the Moon will become more than 50% illuminated and most of the starfields cannot be used for observation. An exception is starfield 192, which is more than 90 deg from the Moon and hence acceptable. From May 25 on, the viewable starfields increase again to eight, but these will be viewable for less than the full night period.

- j) Identical constraints exist for experiment S183. Figure J-11 shows viewing times for twenty-four starfields more or less evenly distributed around the galaxy.

Flight Plan Notes
Experiments

Experiment	Experiment Title	Order of Preference	Functional Objectives Completed	Functional Objectives Not Completed	Remarks
M071	Mineral Balance	N/A	1-5	--	
M073	Bioassay of Body Fluids	N/A	1	--	
M074	Specimen Mass Measurement	N/A	1	--	
M092/M093	Inflight LBNP/Vectorcardiogram	N/A	1	--	
M092/M171	Inflight LBNP/Metabolic Activity	N/A	1	--	
M110	Inflight Blood Sampling	N/A	TBD	--	4 Performances/ crewman
M131	Human Vestibular Function	N/A	1,2	--	
M133	Sleep Monitoring	N/A	1	--	15 Runs scheduled
M151	Time and Motion Study	N/A	1-6	--	
M172	Body Mass Measurement	N/A	1	--	
ATM	---	N/A	--	--	105:00 Hrs. Observation
EREP	---	N/A	15 Passes	--	
S190B	Earth Terrain Camera	N/A	10 Passes	--	
M487	Habitability/Crew Quarters	450	1-9	--	2-9 at CMN convenience
T003	In-Flight Aerosol Analysis	400	1-4	--	4 at CMN convenience
M512 (M551-M555)	Metals Processing in Space	390	1	--	F0-1 completed for each experiment
T027	Sample Array	380	1	--	Deployed 139 hrs.

Table J-1a. Experiment Scheduling Accomplishment (SL-2)
(Sheet 1 of 3 sheets)

Flight Plan Notes
Experiments

Experiment	Experiment Title	Order of Preference	Functional Objectives Completed	Functional Objectives Not Completed	Remarks
M509	Astronaut Maneuvering Equipment	380	1-4	--	
S019	UV Stellar Astronomy	290	1-12	--	
M479	Zero Gravity Flammability	280	--	1-5	SL-4
S073/ T027	Gegenschein/Zodiacal Light/ Contamination Measurement	270	22 scans	23 scans	Insufficient SAL and crew time
D024	Thermal Control Coatings	260	1	--	
S183	UV Panorama	250	1-12	--	
D008	Radiation in Spacecraft	240	1,2	--	
S015	Zero Gravity Single Human Cells	230	1,2	--	
S009	Nuclear Emulsion	220	1	--	Deployed 573 hrs.
T013	Crew Vehicle Disturbances	210	--	1	SL-3
T020	Foot Controlled Maneuvering Unit	200	--	1,2	SL-3/SL-4
T025	Coronagraph Contamination Measurements	180	TBD	--	5 Runs
S149	Particle Collection	140	1,2	--	F01 deployed 35 hrs
S020	X-Ray/UV Solar Photography	130	--	1,2	SL-3
S063	UV Airglow Horizon Photography	120	1	1,2	2 Passes of F0-1
M516	Crew Activity/Maintenance Study	110	1-6	--	Insufficient crew time
T002	Manual Navigation Sightings	100	--	1-6	
M415	Thermal Control Coatings	N/A	1-4		Passive

Table J-1a. Experiment Scheduling Accomplishment (SL-2)
(Sheet 2 Of 3 sheets)

Experiment	Experiment Title	Order of Preference	Functional Objectives Completed	Functional Objectives Not Completed	Remarks
---	Operational Radiation Measurements	TBD	1-6	--	2-5 Require no crew interface
DT0 20.10	Environmental Microbiology	TBD	1-3	--	
---	Student Experiment	TBD	TBD	--	1½ Hours/Week
---	Educational TV	TBD	TBD	--	1 Hour/Week

Table J-1a. Experiment Scheduling Accomplishment (SL-2)
(Sheet 3 of 3 sheets)

Flight Plan Notes
Experiments

Experiment	Experiment Title	Order of Preference	Functional Objectives Completed	Functional Objectives Not Completed	Remarks
M071	Mineral Balance	N/A	1-5	--	
M073	Bioassay of Body Fluids	N/A	1	--	
M074	Specimen Mass Measurement	N/A	1	--	
M092/M093	Inflight LBNP/Vectorcardiogram	N/A	1	--	Extra run on all 3 CMN
M091/M171	Inflight LBNP/Metabolic Activity	N/A	1	--	
M110	Inflight Blood Sampling	N/A	TBD	--	8 performances/CMN
M131	Human Vestibular Function	N/A	1,2	--	
M133	Sleep Monitoring	N/A	1	--	21 runs scheduled
M151	Time and Motion Study	N/A	1-5	--	
M172	Body Mass Measurement	N/A	1	--	
ATM	----	N/A	--	--	238:20 Hrs. Observation Includes 3 SI passes
EREP	----	N/A	28 passes	--	Includes 3 SI passes
S190B	Earth Terrain Camera	N/A	15 passes	--	Includes 3 SI passes
M487	Habitability/Crew Quarters	450	1-9	--	2-9 at CMN convenience
T003	Inflight Aerosol Analysis	400	1-4	--	4 at CMN convenience
M509	Astronaut Maneuvering Equipment	380	1-4	--	

Table J-1b. Experiment Scheduling Accomplishment (SL-3)
(Sheet 1 of 3 sheets)

Flight Plan Notes
Experiments

Experiment	Experiment Title	Order of Preference	Functional Objectives Completed	Functional Objectives Not Completed	Remarks
S019	UV Stellar Astronomy	290	1-13	--	Additional pass to meet observing time
S073/T027	Gegenschein/Zodiacal Light/Contamination Measurement	270	1-10	--	
D024	Thermal Control Coatings	260	1	--	
S183	UV Panorama	250	1-12	--	
T013	Crew Vehicle Disturbances	210	1	--	
T020	Foot Controlled Maneuvering Unit	200	1,2	--	
T025	Coronagraph Contamination Measurement	180	TBD	--	
S149	Particle Collection	140	2,3,4	--	
S020	X-Ray/UV Solar Photography	130	1,2	--	
S063	UV Airglow Horizon Photography	120	1,2	--	
M516	Crew Activity/Maintenance Study	110	1-5	--	
T002	Manual Navigation Sightings	100	6	1-5	
S071/S072	Circadian Rhythm - Pocket Mice/Vinegar Gnat	N/A	1,2	--	
---	Operational Radiation Measurements	TBD	1-6	--	

Table J-1b. Experiment Scheduling Accomplishment (SL-3)
(Sheet 2 of 3 sheets)

Experiment	Experiment Title	Order of Preference	Functional Objectives Completed	Functional Objectives Not Completed	Remarks
DT0 20.10	Environmental Microbiology	TBD	1-3	--	
---	Student Experiments	TBD	TBD	--	1½ Hours per week
---	Educational TV	TBD	TBD	--	1 Hour per week

Table J-1b. Experiment Scheduling Accomplishment (SL-3)
(Sheet 3 of 3 sheets)

Flight Plan Notes
Experiments

Experiment and Functional Objective	Total Time		Experiment and Functional Objective	Total Time	
	Man Hr:Min Per Mission	Mission Hr:Min Per Mission		Man Hr:Min Per Mission	Mission Hr:Min Per Mission
MEDICAL					
M071 1-5	53:35	53:35	S190A		
M073	N/A	N/A	S191		
M074	03:00	03:00	S192	51:40	24:13
M172	02:48	02:48	S193		
M092/M093	35:46	17:53	S194	10:15	10:15
M092/M171	84:42	42:21	S190B		
M131 1	24:00	12:40	EREK TOTAL	61:55	34:28
2	13:24	06:42	INDIVIDUAL		
M133	06:45	06:45	D008		
M110	04:00	04:00	1	01:30	01:30
M151 SU	01:00	01:00	2	01:30	01:30
1-6	TIME INCLUDED IN OTHER EXPERIMENTS		D024	INCLUDED IN EVA TIME	
MEDICAL TOTAL	229:00	150:44	MB10		
ATM					
S052	167:27	167:27	1	01:00	01:00
S054	TOTAL ATM TIME ALLOCATED (INCLUDES		2	00:36	00:12
S055	SETUP AND NIGHT CYCLES)		3	00:30	00:30
S056	105:00	105:00	M415		
S082A			1-4	NO ASTRONAUT ACTIVITY	
S082B			M487		
OBSERVING TIME			1	01:41	01:41
			2-9	AT CREWMAN CONVENIENCE	
			M509		
			SU		
			1	00:50	00:50
			2	04:50	02:25
			3	05:44	02:52
			4	07:06	03:40
				04:35	02:35

Table J-2a. Experiment/Crew Utilization Summary (SL-2)
(Sheet 1 of 3 sheets)

Flight Plan Notes
Experiments

Experiment and Functional Objective	Total Time		Experiment and Functional Objective	Total Time	
	Man Hr:Min Per Mission	Mission Hr:Min Per Mission		Man Hr:Min Per Mission	Mission Hr:Min Per Mission
INDIVIDUAL (CONT'D)					
M551	01:24	01:24	S149	01:05	00:40
M552	00:14	00:14	1, SU	01:00	00:35
M553	00:45	00:45	1, ST	01:20	00:40
M554	00:20	00:20	2, SU		
M555	00:22	00:22			
M516	INCLUDED IN OTHER ACTIVITIES		S183		
RAD			SU	03:46	01:53
1	01:40	01:40	1-12	07:00	07:00
2-5	NO CREW INTERFACE		ST	04:30	02:15
6					
S009	01:47	01:47	T003		
S015			1	10:40	10:40
1	01:24	01:24	2	00:28	00:28
2	01:00	01:00	3	00:48	00:48
S019			4	AT CREWMAN CONVENIENCE	
SU			T025		
1-12	01:41	01:41	SU	02:46	02:46
ST	07:03	07:03	TBD	05:51	05:51
S063	02:08	02:08	ST	01:38	01:38
1	02:12	02:12	T027		
S073/T027			1, SU	01:34	00:47
SU			1, ST	02:04	00:57
1-10	04:53	02:34	INDIVIDUAL TOTAL	115:20	91:47
ST	12:05	12:05	PHYSICAL TRAINING	19:30	19:30
	05:10	02:35	STUDENT ACTIVITIES	06:00	06:00
			EDUCATIONAL TV	04:00	04:00
			UNSCHEDULED TIME (30 MIN)	02:11	02:11

Table J-2a. Experiment/Crew Utilization Summary (SL-2)
(Sheet 2 of 3 sheets)

Flight Plan Notes
Experiments

Experiment and Functional Objective	Total Time	
	Man Hr:Min Per Mission	Mission Hr:Min Per Mission
UNUSABLE TIME (< 30 MIN)	04:50	04:50
SYSTEMS HOUSE-KEEPING TIME	100:59	100:59
OFF DUTY TIME	47:35	----
MISSION PLANNING TIME	47:35	----
REST AND RECREATION TIME	47:15	----
EXPERIMENT TIME AVAILABLE	587:43	----

Table J-2a. Experiment/Crew Utilization Summary (SL-2)
(Sheet 3 of 3 Sheets)

Flight Plan Notes
Experiments

Experiment and Functional Objective	Total Time		Experiment and Functional Objective	Total Time	
	Man Hr:Min Per Mission	Mission Hr:Min Per Mission		Man Hr:Min Per Mission	Mission Hr:Min Per Mission
MEDICAL					
M071 1-5	107:34	107:34	S190A S191 S192 S193 S194 S190B	97:13	47:06
M073	N/A	N/A	ERE P TOTAL	14:38	14:38
M074	03:00	03:00	ERE P TOTAL	111:51	61:44
M172	03:00	03:00	INDIVIDUAL		
M092/M093	110:32	55:16	MBIO		
M092/M171	132:50	66:25	1	01:00	01:00
M110	04:00	04:00	2	00:36	00:12
M131			3	00:30	00:30
1	26:04	13:02	M487		
2	14:42	07:21	1	01:35	01:35
M133	09:45	09:45	2-9	AT CREW CONVENIENCE	
M151			M509		
1-6			1	05:04	02:32
TIME INCLUDED IN OTHER EXPERIMENTS			2	05:44	02:52
MEDICAL TOTAL	411:27	269:23	3	07:14	03:37
ATM			4	05:24	02:42
S052			M516	TIME INCLUDED IN OTHER EXPERIMENTS	
S054			RAD		
S055	396:05	396:05	1	02:10	02:10
S056	TOTAL ATM TIME ALLOCATED (INCLUDES SETUP AND NIGHT CYCLES)		2-5	NO CREW INTERFACE	
S082A			6	01:09	01:09
S082B	238:20	238:20			
OBSERVING TIME					

Table J-2b. Experiment/Crew Utilization Summary (SL-3)
(Sheet 1 Of 3 sheets)

Flight Plan Notes
Experiments

Experiment and Functional Objective	Total Time		Mission Hr:Min Per Mission
	Man Hr:Min Per Mission	Man Hr:Min Per Mission	
INDIVIDUAL (CONT'D)			
S019	01:50	01:50	01:50
SU	06:36	06:36	06:36
1-12	02:18	02:18	02:18
ST			
S020	02:39	02:39	02:39
SU	04:00	04:00	04:00
1	-----	-----	-----
2	02:04	02:04	02:04
ST			
S063	05:43	05:43	05:43
SU	04:56	04:56	04:56
1	16:37	16:37	16:37
2	03:38	03:38	03:38
ST			
S071/S072	NO CREW INTERFACE		
1			
2			
S073/T027	11:47	05:53	05:53
SU	39:48	39:48	39:48
1-10	13:45	06:52	06:52
ST			
S149	01:40	00:50	00:50
2,ST	01:40	00:50	00:50
3,SU	01:40	00:50	00:50
3,ST	01:40	00:50	00:50
4,SU			
INDIVIDUAL (CONT'D)			
S183	04:07	02:03	02:03
SU	08:34	08:34	08:34
1-12	06:28	03:14	03:14
ST			
T002	00:50	00:50	00:50
1	00:55	00:55	00:55
2	00:50	00:50	00:50
3	01:05	01:05	01:05
4	00:50	00:50	00:50
5	AT CREW CONVENIENCE		
6			
T003	11:12	11:12	11:12
1	00:40	00:40	00:40
2	02:00	02:00	02:00
3	AT CREW CONVENIENCE		
4			
T020	05:00	02:30	02:30
1	04:40	02:20	02:20
2			
T025	00:55	00:55	00:55
SU	12:35	12:35	12:35
TBD	00:47	00:47	00:47
ST			
INDIVIDUAL TOTAL		220:06	182:14
PHYSICAL TRAINING		44:44	44:44
STUDENT ACTIVITIES		12:05	12:05

Table J-2b. Experiment/Crew Utilization Summary (SL-3)
(Sheet 2 of 3 sheets)

Flight Plan Notes
Experiments

Experiment and Functional Objective	Total Time	
	Man Hr:Min Per Mission	Mission Hr:Min Per Mission
EDUCATIONAL TV	08:07	08:07
UNSCHEDULED TIME (≥ 30 MIN)	26:51	26:51
UNUSABLE TIME (< 30 MIN)	13:00	13:00
SYSTEMS HOUSE-KEEPING TIME	219:37	-----
OFF DUTY TIME	111:45	-----
MISSION PLANNING TIME	106:21	-----
REST AND RECREATION TIME	91:08	-----
EXPERIMENT TIME AVAILABLE	1773:07	-----

Table J-2b. Experiment/Crew Utilization Summary (SL-3)
(Sheet 3 of 3 sheets)

Flight Plan Notes
Experiments

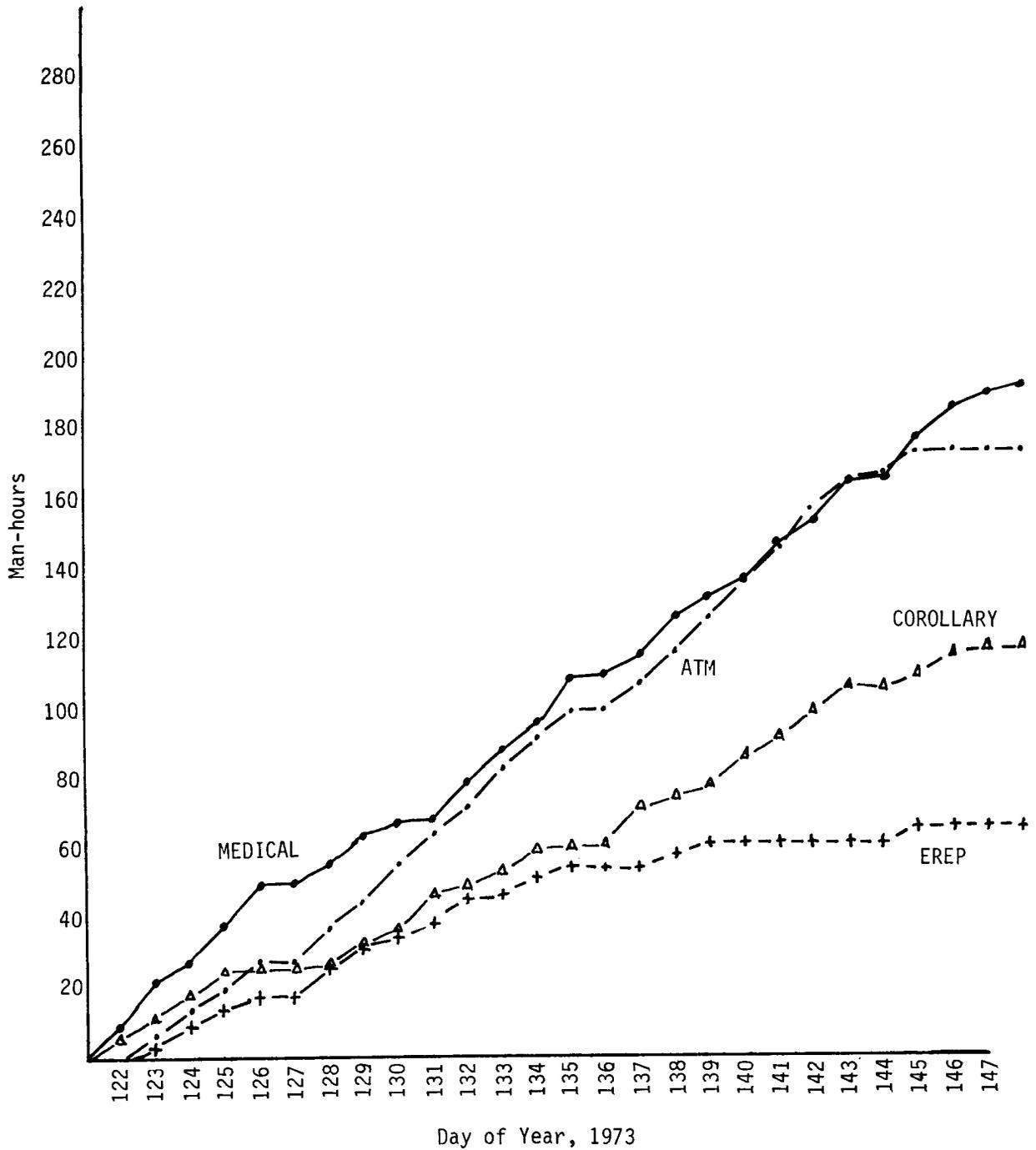


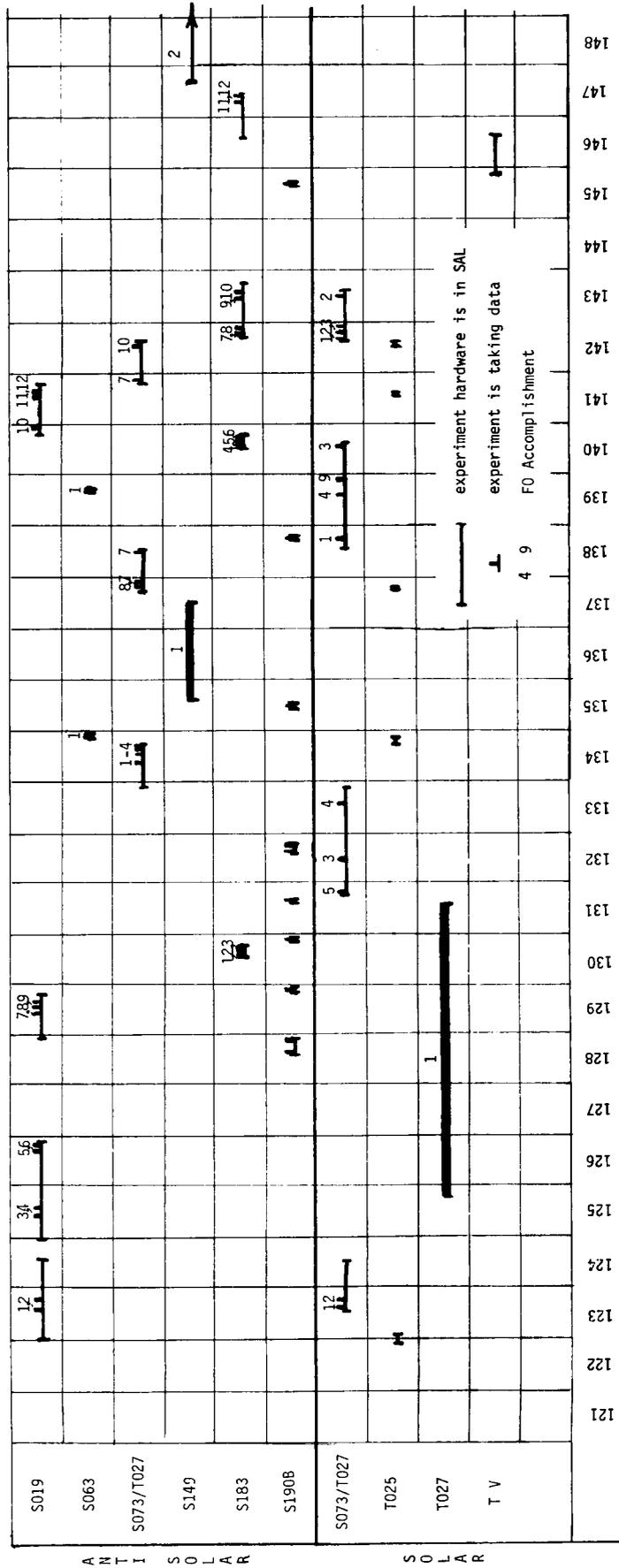
Figure J-1a. Man-hour Utilization (SL-2)

Flight Plan Notes
Experiments

TBS

Figure J-1b. Manhour Utilization (SL-3)

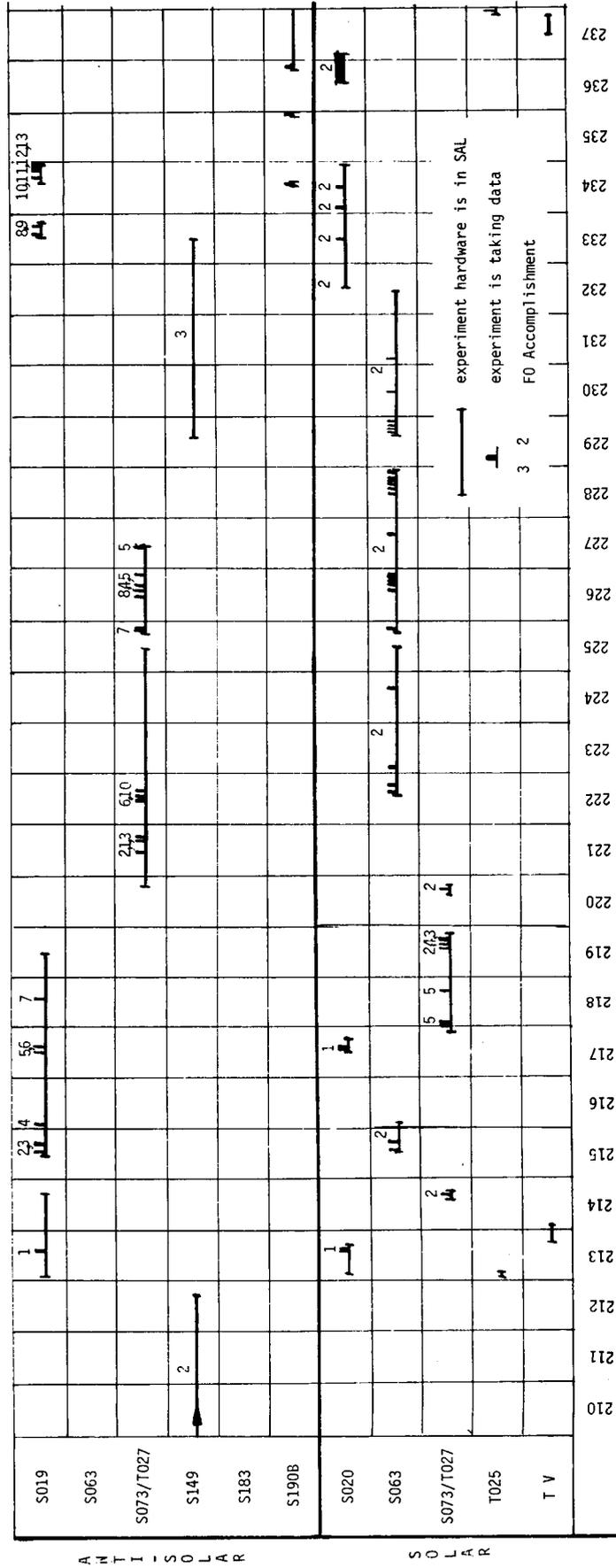
Flight Plan Notes
Experiments



DAY OF YEAR

Figure J-2b. Scientific Airlock Usage SL-2

Flight Plan Notes
Experiments



DAY OF YEAR

Figure J-2a. Scientific Airlock Usage SL-3
(Sheet 1 of 2 sheets)

A M T I - S O L A R

S O L A R

Flight Plan Notes
Experiments

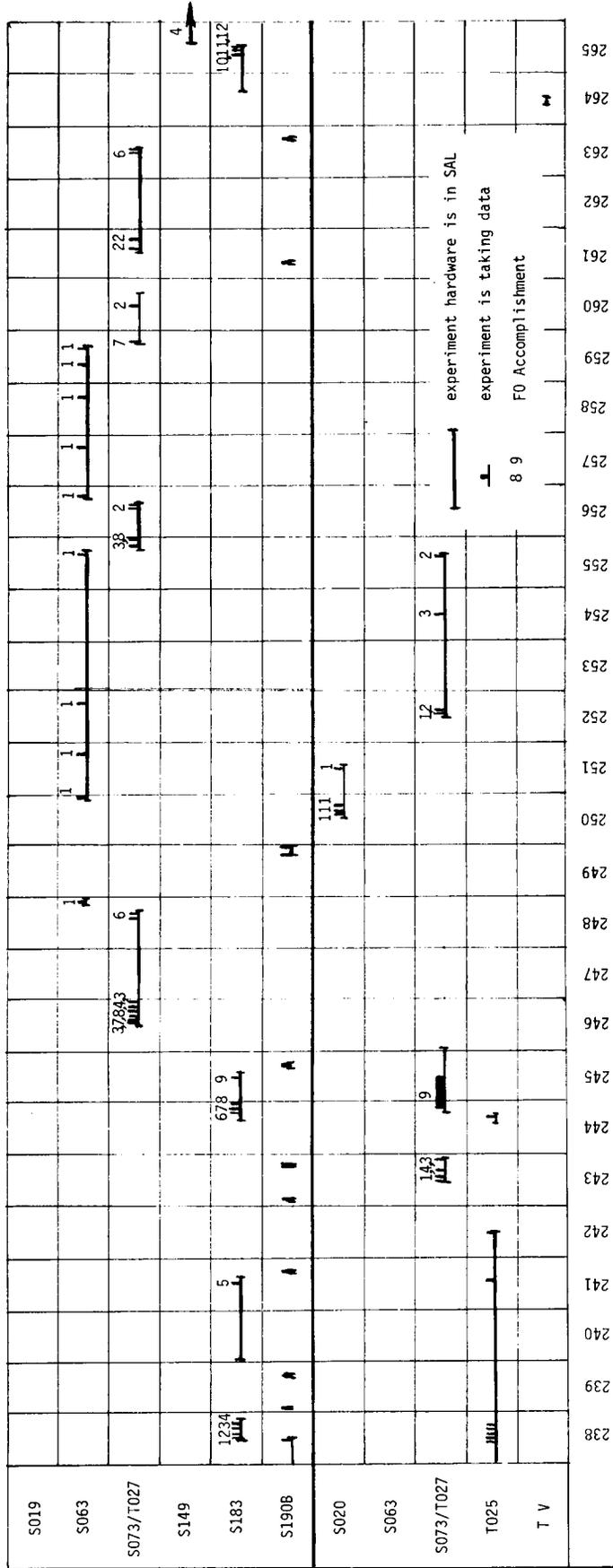


Figure J-2b. Scientific Airlock Usage SL-3
(Sheet 2 of 2 sheets)

ANTISOLAR

SOLAR

Flight Plan Notes
Experiments

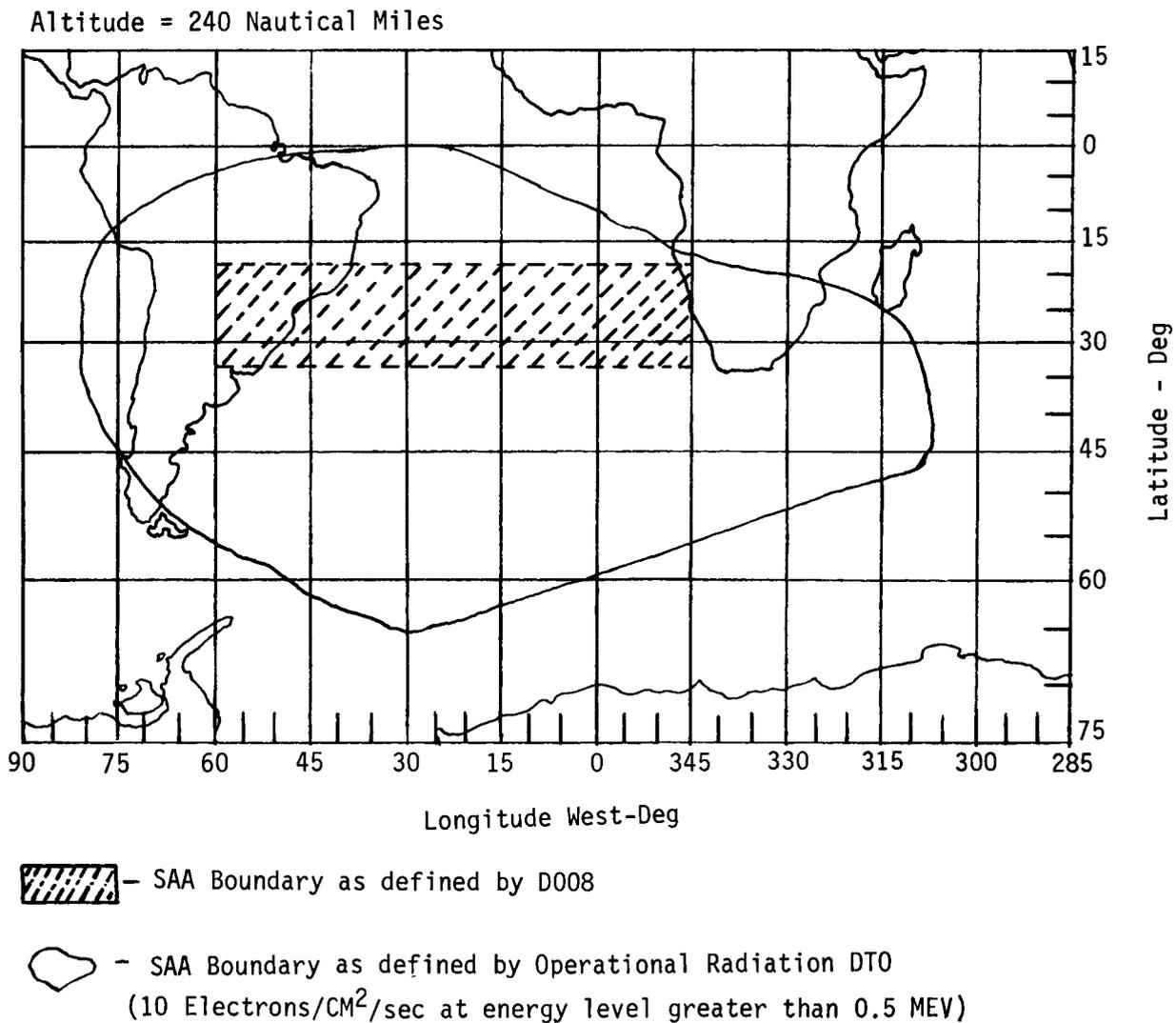


Figure J-3. SAA Boundary defined by D008 and radiation DTO

Flight Plan Notes
Experiments

Joint Observing Program	Building Block
1 Study of the Chromospheric Network and its Coronal Extension	1 Synoptic observations - sun centered
2 Active Regions	2 Corona
3 Flares	3 Spatial resolution, faint objects
4 Prominences and Filaments	4 Spatial and spectral resolution, bright features
5 Constant Latitude Studies	5 Spatial resolution, bright objects
6 Synoptic Observations of the Sun	6 Spectra
7 Atmospheric Extinction	7 Medium high time resolution
8 Coronal transients	8 Flare
9 Solar wind	9 Limb flare
10 Lunar Liberation Points, Lunar Calibration	10 Pre-flare, high time resolution
11 Chromospheric Oscillations and Heating	11 Spectra (film)
12 Program Calibration	13 Limb Study
13 Observation of Night Sky Objects	14 Corona background spectra
14 Solar Eclipse	15 Atmospheric extinction
	16 Coronal transients, fast
	17 Coronal transients, slow
	18 Chromosphere oscillations
	19 Velocities
	20 Night sky sources
	22 Calibration
	23 Active limb study

Table J-4. ATM Joint Observing Programs and Building Blocks

Flight Plan Notes
Experiments

JOP	FO	Building Block																						
		01	02	03	04	05	06	07	08	09	10	11	13	14	15	16	17	18	19	20	22	23		
1	1			X								X												
	2			X								X												
	3			X								X												
2	1		X		X	X	X			X			X									X		
	2		X		X	X	X			X			X											
	3		X		X	X	X			X			X											
	4		X		X	X	X			X			X											
	5		X		X	X	X	X			X			X										
3	1							X	X										X					
	2							X	X															
	3							X	X															
4	1		X	X									X	X										
	2		X	X									X	X										
	3		X	X									X	X										
5	1		X	X									X											
	2		X	X									X											
6	1																							
	2																							
7	1																							
	2																							
8	1																							
	2																							
	3																							
9	1	X	X																					
	2	X	X																					
10	1	X	X																					
	2	X	X																					
11	1		X	X																				
	2		X	X																				
12	1																							
	2																							
	3																							
13	1																							
	2																							
	3																							
14	1	NOT DEFINED AT THIS TIME																						

Table J-5. Joint Observing Program/Building Block Correlation

Flight Plan Notes
Experiments

Building Block	S052	S054	S055	S056	S082A	S082B	H α 1	H α 2
1	X	X	X	X	X		X	X
2	X	X	X	X	X		X	X
3		X	X	X	X	X	X	X
4		X	X	X	X	X	X	X
5		X	X	X	X	X	X	X
6		X	X	X	X	X	X	X
7		X	X	X	X	X	X	X
8		X	X	X	X	X	X	X
9	X	X	X	X	X	X	X	X
10		X	X	X	X	X	X	X
11		X	X	X	X	X	X	X
13		X	X	X	X	X	X	X
14		X	X	X	X	X	X	X
15	X	X	X	X	X	X	X	X
16	X	X	X	X	X	X	X	X
17	X	X	X	X	X	X	X	X
18		X	X	X	X	X	X	X
19		X	X	X	X	X	X	X
20	X	X	X	X	X	X	X	X
22		X	X	X	X	X	X	X
23		X	X	X	X	X	X	X

Table J-6. Building Block/Experiment Correlation

Flight Plan Notes
Experiments

PASS	REV NUMBER	TIME OF AOS (GMT)			DURATION OF PASS (MINUTES)	BETA ANGLE (°)	SUN ELEVATION ANGLE (°)		GROUND TRACK
		MO	DAY	HOUR			MIN	AOS	
1	59/60	5	4	19	28	-22.1	67.6	33.2	California to New Foundland
2	72/73	5	5	17	07	-24.6	57.4	53.3	Baja to Newfoundland
3	87/88	5	6	18	00	-27.1	57.3	46.8	California to Newfoundland
4	114/115	5	8	14	50	-30.4	11.6	59.6	Mexico to Nova Scotia
5	118/119	5	8	21	29	-30.9	52.0	14	British Columbia to Virginia
6	132/133	5	9	20	43	-31.6	58.2	24.2	British Columbia to Delaware
7	146/147	5	10	20	00	-32.0	57.9	31.0	British Columbia to New Jersey
8	160/161	5	11	19	18	-32.0	56.6	43.2	British Columbia to Mass.
9	172/173	5	12	15	17	-31.5	22.8	56.3	So. Calif. to New Foundland
10	175/176	5	12	20	11	-31.3	57.7	33.7	British Columbia to No. Carolina
11	205/206	5	14	21	37	-28.3	7.7	46.5	Japan, No. Pacific to Mexico
12	214/215	5	15	11	52	-27.1	56.13	29.9	France to Turkey
13	261/262	5	18	19	06	-17.7	55.6	63.4	Wash. State to Louisiana
14	275/276	5	19	18	22	-14.3	47.4	29.2	Wash. State to Brazil
15	361/362	5	25	17	17	-10.4	39	35	No. Calif. to Paraguay

Table J-7a. Prime EREP Passes (SL-2)

Flight Plan Notes
Experiments

PASS	REV NUMBER	TIME OF AOS (GMT)		DURATION OF PASS (MINUTES)	BETA ANGLE (°)	SUN ELEVATION ANGLE (°)		GROUND TRACK		
		MO	DAY			AOS	LOS			
1	1546	8	15	17	46	18	42.2	41	34	Chile to Brazil
2	1560	8	16	17	03	17	37.6	40	44	Chile to Brazil
3	1619/20	8	20	20	42	16	17.9	69	24	Mexico to Nova Scotia
4	1645	8	22	14	22	32	9.6	31	27	Chile to Hungary
5	1677/78	8	24	21	06	14	-1.2	69	13	So. Calif. to Newfoundland
6	1701	8	26	11	44	18	-8.3	62	47	Liberia to Greece
7	1719/20	8	27	18	54	16	-14.4	75	34	Mexico to Newfoundland
8	1747/48	8	29	17	24	16	-22.5	50	56	Mexico to New York
9	1767	8	31	01	58	17	-27.8	33	62	Sumatra to Japan
10	1777/78	8	31	19	19	13	-30.4	57	36	No. Calif. to Newfoundland
11	1805/06	9	2	17	52	13	-36.5	54	45	So. Calif. to Newfoundland
12	1850/51	9	5	20	36	15	-42.8	48	15	British Columbia to New Jersey
13	1864/65	9	6	19	53	14	-43.6	48	21	British Columbia to New Jersey
14	1866/67	9	6	23	08	8	-43.7	41	21	No. Calif. to Mexico
15	1880/81	9	7	22	25	13	-43.7	44	13	Oregon to Panama
16	1893/94	9	8	20	04	14	-43.4	47	32	British Columbia to No. Carolina
17	1907/08	9	9	19	21	14	-42.4	45	37	British Columbia to Virginia
18	1949/50	9	12	17	10	13	-36.1	28	56	British Columbia to Connecticut
19	1965/66	9	13	19	42	20	-32.8	49	34	Oregon to Brazil
20	1979/80	9	14	18	58	24	-29.5	43	30	Washington to Brazil

Table J-7b. Prime EREP Passes (SL-3)

(Sheet 1 of 2 sheets)

PASS	REV NUMBER	TIME OF AOS (GMT)		DURATION OF PASS (MINUTES)	BETA ANGLE (°)	SUN ELEVATION ANGLE (°)		GROUND TRACK
		MO	DAY			AOS	LOS	
21	1993/94	9	15 18	22	-25.9	40	41	Washington to Brazil
22	2003	9	16 09	20	-23.5	35	57	France to Saudi Arabia
23	2007/08	9	16 17	14	-22.2	32	68	Washington to Florida
24	2030	9	18 06	24	-16.0	58	27	Afghanistan to Australia
25	2065	9	20 17	10	- 4.6	24	62	Calif. to Mexico

Table J-7b. Prime EREP Passes (SL-3)

(Sheet 2 of 2 sheets)

Flight Plan Notes
Experiments

SL-2			SL-3		
DOY	Prime Rev.	Backup Rev.	DOY	Prime Rev.	Backup Rev.
124	59	130,201	227	1546	1617
125	72	143	223	1560	1631
126	87	158	232	1619	1690, 1731
128	114	185	234	1645	1716, 1787
128	118	189, 260, 331	236	1677	1748, 1819
129	132	203, 274, 345	238	1701	1772, 1843
130	146	217, 288, 359	239	1719	1790, 1861
131	160	231, 302	241	1747	1818
132	172	None	243	1767	1838
132	175	246, 317	243	1777	1848
134	205	276, 347	245	1805	1876
135	214	285	248	1850	1921, 1992
138	261	232	249	1864	1935, 2006
139	275	346	249	1866	1937, 2008
145	361	None	250	2880	1951, 2022
			251	1893	1964, 2035
			252	1907	1978, 2049
			253	1923	1994, 2065
			254	1932	2003
			254	1936	2007
			255	1949	2020
			256	1965	2036
			257	1979	2050
			258	1993	2064
			261	2030	None

Table J-8. Prime and Backup EREP Passes Available for SL-2 and SL-3

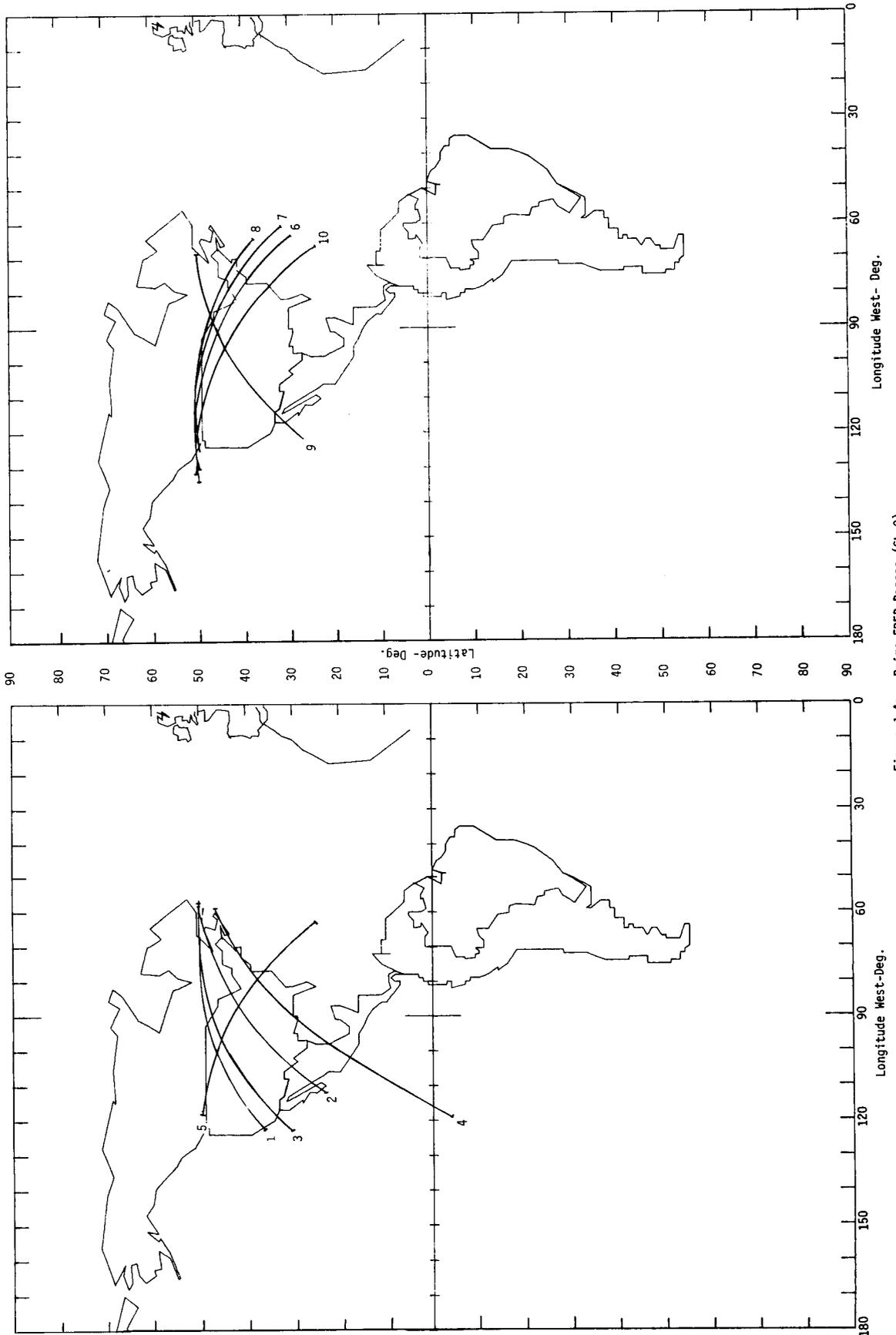


Figure J-4a. Prime EREP Passes (SL-2)
(Sheet 1 of 2 Sheets)

Flight Plan Notes
Experiments

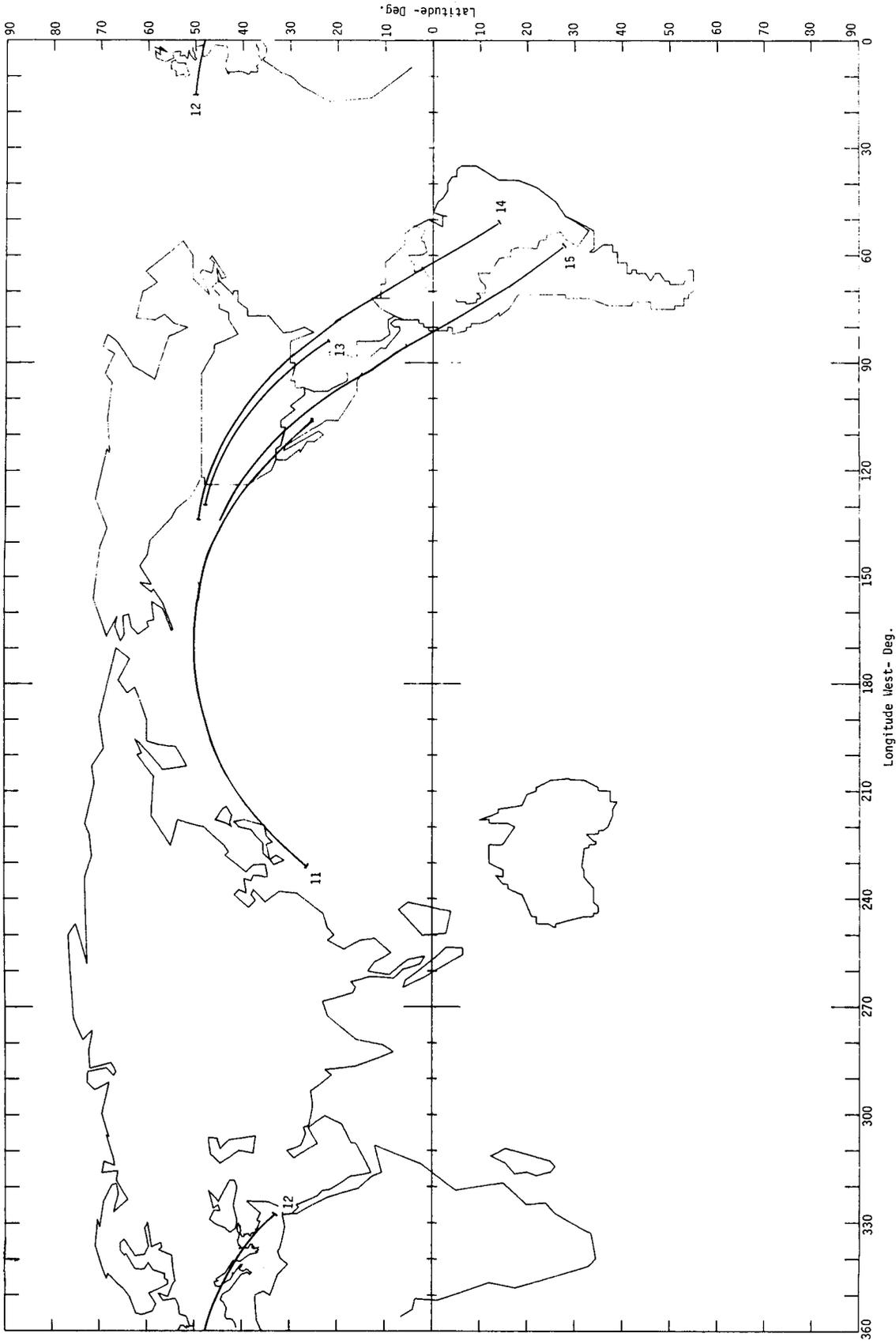


Figure J-4a. Prime EREP Passes (SL-2)
(Sheet 2 of 2 Sheets)

Flight Plan Notes
Experiments

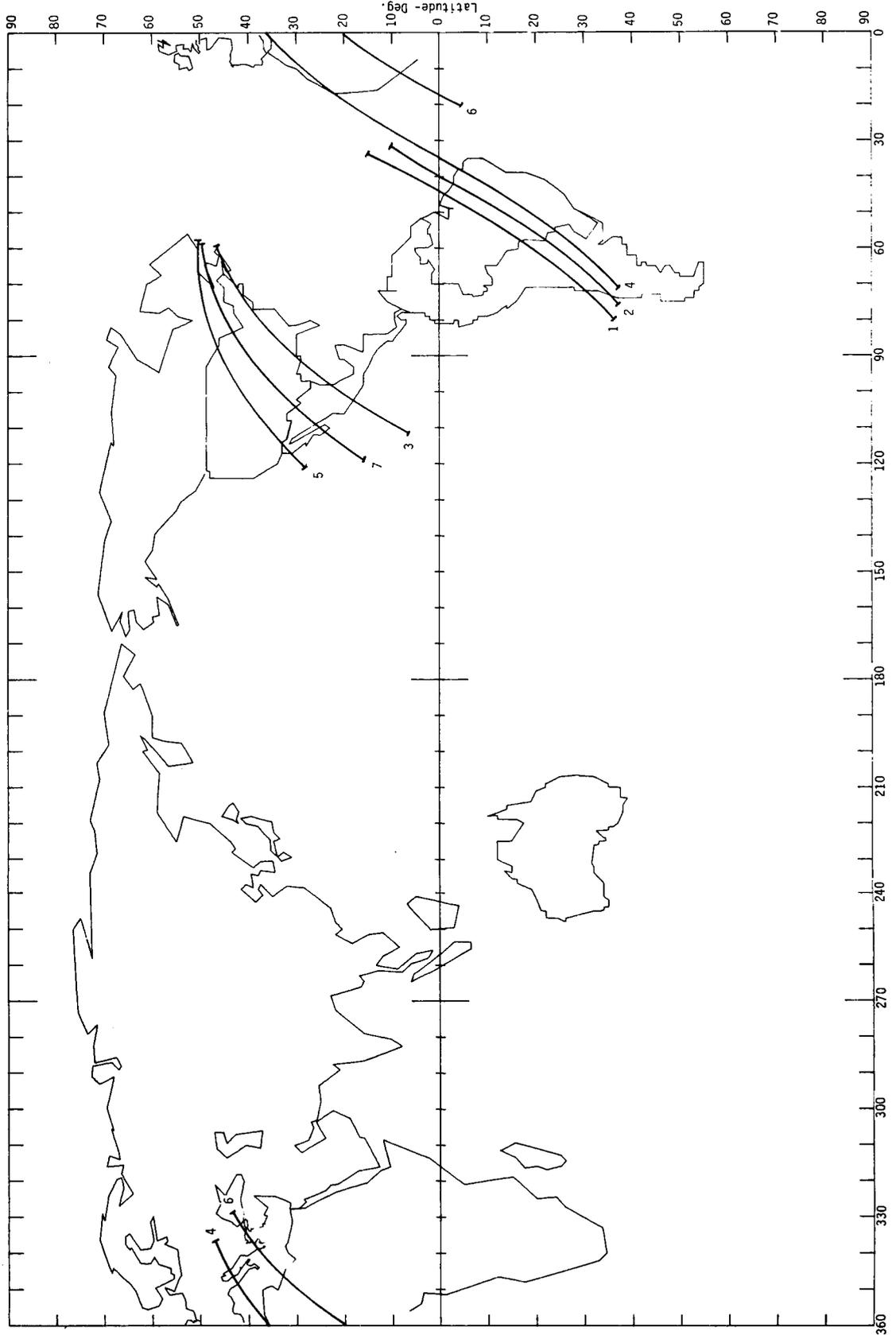


Figure J-4b. Prime EREP Passes (SL-3)
(Sheet 1 of 4 Sheets)

Flight Plan Notes
Experiments

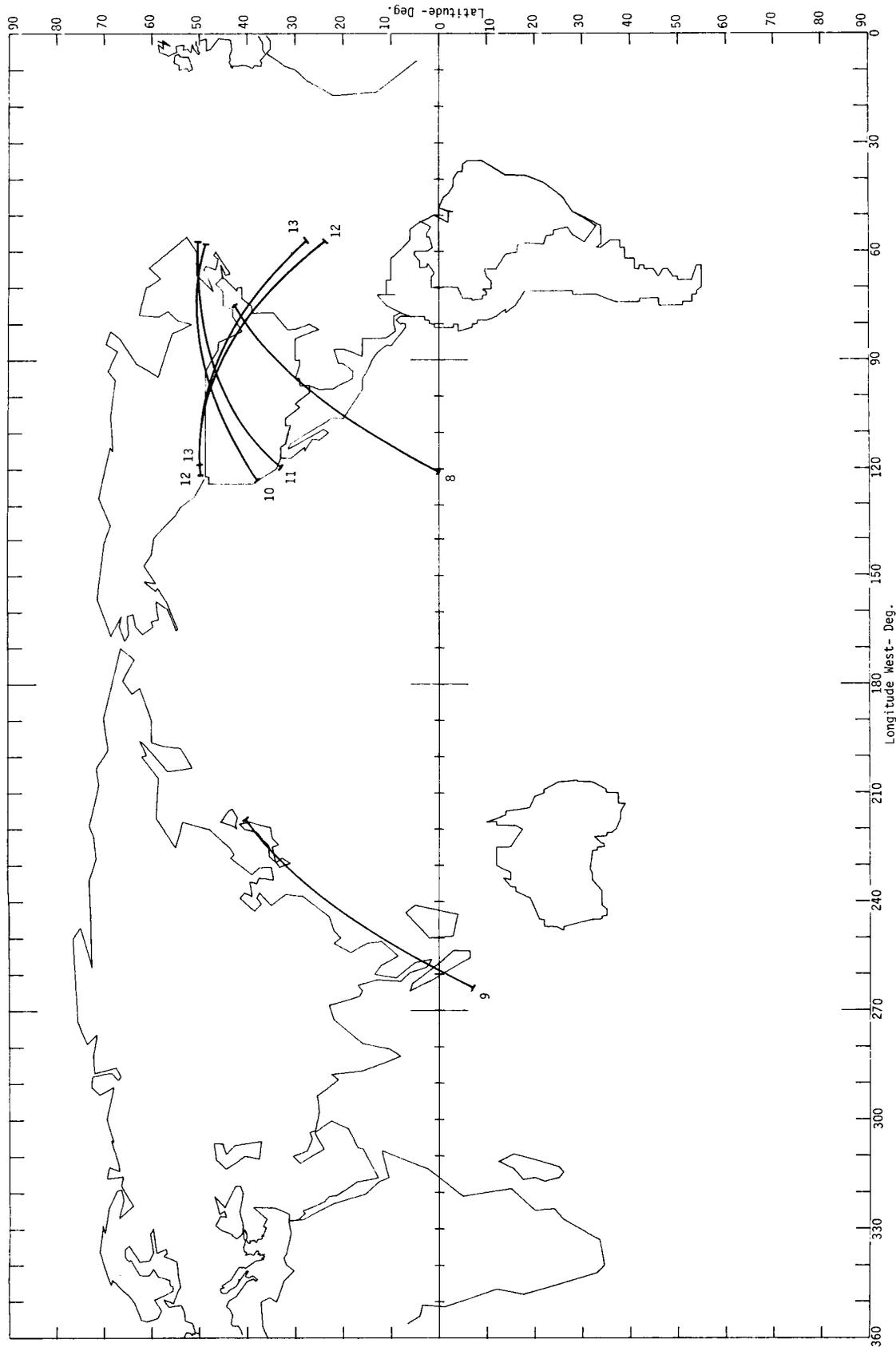


Figure J-4b. Prime EREP Passes (SL-3)
(Sheet 2 of 4 Sheets)

Flight Plan Notes
Experiments

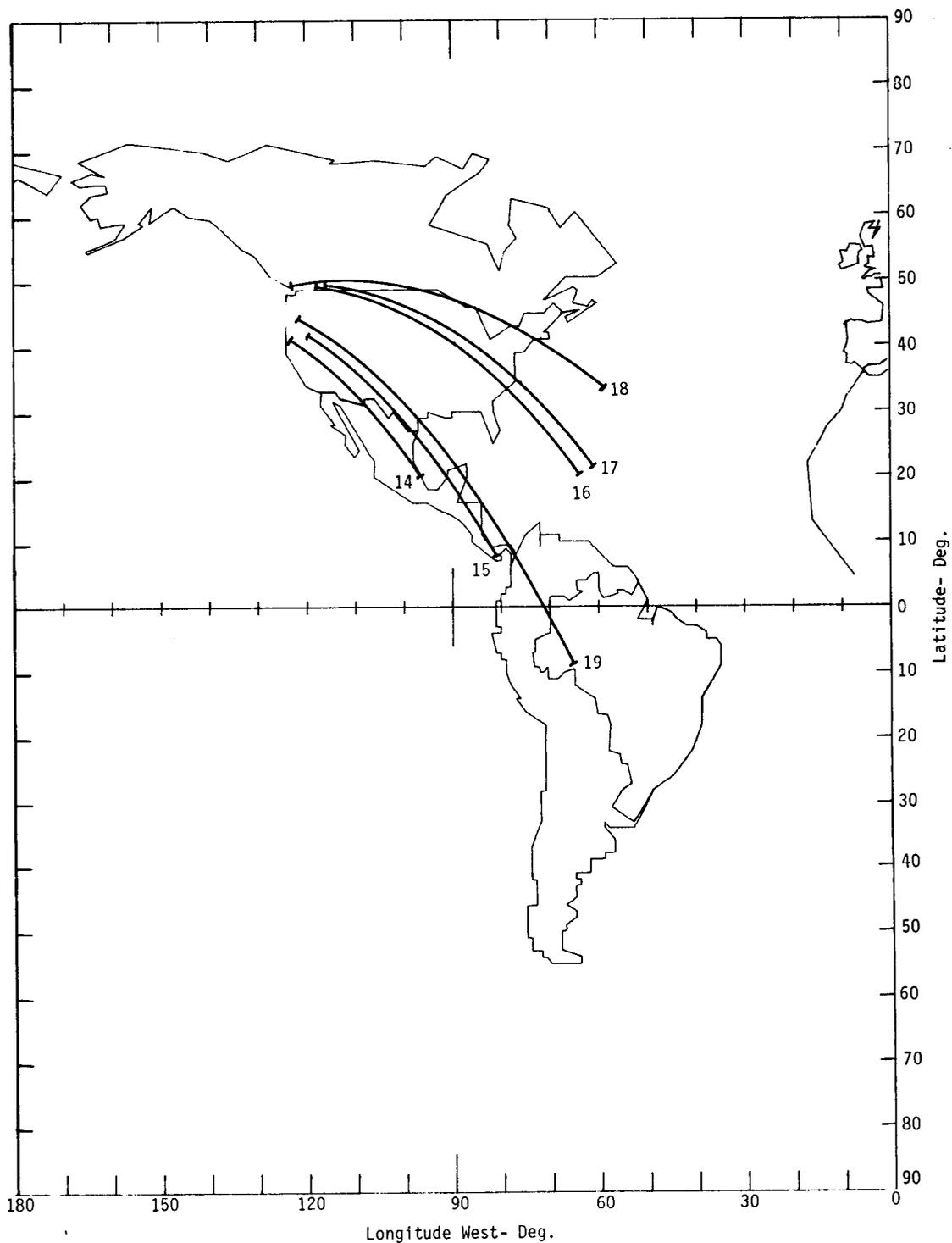


Figure J-4b. Prime EREP Passes (SL-3)
(Sheet 3 of 4 Sheets)

Flight Plan Notes
Experiments

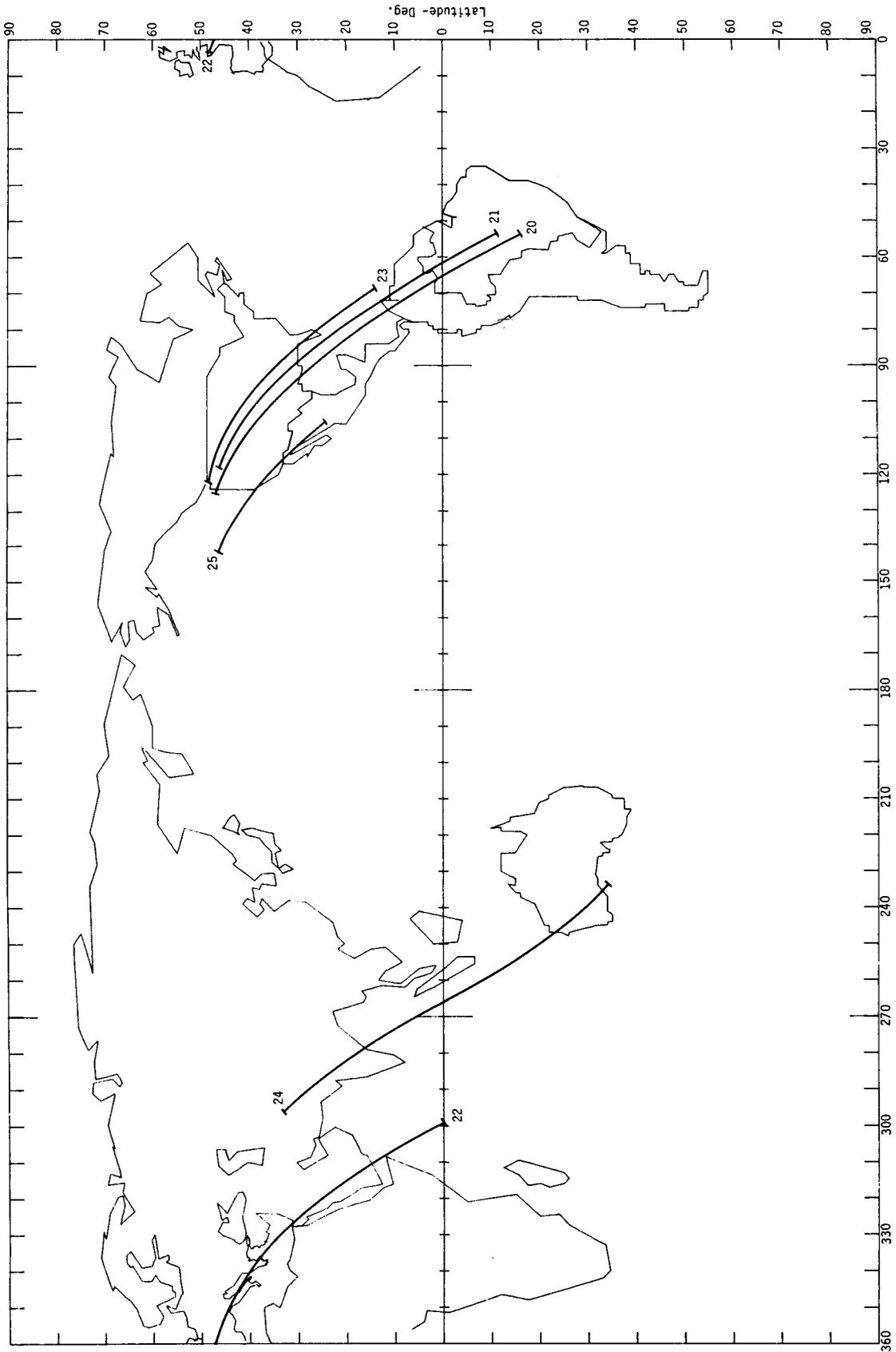
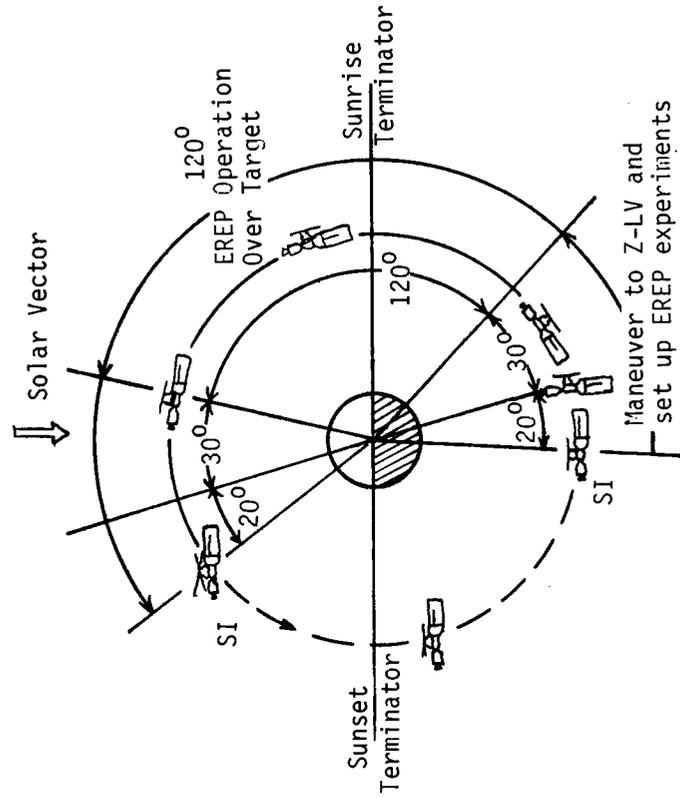


Figure J-4b. Prime EREP Passes (SL-3)
(Sheet 4 of 4 Sheets)

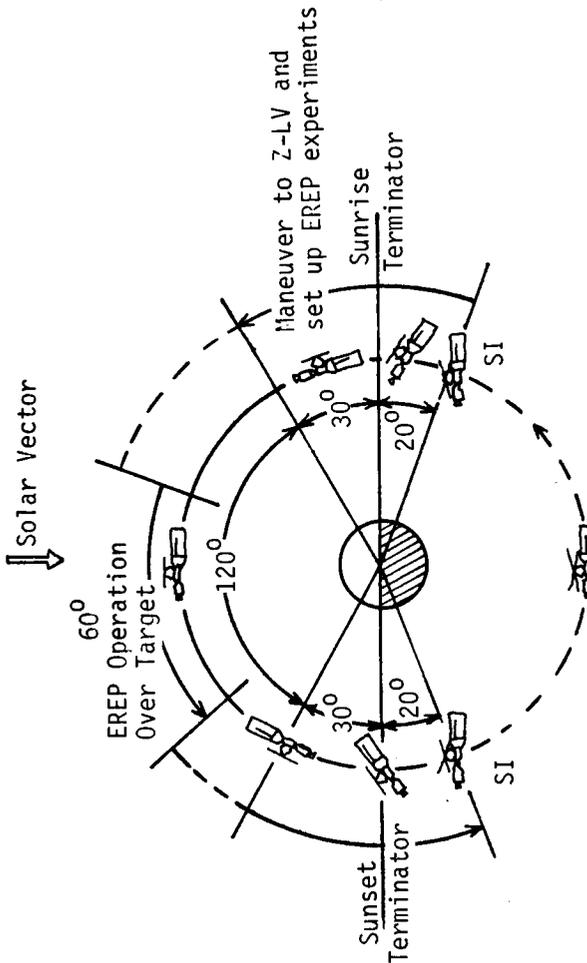


120 orbital degrees centered about any orbital location

This mode must not be performed for more than one orbit at a time.

After one pass in this mode, the SMS must remain in the solar-inertial mode for four continuous revolutions before initiating another Z-LV pass.

Restricted to β angles between -30 and $+50$ degrees



60 orbital degrees centered about any orbital location.

These Z-LV passes must not be performed during more than two consecutive revolutions.

If these Z-LV passes are contained within a 120-degree centered about orbital noon, the SMS must remain in the solar-inertial mode for at least four revolutions before two other consecutive Z-LV passes are initiated.

If the Z-LV passes are at any other orbital location, the SMS must remain in the solar-inertial mode for at least five revolutions before initiating additional Z-LV passes.

During a given 24-hour period, the SMS shall provide a maximum of four continuous sequences of one Z-LV pass followed by one revolution of the solar-inertial mode.

Restricted to β angles between -50 and $+50$ degrees

Figure J-5. Typical EREP Pointing Sequences and Z-LV Capabilities

Flight Plan Notes
Experiments

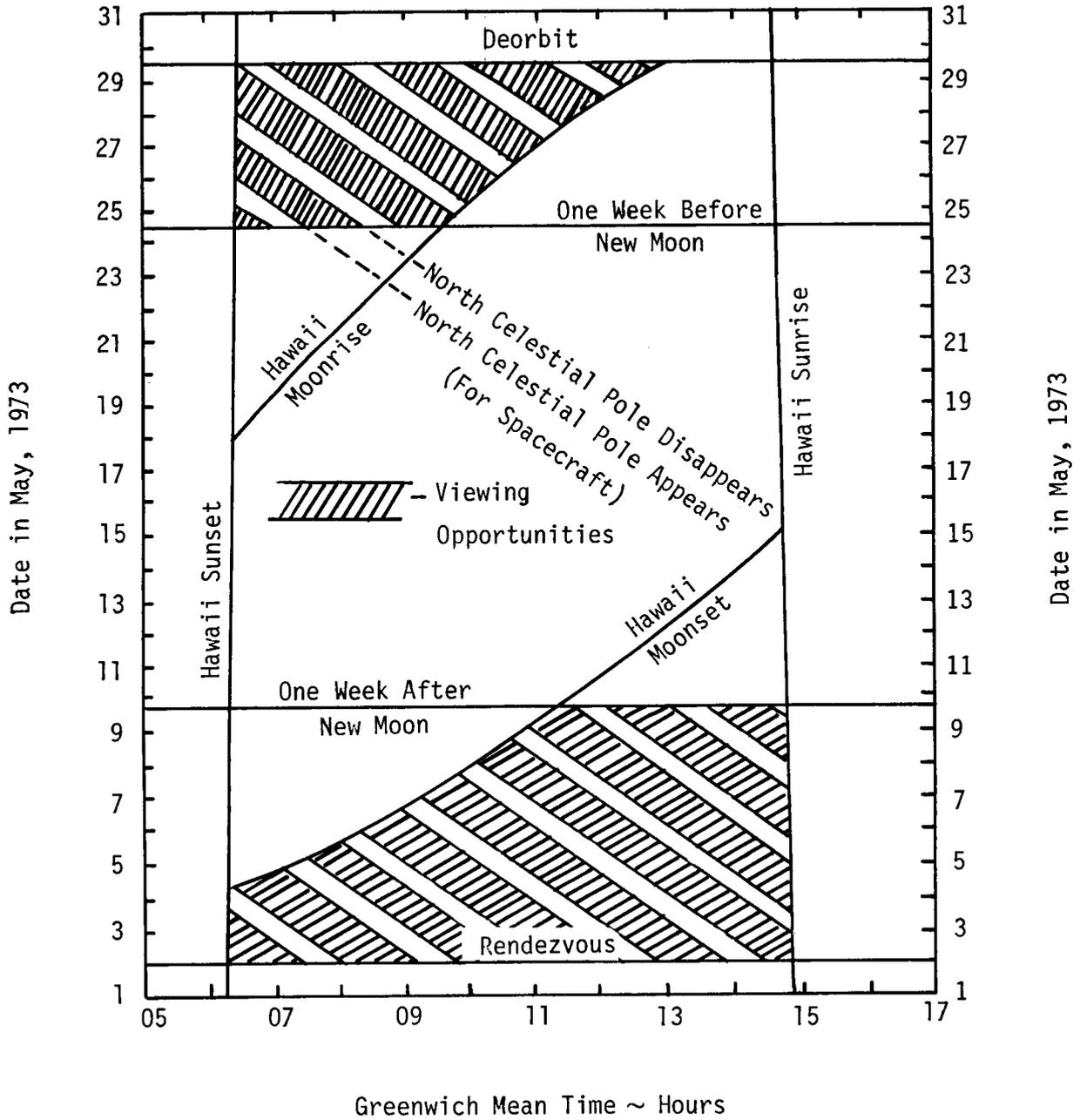


Figure J-6 Concurrent Spacecraft and Ground Station (Hawaii) Viewing Opportunities of North Celestial Pole, One Week Of New Moon

Flight Plan Notes
Experiments

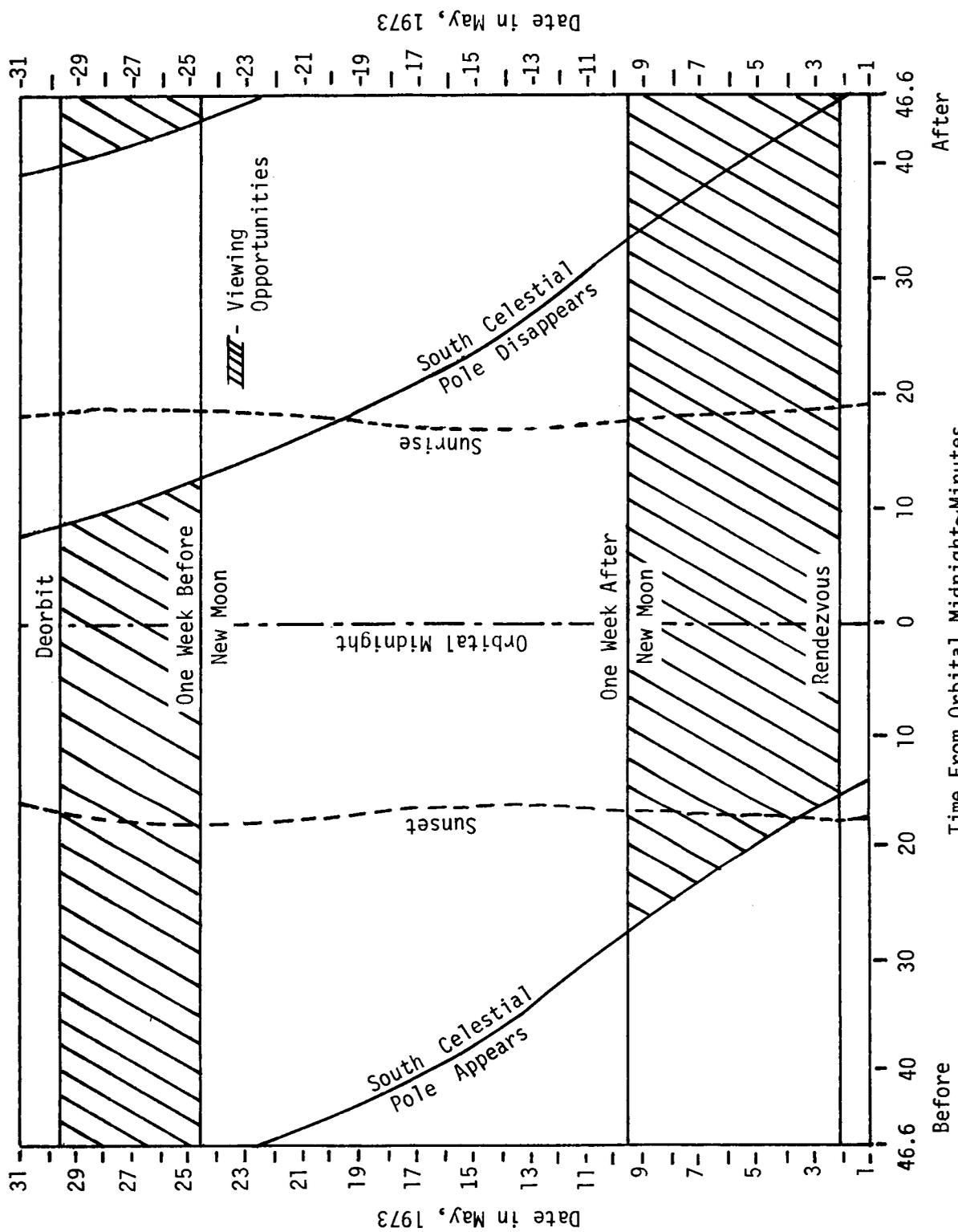


Figure J-7 Spacecraft Viewing Opportunities of South Celestial Pole

Flight Plan Notes
Experiments

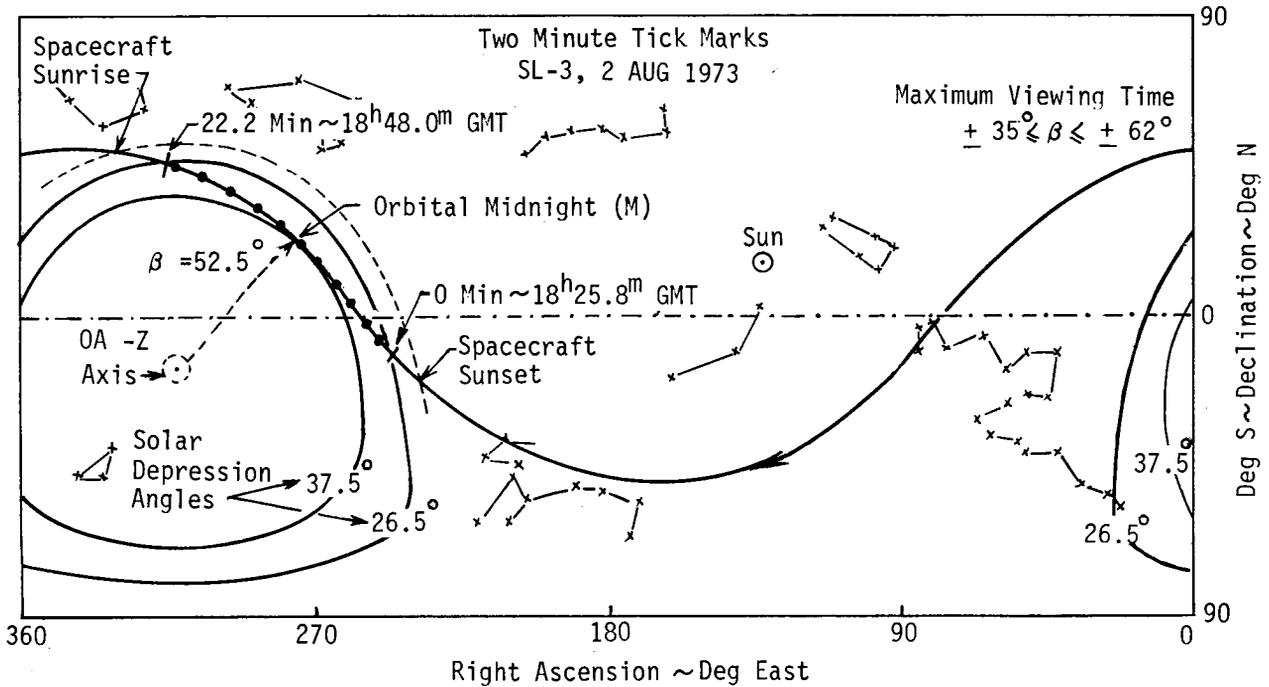
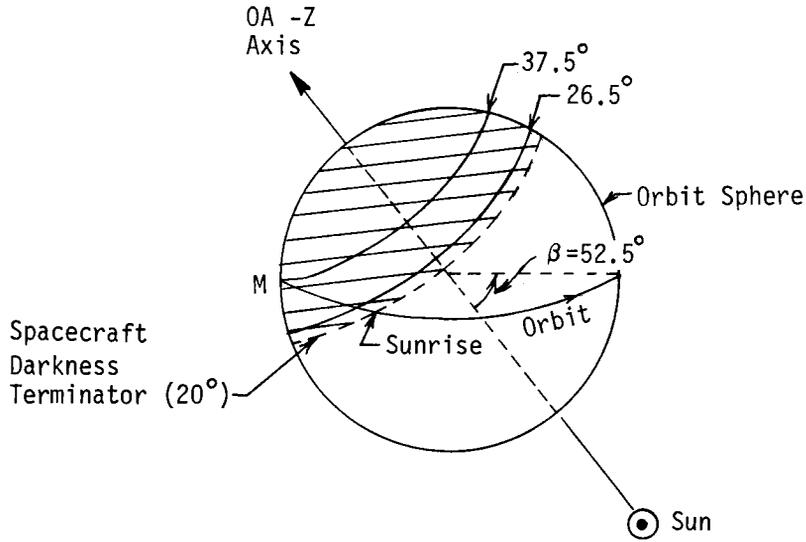
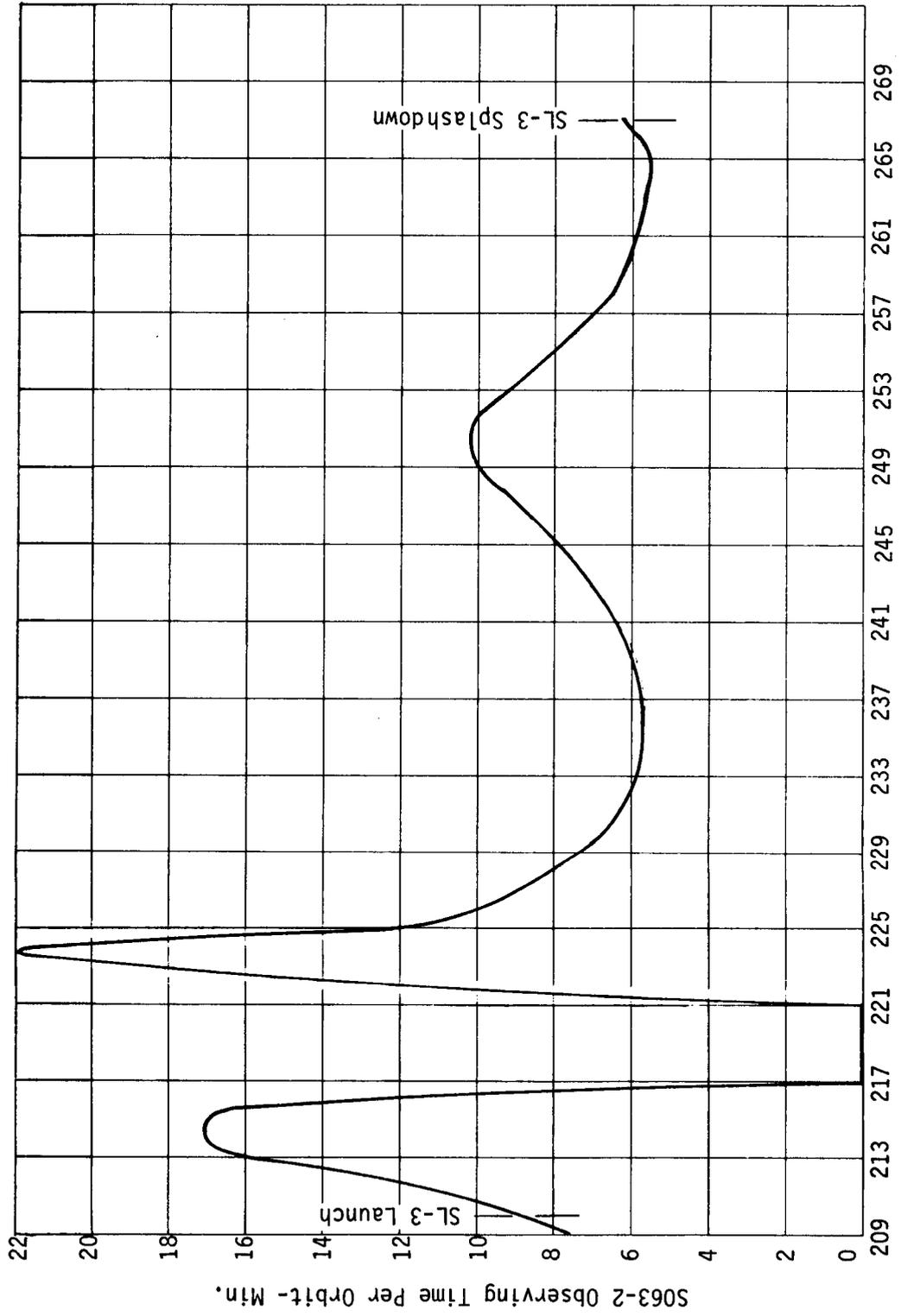


Figure J-8. Solar Depression Angles For S063-F02

Flight Plan Notes
Experiments



Day of Year, 1973

Figure J-9. Time Available for S063-2

Flight Plan Notes
Experiments

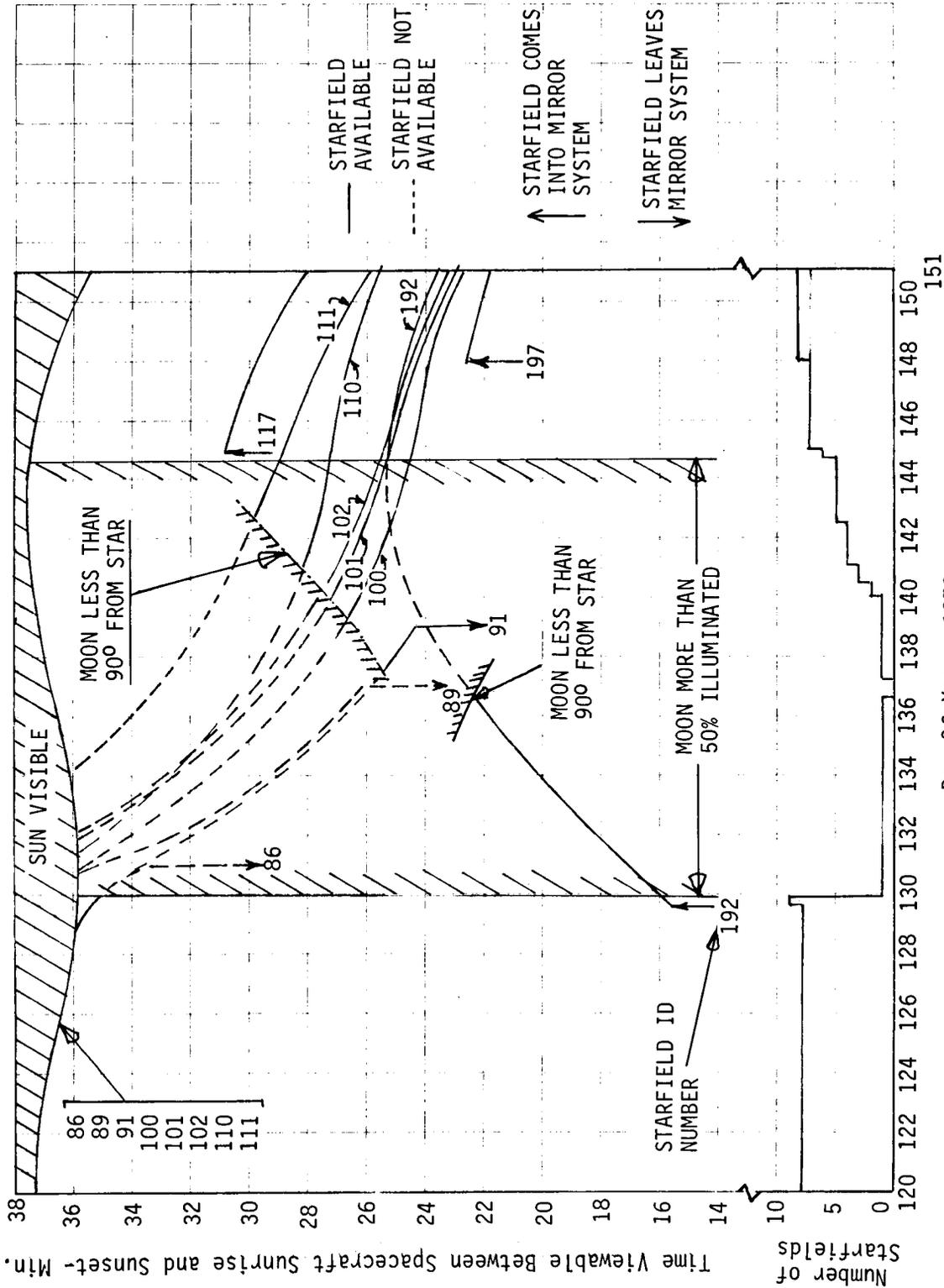
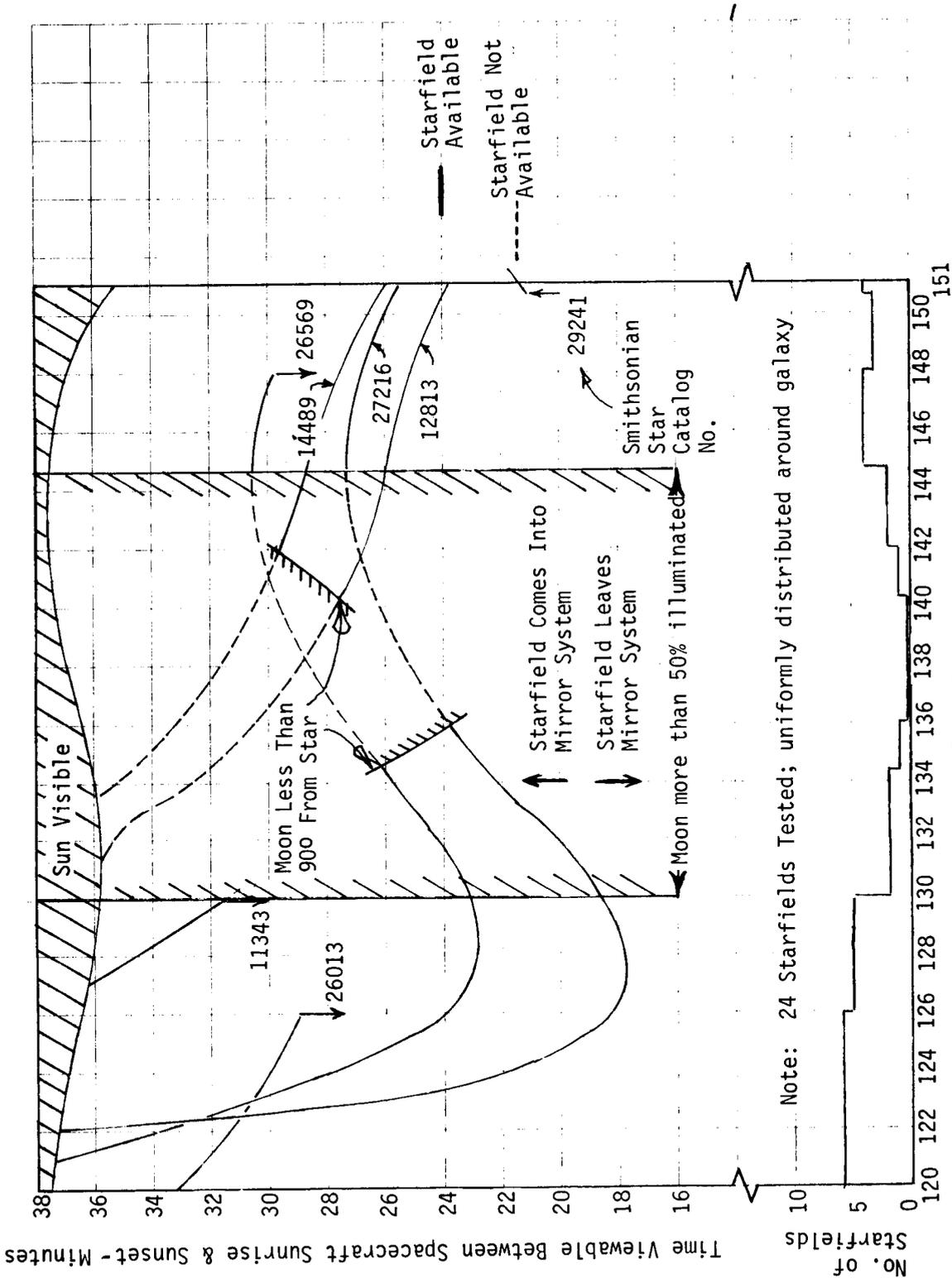


Figure J-10. S019 Viewing Opportunities
(Criteria A Starfields)

Flight Plan Notes
Experiments



Note: 24 Starfields Tested; uniformly distributed around galaxy.

Day of Year, 1973

Figure J-11. S183 Viewing Opportunities

Flight Plan Notes
Photography

K. Photography

1. Details on experiment objectives and operations requiring cameras and film utilization are contained in the "Preliminary Reference Photographic Plan" for SL-2, SL-3, SL-4 (Revision A), October 31, 1971.
2. A sample teleprinter message for inflight use follows. Data contained on these updates consist of the functional objective to be satisfied, camera crew station location, camera identification, lens code, magazine or cassette code, f stop, shutter speed, focus setting, frame rate or frames to be taken, task run time, and activity to be photographed.
3. Teleprinter sample update:

```
Ø197 ØØ6 PHOTO PAD 14Ø 1912  
-----  
M151Ø2 11C DACØ1 1ØMØ1 CIØ4  
(T2.8, 1/6Ø, 1Ø) 6FPS 5MIN M131Ø2  
-----  
Ø197 ØØ6 PHOTO PAD END MSG
```

Flight Plan Notes
Stowage

Stowage

1. The detailed timeline shows transfer of major items between orbital assembly compartments. Items are called out under "Stowage Transfer" and show the location where the item is found and where it is transferred. Transfer of items takes place throughout the mission.
2. All stowage transfers will be noted in the stowage checklist (not yet published).
3. Ground rules for transfer and stowage of items are:
 - . Transfer of items to the SWS will be accomplished as stowage space becomes available.
 - . Any item, if not transferred to another module location or disposed of, will be stowed in its original launch stowage location.
 - . Any deviations from planned transfers or stowage should be noted by crew in order to maintain accurate C.G. definition for the entire orbital assembly and entry configuration for the CSM.
4. The trash airlock is used only for those items which contribute to microbial growth or contamination.
5. The detailed timeline shows inventory items for the OWS, MDA and AM. During the inventory period of each mission the crewman will note the quantity of each item remaining in order that resupply items may be brought on the next mission, should shortages exist.
6. If the status of the crewmen or the success of an experiment depends upon the use of a backup (spare) item or a supplied item for the next mission, then the item should be used. The use of the item should then be noted on the inventory list in order that it may be replaced on the next mission. Items that prove to be defective or are accidentally put out-of-order should also be noted on the inventory list.
7. Dry uncontaminated waste matter in the CM will be deposited in the temporary stowage bags (one for each crewmember - left and right girth shelf and the LEB). Used food bags are stowed in the food stowage compartment (either LEB or LHEB), and transferred to the OWS trash airlock during activation.
8. While in the CM, urine will be collected by the Urine Collection Transfer Assembly (UCTA). After OWS activation, the urine will be transferred from the CM to the urine processor, a sample will be obtained and frozen, and the remainder dumped to the waste tank via the trash airlock. Any urine collection during the re-entry/recovery phase will be collected by the UTCA and processed on the recovery ship.

Flight Plan Notes
Stowage

9. While in the CM, vomitus will be collected in the vomitus collection bag and temporarily stowed. After OWS activation, the vomitus will be transferred to the fecal drier to be dried. Any vomitus collected during the re-entry/recovery phase will be collected in the vomitus collection bag and temporarily stowed to be processed on the recovery ship.
10. Table L-1 shows food types and stowage locations.

Flight Plan Notes

Stowage

FOOD TYPE	LOCATION	AMOUNT
Unrefrigerated Food	11 Food Storage Compartments each containing a 14 day supply (F557, F558, F559, F562, F563, F550, F554, F555, F556, F560, F561)	154 days
Unrefrigerated Food	Wardroom Food Pantry for storage of a 7-day supply of food; (W764, W770, W773, W774, (W775, W776)	7 days
Frozen	Food Storage Freezer divided into three 28 day supply compartments (F553, F552, F551)	84 days
Frozen	Wardroom Food Freezer divided into three compartments; one food chiller and two 28 day supply compartments (W756, W755)	56 days

Table L-1. Food Type and Stowage Location

Flight Plan Notes
Updates

M. Updates

1. Summaries of messages to be sent to and from the SWS are presented in tables M-1 and M-2.
2. Definitions applicable to the above tables are:
 - TPTR, message - teleprinter message Uplink
 - VX, message - voice message Update
 - CMD, message - command telemetry message Uplink
 - PAD - preadvisory data - Uplink or Update messages received in specific tabular order or format
 - Uplink message - non-voice RF communication
 - Update message - Voice RF communication
3. Specific PAD formats are found in SKYLAB FLIGHT PLAN UPDATE MESSAGES, MSC, May 1, 1972.

Flight Plan Notes
Updates

MESSAGE	SCHEDULED	REMARKS	MODE PRIMARY/BACKUP
FLIGHT PLAN UPDATES			
Summary Flight Plan for next day activities	Daily during evening meal	Prior to crew mission planning for next day activities	TPTR/VX
Detail Flight Plan (One per crewman)	Daily during sleep period	Tailored to each crewman's individual activities-use General Message format	TPTR/VX
General Message	Approx. three times daily	This PAD will also be used for short unformatted messages	TPTR/VX
Photo PAD	Daily & as required	Camera settings	TPTR/VX
Selected Consumables	Daily during sleep period	Use General Message format	TPTR/VX
ATM & ATMDC UPDATES			
Solar Activity PAD	Daily during sleep period & as required	Prior to first ATM period of the day & up to once every 3 hours	TPTR/VX
ATM Schedule	Daily during sleep period	Prior to the first ATM period of the day	TPTR/VX
ATMDC Updates Memory & Navigation	NAV PAD daily & prior to Z-LV maneuvers, others as required	First opportunity after breakfast & site contact prior to Z-LV maneuvers	CMD/ TPTR/VX
GMT Clock	Prior to EREP	One hour prior to initial attitude change	TPTR/VX
General Maneuver PAD	At least one rev prior to any maneuver	This is used for EREP maneuver data also	TPTR/VX

Table M-1. Scheduled Updates
(Sheet 1 of 4)

Flight Plan Notes
Updates

MESSAGE	SCHEDULED	REMARKS	MODE PRIMARY/BACKUP
TACS PAD	As required	Thruster data	TPTR/VX
Star Tracker	As required	Prior to ATM activation initially	TPTR/VX
CMG Updates Momentum Dump & Caging	As required		TPTR/VX
EREK UPDATES			
EREK Prep	At least two revs prior to EREK pass	EREK C&D panel configuration for EREK systems at beginning of data- taking pass	TPTR/VX
EREK Operate	At least two revs prior to EREK pass	EREK experiment timeline of switch operations during pass	TPTR/VX
V/TS (Viewfinder Tracking System)	During sleep period before EREK day	Truth site ID, V/TS pointing data, and cloud cover	TPTR/VX
S190B	At least two revs prior to EREK pass	Camera operation and settings	TPTR/VX

Table M-1. Scheduled Updates
(Sheet 2 of 4)

Flight Plan Notes
Updates

MESSAGE	SCHEDULED	REMARKS	MODE PRIMARY/BACKUP
EXPERIMENT UPDATES			
S009	Daily one rev prior to required adjustment	Beta Angle adjustment change to prevent earth occultation and period adjust to prevent exposure in SAA	TPTR/VX
S019	One rev prior to event	Starfield identification and pointing data	TPTR/VX
S020	One rev prior to quiet sun exposure & flare activity data per occurrences	Sunrise/Sunset times and exposure data	TPTR/VX
S063	One rev prior to event	Beginning and ending times of performance session, pointing & exposure/setup data	TPTR/VX
T027/S073 (Photometer System)	One rev prior to event	Photometer scan pointing, exposure/setup data, and scan start time	TPTR/VX

Table M-1. Scheduled Updates
(Sheet 3 of 4)

Flight Plan Notes
Updates

MESSAGE	SCHEDULED	REMARKS	MODE PRIMARY/BACKUP
S183	One rev prior to event	Starfield identification and pointing, exposure/setup data	TPTR/VX
T002	One rev prior to event	Specific target identification and pointing data	TPTR/VX
T025	One rev prior to event	Operational exposure data	TPTR/VX
CSM UPDATES			
Rendezvous Updates	Prelaunch and before burns	NC1, NPC, NC2, NCC, NSR, TPI	VX
P27 Update	When convenient	State Vector (SV) T EPHEM	TPTR/VX/CMD
Maneuver Update (P30)	Prior to undocking	SEP, Shaping, Deorbit	TPTR/VX
Entry Update	Prior to undocking, Daily	Target data	TPTR/VX
Block Data	Twice daily		TPTR/VX
CSM Docked Maneuver Pad	At least one rev prior to mnvr	RCS Trim Burns for 71-rev ground track repeatability	TPTR/VX

Table M-1. Scheduled Updates
(Sheet 4 of 4)

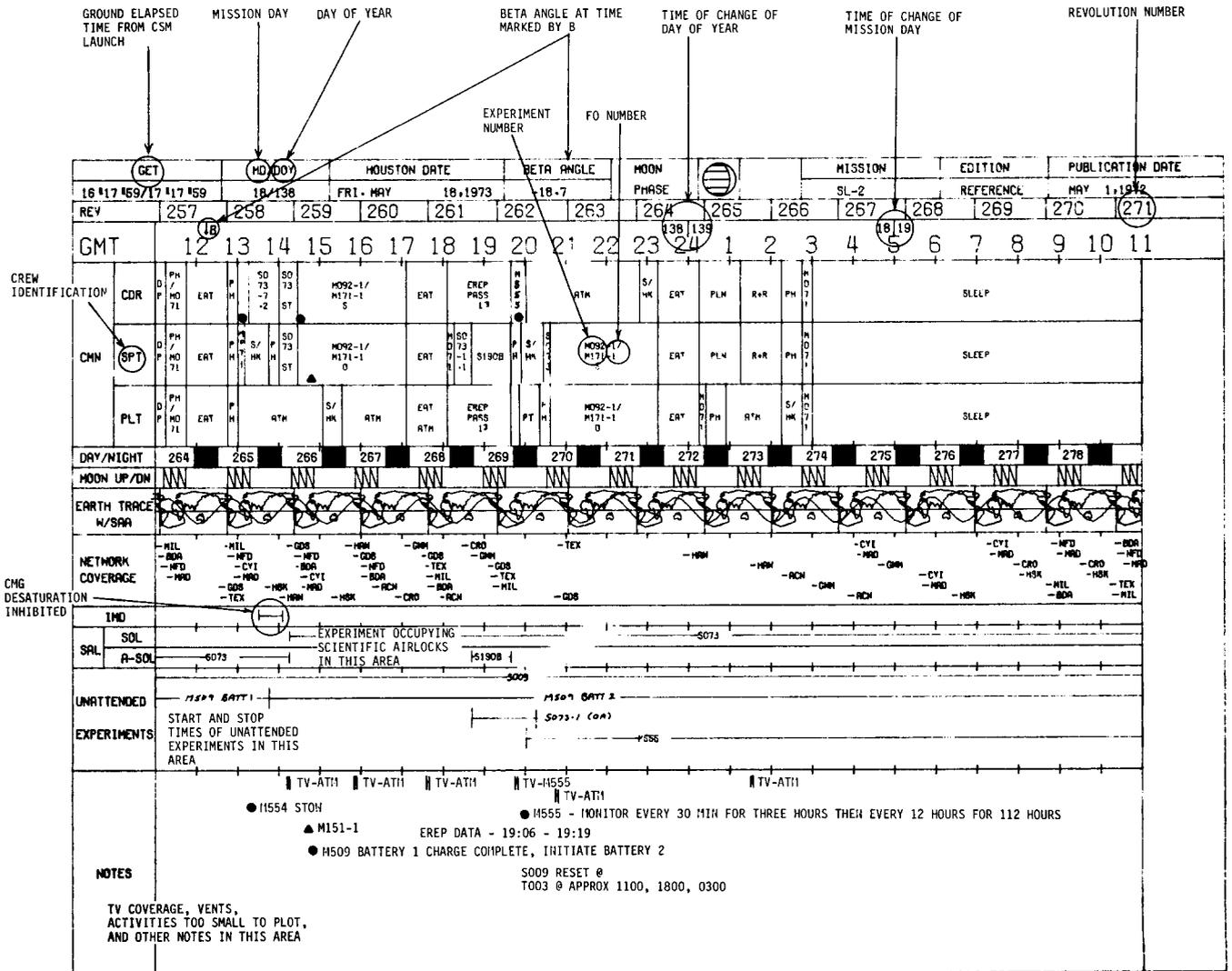
Flight Plan Notes
Updates

VOICE/DATA	SCHEDULED	REMARKS	MODE
EMOD dumps	Prior to CM entry & as required		CSM TM
Film usage	End of each day		VX
ATM film usage	End of each day		VX
EREP film/tape usage	End of each EREP pass		VX
Specific ATM data	End of each day		VX
Selected consumables	End of each day		VX
Crew status	Daily, each morn- ing	Part of post sleep checklist	VX
Expendables status	Weekly		VX
Medical Experiment Summaries	End of each day		VX
Flight Plan accomplishment summary	End of each day	Accomplished vs scheduled	VX
Radiation Accumulation	Daily	Reading of Personal Radiation Dosimeters and radiation survey meter	VX
Iodine Monitoring	Every 14 days	Determine iodine content for each of 10 water tanks	VX
Private communication with Flight Surgeon and family	Daily	Family conversation will neither be monitored nor recorded	VX
Crew Debriefing	Weekly & daily in evening	Contains items not covered by the above & anomalies	VX/ AM Tape Dump

Table M-2. Voice/Data Downlinks

SECTION 2
SUMMARY TIMELINES

SUMMARY TIMELINE FORMAT



- NOTES:
1. THE SUMMARY TIMELINES WERE GENERATED BY COMPUTER CONTROLLED CAL-COMP PLOTTER
 2. THE FOLLOWING ABBREVIATIONS ARE USED IN THE SUMMARY TIMELINE

ACN- ASCENSION	GMM- GUAM	PLT- PILOT
ACT- ACTIVATION	HAW- HAWAII	PT- EXERCISE
A-SOL- ANTI-SOLAR	HSK- HONEYSUCKLE CREEK	R-R- REST AND RECREATION
BDA- BERMUDA	IMD- INHIBIT MOMENTUM DUMP	S- SUBJECT
CDR- COMMANDER	MAD- MADRID	SA- SOUTH ATLANTIC ANOMALY
CMN- CREWMAN	MD- MISSION DAY	SAL- SCIENTIFIC AIRLOCK
C/O- CHECKOUT	MIL- MERRITT ISLAND	S/HK- SYSTEMS HOUSEKEEPING
CRO- CARNARVON	NFD- NEW FOUNDLAND	SOL- SOLAR
CYI- GRAND CANARY ISLAND	O- OBSERVER	SPT-SCIENTIST PILOT
DE-ACT- DEACTIVATION	PH- PERSONAL HYGIENE	ST- STOW
DP- DAILY PREPS	PLN- PLANNING	SU- SETUP
DOY- DAY OF YEAR		SYS- SYSTEM
GDS- GOLDSTONE		TEX- TEXAS
GMT- GREENWICH MEAN TIME		VAN- VANGUARD

GET	MD/DOY	HOUSTON DATE							BETA ANGLE	MOON PHASE	MISSION	EDITION							PUBLICATION DATE					
		TUE - MAY	1	1973	-11-8	18	19	20				21	22	23	24	25								
0	17159	1	121	122	123	124	125	-11-8	121	SL-2	1	2	3	4	5	6	7	8	9	10	11	12	MAY 1, 1972	
REV	12	13	14	15	16	17	18	19	20	21	22	23	24	25										
GMT	12	13	14	15	16	17	18	19	20	21	22	23	24	25										
CDR																								
SPT																								
PLT																								
DRY/NIGHT																								
MOON UP/DN																								
EARTH TRACE W/SAR																								
NETWORK COVERAGE																								
IMD																								
SOL																								
P-SOL																								
UNATTENDED																								
EXPERIMENTS																								
NOTES																								

TV-RENDEVOUS/DOCKING

REPORT: DAILY STATUS/ ACCOMPLISHMENTS

UP LINK: DETAIL FLIGHT PLAN PHOTO CONSUMABLES

GET	MD/DOY	HOUSTON DATE	BETA ANGLE	MOON PHASE	MISSION	EDITION	PUBLICATION DATE
0 17 59/1 17 59	2 /122	WED. MAY 2 1973	-15.4		SL-2	REFERENCE	MAY 1, 1972
REV	26 27 28 29 30 31 32 33	34 35 36 37 38 39 40					
GMT	12 13 14 15 16 17 18 19 20 21 22 23 24	1 2 3 4 5 6 7 8 9 10 11					
CDR	PH ERT	PH / MO 71	ACT	EAT	T025	SLEEP	
SPT	PH ERT	PH / MO 71	ACT	EAT	M092-1/ M171-1	SLEEP	
PLT	PH ERT	PH / MO 71	ACT	EAT	M092-1/ M171-1	SLEEP	
DRY/NIGHT	13 14 15 16 17 18 19 20 21 22 23 24	25 26 27					
MOON UP/DN							
EARTH TRACE W/SAR							
NETWORK COVERAGE							
IMD							
SOL							
R-SOL							
UNATTENDED							
EXPERIMENTS							
NOTES	<p> <input checked="" type="checkbox"/> M516-1 FOOD TRANSFER H VENT H VENT <input checked="" type="checkbox"/> M516-4 M509 UNSTOW H VENT H VENT <input checked="" type="checkbox"/> M151-6 <input checked="" type="checkbox"/> M092, M093, M131, M171, M133 ACTIVATION <input checked="" type="checkbox"/> INITIATE M509 BATTERY #1 CHARGE PRESSURIZE PSS BOTTLE <input checked="" type="checkbox"/> M092, M093, M131, M171, M133 ACTIVATION <input checked="" type="checkbox"/> INITIATE M509 BATTERY #1 CHARGE PRESSURIZE PSS BOTTLE </p> <p> REPORT: DAILY STATUS/ ACCOMPLISHMENTS UP LINK: SUMMARY FLIGHT PLAN/BLOCK DATA UP LINK: CONSUMABLES DETAIL FLIGHT PLAN PHOTO </p>						

GET		HD/DOY		HOUSTON DATE		BETA ANGLE		MOON PHASE		MISSION		EDITION		PUBLICATION DATE																											
3 17 159/A 17 159		5 / 125		SAT. MAY 5 1973		-24.7		SL-2		SL-2		81		MAY 1, 1972																											
REV 70		71		72		73		74		75		76		77		78		79		80		81		82		83															
GMT		12		13		14		15		16		17		18		19		20		21		22		23		24		25		26		27		28		29		30		31	
CDR		PH		D / PH		EAT		ENEP		ATM		S / ATM		PT		R-4		SLEEP																							
SPT		PH		D / PH		EAT		ENEP		ATM		S / ATM		PT		R-4		SLEEP																							
PLT		PH		D / PH		EAT		ENEP		ATM		S / ATM		PT		R-4		SLEEP																							
DRY/NIGHT		60		61		62		63		64		65		66		67		68		69		70		71		72		73		74											
MOON UP/DN																																									
EARTH TRACE W/SRA																																									
NETWORK COVERAGE																																									
IMD																																									
SOL																																									
R-SOL																																									
UNATTENDED																																									
EXPERIMENTS																																									
NOTES																																									

GET	MO/DOY	HOUSTON DATE	BETA ANGLE	MOON PHASE	MISSION	EDITION	PUBLICATION DATE
4 17 59/5 17 59	6 / 126	SUN. MAY 6, 1973	-27.1	PHASE	SL-2	REFERENCE	MAY 1, 1972
REV 84	85	86	87	88	89	90	91
	126	127	128	129	130	131	132
	133	134	135	136	137	138	139
	140	141	142	143	144	145	146
	147	148	149	150	151	152	153
	154	155	156	157	158	159	160
	161	162	163	164	165	166	167
	168	169	170	171	172	173	174
	175	176	177	178	179	180	181
	182	183	184	185	186	187	188
	189	190	191	192	193	194	195
	196	197	198	199	200	201	202
	203	204	205	206	207	208	209
	210	211	212	213	214	215	216
	217	218	219	220	221	222	223
	224	225	226	227	228	229	230
	231	232	233	234	235	236	237
	238	239	240	241	242	243	244
	245	246	247	248	249	250	251
	252	253	254	255	256	257	258
	259	260	261	262	263	264	265
	266	267	268	269	270	271	272
	273	274	275	276	277	278	279
	280	281	282	283	284	285	286
	287	288	289	290	291	292	293
	294	295	296	297	298	299	300
	301	302	303	304	305	306	307
	308	309	310	311	312	313	314
	315	316	317	318	319	320	321
	322	323	324	325	326	327	328
	329	330	331	332	333	334	335
	336	337	338	339	340	341	342
	343	344	345	346	347	348	349
	350	351	352	353	354	355	356
	357	358	359	360	361	362	363
	364	365	366	367	368	369	370
	371	372	373	374	375	376	377
	378	379	380	381	382	383	384
	385	386	387	388	389	390	391
	392	393	394	395	396	397	398
	399	400	401	402	403	404	405
	406	407	408	409	410	411	412
	413	414	415	416	417	418	419
	420	421	422	423	424	425	426
	427	428	429	430	431	432	433
	434	435	436	437	438	439	440
	441	442	443	444	445	446	447
	448	449	450	451	452	453	454
	455	456	457	458	459	460	461
	462	463	464	465	466	467	468
	469	470	471	472	473	474	475
	476	477	478	479	480	481	482
	483	484	485	486	487	488	489
	490	491	492	493	494	495	496
	497	498	499	500	501	502	503
	504	505	506	507	508	509	510
	511	512	513	514	515	516	517
	518	519	520	521	522	523	524
	525	526	527	528	529	530	531
	532	533	534	535	536	537	538
	539	540	541	542	543	544	545
	546	547	548	549	550	551	552
	553	554	555	556	557	558	559
	560	561	562	563	564	565	566
	567	568	569	570	571	572	573
	574	575	576	577	578	579	580
	581	582	583	584	585	586	587
	588	589	590	591	592	593	594
	595	596	597	598	599	600	601
	602	603	604	605	606	607	608
	609	610	611	612	613	614	615
	616	617	618	619	620	621	622
	623	624	625	626	627	628	629
	630	631	632	633	634	635	636
	637	638	639	640	641	642	643
	644	645	646	647	648	649	650
	651	652	653	654	655	656	657
	658	659	660	661	662	663	664
	665	666	667	668	669	670	671
	672	673	674	675	676	677	678
	679	680	681	682	683	684	685
	686	687	688	689	690	691	692
	693	694	695	696	697	698	699
	700	701	702	703	704	705	706
	707	708	709	710	711	712	713
	714	715	716	717	718	719	720
	721	722	723	724	725	726	727
	728	729	730	731	732	733	734
	735	736	737	738	739	740	741
	742	743	744	745	746	747	748
	749	750	751	752	753	754	755
	756	757	758	759	760	761	762
	763	764	765	766	767	768	769
	770	771	772	773	774	775	776
	777	778	779	780	781	782	783
	784	785	786	787	788	789	790
	791	792	793	794	795	796	797
	798	799	800	801	802	803	804
	805	806	807	808	809	810	811
	812	813	814	815	816	817	818
	819	820	821	822	823	824	825
	826	827	828	829	830	831	832
	833	834	835	836	837	838	839
	840	841	842	843	844	845	846
	847	848	849	850	851	852	853
	854	855	856	857	858	859	860
	861	862	863	864	865	866	867
	868	869	870	871	872	873	874
	875	876	877	878	879	880	881
	882	883	884	885	886	887	888
	889	890	891	892	893	894	895
	896	897	898	899	900	901	902
	903	904	905	906	907	908	909
	910	911	912	913	914	915	916
	917	918	919	920	921	922	923
	924	925	926	927	928	929	930
	931	932	933	934	935	936	937
	938	939	940	941	942	943	944
	945	946	947	948	949	950	951
	952	953	954	955	956	957	958
	959	960	961	962	963	964	965
	966	967	968	969	970	971	972
	973	974	975	976	977	978	979
	980	981	982	983	984	985	986
	987	988	989	990	991	992	993
	994	995	996	997	998	999	1000

UPLINK: DETAIL FLIGHT PLAN PHOTO CONSUMABLES

REPORT: DAILY STATUS/ ACCOMPLISHMENTS

UPLINK: SUMMARY FLIGHT PLAN BLOCK DATA

S009 RESET @ 19:59
S015 LAMP CHECK @ 1820
T003 @ APPROX 1100, 1800, 0300

REPORT: CREW STATUS

UPLINK: EREP UPDATE MANEUVER BLOCK DATA ENTRY UPDATE

ATMDC UPDATE

REPORT: CREW STATUS

UPLINK: EREP UPDATE MANEUVER BLOCK DATA ENTRY UPDATE

ATMDC UPDATE

GET	MO/DDOY	HOUSTON DATE	BETA ANGLE	MOON PHASE	MISSION	EDITION	PUBLICATION DATE																				
18 17 59/19 17 59	20/140	SUN. MAY 20, 1973	-11.6		SL-2	REFERENCE	MAY 1, 1972																				
REV 286	287	288	289	290	291	292	293	294	295	296	297	298	299	300													
GMT	12	13	14	15	16	17	18	19	20	21	22	23	24	1	2	3	4	5	6	7	8	9	10	11			
CDR	PH	PH																									
	MO	MO																									
	7I	7I																									
SPT	PH	PH	PH																								
	MO	MO	MO																								
	7I	7I	7I																								
PLT	PH	PH	PH																								
	MO	MO	MO																								
	7I	7I	7I																								
DAY/NIGHT	281	282	283	284	285	286	287	288	289	290	291	292	293	294	295	296	297	298	299	300	301	302	303	304	305		
	MOON UP/DN																										
	EARTH TRACE W/SAA																										
NETWORK COVERAGE	GOSEVY	GOSEVY																									
	NET	NET																									
	COV	COV																									
IMD	SOL	SOL																									
	A-SOL	A-SOL																									
	UNATTENDED	UNATTENDED																									
EXPERIMENTS	NET	NET																									
	NET	NET																									
	NET	NET																									
NOTES	UPLINK :	UPLINK :																									
	UPLINK :	UPLINK :																									
	UPLINK :	UPLINK :																									

GET	MO/DOY		HOUSTON DATE		BETA ANGLE		MOON PHASE		MISSION		EDITION		PUBLICATION DATE											
	19 517 559/20 117 559	21/141	MON - MAY	21-1973	-7-7	301	302	303	304	305	306	307	308	309	310	311	312	313	314					
REV	301	302	303	304	305	306	307	308	309	310	311	312	313	314	315	316	317	318	319	320	321			
GMT	12	13	14	15	16	17	18	19	20	21	22	23	24	1	2	3	4	5	6	7	8	9	10	11
CDR	P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT	
SPT	P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT	
PLT	P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT	
DAY/NIGHT	P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT	
MOON UP/DN	P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT	
EARTH TRACE W/SRA	P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT	
NETWORK COVERAGE	P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT	
IMD	P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT	
SOL	P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT	
A-SOL	P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT	
UNATTENDED	P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT	
EXPERIMENTS	P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT	
NOTES	P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT		P TO P M TO M E RT	

UPLINK :
DETAIL FLIGHT PLAN
PHOTO
CONSUMABLES
ATM SCHED
SOLAR ACT

UPLINK :
BLOCK DATA
SUMMARY FLIGHT PLAN
INCLUDES 15 MIN PT
S/HK-THIRD DAY ACTIVITIES

UPLINK :
INCLUDES 15 MIN PT
S/HK-SEVENTH DAY ACTIVITIES
INCLUDES 15 MIN PT
S/HK-THIRD DAY ACTIVITIES

UPLINK :
ATMDC UPDATE
BLOCK DATA
ENTRY UPDATE
T025
S019
REPORT :
CREW STATUS

UPLINK :
S009
ATMDC UPDATE
BLOCK DATA
ENTRY UPDATE
T025
S019
REPORT :
CREW STATUS

UPLINK :
T027/S073
S/HK-THIRD DAY ACTIVITIES
INCLUDES 15 MIN PT
S009 RESET @ 14:00
T003 @ APPROX 1100, 1800, 0300

5073-7(4R)

5073

5009 + D024

H656

6019

H133

GET	MD/DOY	HOUSTON DATE	BETA ANGLE	MOON PHASE	MISSION	EDITION	PUBLICATION DATE
23 17 159/24 17 159	25/145	FRI. MAY 25, 1973	9.4	364	SL-2	370	MAY 1, 1972
REV	359	360	363	366	367	369	371
	362	365	368	372	375	378	382
	361	364	367	371	374	377	381
	360	363	366	370	373	376	380
	359	362	365	368	371	374	378
	358	361	364	367	370	373	377
	357	360	363	366	369	372	376
	356	359	362	365	368	371	375
	355	358	361	364	367	370	374
	354	357	360	363	366	369	373
	353	356	359	362	365	368	372
	352	355	358	361	364	367	371
	351	354	357	360	363	366	370
	350	353	356	359	362	365	369
	349	352	355	358	361	364	368
	348	351	354	357	360	363	367
	347	350	353	356	359	362	366
	346	349	352	355	358	361	365
	345	348	351	354	357	360	364
	344	347	350	353	356	359	363
	343	346	349	352	355	358	362
	342	345	348	351	354	357	361
	341	344	347	350	353	356	360
	340	343	346	349	352	355	359
	339	342	345	348	351	354	358
	338	341	344	347	350	353	357
	337	340	343	346	349	352	356
	336	339	342	345	348	351	355
	335	338	341	344	347	350	354
	334	337	340	343	346	349	353
	333	336	339	342	345	348	352
	332	335	338	341	344	347	351
	331	334	337	340	343	346	350
	330	333	336	339	342	345	349
	329	332	335	338	341	344	348
	328	331	334	337	340	343	347
	327	330	333	336	339	342	346
	326	329	332	335	338	341	345
	325	328	331	334	337	340	344
	324	327	330	333	336	339	343
	323	326	329	332	335	338	342
	322	325	328	331	334	337	341
	321	324	327	330	333	336	340
	320	323	326	329	332	335	339
	319	322	325	328	331	334	338
	318	321	324	327	330	333	337
	317	320	323	326	329	332	336
	316	319	322	325	328	331	335
	315	318	321	324	327	330	334
	314	317	320	323	326	329	333
	313	316	319	322	325	328	332
	312	315	318	321	324	327	331
	311	314	317	320	323	326	330
	310	313	316	319	322	325	329
	309	312	315	318	321	324	328
	308	311	314	317	320	323	327
	307	310	313	316	319	322	326
	306	309	312	315	318	321	325
	305	308	311	314	317	320	324
	304	307	310	313	316	319	323
	303	306	309	312	315	318	322
	302	305	308	311	314	317	321
	301	304	307	310	313	316	320
	300	303	306	309	312	315	319
	299	302	305	308	311	314	318
	298	301	304	307	310	313	317
	297	300	303	306	309	312	316
	296	299	302	305	308	311	315
	295	298	301	304	307	310	314
	294	297	300	303	306	309	313
	293	296	299	302	305	308	312
	292	295	298	301	304	307	311
	291	294	297	300	303	306	310
	290	293	296	299	302	305	309
	289	292	295	298	301	304	308
	288	291	294	297	300	303	307
	287	290	293	296	299	302	306
	286	289	292	295	298	301	305
	285	288	291	294	297	300	304
	284	287	290	293	296	299	303
	283	286	289	292	295	298	302
	282	285	288	291	294	297	301
	281	284	287	290	293	296	300
	280	283	286	289	292	295	299
	279	282	285	288	291	294	298
	278	281	284	287	290	293	297
	277	280	283	286	289	292	296
	276	279	282	285	288	291	295
	275	278	281	284	287	290	294
	274	277	280	283	286	289	293
	273	276	279	282	285	288	292
	272	275	278	281	284	287	291
	271	274	277	280	283	286	290
	270	273	276	279	282	285	289
	269	272	275	278	281	284	288
	268	271	274	277	280	283	287
	267	270	273	276	279	282	286
	266	269	272	275	278	281	285
	265	268	271	274	277	280	284
	264	267	270	273	276	279	283
	263	266	269	272	275	278	282
	262	265	268	271	274	277	281
	261	264	267	270	273	276	280
	260	263	266	269	272	275	279
	259	262	265	268	271	274	278
	258	261	264	267	270	273	277
	257	260	263	266	269	272	276
	256	259	262	265	268	271	275
	255	258	261	264	267	270	274
	254	257	260	263	266	269	273
	253	256	259	262	265	268	272
	252	255	258	261	264	267	271
	251	254	257	260	263	266	270
	250	253	256	259	262	265	269
	249	252	255	258	261	264	268
	248	251	254	257	260	263	267
	247	250	253	256	259	262	266
	246	249	252	255	258	261	265
	245	248	251	254	257	260	264
	244	247	250	253	256	259	263
	243	246	249	252	255	258	262
	242	245	248	251	254	257	261
	241	244	247	250	253	256	260
	240	243	246	249	252	255	259
	239	242	245	248	251	254	258
	238	241	244	247	250	253	257
	237	240	243	246	249	252	256
	236	239	242	245	248	251	255
	235	238	241	244	247	250	254
	234	237	240	243	246	249	253
	233	236	239	242	245	248	252
	232	235	238	241	244	247	251
	231	234	237	240	243	246	250
	230	233	236	239	242	245	249
	229	232	235	238	241	244	248
	228	231	234	237	240	243	247
	227	230	233	236	239	242	246
	226	229	232	235	238	241	245
	225	228	231	234	237	240	244
	224	227	230	233	236	239	243
	223	226	229	232	235	238	242
	222	225	228	231	234	237	241
	221	224	227	230	233	236	240
	220	223	226	229	232	235	239
	219	222	225	228	231	234	238
	218	221	224	227	230	233	237
	217	220	223	226	229	232	236
	216	219	222	225	228	231	235
	215	218	221	224	227	230	234
	214	217	220	223	226	229	233
	213	216	219	222	225	228	232
	212	215	218	221	224	227	231
	211	214	217	220	223	226	230
	210	213	216	219	222	225	229
	209	212	215	218	221	224	228
	208	211	214	217	220	223	227
	207	210	213	216	219	222	226
	206	209	212	215	218	221	225
	205	208	211	214	217	220	224
	204	207	210	213	216	219	223
	203	206	209	212	215	218	222
	202	205	208	211	214	217	221

GET	MD/DOY	HOUSTON DATE	BETA ANGLE	MOON PHASE	MISSION	EDITION	PUBLICATION DATE
27 17 159/28 17 159	29/149	TUE . MAY 29, 1973	28.0		SL-2	REFERENCE	MAY 1, 1972
REV	416	417					
GMT	12 13 14 15 16 17 18 19 20 21 22 23 24			149/150	29/30	5 6 7 8 9 10 11	
COR							
CNN							
PLT							
DRY/NIGHT							
MOON UP/DN							
EARTH TRACE W/SAA							
NETWORK COVERAGE							
IMD							
SOL							
P-SOL							
UNATTENDED							
EXPERIMENTS							
NOTES							

GET	MD/DOY	HOUSTON DATE	BETA ANGLE	MOON PHASE	MISSION	EDITION	PUBLICATION DATE
0 1-1744/D 16 #16	1 /210	SUN. JULY 29, 1973	38.2		SL-3	REFERENCE	MAY 1, 1972
REV					1305	1306	1307
					1308	1309	1310
					1311		
GMT	12	13	14	15	16	17	18
	19	20	21	22	23	24	1
	2	3	4	5	6	7	8
	9	10	11				
CDR							
SPT							
PLT							
DAY/NIGHT							
MOON UP/DN							
EARTH TRACE W/SRA							
NETWORK COVERAGE							
IMD							
SOL							
P-SOL							
UNATTENDED							
EXPERIMENTS							
NOTES							

SL-3 LAUNCH-0443

6071

6149-2

6149-2

0024

GET	MO/DOY	HOUSTON DATE	BETA ANGLE	MOON PHASE	MISSION	EDITION	PUBLICATION DATE
0 #6 116/1 #6 116	2 /211	MON. JULY 30, 1973	42.7		SL-3	REFERENCE	MAY 1, 1972
REV	1311 1312	1313 1314 1315 1316 1317 1318			1320 1321 1322 1323		1324 1325
GMT	12 13 14 15 16 17 18 19 20 21 22 23 24				2 3 4 5 6 7 8 9 10 11		
CDR		ACT	SLEEP		ACT	EAT	SLEEP
SPT		ACT	SLEEP		ACT	EAT	SLEEP
PLT		ACT	SLEEP		ACT	EAT	SLEEP
DRY/NIGHT	5	6	7	8	9	10	11
MOON UP/DN							
EARTH TRACE W/SRR							
NETWORK COVERAGE							
IMD							
SOL							
R-SOL							
UNATTENDED							
EXPERIMENTS							
NOTES							

GET	MO/DOY	HOUSTON DATE	BETA ANGLE	MOON PHASE	MISSION	EDITION	PUBLICATION DATE
2 46 16/3 16	4 /213	MED. AUGUST 1, 1973	51.2		SL-3	REFERENCE	MAY 1, 1972
REV	1340	1342 1343 1344	1345 1346 1347	1348	1349	1351 1352	1353 1354
GMT	12 13 14 15 16 17 18 19 20 21 22 23 24	213 214	4 5 6 7 8 9 10 11				
CDR	SLEEP	PH/NO P 71	EAT	PH/NO P 71	PH/NO P 71	PH/NO P 71	SLEEP
	SLEEP MISS	PH/NO P 71	EAT	PH/NO P 71	PH/NO P 71	PH/NO P 71	SLEEP MISS
	SLEEP	PH/NO P 71	EAT	PH/NO P 71	PH/NO P 71	PH/NO P 71	SLEEP
DRY/NIGHT	36 37 38 39 40 41 42 43 44 45 46 47 48 49 50						
MOON UP/DN							
EARTH TRACE W/SRA							
NETWORK COVERAGE							
IMD							
SOL							
UNATTENDED							
EXPERIMENTS							
NOTES							

GET	MO/DOY	HOUSTON DATE	BETA ANGLE	MOON PHASE	MISSION	EDITION	PUBLICATION DATE																		
3 16/4 16	5 /214	THU. AUGUST 2, 1973	55.2	PHASE	SL-3	REFERENCE	MAY 1, 1972																		
REV	1355	1356	1357	1358	1359	1360	1361	1362	1363	1364	1365	1366	1367	1368											
GMT	12	13	14	15	16	17	18	19	20	21	22	23	24	1	2	3	4	5	6	7	8	9	10	11	
CDR	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP
SPT	SLEEP M133	SLEEP M133	SLEEP M133	SLEEP M133	SLEEP M133	SLEEP M133	SLEEP M133	SLEEP M133	SLEEP M133	SLEEP M133	SLEEP M133	SLEEP M133	SLEEP M133	SLEEP M133	SLEEP M133	SLEEP M133	SLEEP M133	SLEEP M133	SLEEP M133	SLEEP M133	SLEEP M133	SLEEP M133	SLEEP M133	SLEEP M133	SLEEP M133
PLT	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP	SLEEP
DRY/NIGHT	52	53	54	55	56	57	58	59	60	61	62	63	64	65	66										
MOON UP/DN																									
EARTH TRACE W/SRR																									
NETWORK COVERAGE																									
IMD																									
SOL																									
R-SOL																									
UNATTENDED																									
EXPERIMENTS																									
NOTES																									

● M110 BLOOD SAMPLE DURING PH
▲ M151-2 ▲ M151-5

T003 @ APPROX 1110, 1800, 0300

GET	MD/DOY	HOUSTON DATE	BETA ANGLE	MOON PHASE	MISSION	EDITION	PUBLICATION DATE								
7 16 116/8 16	9 /218	MON. AUGUST 6, 1973	65.8	PHASE	SL-3	REFERENCE	MAY 1, 1972								
REV	1412	1413 1414 1415 1416 1417 1418 1419 1420 1421 1422 1423 1424 1425 1426													
GMT	12 13 14 15 16 17 18 19 20 21 22 23 24	218 219													
CMN	CDR	PH / MO P 71	PH / MO P 71	PH / MO P 71	PH / MO P 71	PH / MO P 71	PH / MO P 71								
	SPT	PH / MO P 71	PH / MO P 71	PH / MO P 71	PH / MO P 71	PH / MO P 71	PH / MO P 71								
	PLT	PH / MO P 71	PH / MO P 71	PH / MO P 71	PH / MO P 71	PH / MO P 71	PH / MO P 71								
DAY/NIGHT	113	114	115	116	117	118	119	120	121	122	123	124	125	126	127
MOON UP/DN															
W/SRR															
NETWORK COVERAGE															
IMD															
SOL															
A-SOL															
UNATTENDED															
EXPERIMENTS															
NOTES															

GET	MD/DDY	HOUSTON DATE	BETA ANGLE	MOON	MISSION	EDITION	PUBLICATION DATE
12 16 116/13 16 116	14/223	SAT AUGUST 11, 1973	57.8	PHASE	SL-3	REFERENCE	MAY 1, 1972
REV	1485	1486	1487	1488	1489	1490	1491
	1492	1493	1494	1495	1496	1497	1498
GMT	12	13	14	15	16	17	18
	19	20	21	22	23	24	19 15
	1	2	3	4	5	6	7
	8	9	10	11			
CDR	PH / MD / 71	EAT	S/A/K	OFF DUTY	OFF DUTY	OFF DUTY	OFF DUTY
	PH / MD / 71	EAT	S/A/K	OFF DUTY	OFF DUTY	OFF DUTY	OFF DUTY
SPT	PH / MD / 71	EAT	S/A/K	OFF DUTY	OFF DUTY	OFF DUTY	OFF DUTY
	PH / MD / 71	EAT	S/A/K	OFF DUTY	OFF DUTY	OFF DUTY	OFF DUTY
PLT	PH / MD / 71	EAT	S/A/K	OFF DUTY	OFF DUTY	OFF DUTY	OFF DUTY
	PH / MD / 71	EAT	S/A/K	OFF DUTY	OFF DUTY	OFF DUTY	OFF DUTY
DRY/NIGHT	190	191	192	193	194	195	196
	197	198	199	200	201	202	203
	204						
MOON UP/DN							
EARTH TRACE W/SRR							
NETWORK COVERAGE							
IMD							
SOL							
R-SOL							
UNATTENDED EXPERIMENTS							
NOTES							

GET	MD/DOY	HOUSTON DATE		BETA ANGLE	MOON PHASE		MISSION		EDITION	PUBLICATION DATE					
		SUN - AUGUST	12-1973		54.0	PHASE	SL-3	REFERENCE							
13 16	15/224	1501	1502	1503	1504	1505	1506	1507	1508	1509	1510	1511	1512		
REV	1499	1500	1501	1502	1503	1504	1505	1506	1507	1508	1509	1510	1511		
GMT	12	13	14	15	16	17	18	19	20	21	22	23	24		
CDR	PH D / P MO / P 71														
CHN	PH D / P MO / P 71														
PLT	PH D / P MO / P 71														
DAY/NIGHT	206	207	208	209	210	211	212	213	214	215	216	217	218	219	220
MOON UP/DN															
EARTH TRACE W/SRR															
NETWORK COVERAGE															
IMD															
SOL															
A-SOL															
UNATTENDED EXPERIMENTS															
NOTES															

GET	MD/DOY	HOUSTON DATE	BETA ANGLE	MOON PHASE	MISSION	EDITION	PUBLICATION DATE
14 46	16/225	MON. AUGUST 13, 1973	49-9		SL-3		MAY 1, 1972
REV 1513	1514 1515 1516 1517 1518 1519	1517 1518 1519		1520	1521 1522 1523 1524 1525		1526 1527
GMT	12 13 14 15 16 17 18 19 20 21 22 23 24	1525 1526		1527	1528 1529 1530 1531 1532 1533 1534 1535		1536 1537 1538 1539 1540 1541 1542 1543 1544 1545 1546 1547 1548 1549 1550 1551 1552 1553 1554 1555 1556 1557 1558 1559 1560 1561 1562 1563 1564 1565 1566 1567 1568 1569 1570 1571 1572 1573 1574 1575 1576 1577 1578 1579 1580 1581 1582 1583 1584 1585 1586 1587 1588 1589 1590 1591 1592 1593 1594 1595 1596 1597 1598 1599 1600
CDR	SC P 73 P 71	M609-3 S	ATM	P H EAT	R+R PH I	SLEEP	
SPT	SC P 73 P 71	ATM	SO 53 S S/ 53 TV -2-2 U PT H UV VTS	ATM	R+R PH I	SLEEP	
PLT	SC P 73 P 71	M609-3 D	SO 53 S S/ 53 TV -2-2 U PT H UV VTS	ATM	R+R PH I	SLEEP	
DAY/NIGHT	221 222 223 224 225 226 227 228 229 230 231 232 233 234 235						
MOON UP/DN							
EARTH TRACE M/SRA							
NETWORK COVERAGE							
IMD							
SOL							
R-SOL							
UNATTENDED							
EXPERIMENTS							
NOTES							

T003 @ APPROX 1110, 1800, 0300

GET	MD/DOY	HOUSTON DATE	BETA ANGLE	MOON	MISSION	EDITION	PUBLICATION DATE
15 46	16/16 16	17/226	45-5	PHASE	SL-3	REFERENCE	MAY 1, 1972
REV	1528	1529	1530	1531	1532	1533	1534
	1530	1531	1532	1533	1534	1535	1536
	1537	1538	1539	1540	1541		
GMT	12	13	14	15	16	17	18
	19	20	21	22	23	24	1
	2	3	4	5	6	7	8
	9	10	11				
CDR	PH D/NO P/71	EAT	ATM	ATM	ATM	ATM	ATM
SPT	PH D/NO P/71	EAT	ATM	ATM	ATM	ATM	ATM
PLT	PH D/NO P/71	EAT	ATM	ATM	ATM	ATM	ATM
DRY/NIGHT	237	238	239	240	241	242	243
MOON UP/DN							
EARTH TRACE W/SAR							
NETWORK COVERAGE							
IMD							
SOL							
R-SOL							
UNATTENDED							
EXPERIMENTS							
NOTES							

T003 @ APPROX 1110, 1800, 0300

GET	MD/DOY	HOUSTON DATE	BETA ANGLE	MOON PHASE	MISSION	EDITION	PUBLICATION DATE									
1616	18/227	MED. AUGUST 15.1973	41.0		SL-3	REFERENCE	MAY 1, 1972									
REV	1542	1543 1544 1545 1546	1547 1548	1549	1550 1551 1552 1553	1554	1555 1556									
GMT	12 13 14 15 16 17 18 19 20 21 22 23 24	1 2 3 4 5 6 7 8 9 10 11														
COR	P S/ H MK	EREPR	ATM	M052-1/ M053-1	R-R		SLEEP									
	P S/ H MK	EREPR	ATM	M052-1/ M053-1	R-R		SLEEP									
	P S/ H MK	EREPR	ATM	ATM	ATM		SLEEP									
SPT	P S/ H MK	EREPR	M052-1/ M053-1	M052-1/ M053-1												
	P S/ H MK	EREPR	M052-1/ M053-1	M052-1/ M053-1												
	P S/ H MK	EREPR	ATM	ATM												
PLT	P S/ H MK	EREPR	M052-1/ M053-1	ATM												
	P S/ H MK	EREPR	M052-1/ M053-1	ATM												
	P S/ H MK	EREPR	ATM	ATM												
DRY/NIGHT	252	253	254	255	256	257	258	259	260	261	262	263	264	265	266	
MOON UP/DN																
EARTH TRACE W/SAR																
NETWORK COVERAGE																
IMD																
SOL																
R-SOL																
UNATTENDED EXPERIMENTS																
NOTES																

GET		HOUSTON DATE		BETA ANGLE		MOON PHASE		MISSION		EDITION		PUBLICATION DATE	
1746 16/18 16 16		THU. AUGUST 16, 1973				1563 1564		SL-3		REFERENCE		MAY 1, 1972	
REV 1557 1558		1559 1560 1561 1562		1563 1564		1565 1566 1567 1568		1569 1570					
GMT 12 13 14 15 16 17 18 19 20 21 22 23 24 1 2 3 4 5 6 7 8 9 10 11													
CDR		S/MK		ATM		PT H		PLN		R-R		SLEEP	
SPT		S/MK		S/MK		S/ M-2 VIS		EAT		R-R		SLEEP	
PLT		ATM		S/ M-2 VIS		S/ M-2 VIS		EAT		S/ M-2 VIS		SLEEP	
DAY/NIGHT		268 269 270 271 272 273 274 275 276 277 278 279 280 281 282											
MOON UP/DN													
EARTH TRACE													
W/SAR													
NETWORK COVERAGE													
IMD													
SOL													
R-SOL													
UNATTENDED EXPERIMENTS													
NOTES													

T003 @ APPROX 1110, 1800, 0300

GET	MD/DOY	HOUSTON DATE	BETA ANGLE	MOON PHASE	MISSION	EDITION	PUBLICATION DATE
21 16	16/22 16	MON - AUGUST 20, 1973	19-6	SL-3	SL-3	REFERENCE	MAY 1, 1972
REV	1615 1616 1617 1618 1619 1620 1621 1622 1623 1624 1625 1626 1627 1628						
GMT	12 13 14 15 16 17 18 19 20 21 22 23 24	232 233					
CDR	PH / NO / 71 EAT	PH / NO / 71 EAT	PH / NO / 71 EAT	PH / NO / 71 EAT	PH / NO / 71 EAT	PH / NO / 71 EAT	PH / NO / 71 EAT
SPT	PH / NO / 71 EAT	PH / NO / 71 EAT	PH / NO / 71 EAT	PH / NO / 71 EAT	PH / NO / 71 EAT	PH / NO / 71 EAT	PH / NO / 71 EAT
PLT	PH / NO / 71 EAT	PH / NO / 71 EAT	PH / NO / 71 EAT	PH / NO / 71 EAT	PH / NO / 71 EAT	PH / NO / 71 EAT	PH / NO / 71 EAT
DAY/NIGHT	329	330	331	332	333	334	335
MOON UP/DN							
EARTH TRACE							
W/SRA							
NETWORK COVERAGE							
IMD							
SOL							
A-SOL							
UNATTENDED							
EXPERIMENTS							
NOTES							

T003 @ APPROX 1110, 1800, 0300

GET	MO/DOY	HOUSTON DATE	BETA ANGLE	MOON PHASE	MISSION	EDITION	PUBLICATION DATE																			
22 16	16/23 16	TUE. AUGUST 21, 1973	14.8		SL-3	REFERENCE	MAY 1, 1972																			
REV	1629	1630	1631	1632	1633	1634	1635	1636	1637	1638	1639	1640	1641	1642												
GMT	12	13	14	15	16	17	18	19	20	21	22	23	24	1	2	3	4	5	6	7	8	9	10	11		
CDR	PH D/P 71	PH D/P 71	PH D/P 71	PH D/P 71	PH D/P 71	PH D/P 71	PH D/P 71	PH D/P 71	PH D/P 71	PH D/P 71	PH D/P 71	PH D/P 71	PH D/P 71	PH D/P 71	PH D/P 71	PH D/P 71	PH D/P 71	PH D/P 71	PH D/P 71	PH D/P 71	PH D/P 71	PH D/P 71	PH D/P 71	PH D/P 71	PH D/P 71	
CHN	SPT	PLT	DRY/NIGHT	MOON UP/DN	EARTH TRACE W/SRA	NETWORK COVERAGE	IMD	SOL	A-SOL	UNATTENDED	EXPERIMENTS	NOTES														

GET	MD/DDYY	HOUSTON DATE	BETA ANGLE	MOON PHASE	MISSION	EDITION	PUBLICATION DATE																		
25 16	16/26 16	FRI. AUGUST 24, 1973	0-5		SL-3	REFERENCE	MAY 1, 1972																		
REV	1672	1673	1674	1675	1676	1677	1678	1679	1680	1681	1682	1683	1684	1685	1686										
GMT	12	13	14	15	16	17	18	19	20	21	22	23	24	1	2	3	4	5	6	7	8	9	10	11	
CDR	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	
SPT	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	
PLT	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	PH D P	
DRY/NIGHT	391	392	393	394	395	396	397	398	399	400	401	402	403	404	405										
MOON UP/DN																									
EARTH TRACE																									
N/S/R																									
NETWORK COVERAGE																									
IMD																									
SOL																									
A-SOL																									
UNATTENDED																									
EXPERIMENTS																									
NOTES																									

T003 @ APPROX 1110, 1800, 0300

GET	MD/DOY	HOUSTON DATE	BETA ANGLE	MOON PHASE	MISSION	EDITION	PUBLICATION DATE								
2646	16/2746 116	SAT. AUGUST 25, 1973	-4.2	PHASE	SL-3	REFERENCE	MAY 1, 1972								
REV	1687 1688	1689 1690 1691 1692	1693 1694	1695 1696 1697	1698 1699	1700									
GMT	12 13 14 15 16 17 18 19 20 21 22 23 24 1 2 3 4 5 6 7 8 9 10 11	28/29													
CMN	CDR	EVA PREPS	ATM FILM REPLACE	POST EVA	EAT	S/FO MK 25 BU	P H	PH D / MO P 71	EAT						
	SPT	EVA PREPS	ATM FILM REPLACE	POST EVA	EAT	ATH	P H	PH D / MO P 71	EAT						
	PLT	EVA PREPS	MONITOR EVA	POST EVA	EAT	S/MK	P H	PH D / MO P 71	EAT						
DRY/NIGHT	407	408	409	410	411	412	413	414	415	416	417	418	419	420	421
MOON UP/DN															
EARTH TRACE W/SRA															
NETWORK COVERAGE															
IMD															
SOL															
A-SOL															
UNATTENDED															
EXPERIMENTS															
NOTES															

GET	MD/DOY	HOUSTON DATE	BETA ANGLE	MOON PHASE	MISSION	EDITION	PUBLICATION DATE
31 16	16/32 16	THU - AUGUST 30, 1973	-25.6		SL-3	REFERENCE	MAY 1, 1972
REV	1759	1760	1761	1762	1763	1764	1765
	1759	1760	1761	1762	1763	1764	1765
	1766	1767	1768	1769	1770	1771	1772
GMT	12	13	14	15	16	17	18
	19	20	21	22	23	24	1
	2	3	4	5	6	7	8
	9	10	11				
CDR	PH / D / P / 71	PH / D / P / 71	PH / D / P / 71	PH / D / P / 71	PH / D / P / 71	PH / D / P / 71	PH / D / P / 71
SPT	PH / D / P / 71	PH / D / P / 71	PH / D / P / 71	PH / D / P / 71	PH / D / P / 71	PH / D / P / 71	PH / D / P / 71
PLT	PH / D / P / 71	PH / D / P / 71	PH / D / P / 71	PH / D / P / 71	PH / D / P / 71	PH / D / P / 71	PH / D / P / 71
DRY/NIGHT	484	485	486	487	488	489	490
MOON UP/DN	484	485	486	487	488	489	490
EARTH TRACE	484	485	486	487	488	489	490
NETWORK COVERAGE	484	485	486	487	488	489	490
IMD	484	485	486	487	488	489	490
SOL	484	485	486	487	488	489	490
A-SOL	484	485	486	487	488	489	490
UNATTENDED EXPERIMENTS	484	485	486	487	488	489	490
NOTES	484	485	486	487	488	489	490

T003 @ APPROX 1110, 1800, 0300

GET	MD/DOY	HOUSTON DATE	BETA ANGLE	MOON PHASE	MISSION	EDITION	PUBLICATION DATE																	
3246 116/3346 116	34/243	FRI. AUGUST 31.1973	-29.4		SL-3		MAY 1.1972																	
REV	1773	1774	1775	1776	1777	1781	1782	1783	1784	1785	1786	1787												
GMT	12	13	14	15	16	17	18	19	20	21	22	23	24	1	2	3	4	5	6	7	8	9	10	11
CDR	PH D/P MO/71	PH D/P MO/71	PH D/P MO/71	PH D/P MO/71	PH D/P MO/71	PH D/P MO/71	PH D/P MO/71	PH D/P MO/71	PH D/P MO/71	PH D/P MO/71	PH D/P MO/71	PH D/P MO/71	PH D/P MO/71	PH D/P MO/71	PH D/P MO/71	PH D/P MO/71	PH D/P MO/71	PH D/P MO/71	PH D/P MO/71	PH D/P MO/71	PH D/P MO/71	PH D/P MO/71	PH D/P MO/71	PH D/P MO/71
CHN	SPT	SPT	SPT	SPT	SPT	SPT	SPT	SPT	SPT	SPT	SPT	SPT	SPT	SPT	SPT	SPT	SPT	SPT	SPT	SPT	SPT	SPT	SPT	SPT
PLT	PLT	PLT	PLT	PLT	PLT	PLT	PLT	PLT	PLT	PLT	PLT	PLT	PLT	PLT	PLT	PLT	PLT	PLT	PLT	PLT	PLT	PLT	PLT	PLT
DRY/NIGHT	499	500	501	502	503	504	505	506	507	508	509	510	511	512	513									
MOON UP/DN																								
EARTH TRACE																								
M/SRR																								
NETWORK COVERAGE																								
IMD																								
SOL																								
R-SOL																								
UNATTENDED EXPERIMENTS																								
NOTES																								

T003 @ APPROX 1110, 1800, 0300

GET	MD/DOY		HOUSTON DATE		BETA ANGLE		MOON PHASE		MISSION		EDITION		PUBLICATION DATE												
	34 46	116/35 46	116	36/245	SUN- SEPTEMBER 2	1973	-35-9	36/245	SL-3	1811	1812	1813	1814	1815	1816										
REV	1802	1803	1804	1805	1806	1807	1808	1809	1810	1811	1812	1813	1814	1815	1816										
GMT	12	13	14	15	16	17	18	19	20	21	22	23	24	1	2	3	4	5	6	7	8	9	10	11	
COR	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH
	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD
	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL
SPT	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH
	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD
	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL
PLT	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH
	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD
	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL
DAY/NIGHT	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH
	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD
	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL
MOON UP/DN	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH
	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD
	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL
EARTH TRACE W/SAR	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH
	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD
	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL
NETWORK COVERAGE	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH
	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD
	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL
IMD	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH
	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD
	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL
SOL	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH
	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD
	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL
SAL	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH
	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD
	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL
UNATTENDED	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH
	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD
	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL
EXPERIMENTS	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH
	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD
	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL
NOTES	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH
	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD	MD
	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL	TL

GET		MD/DOY		HOUSTON DATE		BETA ANGLE		MOON		MISSION		EDITION		PUBLICATION DATE										
41 16	16/42 16	43/252	SUN -	SEPTEMBER 9	1973	-42-8		PHASE		SL-3		REFERENCE		MAY 1, 1972										
REV	1903	1904	1905	1906	1907	1908	1909	1910	1911	1912	1913	1914	1915	1916	1917									
GMT	12	13	14	15	16	17	18	19	20	21	22	23	24	1	2	3	4	5	6	7	8	9	10	11
CDR	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH
SPT	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH
PLT	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH
DAY/NIGHT	638	639	640	641	642	643	644	645	646	647	648	649	650	651	652									
MOON UP/DN																								
EARTH TRACE																								
W/SRR																								
NETWORK COVERAGE																								
IMD																								
SOL																								
A-SOL																								
UNATTENDED EXPERIMENTS																								
NOTES																								

T003 @ APPROX 1110, 1800, 0300

GET	MO/DOY	HOUSTON DATE	BETA ANGLE	MOON PHASE	MISSION	EDITION	PUBLICATION DATE																		
44 16 116/45 16	46/255	WED - SEPTEMBER 12, 1973	-36.6		SL-3	REFERENCE	MAY 1, 1972.																		
REV	1947	1948	1949	1950	1951	1952	1953	1954	1955	1956	1957	1958	1959	1960											
GMT	12	13	14	15	16	17	18	19	20	21	22	23	24	1	2	3	4	5	6	7	8	9	10	11	
CDR	PH D/NO P/71	EAT	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU
SPT	PH D/NO P/71	EAT	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU
PLT	PH D/NO P/71	EAT	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU	SD 73 SU
DRY/NIGHT	684	685	686	687	688	689	690	691	692	693	694	695	696	697	698	699									
MOON UP/DN																									
EARTH TRACE																									
W/SRR																									
NETWORK COVERAGE																									
IMD																									
SOL																									
A-SOL																									
UNATTENDED																									
EXPERIMENTS																									
NOTES																									

GET	MD/DDY	HOUSTON DATE	BETA ANGLE	MOON PHASE	MISSION	EDITION	PUBLICATION DATE																		
48 16	16/19 16	16	-23.0	50/259	SL-3	REFERENCE	MAY 1, 1972																		
REV	2004	2005	2006	2007	2008	2009	2010	2011	2012	2013	2014	2015	2016	2017	2018										
GMT	12	13	14	15	16	17	18	19	20	21	22	23	24	1	2	3	4	5	6	7	8	9	10	11	
CDR	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH
SPT	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH
PLT	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH	PH
DAY/NIGHT	746	747	748	749	750	751	752	753	754	755	756	757	758	759	760										
MOON UP/DN																									
EARTH TRACE W/SAR																									
NETWORK COVERAGE	MIL -BDR -CYL	-GDS -TEX -CYL -HNSK -HNSK	-GDS -BOR -CYL -RCM -HNSK -HNSK	-HNSK -GDS -MIL -BOR -RCM -CRD -HNSK	-GDS -TEX -MIL -BOR -CRD -HNSK																				
IMD																									
SOL																									
R-SOL																									
UNATTENDED EXPERIMENTS																									
NOTES																									

T003 @ APPROX 1110, 1800, 0300

GET	MD/DOY	HOUSTON DATE		BETA ANGLE	MOON PHASE		MISSION					EDITION	PUBLICATION DATE		
		MON - SEPTEMBER 17, 1973	-19.1		MOON	PHASE	SL-3	2026	2027	2028	2029			2030	2031
49 16 16 16	51/260	MON - SEPTEMBER 17, 1973	-19.1										MAY 1, 1972		
REV	2019	2020	2021	2022	2023	2024	2025	2026	2027	2028	2029	2030	2031	2032	
GMT	12	13	14	15	16	17	18	19	20	21	22	23	24	1 2 3 4 5 6 7 8 9 10 11	
CDR	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	
SPT	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	
PLT	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	PH D/P	
DAY/NIGHT	762	763	764	765	766	767	768	769	770	771	772	773	774	775	776
MOON UP/DN															
EARTH TRACE															
W/SAR															
NETWORK COVERAGE															
IHD															
SOL															
R-SOL															
UNATTENDED EXPERIMENTS															
NOTES															

GET	MD/DOY	HOUSTON DATE	BETA ANGLE	MOON PHASE	MISSION	EDITION	PUBLICATION DATE																	
S1 16 116/52 16 116	53/262	WED. SEPTEMBER 19, 1973	-10.8	PHASE	SL-3		MAY 1, 1972																	
REV	2048	2049	2050	2051	2052	2053	2054	2055	2056	2057	2058	2059	2060	2061										
GMT	12	13	14	15	16	17	18	19	20	21	22	23	24	1	2	3	4	5	6	7	8	9	10	11
CDR	REENTRY SIMULATION																							
SPT	REENTRY SIMULATION																							
PLT	REENTRY SIMULATION																							
DRY/NIGHT	793	794	795	796	797	798	799	800	801	802	803	804	805	806	807									
MOON UP/DN	[MOON PHASE INDICATORS]																							
EARTH TRACE N/S/A	[EARTH TRACE WAVEFORM]																							
NETWORK COVERAGE	[NETWORK COVERAGE INDICATORS]																							
IMD	[IMD DATA]																							
SOL	[SOL DATA]																							
A-SOL	[A-SOL DATA]																							
UNATTENDED EXPERIMENTS	[UNATTENDED EXPERIMENTS DATA]																							
NOTES	[NOTES DATA]																							

T003 @ APPROX 1100, 1800, 0300

GET	MD/DOY	HOUSTON DATE	BETA ANGLE	MOON PHASE	MISSION	EDITION	PUBLICATION DATE
56 16 116/57 16 16	58/267	MON. SEPTEMBER 24, 1973	10-8		SL-3	REFERENCE	MAY 1, 1972
REV	2120						
GMT	12 13 14 15 16 17 18 19 20 21 22 23 24				58 59	1 2 3 4 5 6 7 8 9 10 11	
COR							
CNN SPT							
PLT							
DRY/NIGHT							
MOON UP/DN							
EARTH TRACE W/SRA							
NETWORK COVERAGE							
IMD							
SOL							
A-SOL							
UNATTENDED EXPERIMENTS							
NOTES							

SL-3 SPLASHDOWN-1336

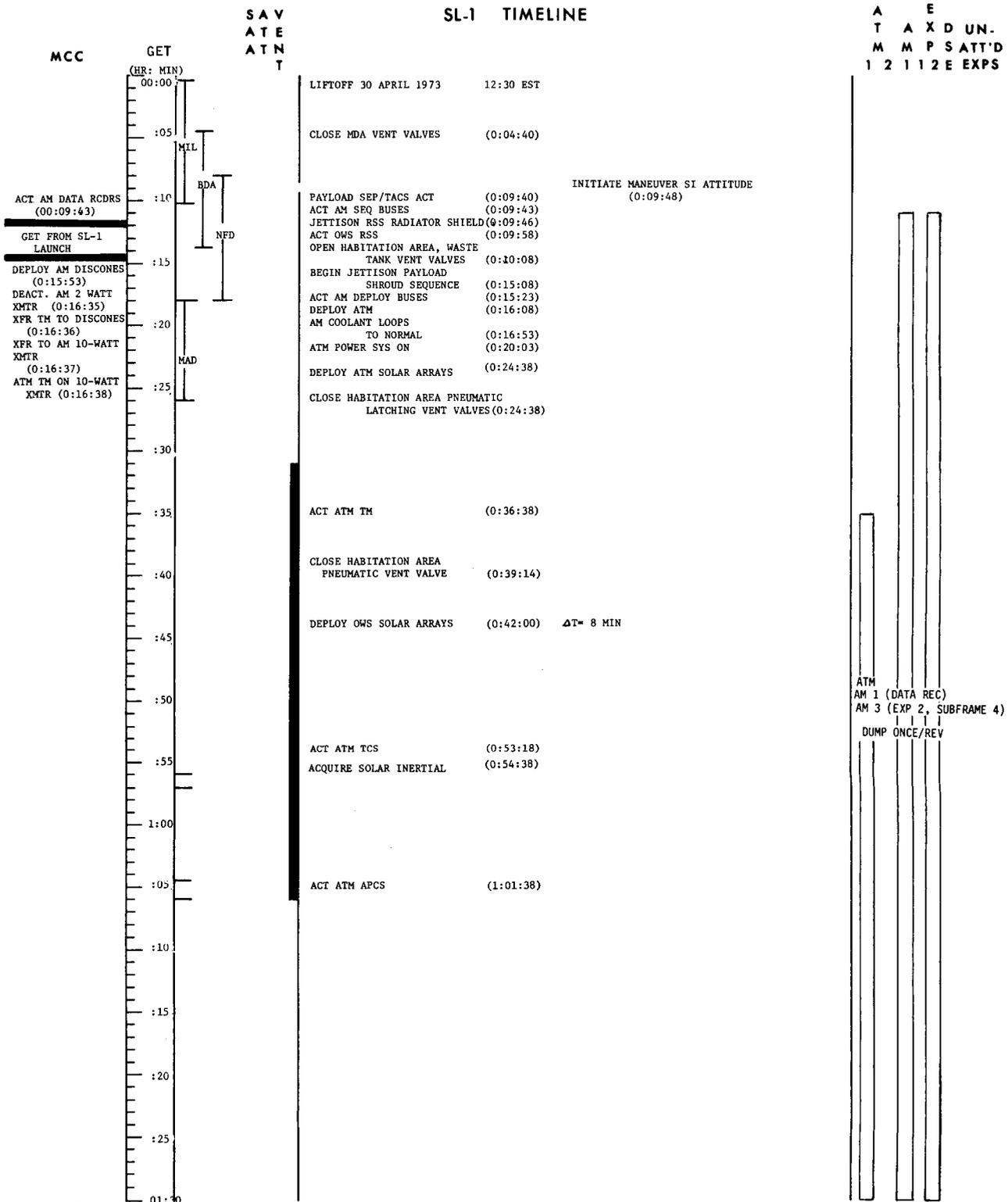
SECTION 3
DETAILED TIMELINES

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA ϕ	MOON PHASE
SL-1 (0-1.30)		-1/120	APRIL 30, 1973			

SL-1 TIMELINE

RCDR
 A E
 T A X D UN.
 M M P S ATT'D
 1 2 1 1 2 E EXPS



MISSION	EDITION	PUBLICATION DATE
SL-1	REFERENCE	MAY 1, 1972

FLIGHT PLANNING BRANCH

S	A	L	SOL	A-SOL

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA	MOON PHASE
01:30 - 19:00		-1/120	APRIL 30, 1973			

S
A
V
A
T
E
A
T
N
T

SL-1 TIMELINE

RCDR
A
E
T
A
X
D
U
N
M
M
P
S
A
T
T
D
1
2
1
1
2
E
E
X
P
S

MCC

GET (HR: MIN)

01:30 ACT MDA WALL, TUNNEL HTRS (1:35:56)

02:00 ACT MDA PORT HTRS (1:35:59)

GET FROM SL-1 LAUNCH

03:00

04:00

05:00 OPEN OWS SOLENOID VENT VALVES (5:9:38) (SECOND VENT)

06:00

07:00

08:00

09:00

10:00 CLOSE OWS SOLENOID VENT VALVES (9:49:38)

PRESS OWS WITH O2 (9:49:38)

11:00

12:00

13:00 XFER ATT CNTL TO CMG's (12:03:00)

PRESS OWS WITH O2 (9:49:38)

$\Delta T=8$ HRS

14:00

15:00

16:00

17:00

18:00 MONITOR OWS PRESS (17:50:00)

PRESS MDA/AM WITH O2 (17:56:38)

$\Delta T=1.3$ HRS

MONITOR MDA/AM PRESS (19:00:00)

19:00

GDS
TEX
MIL
BDA
NED
MAD

HAW

GDS
ACN

GDS

DEPLOY METEOROID SHIELD (1:35:48)

PRESSURIZE OWS WITH O2 (1:36:38) $\Delta T=9$ hr 22 min

DEACT. AM DEPLOY BUSES (1:39:38)

DUMP OWS PNEUMATIC SYS BOTTLES (3:09:38) $\Delta T=3$ hr 45 min

PARALLEL AM/ATM EPS (4:39:38)

TRANSFER TAGS TO APCS (4:40:0)

TERMINATE OWS PNEUMATIC SYS BOTTLES DUMP (6:54:38)

(END OF GUARANTEED IU LIFE) (7:24:00)

ATM
AM 1 (DATA REC)
AM 3 (EXP 2, SUBFRAME 4)

DUMP ONCE/REV

MISSION	EDITION	PUBLICATION DATE
SL - 1	REFERENCE	MAY 1, 1972

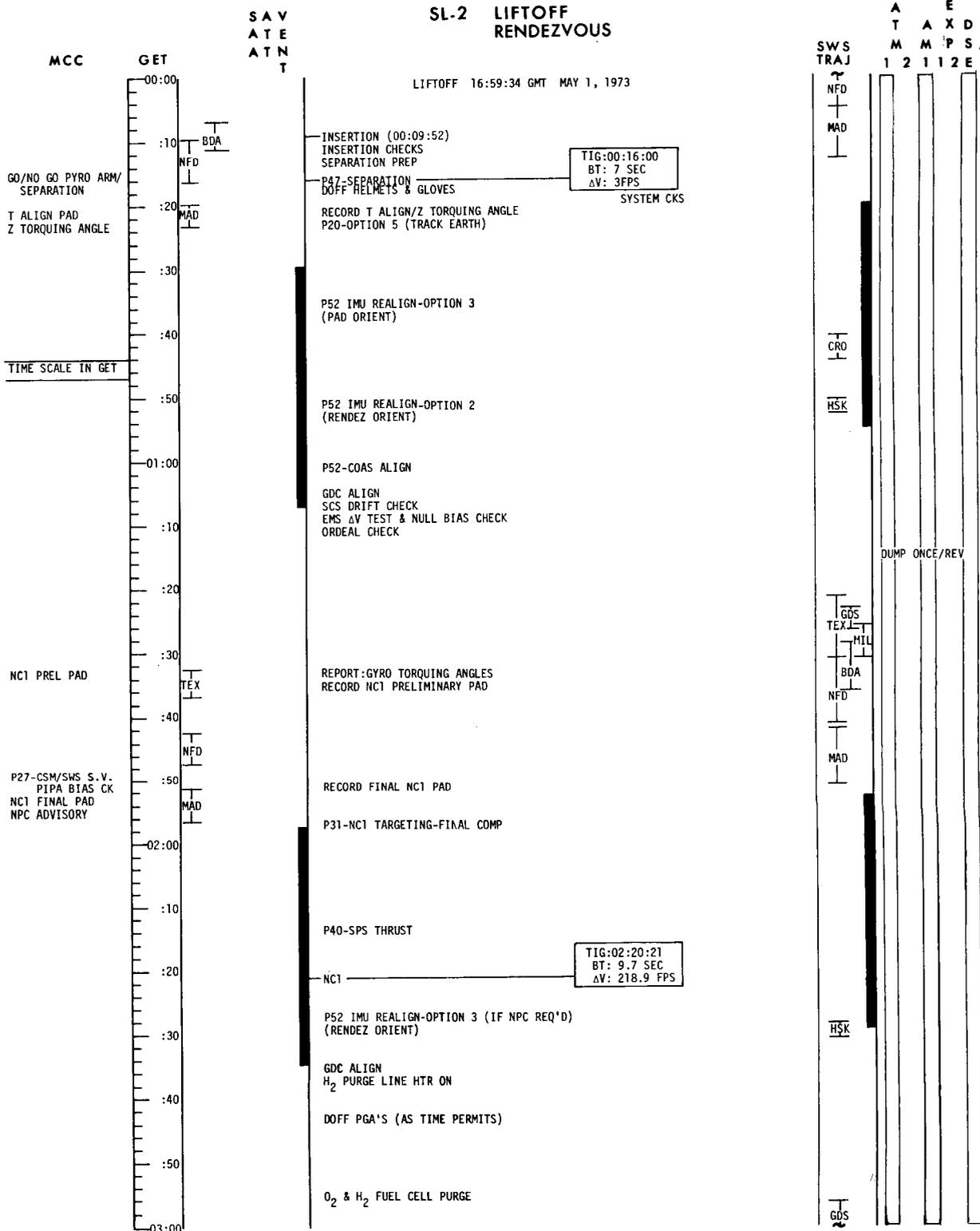
S	SOL A-SOL
A	
L	

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA	MOON PHASE
00:00 - 03:00	17:00 - 20:00	1/121	MAY 1, 1973			

SL-2 LIFTOFF RENDEZVOUS

RCDR
A E
T A X D UN-
M M P S ATT'D
1 2 1 1 2 E EXPS



MISSION	EDITION	PUBLICATION DATE
SL-2	REFERENCE	MAY 1, 1972

3-3

S	SOL A.SOL
A	
L	

FLIGHT PLANNING BRANCH

FLIGHT PLAN

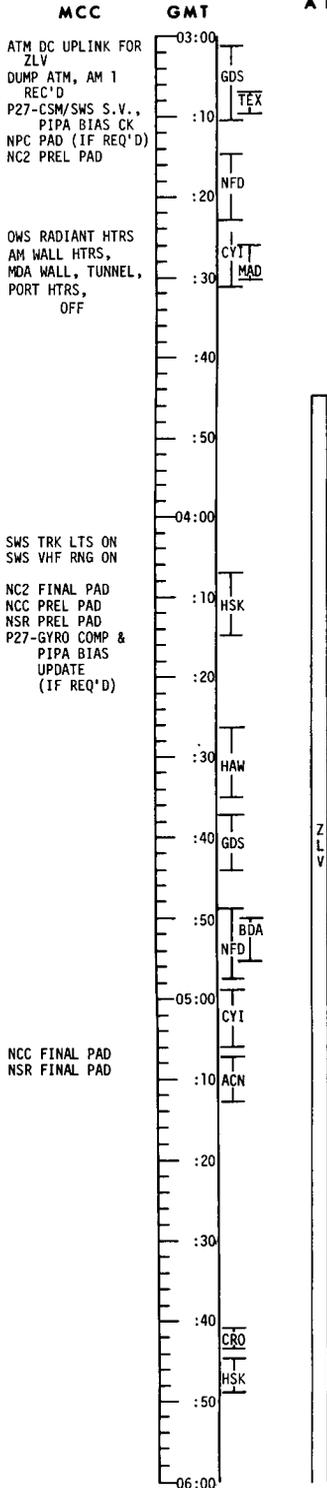
GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA	MOON PHASE
03:00 - 06:00		1/121	MAY 1, 1973			

SL-2 RENDEZVOUS

S
A
V
E
A
T
N
T

R
C
D
R

A
T
M
M
P
S
A
T
T
'
D
E
X
D
U
N
-
E
X
P
S



NC1 BURN STATUS REPORT

RECORD NPC BURN PAD, (IF REQ'D)
NC2 PRELIMINARY PAD

P38-NPC TARGETING (IF REQ'D)

NOTE: NPC MANEUVER PERFORMED IF MGA >60°
AT NCC OR NSR

NPC (IF REQUIRED) TIG: 03:28:00
BT: 0.0 SEC
ΔV: 0.0 FPS

P52-IMU REALIGN-OPTION 3
(RENDEZ ORIENT)

GDC ALIGN

VHF POWER UP

S
X
T

M
A
R
K
S

REPORT GYRO TORQUING ANGLES

RECORD NC2 FINAL PAD,
NCC PRELIMINARY PAD,
NSR PRELIMINARY PAD

P32-NC2 TARGETING - FINAL COMP

P40-SPS THRUST

NC2 (IF REQUIRED) TIG: 04:36:14
BT: 7.0 SEC
ΔV: 160.8 FPS

NC2 BURN STATUS REPORT

S
X
T
BACKUP CHART MEASUREMENTS ARE TAKEN AT
NCC-36,-32,-28,-20,-16, AND -12 MIN USING
EMS OR N77

M
A
R
K
S

RECORD NCC, NSR FINAL PAD

P33-NCC TARGETING - FINAL COMP

P40-SPS THRUST

NCC (IF REQUIRED) TIG: 05:22:21
BT: 1.2 SEC
ΔV: 30.4 FPS

S
X
T
BACKUP CHART MEASUREMENTS ARE TAKEN AT
NSR-28,-24,-20, AND -12 MIN USING
EMS OR N77

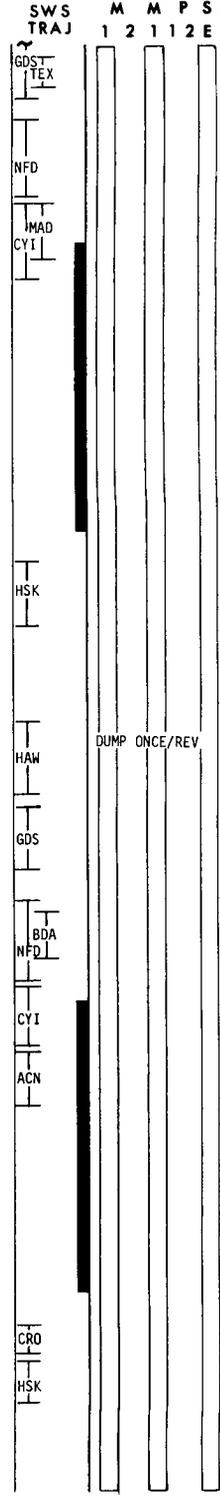
M
A
R
K
S

NCC BURN STATUS REPORT

P34-NSR TARGETING

P40-SPS THRUST TIG: 05:59:21
BT: 0.7 SEC
ΔV: 19.3 FPS

NSR



MISSION	EDITION	PUBLICATION DATE
SL-2	REFERENCE	MAY 1, 1972

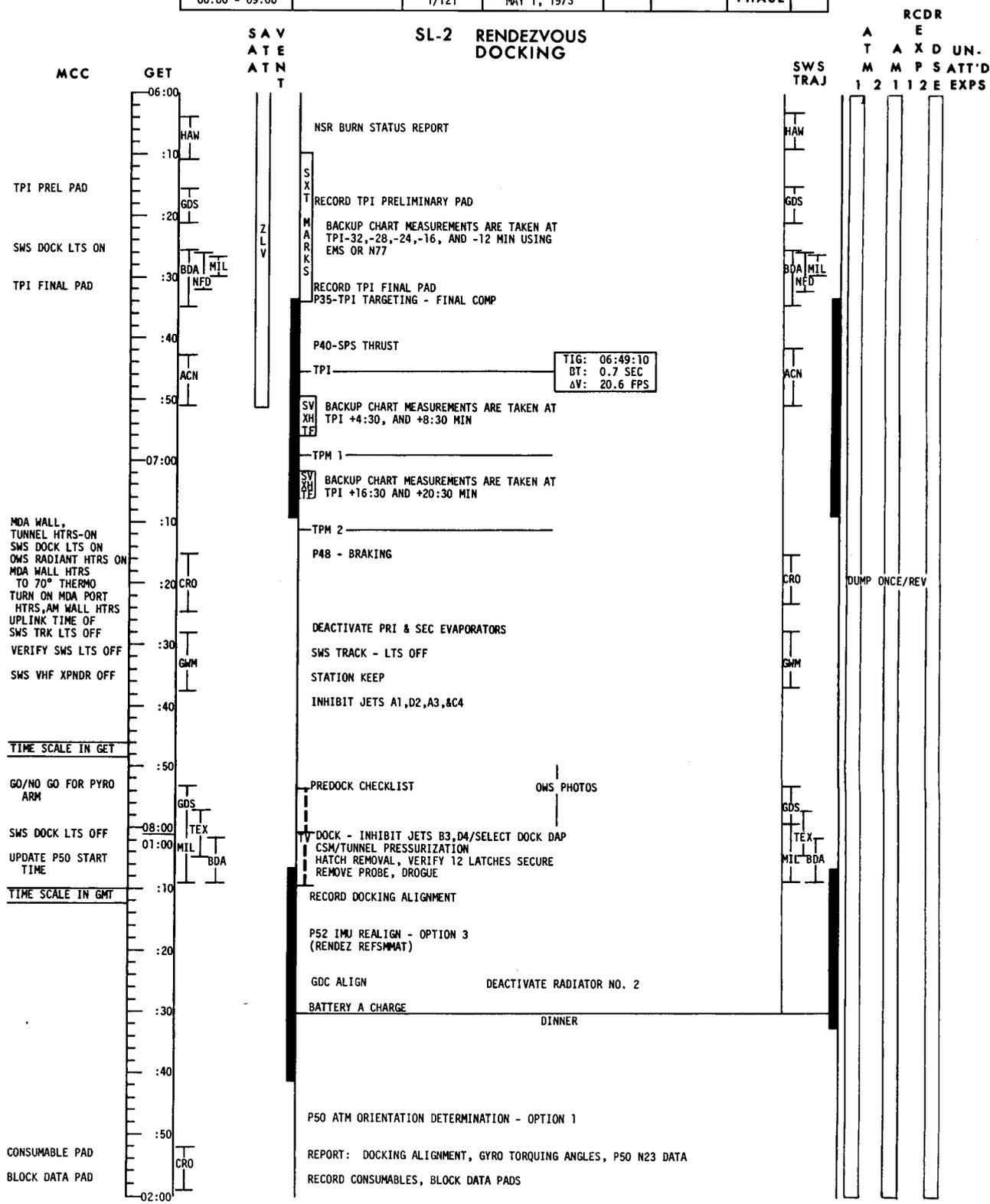
FLIGHT PLANNING BRANCH

S A L	S O L A - S O L
-------------	--------------------------------------

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA \downarrow	MOON
06:00 - 09:00		1/121	MAY 1, 1973			PHASE

SL-2 RENDEZVOUS DOCKING



MISSION	EDITION	PUBLICATION DATE
SL-2	REFERENCE	MAY 1, 1972

FLIGHT PLANNING BRANCH

S	A	L	SOL	A	SOL

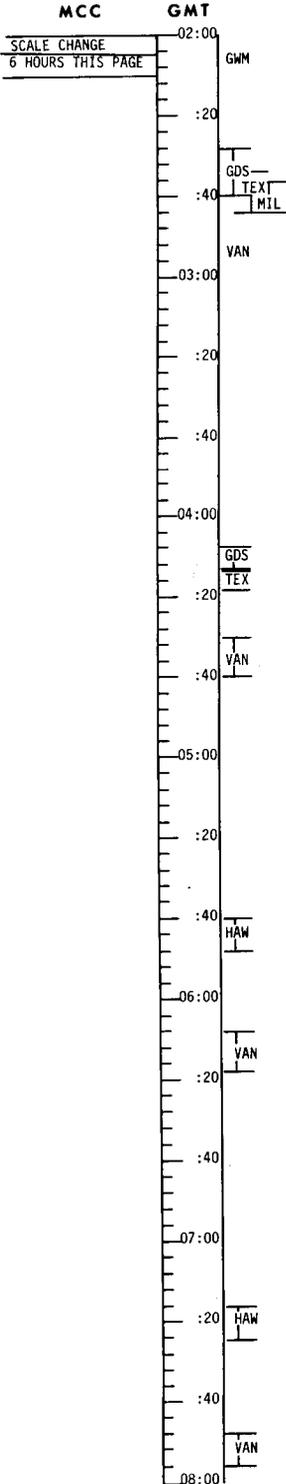
FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA	MOON PHASE
	02:00 - 08:00	1/122	MAY 1, 1973			

SAV
ATE
ATN
T

SL-2 DOCKED

RCDR
E
T A X D UN-
M M P S ATT'D
1 2 1 1 2 E EXPS



DINNER

REINSTALL TUNNEL HATCH

HATCH CABIN LEAK CHECK
L10H CANNISTER CHANGE

M071

REPORT: RADIATION ACCUMULATION
PERSONAL HYGIENE
CHLORINATE POTABLE WATER
PRESLEEP CHECKLIST

8 HOUR REST PERIOD

DUMP ONCE/REV

MISSION	EDITION	PUBLICATION DATE
SL2	REFERENCE	MAY 1, 1972

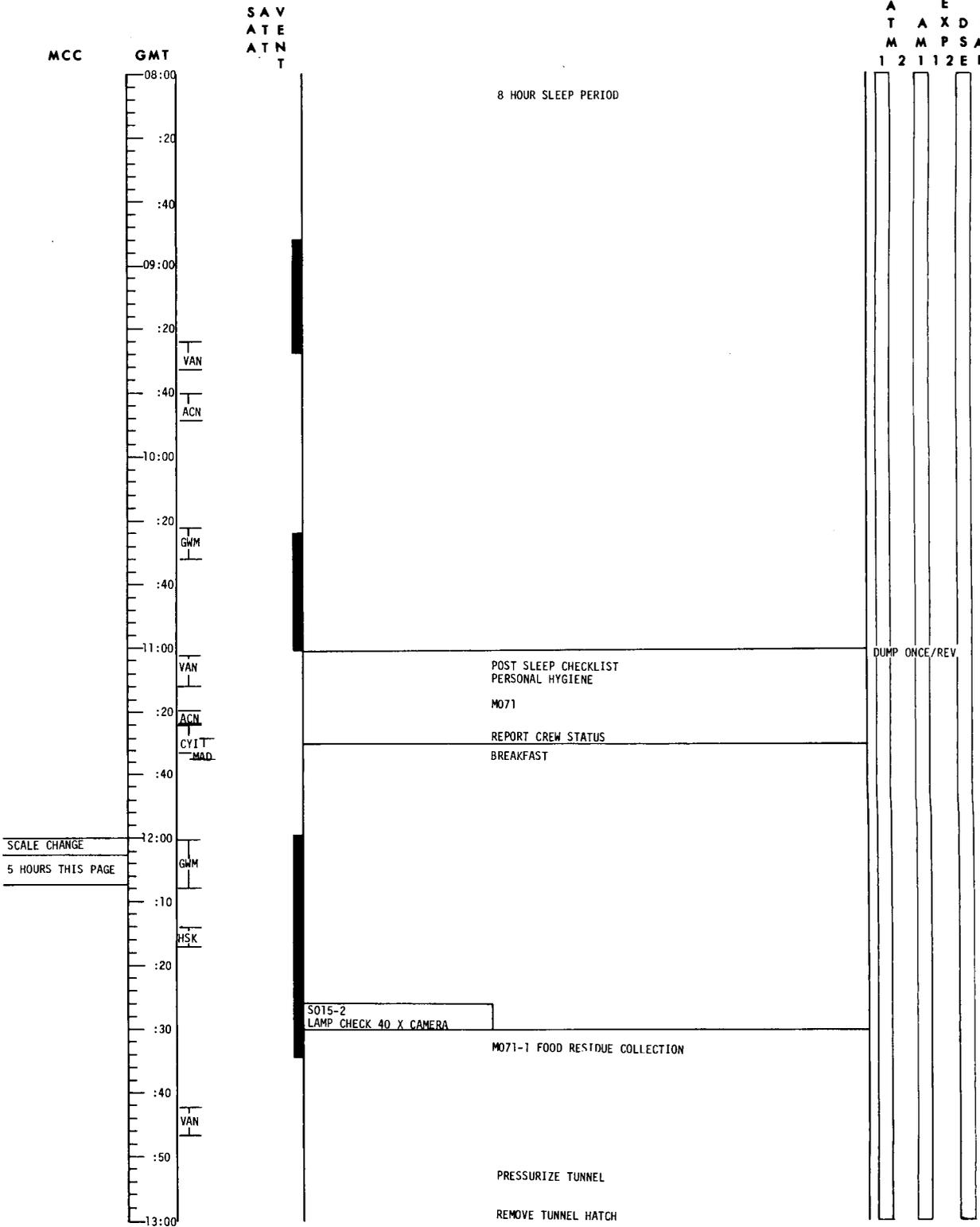
FLIGHT PLANNING BRANCH

S	SOL A-SOL
A	
L	

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA ϕ	MOON PHASE
	08:00 - 13:00	2/122	MAY 2, 1973			

RCDR
E
A X D UN-
M M P S ATT'D
1 2 1 1 2 E EXPS



MISSION	EDITION	PUBLICATION DATE
SL-2	REFERENCE	MAY 1, 1972

S A L	SOL A-SOL	
-------------	-----------	--

FLIGHT PLAN

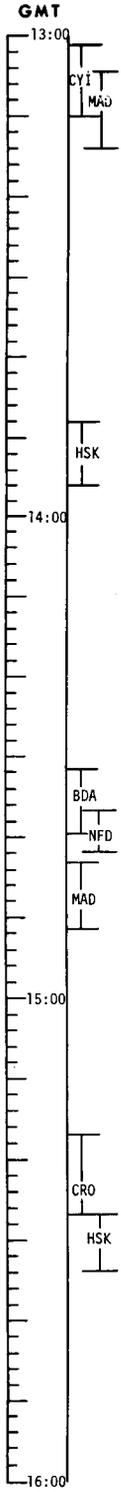
GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA	MOON PHASE
	13:00 - 16:00	2/122	MAY 2, 1973			

SL-2 ACTIVATION

S
A
V
A
T
E
A
T
N
T

R
C
D
R
E
A
T
X
D
U
N
M
M
P
S
A
T
T
D
1
2
1
1
2
E
E
X
P
S

MCC
CDT = 0800



CDR	SPT	PLT
MONITOR CSM	MDA/CSM PRESSURE EQUALIZATION	
DEACT SUIT CIRCUIT (CSM)	MDA HATCH - OPEN, SECURE	
SEC. GLY EVAP DRYOUT	MDA LIGHTS ON, VISUAL INSPECTION	
FLUSH WMS URA WASTE DUMP HTRS OFF	INSTALL S-190 WINDOW PROTECTOR (M)	SEAL MDA VENT
CONFIG CABIN TEMP VLVS	CONFIGURE SOP, SOMA (M)	REMOVE FIRE EXTING. RESTRAINTS (M)
CONFIG H ₂ O SYS. (CSM)		STS LIGHTS - ON
CM PREP FOR UMBILICAL CONNECTION	CONNECT COMM, EPS UMBILICALS (M)	VERIFY PPO ₂
CONFIGURE CSM COMM SYSTEM	CONFIGURE/VERIFY ATM C&D PANEL	MOL SW FANS - ON
END SEC GLY EVAP START PRI GLY EVAP		VERIFY EPS STATUS (A)
RESTOW PROBE, DROUGE IN MDA		CONFIGURE STS cb AND sw PANELS
INSTALL AIR INTERCHANGE DUCT (CSM, M)		RELOCATE MDA ICOM BOX, ACTIVATE COMM SYSTEM (M,A)
END PRI GLY EVAP D/O	ACTIVATE AM C&D STAR TRACKER, ATM C&D COOLANT LOOP	ACTIVATE TELEPRINTER
C/W CHECK (CSM)		ACTIVATE AM DUCT FAN (A)
CONFIG. O2 SYS (CSM)	AM LOCK ENTRY AND CONFIGURATION	ACTIVATE AM C/W RAPID ΔP AND FIRE TEST
P52 OPTION 3	AM AFT ENTRY AND CONFIGURATION	ACTIVATE CM CIRC FAN
RCS TRIM BURN 1A	MONITOR ATM C&D PANEL	FIRE SENSOR TEST
		WARNING AND CAUTION SYSTEMS TEST
P50 OPTION 2 (SUN SENSOR & STAR TRACKER)		O2/N2 SYSTEM ACTIVATION (A)
P52 OPTION 3 (SUN SENSOR & STAR TRACKER)		
		OWS ENTRY SEAL VENT
		INSTALL GAS INTERCHANGE DUCT (D)
		REMOVE FIRE EXTINGUISHER LAUNCH RESTRAINT
		OWS VISUAL INSPECTION
		ACTIVATE OWS CIRC FANS (RETURN TO AM) (E)
		CONFIGURE/VERIFY OWS SYSTEMS -

1	2	1	1	2	E	EXPS
DUMP ONCE/REV						

CDT = 1000

MISSION	EDITION	PUBLICATION DATE
SL-2	REFERENCE	MAY 1, 1972

3-8

S	A	L	SOL	A	SOL
---	---	---	-----	---	-----

FLIGHT PLANNING BRANCH

FLIGHT PLAN

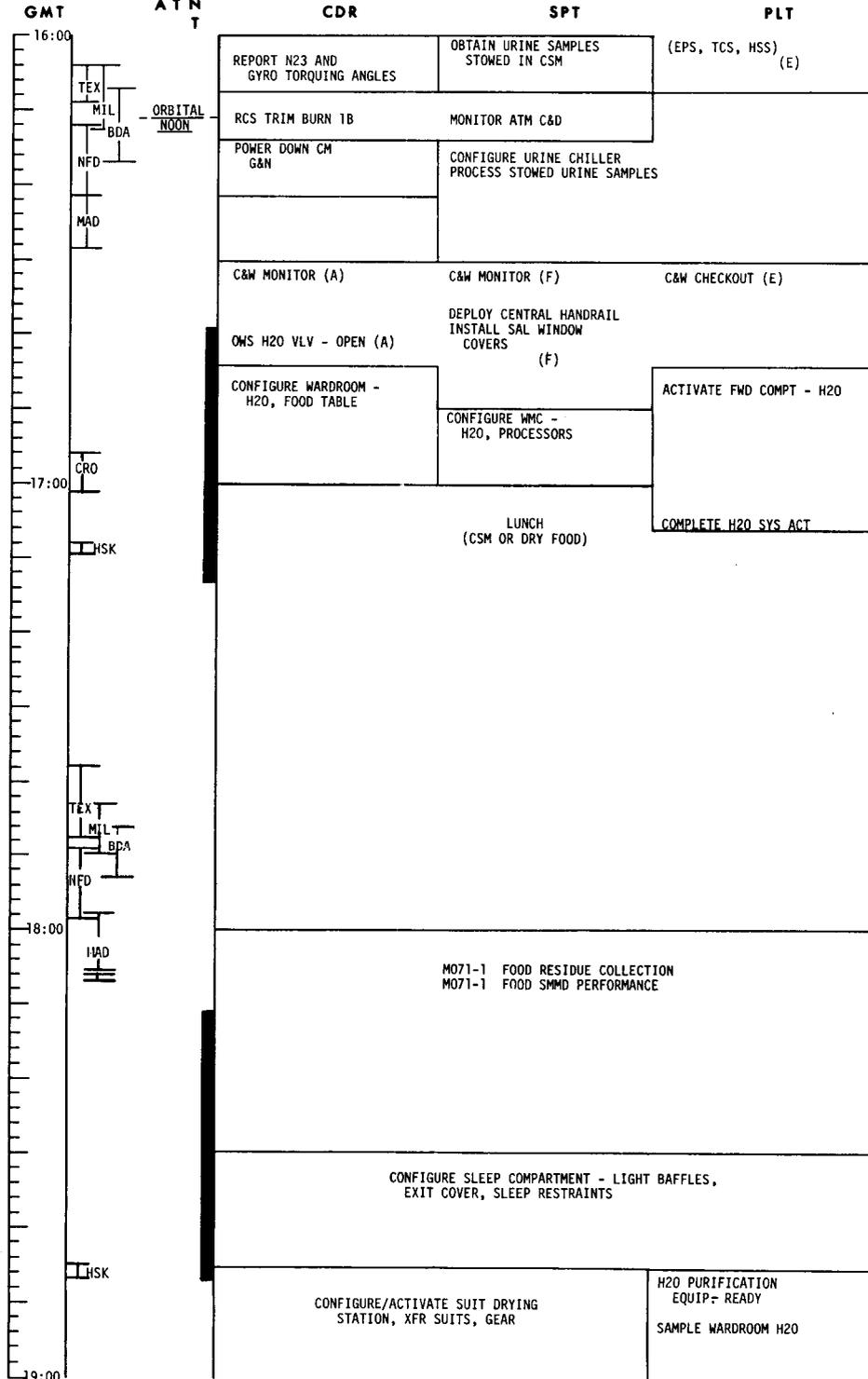
GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA	MOON PHASE
	16:00 - 19:00	2/122	MAY 2, 1973			

SAV
ATE
ATN
T

SL-2 ACTIVATION

RCDR
E
A T A X D UN-
M M P S ATT'D
1 2 1 1 2 E EXPS

MCC
GET = 2300



MISSION	EDITION	PUBLICATION DATE
SL-2	REFERENCE	MAY 1, 1972

FLIGHT PLANNING BRANCH

3-9

S	SOL A-SOL
A	
L	

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA	MOON PHASE
	19:00 - 22:00	2/122	MAY 2, 1973			

S
A
V
A
T
E
A
T
N
T

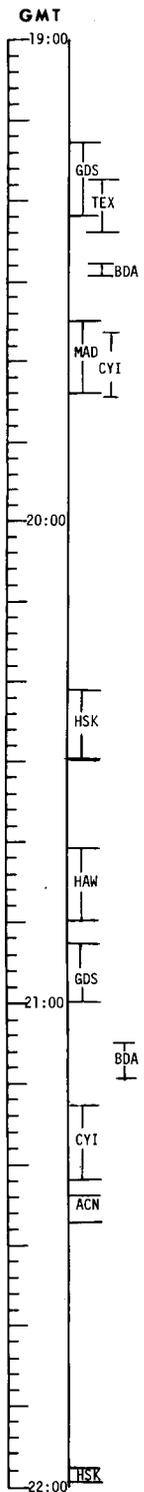
SL-2 ACTIVATION

RCDR
A
T
M
1
2
A
X
D
U
N
-
M
M
P
S
A
T
T
'
D
E
X
P
S

MCC

CDT = 1400

GMT	CDR	SPT	PLT
19:00			ACTIVATE TRASHLOCK
	VERIFY SWS/CSM POWER TIE CAPABILITY (C)	TRANSFER SOP/SOMAS	VERIFY SWS/CSM POWER TIE CAPABILITY (A)
	RELOCATE M168 STOWAGE CONTAINER (AID PLT)	REMOVE, STOW ITEMS FROM EXP. COMPT. FLOOR GRID.	RELOCATE M168 STOWAGE CONTAINER
	TRANSFER, STOW SPARE CONDENSATE MODULE (A TO F)	REMOVE VCS DUCT LAUNCH RESTRAINT	RECONFIGURE FOOD STOWAGE CONTAINERS
20:00			
	INSTALL CONDENSATION BLANKET (CSM)	CONFIGURE WARDROOM FOOD PREP AREA, WINDOW CONFIG. FOOD PREP FOR DINNER	ACTIVATE S190 WINDOW HEATER
	TRANSFER FOOD FROM CSM	OWS LAUNCH RESTRAINT REMOVAL	REPLACE SOLIDS TRAP ASS'Y
		CSM STOWAGE TRANSFERS	VENT STS WINDOWS
			MDA/AM LAUNCH RESTRAINTS REMOVAL
21:00	(COMPLETION OF OWS SYSTEMS ACTIVATION)		FILM VAULT PREP. - COVER DESSICANT PADS
22:00			



1	2	1	1	2	1	1	2	E	EXPS
DUMP ONCE/REV									

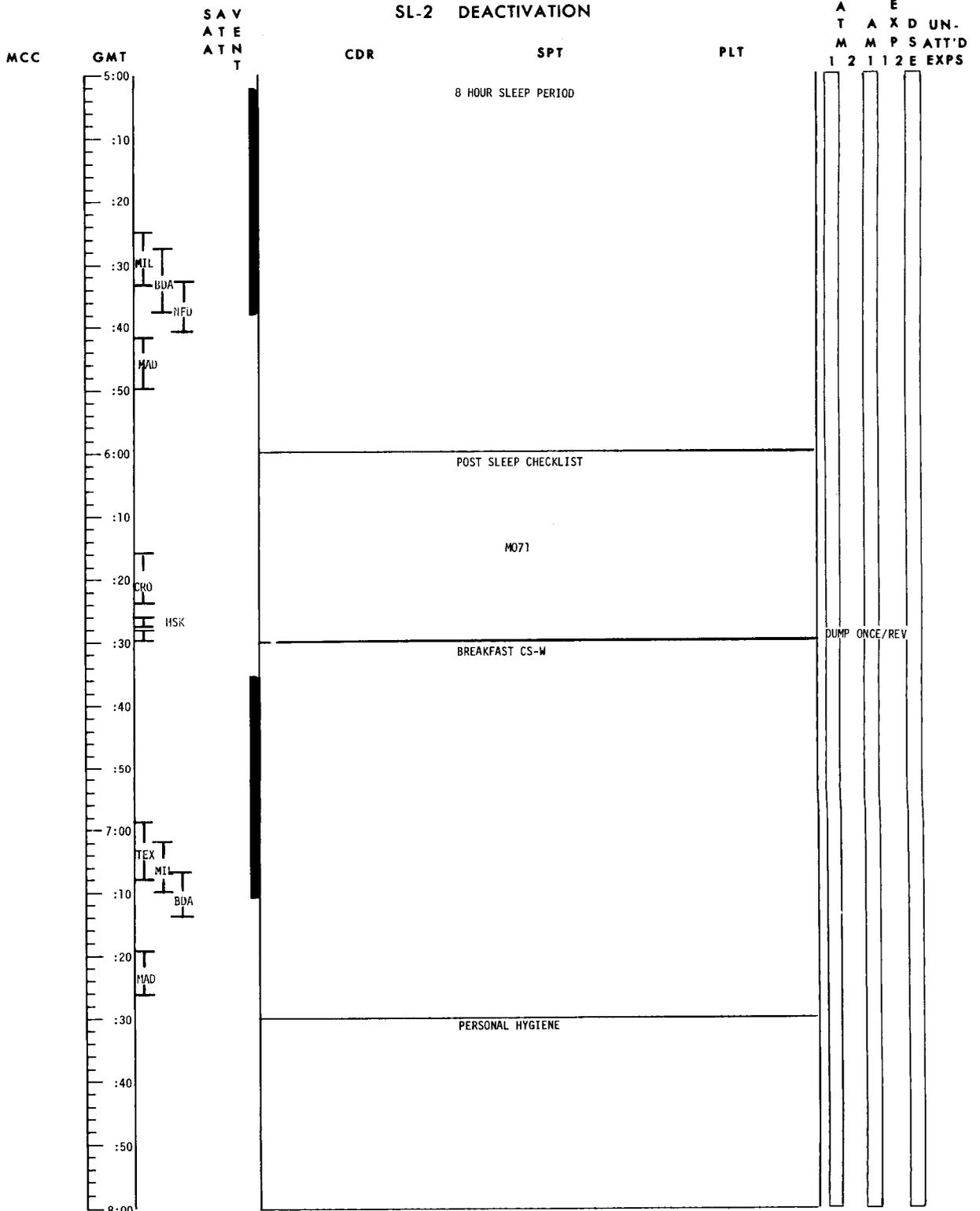
MISSION	EDITION	PUBLICATION DATE
SL-2	REFERENCE	MAY 1, 1972

S	A	L	SOL	A-SOL
---	---	---	-----	-------

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA ζ	MOON PHASE
	05:00 - 08:00	28/148	MAY 28, 1973			

SL-2 DEACTIVATION



MISSION	EDITION	PUBLICATION DATE
SL-2	REFERENCE	MAY 1, 1972

S	A	L	SOL	A-SOL

3-11

FLIGHT PLANNING BRANCH

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA	MOON PHASE
	08:00 - 11:00	28/148	MAY 28, 1973			

SL-2 DEACTIVATION

S
A
V
A
T
E
A
T
N
T

R
C
D
R
E
A
T
A
X
D
U
N
M
M
P
S
A
T
T
'D
1
2
1
1
2
E
E
X
P
S

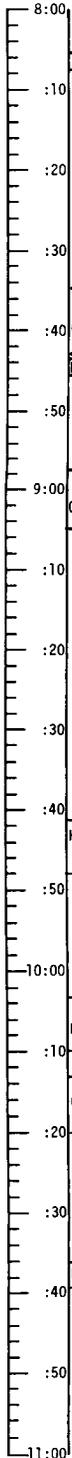
MCC

GMT

CDR

SPT

PLT



CSM STORAGE XFER XFER A7 CONTAINER TO F517 UNLOAD FECAL BUNDLE RETURN BAGS FROM A7 FLIGHT DATA FILE TO R1,R2,R3	DEACTIVATE SHOWER	DISASSEMBLE FOOD FREEZER BUNDLES
	SECURE BMMO	DEACTIVATE URINE FLUSH WATER SYSTEM
	REPLACE COLLECTION MODULE ODOR CONTROL FILTER	
GENERAL HOUSEKEEPING TASKS CLEANING AND DISINFECTING		REPLACE WMC CHARCOAL FILTER
DUMP TRASH AIRLOCK		
REVIEW ENTRY PROCEDURES		

DUMP ONCE/REV

MISSION	EDITION	PUBLICATION DATE
SL-2	REFERENCE	MAY 1, 1972

S	SOL	A-SOL
A		
L		

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA	MOON PHASE
	11:00 - 14:00	28/148	MAY 28, 1973			

SL-2 DEACTIVATION

MCC

GMT

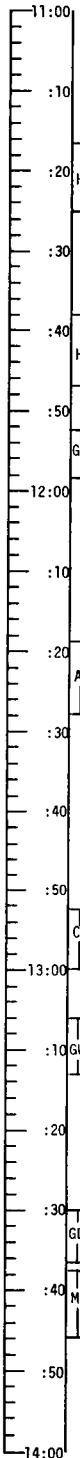
SAV
ATE
ATN
T

CDR

SPT

PLT

RCDR
E
A X D UN.
M M P S ATT'D
1 2 1 1 2 E EXPS



LUNCH	
M071-1 FOOD RESIDUE COLLECTION CS-W M071-1 FOOD SMMD PERFORMANCE CS-W	
STORAGE INVENTORY	
TRANSFER RETURN ITEMS TO CM	<div style="border: 1px solid black; padding: 5px; margin-bottom: 5px;">REPLACE MOLE SIEVES A AND B SOLIDS TRAP</div> <div style="border: 1px solid black; padding: 5px; margin-bottom: 5px;">DUMP CONDENSATE SYSTEM</div> <div style="border: 1px solid black; padding: 5px;">CONFIGURE CONDENSATE SYSTEM FOR STORAGE START SWS OVERPRESS (IF REQUIRED)</div>

DUMP ONCE/REV

MISSION	EDITION	PUBLICATION DATE
SL-2	REFERENCE	MAY 1, 1972

FLIGHT PLANNING BRANCH

3-13

S A L	SOL A-SOL	
-------------	-----------	--

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA	MOON PHASE
	14:00 - 17:00	28/148	MAY 28, 1973			

SL-2 DEACTIVATION

RCDR
 E
 T A X D U N
 M M P S A T T ' D
 1 2 1 1 2 E X P S

S A V
 A T E
 A T N
 T

MCC

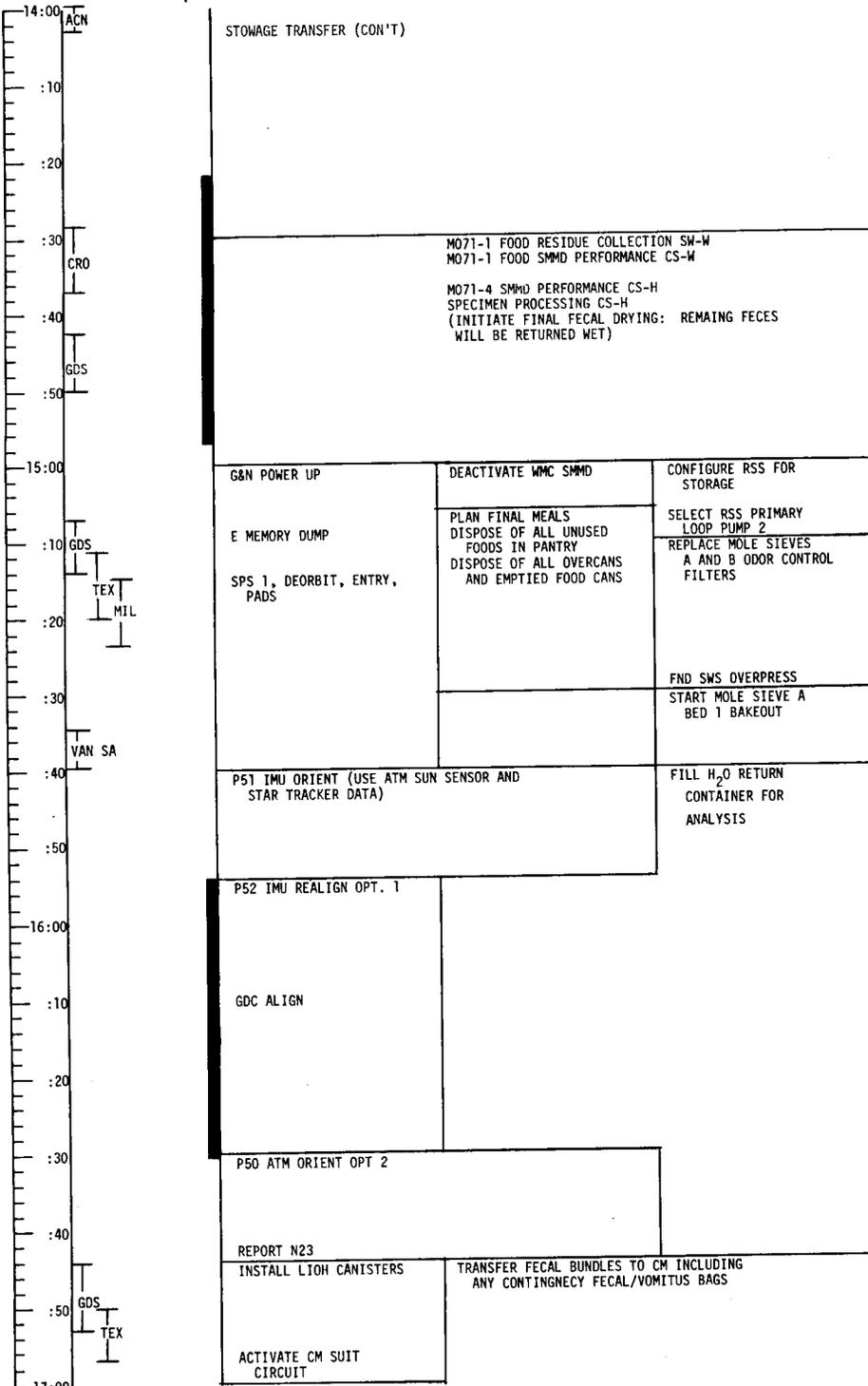
GMT

CDR

SPT

PLT

P-27 REFS/MAT,
CLOCK SYNC
STATE VECTOR
SPS 1 DEORBIT
& ENTRY PADS



DUMP ONCE/REV

MISSION	EDITION	PUBLICATION DATE
SL-2	REFERENCE	MAY 1, 1972

3-14

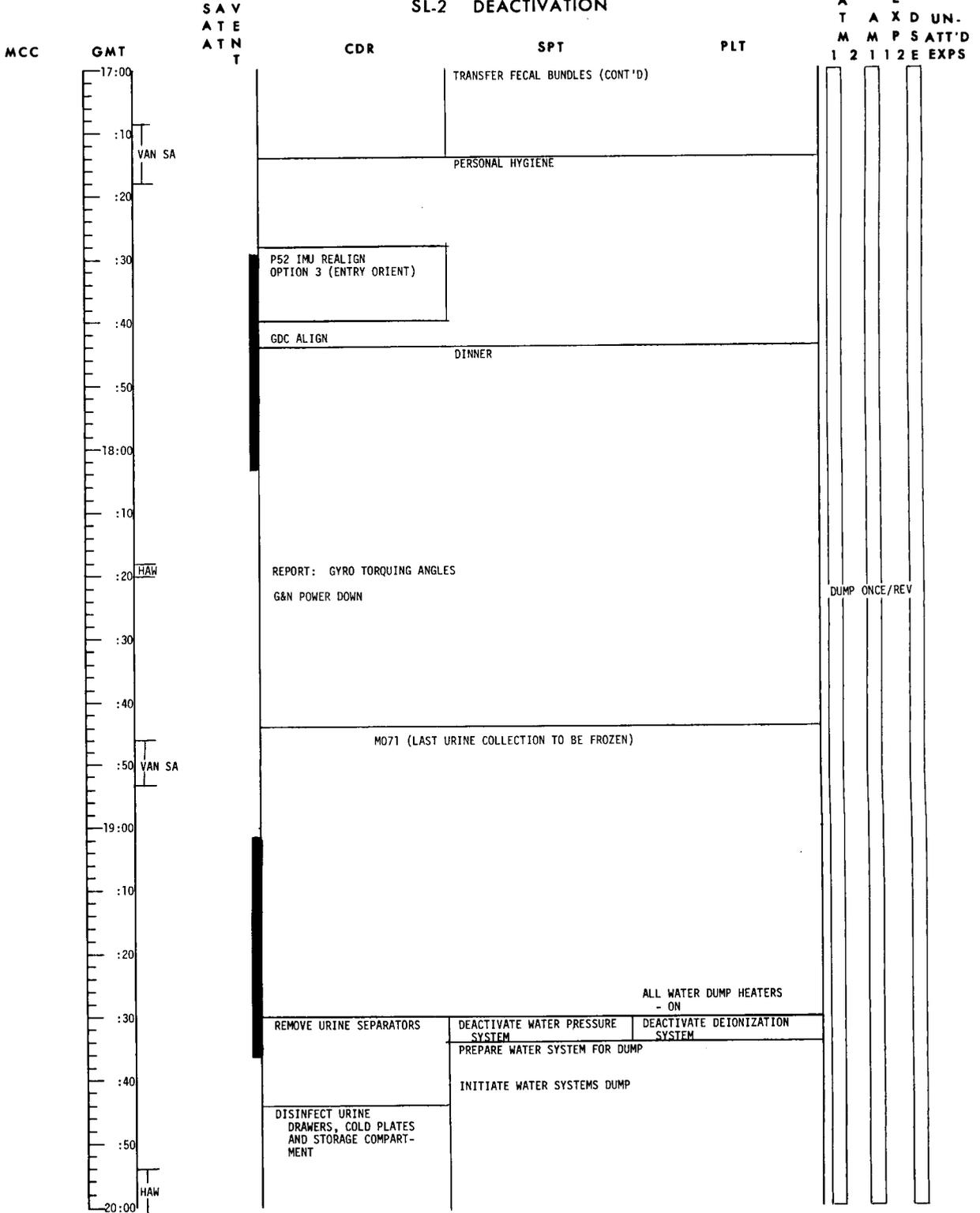
S A L	SOL A-SOL
-------------	-----------

FLIGHT PLANNING BRANCH

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA δ	MOON PHASE
	17:00 - 20:00	28/148	MAY 28, 1973			

SL-2 DEACTIVATION



MISSION	EDITION	PUBLICATION DATE
SL-2	REFERENCE	MAY 1, 1972

3-15

FLIGHT PLANNING BRANCH

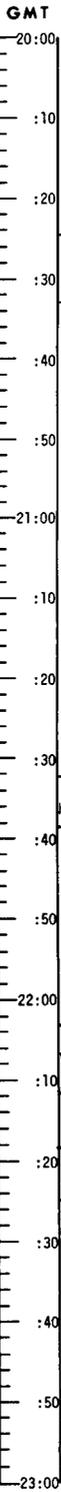
S	SOL	A-SOL					
A							

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA ↕	MOON PHASE
	20:00 - 23:00	28/148	MAY 28, 1973			

SL-2 DEACTIVATION

MCC



SAV
ATE
ATN
T

CDR

SPT

PLT

DEACTIVATE FOOD COMPARTMENT SMMO	TERMINATE WATER SYSTEMS DUMP CONFIGURE WATER SYSTEMS FOR VACUUM DRYING
PRESLEEP CHECKLIST	
MOLE SIEVE A BED 1 HEATER - OFF	
TERMINATE BED 1 BAKEOUT START BED 2 BAKEOUT	
6 HOUR SLEEP PERIOD	

RCDR
A E
T A X D UN.
M M P S ATT'D
1 2 1 1 2 E EXPS

--	--	--	--	--	--	--	--

DUMP ONCE/REV.

MISSION	EDITION	PUBLICATION DATE
SL-2	REFERENCE	MAY 1, 1972

3-16

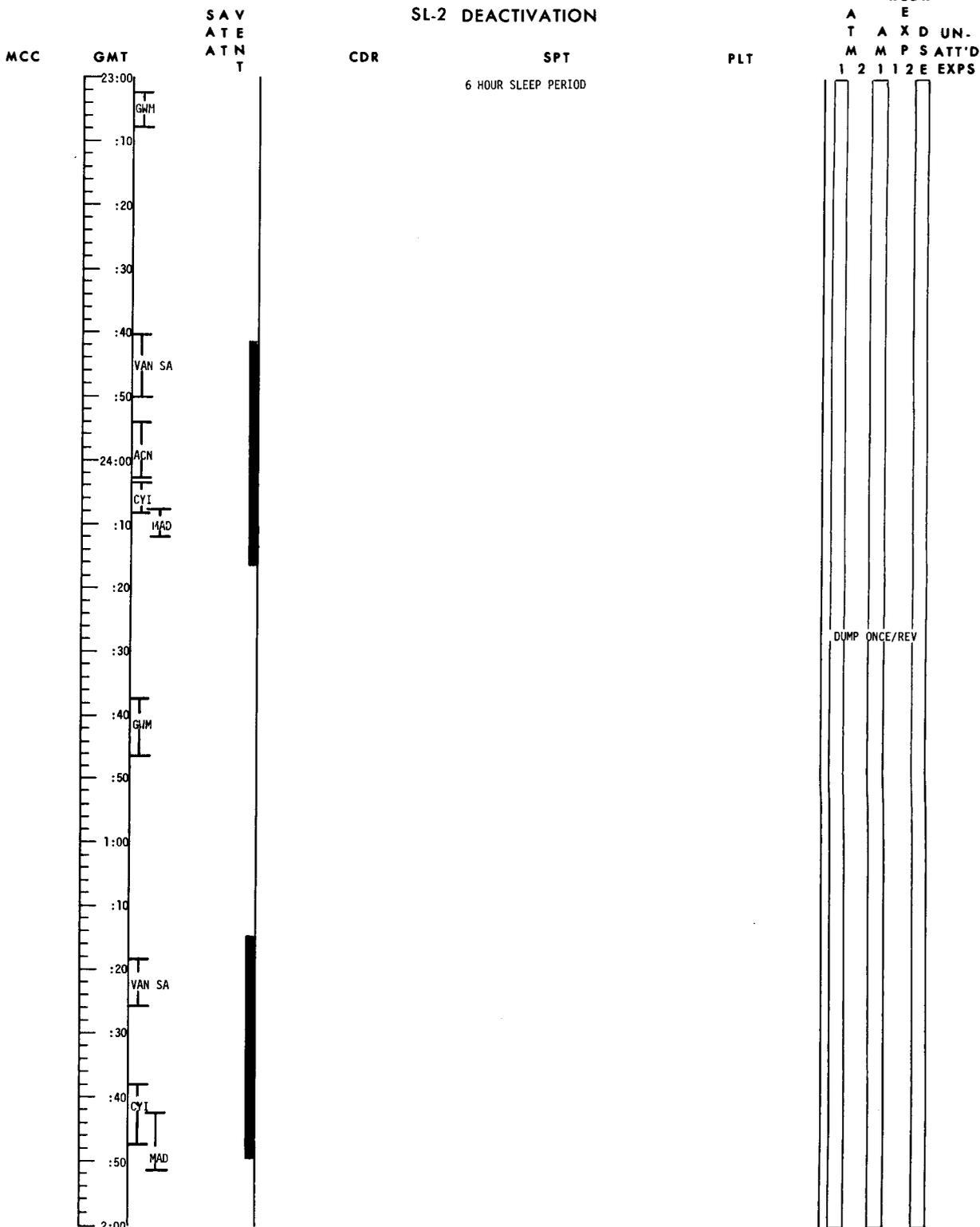
FLIGHT PLANNING BRANCH

S	A	L	SOL	A-SOL

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA	MOON PHASE
	23:00 - 02:00	28/148	MAY 28, 1973			

SL-2 DEACTIVATION



MISSION	EDITION	PUBLICATION DATE
SL-2	REFERENCE	MAY 1, 1972

FLIGHT PLANNING BRANCH

S	A	L	SOL	A-SOL

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA	MOON PHASE
	02:00 - 05:00	28/149	MAY 28, 1973			

SL-2 DEACTIVATION

RCDR

A
T
M
1
2
1
1
2
E
UN.
S
ATT'D
EXPS

S
A
V
A
T
E
A
T
N
T

MCC

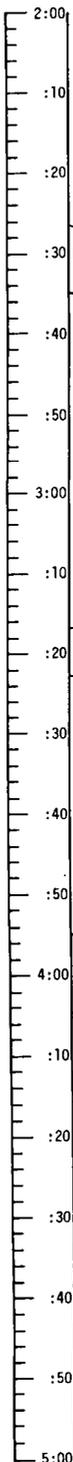
GMT

CDR

SPT

PLT

6 HOUR SLEEP PERIOD



P27 REFSMMAT,
CLOCK SYNC,
STATE VECTOR
E MEMORY DUMP

SPS1, DEORBIT,
ENTRY PADS

	FINAL M071 (CONTINGENCY BAGS)	MOLE SIEVE A BED 2 HEATER - OFF
		TERMINATE BED 2 BAKE- OUT, CLOSE OUT MOLE SIEVE
		TERMINATE WATER SYSTEMS VACUUM DRYING
GAN POWER UP E MEMORY DUMP	BREAKFAST	
SPS 1 DEORBIT ENTRY PADS	DEACTIVATE SLEEP COMPARTMENT	TRANSFER URINE SAMPLES FROM F577 TO A7 TRANSFER FINAL FECAL BUNDLE TRANSFER LWHS FROM U2 TO W720
VHF CHECK	FINAL TRASH LOCK USE CLOSE OUT TRASH LOCK	CLOSE OUT WASTE PROCESSOR
P51 IMU ORIENT (USE ATM SUN SENSOR AND STAR TRACKER DATA)		VERIFY EPS STATUS CONFIGURE OWS PANELS
P52 IMU REALIGN OPT 1 PREF (ENTRY ORIENT)	TRANSFER IMSS CONTAINER	
	DEACTIVATE SIA'S UNPLUG OWS SELENOID VENT	CONFIGURE OWS LIGHTING DEACTIVATE AFT & LOCK COMPARTMENT
TERMINATE AM/ATM - CSM POWER TRANSFER		TERMINATE AM/ATM - CSM POWER TRANSFER

DUMP ONCE/REV

MISSION	EDITION	PUBLICATION DATE
SL-2	REFERENCE	MAY 1, 1972

S	A	L	S	O	L	A	S	O	L
---	---	---	---	---	---	---	---	---	---

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA	MOON PHASE
	05:00 - 08:00	29/149	MAY 29, 1973			

SL - 2 DEACTIVATION

MCC

GMT

SAV
ATE
ATT
T

CDR

SPT

PLT

RCDR
E
A T A X D UN-
M M P S ATT'D
1 2 1 1 2 E EXPS

5:00	MAD	TERMINATE AM/ATM - CSM POWER TRANSFER (CONT'D)	CONFIGURE S190A	TERMINATE AM/ATM - CSM POWER TRANSFER (CONT'D)
:10		CHANGE LIQH CANISTERS		VERIFY LOAD SHARING AM/ATM BUSES
:20		ATD IN REMOVAL OF AIR INTERCHANGE DUCT	DEACTIVATE CM CIRCULATION FAN	
:30		PREHEAT CM RCS (IF REQUIRED) EPS, ECS, SPS QUIESCENT MODE TERMINATION	REMOVE AIR INTERCHANGE DUCT	INSTALL SOP DOCKING LOAD STRAPS
:40	CRO HSK		TRANSFER AND STOW THERMAL BLANKET	DEACTIVATE COMM. SYS DEACTIVATE CLOCK
:50			REMOVE POWER AND COMM UMBILICAL, AND STOW	DEACTIVATE TELEPRINTER
6:00		TRANSFER PROBE FROM M107 TO CM		DEACTIVATE O ₂ /N ₂ CONTROL SYSTEM
:10		DON SUIT		DON SUIT
:20	TEX MIL			
:30	BDA NFD		DON SUIT	
:40	MAD			
:50		EMS DEORBIT ENTRY TEST		CONFIGURE STS CONTROL PANELS
7:00		EMS DEORBIT/ENTRY TEST		CONFIGURE STS CIRCUIT BREAKER PANELS
:10		TRANSFER DROGUE FROM M105 TO CM		
:20	HSK		CONFIGURE STS/MDA LIGHTING	CLOSE MDA HATCH AND VERIFY EQUILIZATION VALVE - CLOSED INSTALL PROBE AND DROGUE INSTALL CM HATCH
:30			DEPRESS TUNNEL VERIFY CM MDA HATCH INTEGRITY	
:40		IMU REALIGN OPT 3 (ENTRY ORIENT)		
:50	GDS TEX			
8:00				

P27 S.V., SPS 1
TARGET LOAD

DUMP ONCE/REV

MISSION	EDITION	PUBLICATION DATE
SL-2	REFERENCE	MAY 1, 1972

3-19

S	A	SOL A-SOL

FLIGHT PLANNING BRANCH

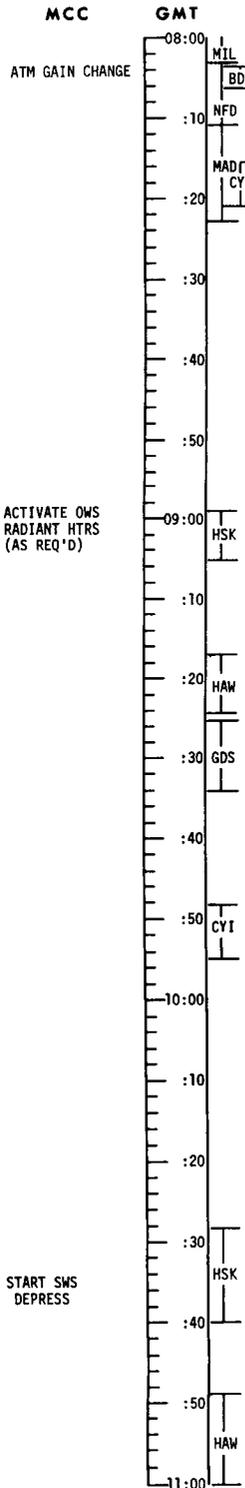
FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA	MOON PHASE
	08:00 - 11:00	29/149	MAY 29, 1973			

SL-2 ENTRY

SAVE ATENT

RCDR
A
E
T A X D UN.
M M P S ATT'D
1 2 1 1 2 E EXPS



SELECT UNDOCK DAP
UNDOCK → 08:01:06 GMT

FLY AROUND
OWS PHOTOS

P41-RCS THRUST

SEPARATION
P30 - EXTERNAL ΔV
V49 - MNVR TO PAD BURN ATT

P40 - SPS THRUST
TVC CHECK & PREP
HORIZON CHECK

SHAPING

DOFF PGA'S, HELMET & GLOVES

P52 - IMU ORIENT - OPTION 3
(DEORBIT ORIENT)

GDC ALIGN
LOGIC SEQUENCE CHECK
PRIMARY WATER EVAP. ACT
SECONDARY WATER EVAP. ACT

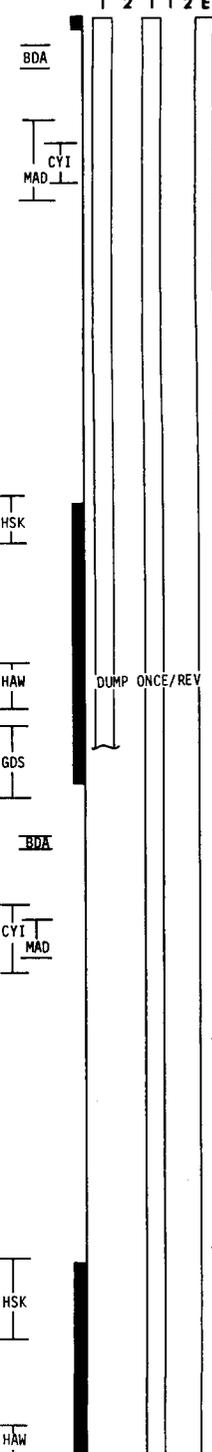
TIG: 09:00:06
BT: 11.5 SEC
ΔV: 5.0 FPS

TIG: 09:19:02
BT: 10.7 SEC
ΔV: 257.8 FPS

OWS RADIANT
HTRS (AS REQ'D)

SWS DEPRESS

SWS TRAJ



DUMP ONCE/REV

MISSION	EDITION	PUBLICATION DATE
SL-2	REFERENCE	MAY 1, 1972

3-20

FLIGHT PLANNING BRANCH

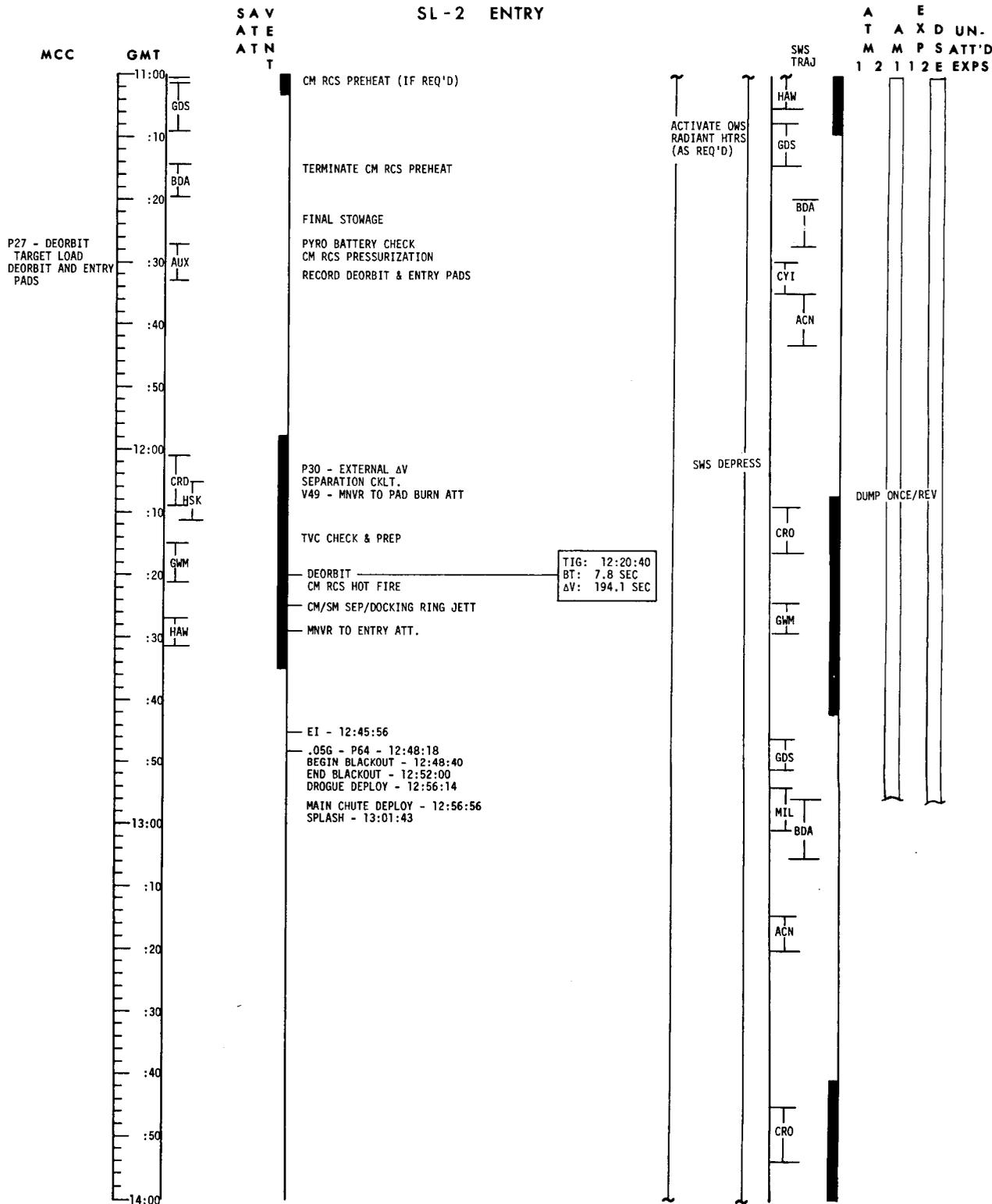
SOL

SOL A-SOL	
-----------	--

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA	MOON PHASE
	11:00 - 16:00	29/149	MAY 29, 1972			

SL-2 ENTRY



MISSION	EDITION	PUBLICATION DATE
SL-2	REFERENCE	MAY 1, 1972

3-21

S A L	SOL A-SOL
-------------	-----------

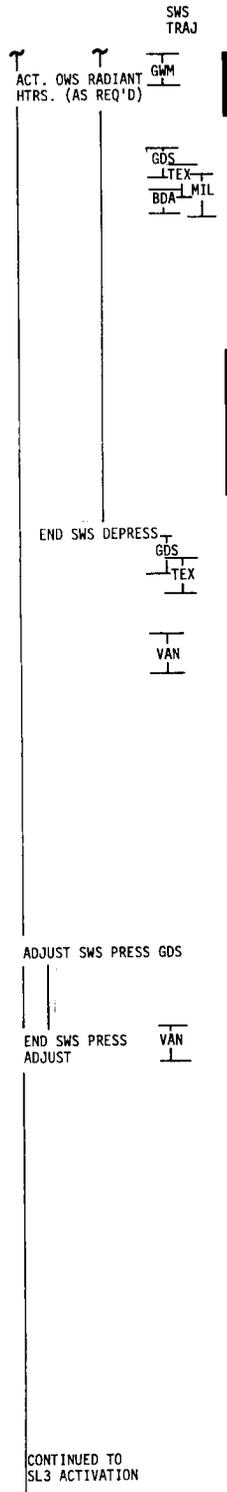
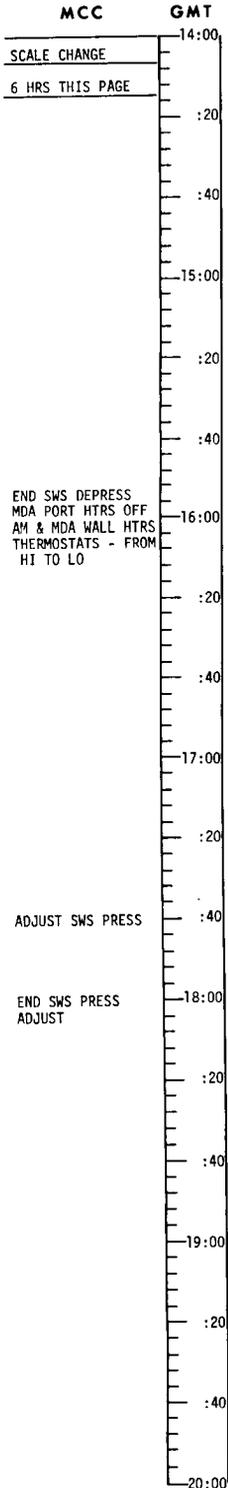
FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA	MOON PHASE

SL-3 UNMANNED

S
A
V
A
T
E
A
T
N
T

RCDR
A
T
M
1
2
1
1
2
E
X
P
S
A
X
D
U
N
M
P
S
A
T
T
'
D



MISSION	EDITION	PUBLICATION DATE
SL-2	REFERENCE	MAY 1, 1972

3-22

S	A	L	SOL	A-SOL

FLIGHT PLANNING BRANCH

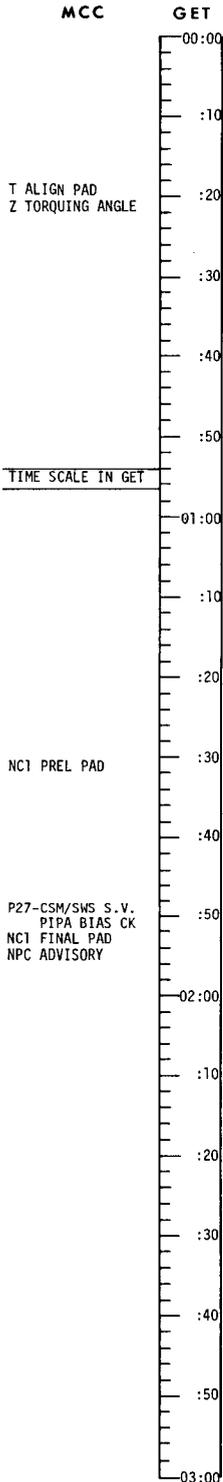
FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA ϕ	MOON PHASE
0:00 - 03:00		1/211	JULY 29, 1973			

SAV
ATE
ATN
T

SL-3 LIFTOFF RENDEZVOUS

RCDR
E
A T A X D UN-
M M P S ATT'D
1 2 1 1 2 E EXPS



LIFTOFF 04:43:57.9 GMT JULY 30, 1973

INSERTION (00:10:10)
INSERTION CHECKS
SEPARATION PREP

P47-SEPARATION TIG: 00:16:00
BT: 7.5 SEC
ΔV: 3.0 FPS

DOFF HELMET & GLOVES
RECORD T ALIGN/Z TORQUING ANGLE
P20-OPTION 5 (TRACK EARTH) SYSTEMS CKS

DOFF PGAS

P52 IMU REALIGN-OPTION 3 (PAD ORIENT)

P52 IMU REALIGN-OPTION 2 (RENDEZ ORIENT)

REPORT GYRO TORQUING ANGLES
RECORD NC1 PRELIMINARY PAD

P52-COAS ALIGN

GDC ALIGN

RECORD NC1 FINAL PAD

SCS DRIFT CHECK
EMS ΔV TEST & NULL BIAS CHECK
ORDEAL CHECK
P31-NC1 TARGETING-FINAL COMP

P40-SPS THRUST

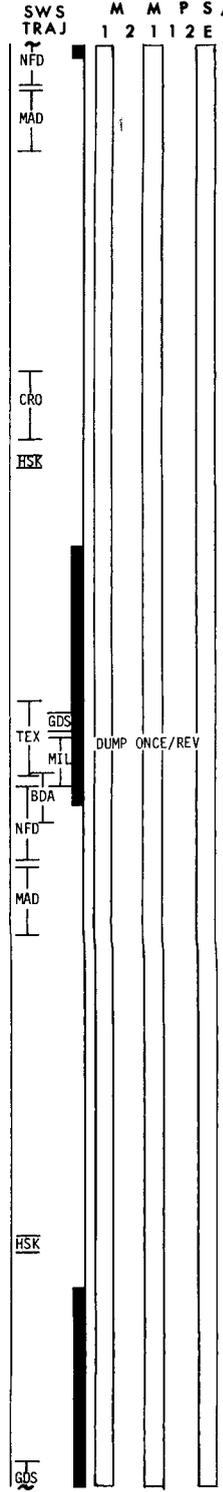
NC1 TIG: 02:21:18
BT: 9.9 SEC
ΔV: 219 FPS

H₂ PURGE LINE HTR ON

P52 IMU REALIGN-OPTION 3 (IF NPC REQ'D)
(RENDEZVOUS ORIENT)

GDC ALIGN

O₂ & H₂ FUEL CELL PURGE



MISSION	EDITION	PUBLICATION DATE
SL-3	REFERENCE	MAY 1, 1972

3-23

FLIGHT PLANNING BRANCH

S
A
L

SOL	A-SOL
-----	-------

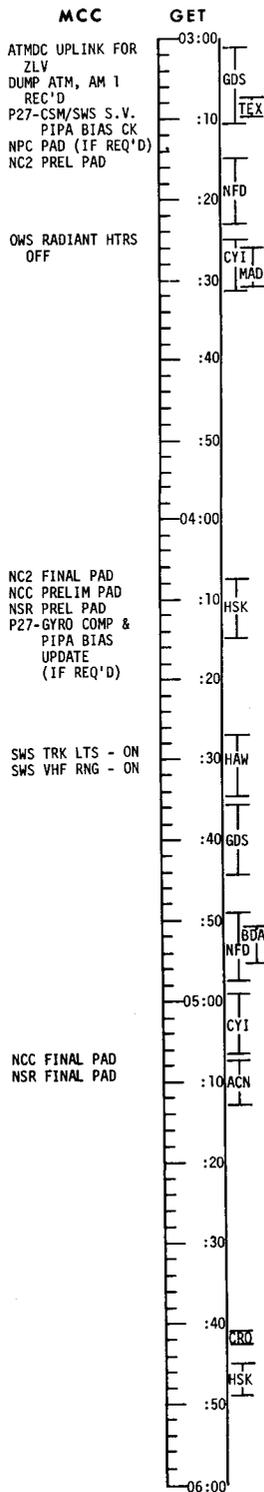
FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA ↔	MOON PHASE
03:00 - 06:00		2/211	JULY 30, 1973			

SL-3 RENDEZVOUS

RCDR
A E
T A X D UN-
M M P S ATT'D
1 2 1 1 2 E EXPS

S A V
A T E
A T N T
T



NC1 BURN STATUS REPORT
 RECORD NPC BURN PAD, (IF REQUIRED),
 NC2 PRELIMINARY PAD

P38-NPC TARGETING (IF REQ'D)

NOTE: NPC MANEUVER PERFORMED IF
 MGA >60° AT NCC OR NSR

NPC (IF REQUIRED) TIG: 03:28:00
BT: 0.0 SEC
ΔV: 0.0 FPS

S X T

VHF POWER UP

P52-IMU REALIGN - OPTION 3 (RENDEZVOUS ORIENT) RECORD NC2 FINAL PAD
NCC PRELIMINARY PAD
NSR PRELIMINARY PAD

GDC ALIGN

P32-NC2 TARGETING - FINAL COMP

P40-SPS THRUST

NC2 TIG: 4:37:10
BT: 7.1 SEC
ΔV: 161.0 FPS

BACKUP CHART MEASUREMENTS ARE TAKEN AT
 NCC-36,-32,-28,-20,-16, AND -12 MIN USING
 EMS OR N77

S X T

RECORD NCC, NSR FINAL PAD

P33-NCC TARGETING - FINAL COMP

P40-SPS THRUST

NCC TIG: 5:23:19
BT: 1.2 SEC
ΔV: 30.0 FPS

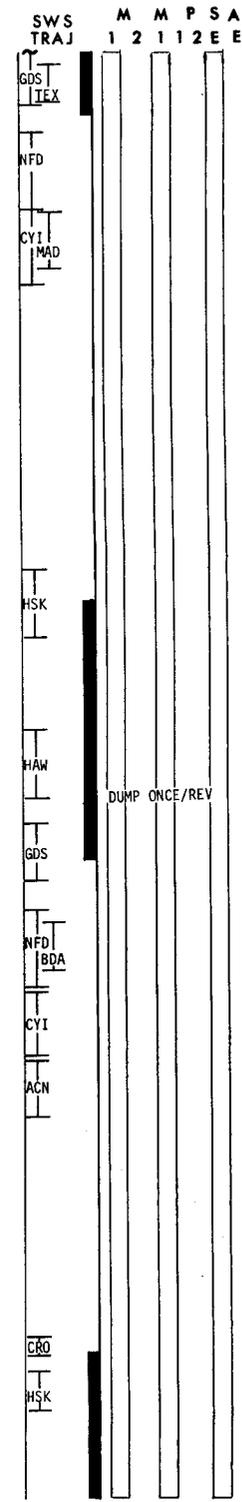
BACKUP CHART MEASUREMENTS ARE TAKEN AT
 NSR-28,-24,-20, AND -12 MIN USING
 EMS OR N77

S X T

HCC BURN STATUS REPORT

P34-NSR TARGETING - FINAL COMP

P40-SPS THRUST



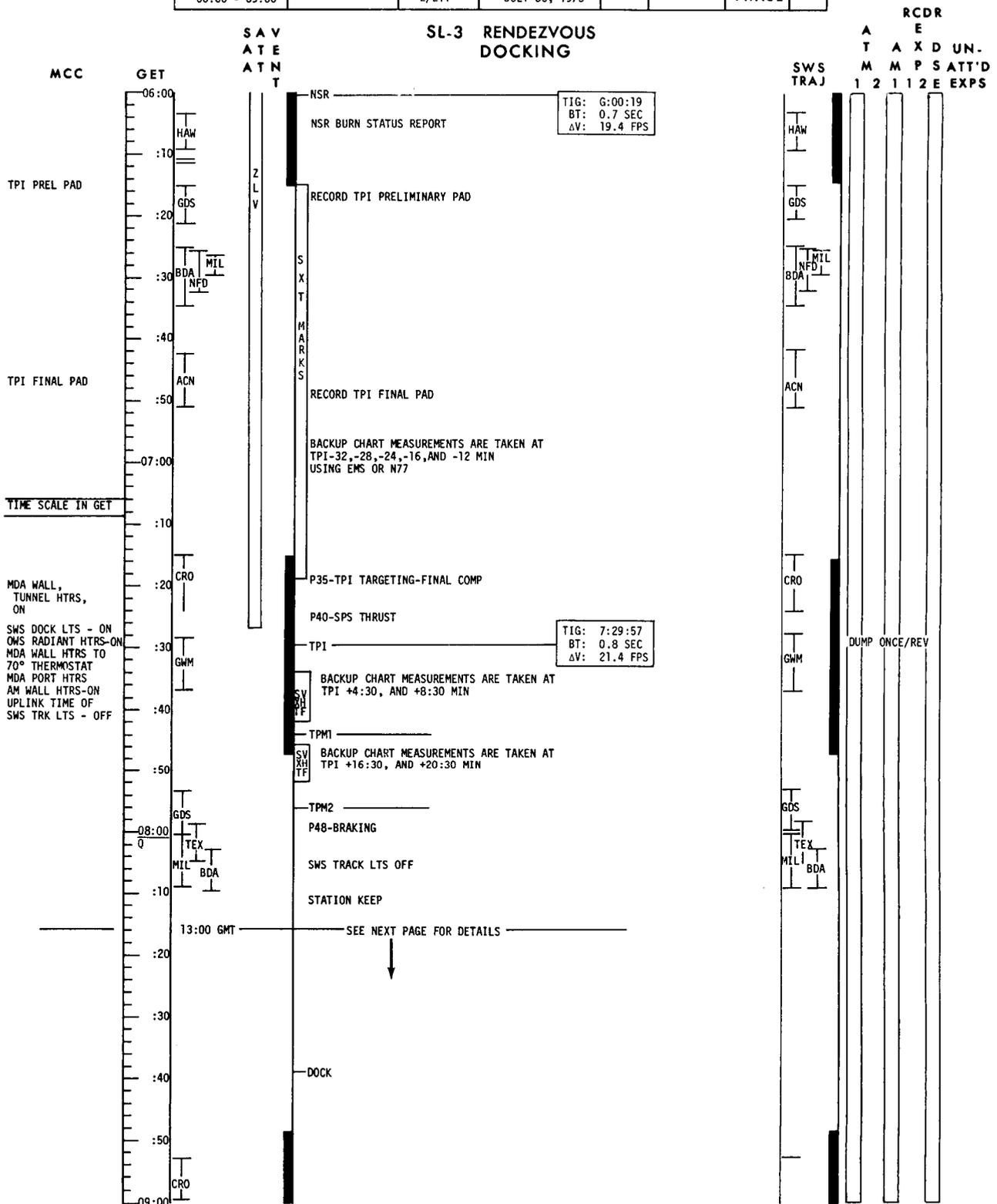
S A L
SOL A.SOL

MISSION	EDITION	PUBLICATION DATE
SL-3	REFERENCE	MAY 1, 1972

FLIGHT PLANNING BRANCH

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA	MOON	PHASE
06:00 - 09:00		2/211	JULY 30, 1973				



MDA WALL,
TUNNEL HTRS,
ON

SWS DOCK LTS - ON
OMS RADIANT HTRS-ON
MDA WALL HTRS TO
70° THERMOSTAT
MDA PORT HTRS
AM WALL HTRS-ON
UPLINK TIME OF
SWS TRK LTS - OFF

MISSION	EDITION	PUBLICATION DATE
SL-3	REFERENCE	MAY 1, 1972

FLIGHT PLANNING BRANCH

S	SOL	A-SOL
A		
L		

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA	MOON PHASE
	13:00 - 16:00	2/211	JULY 30, 1973			

SL-3 ACTIVATION

RCDR

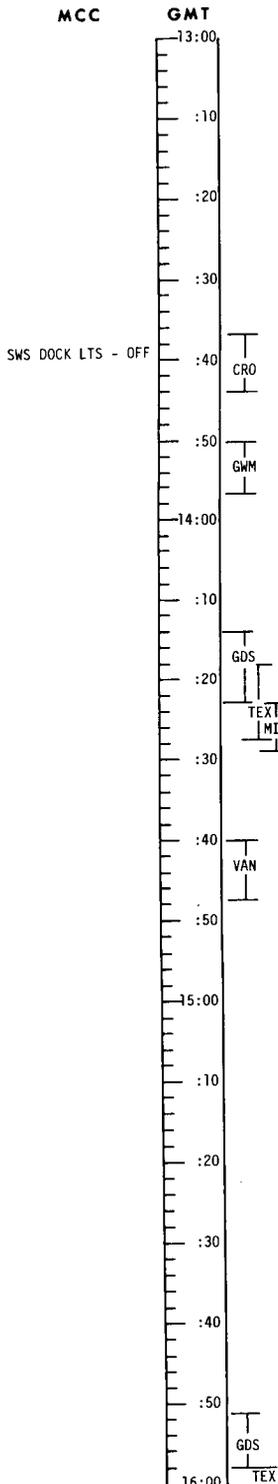
A
E
T A X D UN.
M M P S ATT'D
1 2 1 1 2 E EXPS

S
A
V
A
T
E
A
T
N
T

CDR

SPT

PLT



DEACTIVATE PRI & SEC EVAPORATORS INHIBIT JETS A1, D2, A3, & C4 STATION KEEP PREDOCK CHECKLIST OWS PHOTOS DOCK - INHIBIT JETS B3, D4/SELECT DOCK DAP CSM/TUNNEL PRESSURIZATION HATCH REMOVAL, VERIFY 12 LATCHES SECURE REMOVE PROBE, DROGUE RECORD DOCKING ALIGNMENT													
PS2 IMU REALIGN - OPTION 3 (RENDEZ ORIENT) GDC ALIGN BATTERY A CHARGE LUNCH													
M071													
DEACTIVATE CSM SUIT CIRCUIT (REMOVE BOTH LIQH CAN) SEC GLY EVAP DRYOUT FLUSH WMS URA WASTE DUMP HTRS OFF CONFIG CABIN TEMP VLVS CONFIG CSM H ₂ O SYS	<table border="1" style="width: 100%; border-collapse: collapse;"> <tr> <td colspan="2" style="text-align: center;">MDA/CSM PRESSURE EQUALIZATION MDA HATCH OPEN, SECURE</td> </tr> <tr> <td style="width: 50%; text-align: center;">MDA VISUAL INSPECTION</td> <td style="width: 50%; text-align: center;">AM VISUAL INSPECTION</td> </tr> <tr> <td style="text-align: center;">INSTALL S190 WINDOW COVER</td> <td style="text-align: center;">VERIFY PPO₂, REPLACE PPO₂ SENSORS MOLE SIEVE FANS - ON</td> </tr> <tr> <td style="text-align: center;">CONFIGURE, STOW SOP - SOMA</td> <td style="text-align: center;">CONFIGURE STS CB, SWITCH PANELS VERIFY EPS STATUS</td> </tr> <tr> <td style="text-align: center;">CONNECT COMM, EPS UMBILICALS</td> <td style="text-align: center;">RELOCATE MDA ICOM ACTIVATE TELEPRINTER AM DUCT FAN ACTIVATE COMM SYSTEM</td> </tr> <tr> <td style="text-align: center;">CONFIGURE AND ACTIVATE ATM C&D PANEL, COOLANT LOOP</td> <td style="text-align: center;">ACTIVATE O₂/N₂ CONTROL SYSTEM</td> </tr> </table>	MDA/CSM PRESSURE EQUALIZATION MDA HATCH OPEN, SECURE		MDA VISUAL INSPECTION	AM VISUAL INSPECTION	INSTALL S190 WINDOW COVER	VERIFY PPO ₂ , REPLACE PPO ₂ SENSORS MOLE SIEVE FANS - ON	CONFIGURE, STOW SOP - SOMA	CONFIGURE STS CB, SWITCH PANELS VERIFY EPS STATUS	CONNECT COMM, EPS UMBILICALS	RELOCATE MDA ICOM ACTIVATE TELEPRINTER AM DUCT FAN ACTIVATE COMM SYSTEM	CONFIGURE AND ACTIVATE ATM C&D PANEL, COOLANT LOOP	ACTIVATE O ₂ /N ₂ CONTROL SYSTEM
MDA/CSM PRESSURE EQUALIZATION MDA HATCH OPEN, SECURE													
MDA VISUAL INSPECTION	AM VISUAL INSPECTION												
INSTALL S190 WINDOW COVER	VERIFY PPO ₂ , REPLACE PPO ₂ SENSORS MOLE SIEVE FANS - ON												
CONFIGURE, STOW SOP - SOMA	CONFIGURE STS CB, SWITCH PANELS VERIFY EPS STATUS												
CONNECT COMM, EPS UMBILICALS	RELOCATE MDA ICOM ACTIVATE TELEPRINTER AM DUCT FAN ACTIVATE COMM SYSTEM												
CONFIGURE AND ACTIVATE ATM C&D PANEL, COOLANT LOOP	ACTIVATE O ₂ /N ₂ CONTROL SYSTEM												
STOW PROBE, DROGUE IN MDA													

DUMP ONCE/REV

MISSION	EDITION	PUBLICATION DATE
SL-3	REFERENCE	MAY 1, 1972

3-26

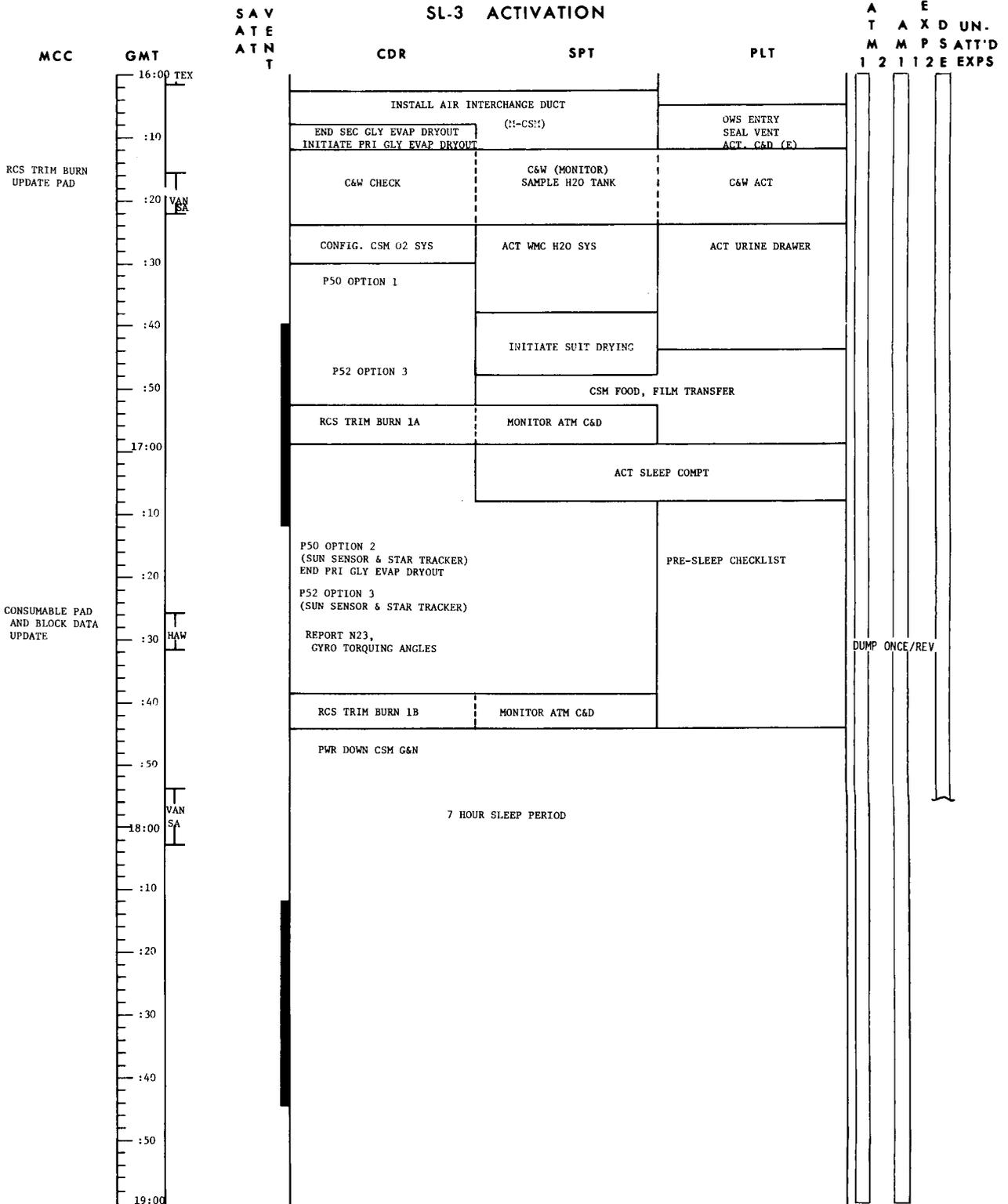
S	SOL A-SOL
A	
L	

FLIGHT PLANNING BRANCH

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA ↕	MOON PHASE
	16:00 - 19:00	2/211	JULY 30, 1973			

SL-3 ACTIVATION



MISSION	EDITION	PUBLICATION DATE
SL-3	REFERENCE	MAY 1, 1972

FLIGHT PLANNING BRANCH

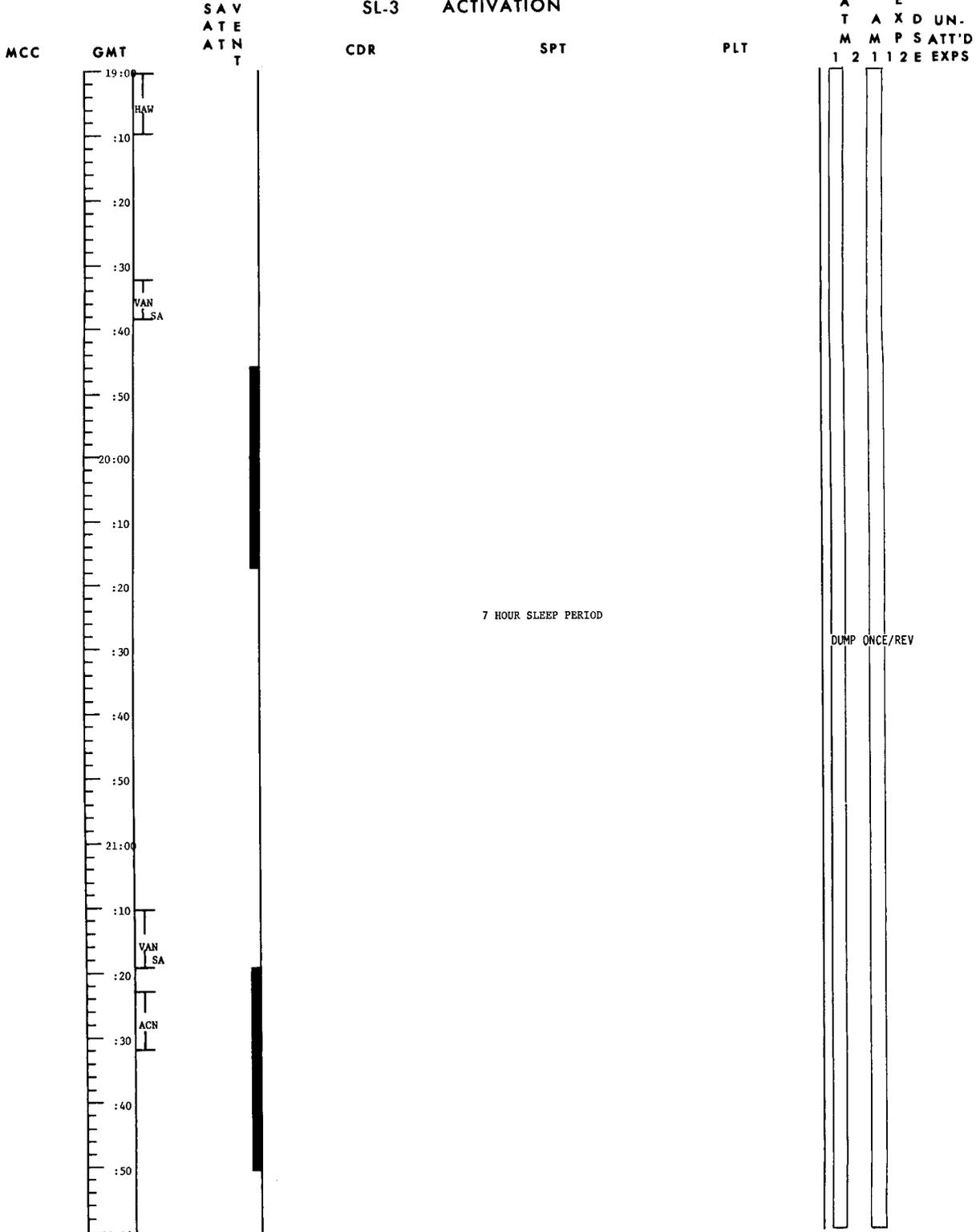
3-27

S A L	SOL A-SOL	SOL A-SOL

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA ϕ	MOON PHASE
	19:00 - 22:00	2/211	JULY 30, 1973			

SL-3 ACTIVATION



MISSION	EDITION	PUBLICATION DATE
SL-3	REFERENCE	MAY 1, 1972

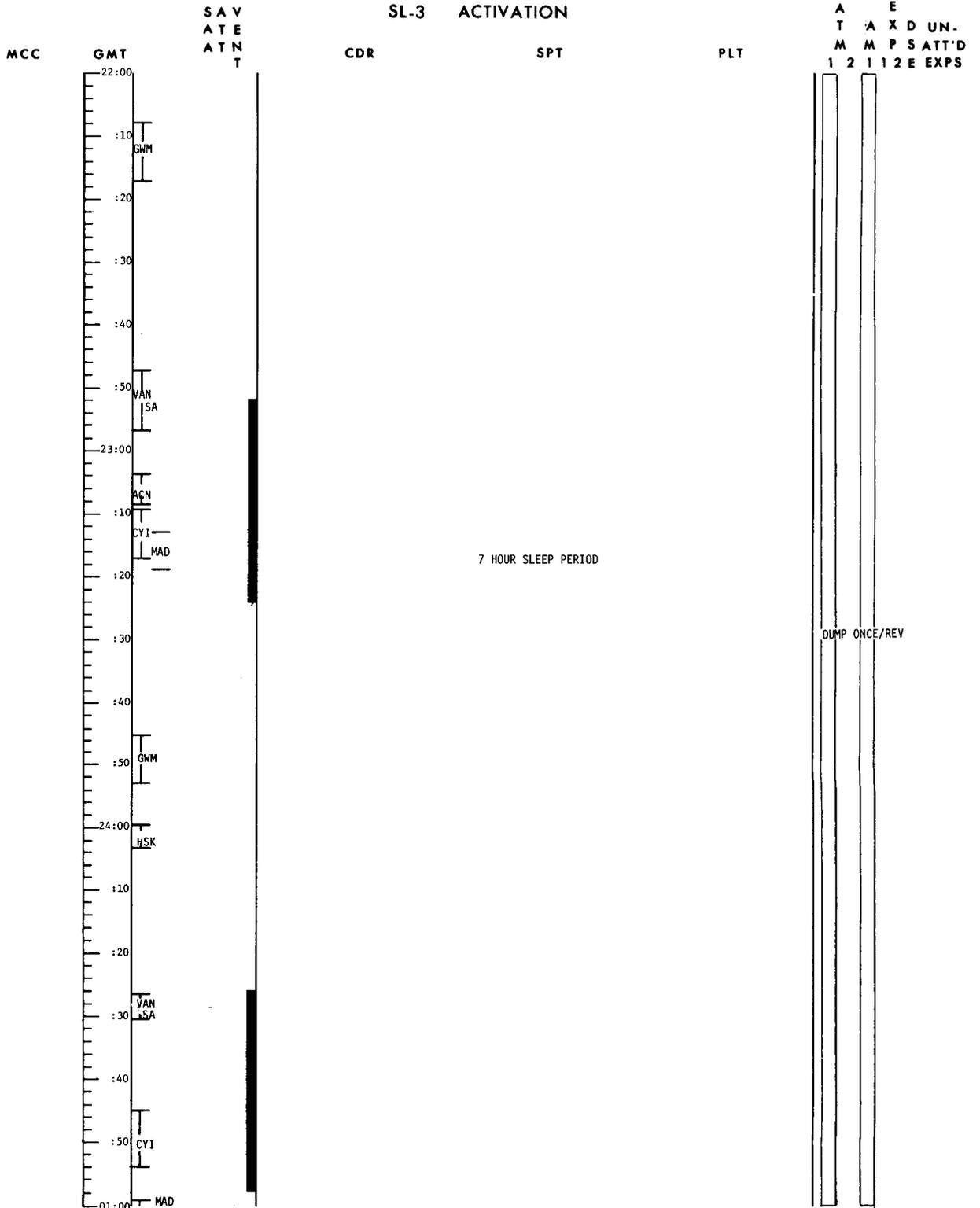
FLIGHT PLANNING BRANCH

SOL	A-SOL

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA	MOON PHASE
	22:00 - 01:00	2/211	JULY 30, 1973			

SL-3 ACTIVATION



MISSION	EDITION	PUBLICATION DATE
SL-3	REFERENCE	MAY 1, 1972

FLIGHT PLANNING BRANCH

3-29

S	SOL A-SOL
A	
L	

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA	MOON PHASE
	04:00 - 07:00	2/212	JULY 30, 1973			

SL-3 ACTIVATION

RCDR
 A E
 T A X D UN..
 M M P S ATT'D
 1 2 1 1 2 E EXPS

MCC

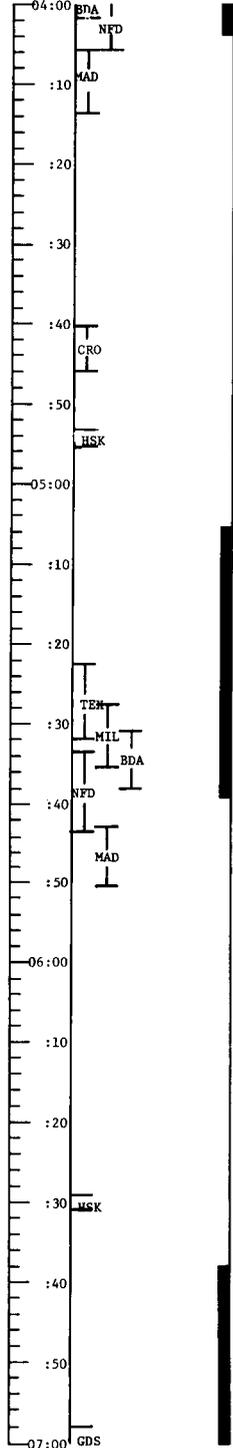
GMT

SAV
 ATE
 ATN
 T

CDR

SPT

PLT



ACT MOL SV A (CONT'D)
ACT S190 HTR INSTALL CSM CONDENSATION BLANKET
CONTINUE WARDROOM H2O SYS ACT
COMPLETE WARDROOM H2O SYS ACT
H2O CHILLER SAMPLE FILL DEORBIT H2O BAGS

DUMP ONCE/REV

MISSION	EDITION	PUBLICATION DATE
SL-3	REFERENCE	MAY 1, 1972

FLIGHT PLANNING BRANCH

3-31

S A L	SOL A-SOL	
-------------	-----------	--

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA ↕	MOON PHASE
	05:00 - 08:00	56/266	SEPT. 23, 1973			

SL-3 DEACTIVATION

RCDR
 A E
 T A X D UN.
 M M P S ATT'D
 1 2 1 1 2 E EXPS

MCC

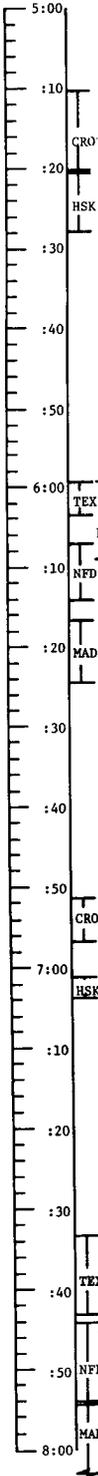
GMT

S
A
V
A
T
E
A
T
N
T

CDR

SPT

PLT



HOUR SLEEP PERIOD CS-S
POST SLEEP CHECKLIST
BREAKFAST CS-W
PERSONAL HYGIENE

MISSION	EDITION	PUBLICATION DATE
SL-3	REFERENCE	MAY 1, 1972

3-32

S	SOL	A-SOL
A		
L		

FLIGHT PLANNING BRANCH

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA	MOON
	08:00 - 11:00	56/266	SEPT. 23, 1973			PHASE

SL-3 DEACTIVATION

MCC

GMT

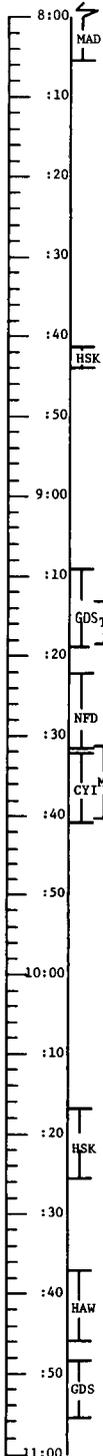
SAV
ATE
ATN
T

CDR

SPT

PLT

RCDR
E
A
T
A
X
D
UN.
M
M
P
S
ATT'D
1
2
1
1
2
E
EXPS



CSM STORAGE XPER XPER A7 CONTAINER TO F517 UNLOAD FECAL BUNDLE RETURN BAGS FROM A7 XPER FLIGHT DATA FILE TO R1, R2, R3	DEACTIVATE SHOWER	DISASSEMBLE FOOD FREEZER BUNDLES
	SECURE BMD	DEACTIVATE URINE FLUSH WATER SYSTEM
	REPLACE COLLECTION MODULE ODOR CONTROL FILTER	
GENERAL HOUSEKEEPING TASKS CLEANING AND DISINFECTING		REPLACE WMC CHARCOAL FILTER

DUMP TRASH AIRLOCK

REVIEW ENTRY PROCEDURES

DUMP ONCE/REV

MISSION	EDITION	PUBLICATION DATE
SL-3	REFERENCE	MAY 1, 1972

3-33

S	SOL A-SOL
A	
L	

FLIGHT PLANNING BRANCH

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA ↕	MOON PHASE
	11:00 - 14:00	56/266	SEPT. 23, 1973			

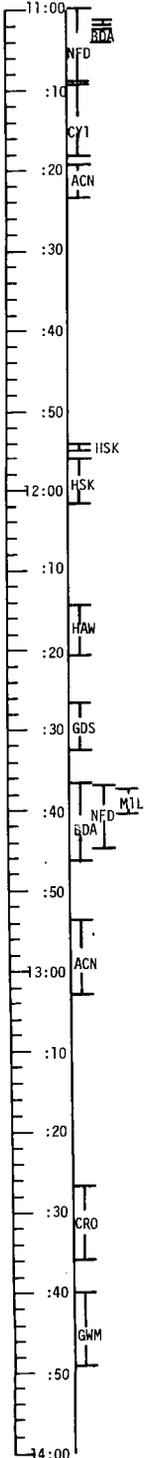
SL-3 DEACTIVATION

S
A
V
A
T
E
A
T
N
T

RCDR
E
T
A
X
D
U
N.
M
M
P
S
A
T
T
'
D
1
2
1
1
2
E
X
P
S

MCC

GMT



CDR	SPT	PLT
LUNCH CS-W		
M071-1 FOOD RESIDUE COLLECTION CS-W M071-1 FOOD SMMD PERFORMANCE CS-W		
AM/MDA HOUSEKEEPING		
		REPLACE MOLE SIEVES A AND B SOLIDS TRAP
		DUMP CONDENSATE SYSTEM
		CONFIGURE CONDENSATE SYSTEM FOR STOWAGE
		START SMS OVERPRESS (IF REQUIRED)

DUMP ONCE/REV				
DUMP ONCE/REV				

MISSION	EDITION	PUBLICATION DATE
SL-3	REFERENCE	MAY 1, 1972

3-34

S A L	SOL A-SOL
-------------	-----------

FLIGHT PLANNING BRANCH

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA \pm	MOON PHASE
	14:00 - 17:00	56/266	SEPT. 23, 1973			

SL-3 DEACTIVATION

MCC

P-27 REFSMMAT
CLOCK SYNC STATE
VECTOR
SPS 1 DEORBIT &
ENTRY PADS
E MEMORY DUMP

GMT

14:00
:10
:20
:30
:40
:50
15:00
:10
:20
:30
:40
:50
16:00
:10
:20
:30
:40
:50
17:00

**SAV
ATE
ATN
T**

GDS
TEX
MIL
BDA

CRO
GWM

GDS
TEX
MIL

VAN SA

CDR	SPT	PLT
G&N POWER UP	TRANSFER FF&AL BUNDLES TO CM INCLUDING ANY CONTINGENCY FECAL/VOMITUS BAGS	
SPS 1, DEORBIT, ENTRY PADS	DUMP ONCE/REV	
P51 IMU ORIENT (USE ATM SUN SENSOR AND STAR TRACKER DATA)	FILL WATER RETURN CONTAINER (FOR ANALYSIS)	
P52 IMU REALIGN OPT. 1	CONFIGURE RSS FOR STORAGE	
GDC ALIGN	REPLACE MOLE SIEVES A AND B ODOR CONTROL FILTERS	
M071-4 FECAL COLLECTION CS-H M071-4 SMMD PERFORMANCE CS-H (INITIATE FINAL FECAL DRYING: REMAINING FECES WILL BE RETURNED WET		DUMP ONCE/REV
P50 ATM ORIENT OPT 2	REPORT N23	
INSTALL LIQH CANISTERS	DEACTIVATE WMC SMMD	
ACTIVATE CM SUIT CIRCUIT	PLAN FINAL MEALS DISPOSE OF ALL UNUSED FOODS IN PANTRY DISPOSE OF ALL OVERCANS AND EMPTIED FOOD CANS	
P52 IMU REALIGN OPTION 3 (ENTRY ORIENT)	GDC ALIGN	

RCDR

A
T
M
M
1

**E
X
D
UN-
S
ATT'D**

2
1
1
2
E
EXPS

MISSION	EDITION	PUBLICATION DATE
SL-3	REFERENCE	MAY 1, 1972

3-35

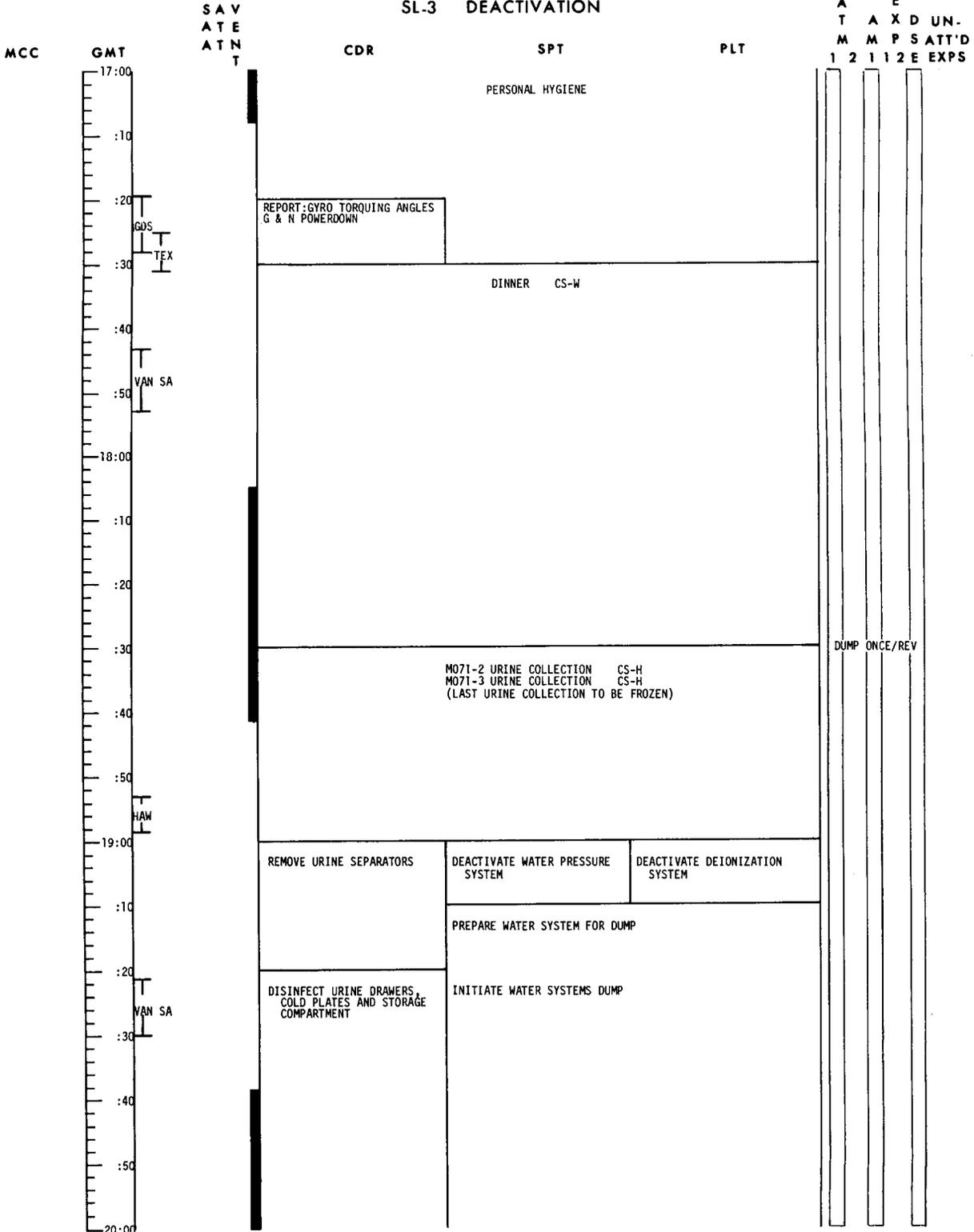
S A L	SOL A-SOL
-------------	-----------

FLIGHT PLANNING BRANCH

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA Δ	MOON PHASE
	17:00 - 20:00	56/266	SEPT. 23, 1973			

SL-3 DEACTIVATION



MISSION	EDITION	PUBLICATION DATE
SL-3	REFERENCE	MAY 1, 1972

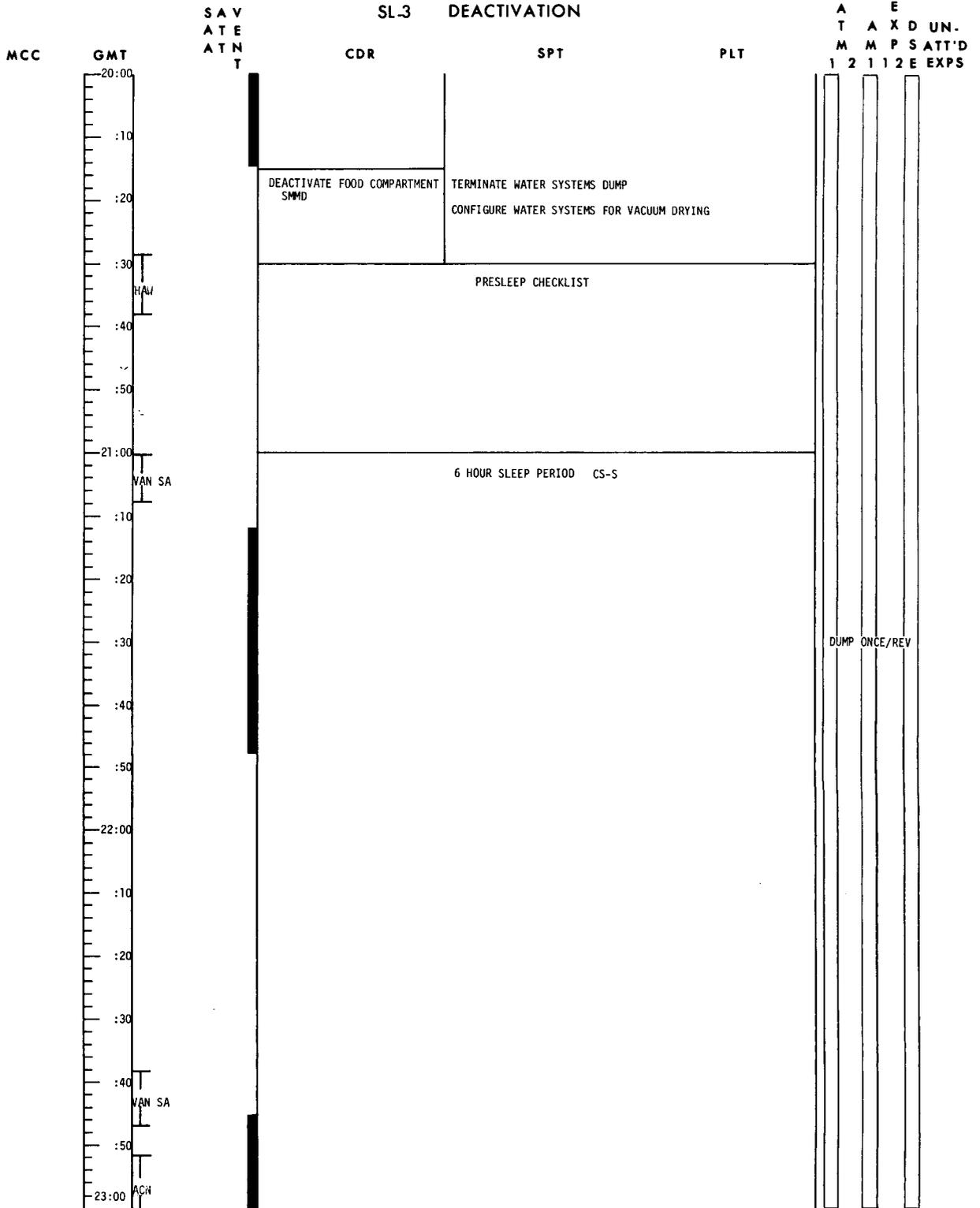
S	SOL	A-SOL
A		
L		

FLIGHT PLANNING BRANCH

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA ↖	MOON PHASE
	20:00 - 23:00	56/266	SEPT. 23, 1973			

SL-3 DEACTIVATION



MISSION	EDITION	PUBLICATION DATE
SL-3	REFERENCE	MAY 1, 1972

FLIGHT PLANNING BRANCH

3-37

S A L	SOL	A-SOL

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA ↕	MOON PHASE
	23:00 - 02:00	56/266	SEPT. 23, 1973			

SL-3 DEACTIVATION

MCC

GMT

S
A
V
A
T
E
A
T
N
T

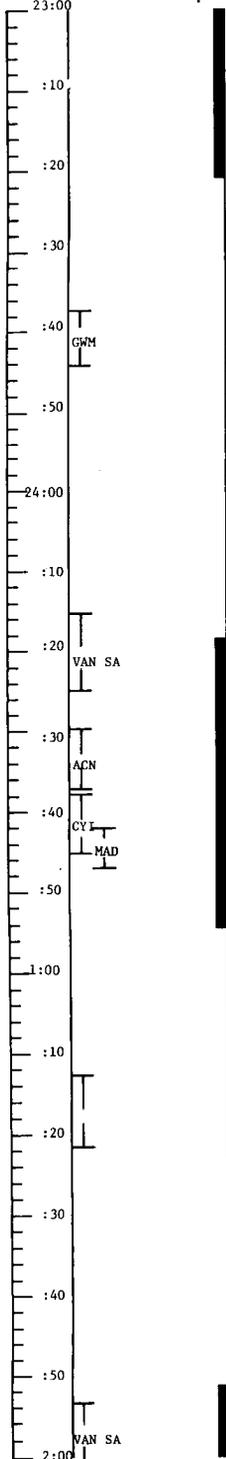
CDR

SPT

PLT

RCDR
A
E
T
A
X
D
U
N
M
P
S
A
T
T
'
D
1
2
1
1
2
E
X
P
S

6 HOUR SLEEP PERIOD CS-S



MISSION	EDITION	PUBLICATION DATE
SL-3	REFERENCE	MAY 1, 1972

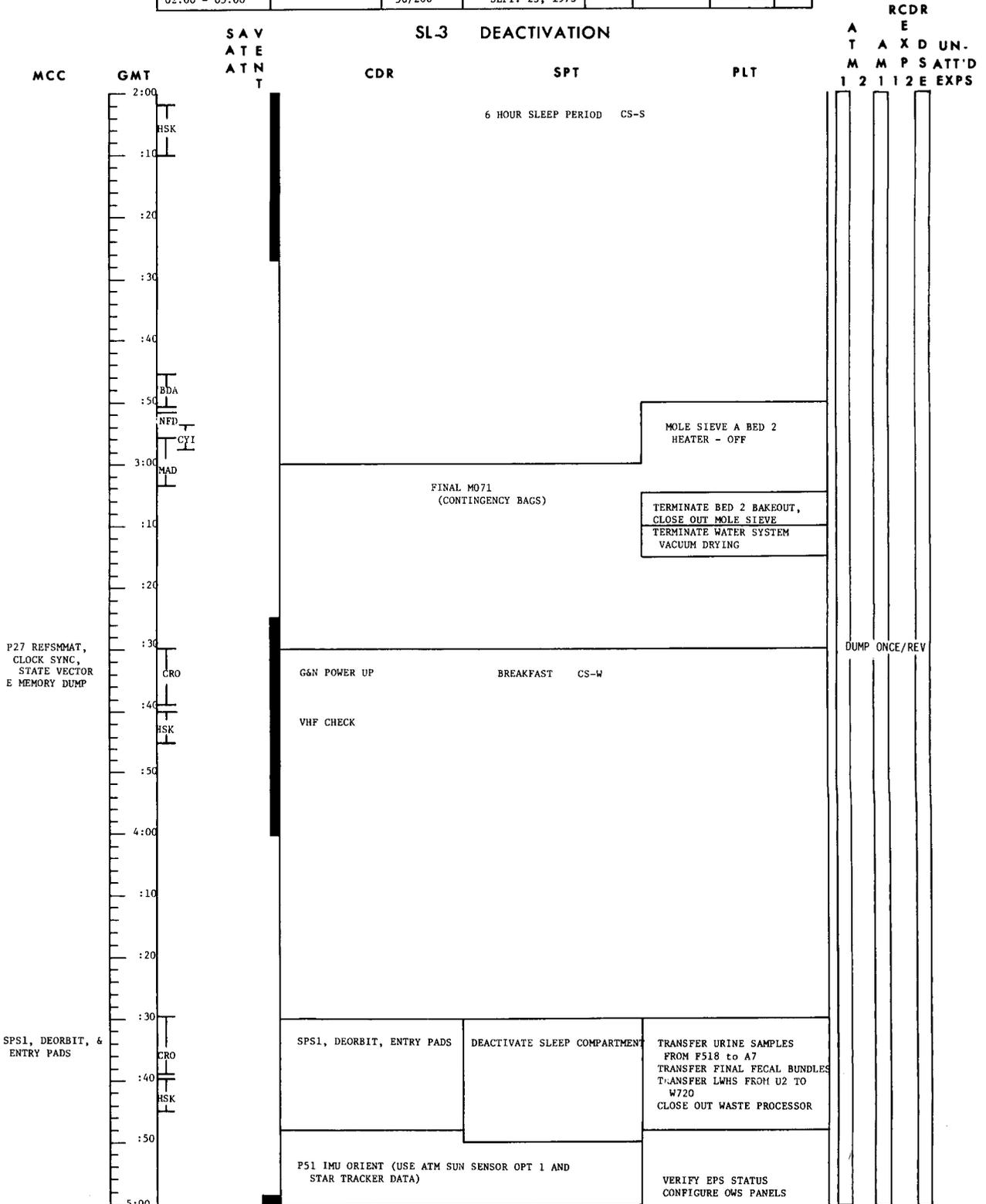
S A L	SOL	A-SOL

FLIGHT PLANNING BRANCH

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA	MOON PHASE
02:00 - 05:00		56/266	SEPT. 23, 1973			

SL-3 DEACTIVATION



MISSION	EDITION	PUBLICATION DATE
SL-3	REFERENCE	MAY 1, 1972

3-39

S	SOL A-SOL
A	
L	

FLIGHT PLANNING BRANCH

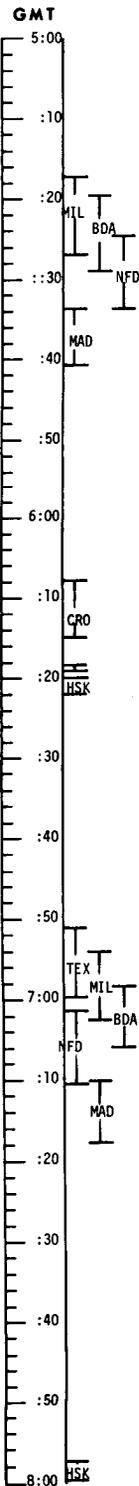
FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA	MOON PHASE
05:00 - 08:00		57/266	SEPT. 24, 1973			

SL-3 DEACTIVATION

RCDR
E
T A X D UN-
M M P S ATT'D
1 2 1 1 2 E EXPS

MCC



S A V
A T E
A T N
T

CDR	SPT	PLT
P52 IMU REALIGN OPT 1 REF (ENTRY ORIENT)	TRANSFER IMSS CONTAINER	VERIFY EPS STATUS CONFIGURE OWS LIGHTING (CONT'D)
	DEACTIVATE SIA's UNPLUG OWS SOLENOID VENT	CONFIGURE OWS LIGHTING
TERMINATE AM/ATM-CSM POWER TRANSFER	CONFIGURE S190A	TERMINATE AM ATM-CSM POWER TRANSFER
CHANGE LIQH CANISTERS		INSTALL SOP DOCKING LOAD STRAPS
PREHEAT CM RCS (IF REQUIRED) EPS, ECS, SPS QUIESCENT MODE TERMINATION	DEACTIVATE CM CIRCULATION FAN REMOVE AIR INTERCHANGE DUCT	VERIFY LOAD SHARING AM/ATM BUSES
	TRANSFER AND STOW THERMAL BLANKET	DEACTIVATE COMM. SYS.
	REMOVE POWER AND COMM UMBILICAL AND STOW	DEACTIVATE CLOCK AND TELEPRINTER
TRANSFER PROBE FROM M107 TO CM		DEACTIVATE O2/N2 CONTROL SYSTEM
DON SUITS		DON SUITS
	DON SUITS	
EMS DEORBIT ENTRY TEST		CONFIGURE STS CONTROL PANELS
TRANSFER DROGUE FROM M105 TO CM CONFIGURE STS/MDA LIGHTING		

1	2	1	1	2	E	EXPS
DUMP ONCE/REV						

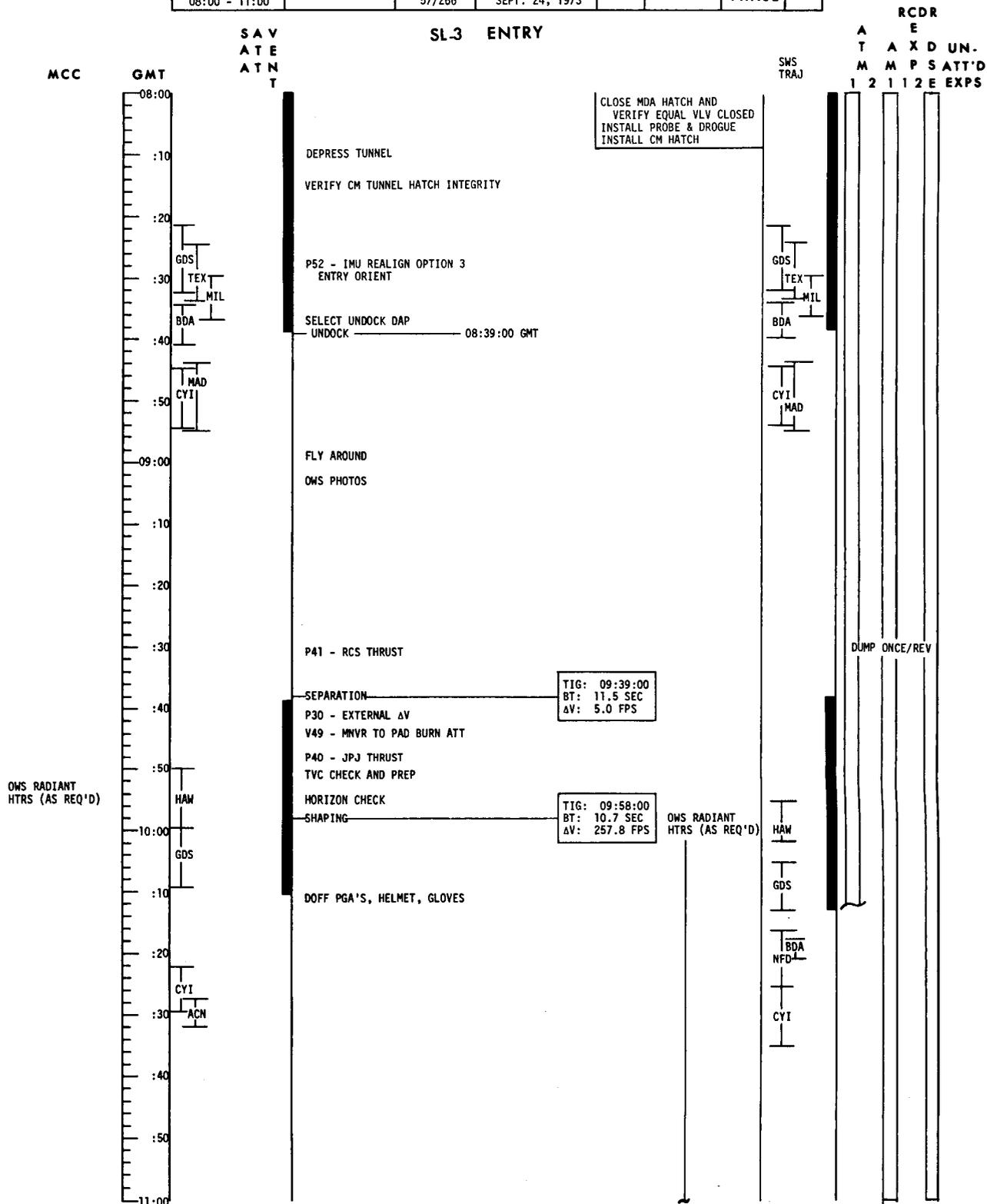
MISSION	EDITION	PUBLICATION DATE
SL-3	REFERENCE	MAY 1, 1972

S	A	L	SOL	A-SOL

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA	MOON PHASE
08:00 - 11:00		57/266	SEPT. 24, 1973			

SL-3 ENTRY



MISSION	EDITION	PUBLICATION DATE
SL3	REFERENCE	MAY 1, 1972

3-41

S	A	L	SOL	A-SOL

FLIGHT PLANNING BRANCH

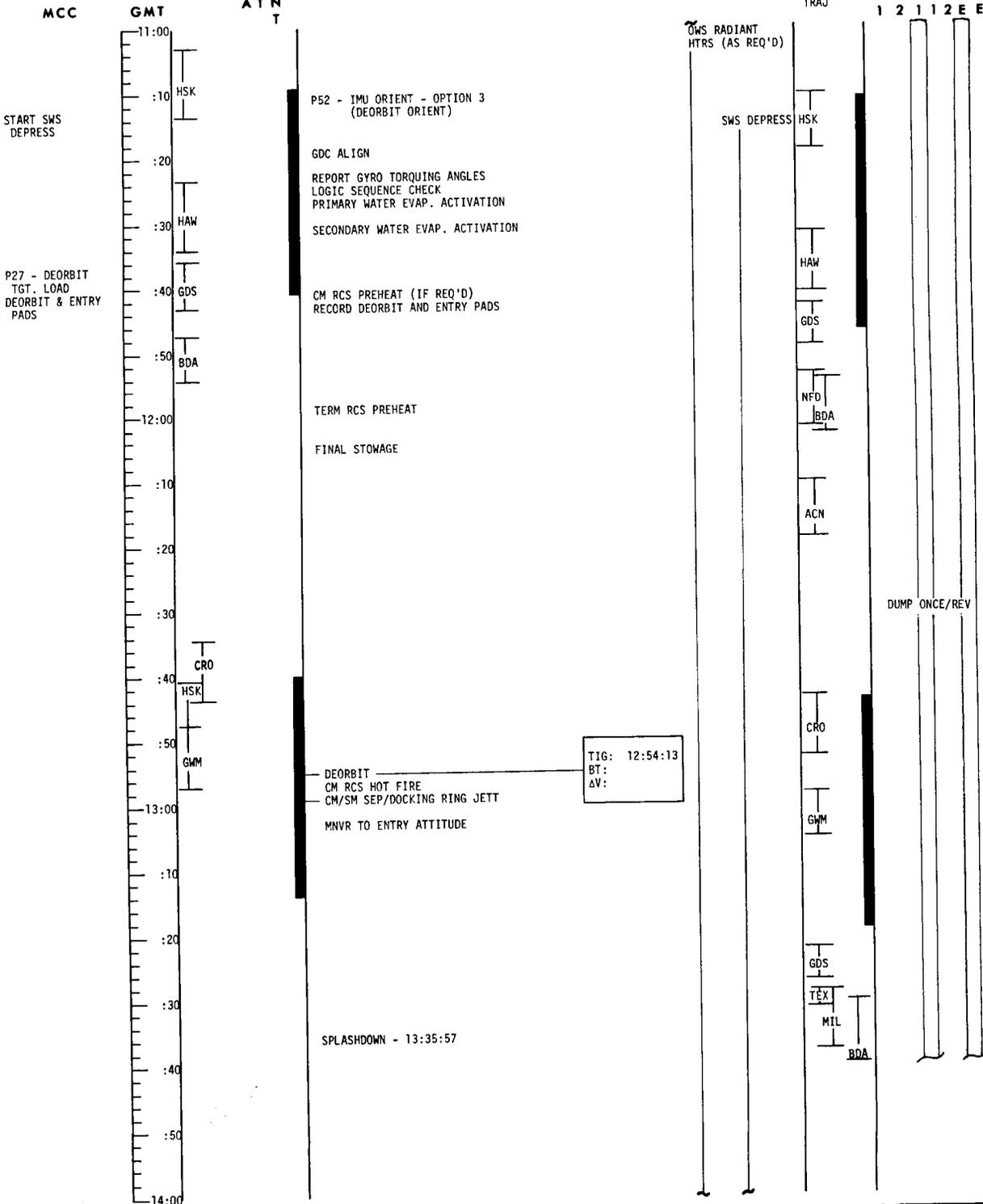
FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA	MOON PHASE
11:00 - 14:00		57/266	SEPT. 24, 1973			

SL-3 ENTRY

S
A
V
A
T
E
A
T
N
T

RCDR
A
E
T
A
X
D
U
N.
M
M
P
S
A
T
T
'
D
1
2
1
1
2
E
X
P
S



MISSION	EDITION	PUBLICATION DATE
SL3	REFERENCE	MAY 1, 1972

3-42

S A L	SOL A-SOL
-------------	-----------

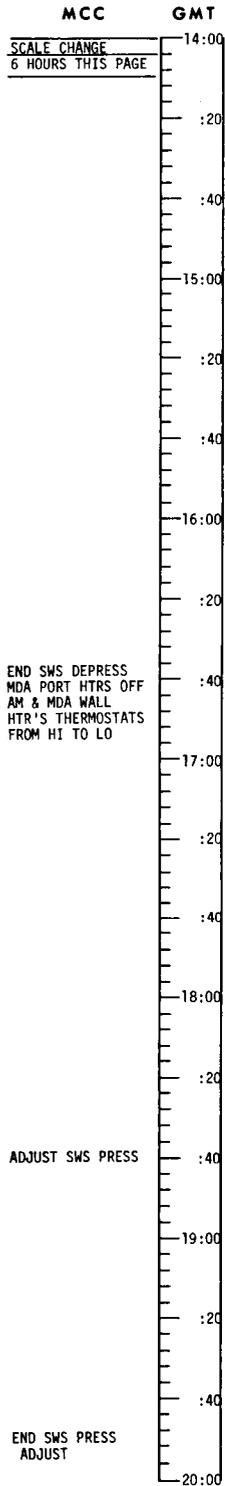
FLIGHT PLANNING BRANCH

FLIGHT PLAN

GET	GMT	MD/DOY	HOUSTON DATE	REV	BETA	MOON PHASE
14:00 - 20:00		57/266	SEPT. 24, 1973			

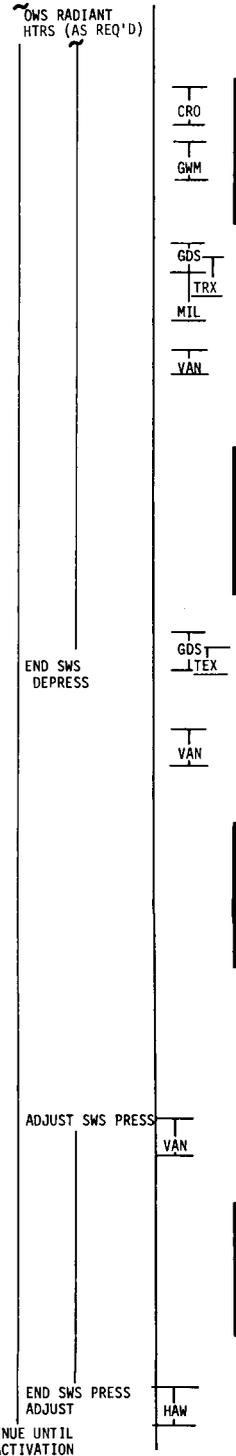
SL-4 UNMANNED

RCDR
A E
T A X D UN.
M M P S ATT'D
1 2 1 1 2 E EXPS



S A V
A T E
A T N
T

SWS TRAJECTORY



MISSION	EDITION	PUBLICATION DATE
SL-3	REFERENCE	MAY 1, 1972

3-43

S	A	L
	SOL A-SOL	

FLIGHT PLANNING BRANCH

SECTION 4
EXPERIMENT AND TEST OBJECTIVES

4 EXPERIMENTS

A. Foreword

The experiment planning sheets shown on the following pages summarize the experiment constraints and scheduling criteria extracted from the Mission Requirements document and information concerning changes to the Mission Requirements document which was available as of April 17, 1972. In addition, highlights of the required crew activity, as will be presented in the detailed timelines, are shown in a timeline fashion for each planned experiment.

The purpose of these sheets is to provide premission information for flight plan timelines and for real-time rescheduling of activities during the mission.

In general, the columns on the left present the timeline of crew activities and the notes on the right present information and constraints on how the experiment is to be scheduled. Experiments which require no crew activity are not listed.

Additional information has also been added to these sheets for the Skylab Flight Planning Data Base as shown in the sample below

ET	AM				*δ	CREWMAN
	1	2	3	v		
0:00					05	[//S073 RETRAC] [11.08.05-00 //] S073-X OPERATE S-SAL RETRACT PHOTOMETER DON GLOVES, RETRACT PHOTOMETER INTO SAL CLOSE OUTER SAL DOOR
0:10						

— Time where readout will start
 - - - Alias
 ← Activity Element Designator

B. Index

Page

MEDICAL EXPERIMENTS

1. M071 - Mineral Balance 4-4
2. M073 - Bio-Assay of Body Fluids 4-5
3. M074 - Specimen Mass Measurement (Calibration) 4-6
4. M092/M093 - In-Flight Lower Body Negative Pressure/
Vectorcardiogram 4-7
5. M092/M171 - In-Flight Lower Body Negative Pressure/
Metabolic Activity 4-8
6. M110 - Hematology and Immunology 4-9
7. M131 - Human Vestibular Function 4-10

	<u>Page</u>
8. M133 - Sleep Monitoring	4-11
9. M151 - Time and Motion Study	4-12
10. M172 - Body Mass Measurement (Calibration)	4-13

APOLLO TELESCOPE MOUNT EXPERIMENTS

1. APOLLO TELESCOPE MOUNT	4-14
-------------------------------------	------

EARTH RESOURCES EXPERIMENTS

1. EREP FLIGHT PLAN PRESENTATION	4-15
--	------

COROLLARY EXPERIMENTS

1. D008 - Radiation in Spacecraft.	4-16
2. D021/D024 - Expandable Airlock Technology/ Thermal Control Coatings	4-17
3. M479 - Zero Gravity Flammability	4-18
4. M487 - Habitability/Crew Quarters.	4-19
5. M509 - Astronaut Maneuvering Equipment	4-20
6. M516 - Crew Activities/Maintenance	4-21
7. M551 - Metal Melting	4-22
8. M552 - Exothermic Heating	4-23
9. M553 - Sphere Forming	4-24
10. M554 - Composite Casting Experiment	4-25
11. M555 - Single Crystal Growth	4-26
12. S009 - Nuclear Emulsion	4-27
13. S015 - Zero-G Single Human Cells	4-28
14. S019 - UV Stellar Astronomy	4-29
15. S020 - UV/X-Ray Solar Photography	4-30
16. S063 - UV Airglow Horizon Photography	4-31

	<u>Page</u>
17. S073/T027 - Gegenschein/Zodiacal Light/Contamination Measurements	4-32
18. S149 - Particle Collection	4-33
19. S183 - UV Pancrama	4-34
20. S190B - Earth Terrain Camera	4-35
21. T002 - Manual Navigation Sightings	4-36
22. T003 - In-Flight Aerosol Analysis	4-37
23. T013 - Crew Vehicle Disturbances	4-38
24. T020 - Foot Controlled Maneuvering Unit	4-39
25. T025 - Coronagraph Contamination Measurements	4-40
26. T027 - Contamination Measurement	4-41

PHOTOGRAPHY

1. M151, M487, M516, OPS Photography	4-42
--	------

OPERATIONAL

1. Operational Radiation Measurements	4-43
2. 20.10 Environmental Microbiology	4-44

STUDENT ACTIVITIES

1.0 TBS	4-45
-------------------	------

M071 MINERAL BALANCE
FO-1 MEASURE AND RECORD THE TOTAL DIET RESIDUE, MENU DEVIATIONS,
AND FLUID INTAKE FOR EACH CREWMAN THROUGHOUT THE MISSION

MRD 2/1/72
EOH VOL I 4/9/71
EOH VOL II 2/1/71
ODB 10/2/70
AE 7/3/71

ET	AM		CREWMAN	*6	CREWMAN	*6	CREWMAN	SCHEDULE CRITERIA
	1	2						
0:00			// M071 CALIB	5	//			<ul style="list-style-type: none"> No SMMD operations will take place during momentum dumps. The experiment will be performed by each crewman throughout the mission. Logs of solid/liquid intake will include the ascent and recovery phase of the mission. The logging of each crewman's water dispenser counter will be accomplished once each day. The measurement of any food residue and logging of menu deviations will be accomplished where applicable. <p><u>CONSTRAINTS</u></p> <ul style="list-style-type: none"> The SMMD will not be operated during spacecraft linear or angular maneuvers. The SMMD must be calibrated prior to use and calibration data recorded. <p><u>RETURN WEIGHT</u></p> <ul style="list-style-type: none"> Astronauts log book <u>TBD</u> lbs. <p><u>NOTES</u></p> <ul style="list-style-type: none"> For all M071 FO's a record of general crew activity, temp. relative humidity profile throughout the OA, crew radiation exposure data, and OA partial pressure will be logged.
0:10			M071-1 CALIBRATION RECORD DATA		CS-W			
0:00			// M071-1 OPER	4	//			
0:08			M071-1 LOG FOOD RESIDUE AND WATER DISP READING		CS-W			

M071 MINERAL BALANCE		MRD		2/1/72					
FO-2 COLLECT, IDENTIFY, MEASURE, AND PROCESS ALL URINE ELIMINATIONS OF EACH CREWMAN		EOH VOL I		4/9/71					
		EOH VOL II		1/71					
		ODB		10/2/70					
		AE		7/31/71					
ET	AM	1	2	3	V	*6	CREWMAN	*6	CREWMAN
0:00						2	//M071-2 OP	//	
0:04							M071-2 VOL MEAS URINE SAMPLE		
<p><u>SCHEDULE CRITERIA</u></p> <ul style="list-style-type: none"> The experiment will be performed by each crewman throughout the mission. Logs of solid/liquid intake will include the ascent and recovery phase of the mission. Urine voids for each crewman are to be accumulated in 24 hour cycles with the closeout time period to be within +30 minutes of the same hour each day. <p><u>CONSTRAINTS</u></p> <ul style="list-style-type: none"> The cross-mixing of crewman voids shall not exceed 0.1 percent maximum. Collected urine must be maintained 10° +5°C. 120 + 10-0 ml homogeneous urine sample must be withdrawn and preserved. Samples must be properly identified. <p><u>RETURN WEIGHT</u></p> <ul style="list-style-type: none"> See FO-3. <p><u>NOTES</u></p> <ul style="list-style-type: none"> See FO-1. All urine passed in flight must be collected and sampled. 									

M071 MINERAL BALANCE
 FO-3 COLLECT, IDENTIFY, MEASURE, PROCESS, STORE FOR RETURN A HOMOGENEOUS URINE SAMPLE OF AT LEAST 45 ML EVERY 24 HOURS FOR EACH CREWMAN, THROUGHOUT THE MISSION.

ET	A.M.		CREWMAN	*6	CREWMAN	*6	CREWMAN	SCHEDULE CRITERIA
	1	2						
0:00								<p><u>SCHEDULE CRITERIA</u></p> <ul style="list-style-type: none"> The experiment will be performed by each crewman throughout the mission. Logs of solid/liquid intake will include the ascent and recovery phase of the mission. The 120 ml. urine samples taken per crewman must be frozen within 3 hours of the last void collection in the 24 hour cycle. FO-2 must be completed prior to FO-3 performance. <p><u>CONSTRAINTS</u></p> <ul style="list-style-type: none"> Volume measurements of the 24-hour urine pool shall be within 2% accuracy. <p><u>RETURN WEIGHT</u></p> <ul style="list-style-type: none"> TBD lbs. <p><u>NOTES</u></p> <ul style="list-style-type: none"> The 120 ml. homogeneous urine sample taken daily (per crewman) satisfies both the M071-3 requirement (for a 45 ml. sample) and the M073-1 requirement for a 75 ml. sample. See FO-1.
0:04								

MRD 2/1/72
 EOH VOL I 4/9/71
 EOH VOL II 1/71
 ODB 10/2/70
 AE 7/31/71

M071 MINERAL BALANCE		MRD		2/1/72						
FO-4 COLLECT, IDENTIFY, MEASURE, PROCESS, STORE AND RETURN ALL FECAL AND VOMITUS ELIMINATIONS OF EACH CREWMAN THROUGHOUT THE MISSION		EOH VOL I		4/9/71						
		EOH VOL II		1/71						
		ODB		10/2/71						
		AE		7/31/71						
ET	AM	1	2	3	V	*6	*6	CREWMAN	SCHEDULE CRITERIA	
0:00						2	//	TBS	//	<ul style="list-style-type: none"> No SMMD operations will take place during momentum dumps. Collection and identification of samples will be performed by each crewman throughout the mission including the ascent and recovery phase of the mission. Measuring, processing, and stowage will be accomplished in the OWS or on the recovery ship as required. Each crewman will be scheduled to take a dye marker pill every six days. Schedule all feces/vomit to be weighed (wet) and processed no later than 24 hours after collection. The removal and stowage of the processed feces can only be accomplished after the automatic drying cycle has ended. <p><u>CONSTRAINTS</u></p> <ul style="list-style-type: none"> The SMMD will not be operated during spacecraft linear or angular maneuvers. <p><u>RETURN WEIGHT</u></p> <ul style="list-style-type: none"> Specimen bag and contents <u>TBD</u> lbs. <p><u>NOTES</u></p> <ul style="list-style-type: none"> 2 types of dye marker pills will be taken at 6 day intervals alternately. It appears that a convenient time to re-move and stow the processed sample would be for the next crewman who uses the waste processor to stow the processed specimens. See FO-1.
0:04							M071-4 INGEST DYE MARKER PILL		CS-H	
0:00						3	//	TBS	//	
0:06							M071-4 TAG AND WEIGH SAMPLE BAG PLACE IN DRYER, SET TIMER		CS-H	
0:00						2	//	TBS	//	
0:04							M071-4 REMOVE PROCESSED FECES		CS-H	

MRD 4/30/71
 EOH VOL I 4/9/71
 EOH VOL II 11/1/71
 ODB 9/70
 AE 7/31/71

M073 BIOASSAY OF BODY FLUIDS
 FO-1 COLLECT, IDENTIFY, MEASURE AND PROCESS, STORE FOR RETURN A HOMOGENEOUS URINE
 SAMPLE OF AT LEAST 75 ML EVERY 24 HOURS FOR EACH CREWMAN THROUGHOUT THE
 MISSION

ET	1	2	3	V	*6	*6	SCHEDULE CRITERIA
							<p> o M073 requirements are satisfied by the same experiment performance that ful- fills M071-3 requirements (see page 4-4B). </p>

M074 SPECIMEN MASS MEASUREMENT (CALIBRATION)
 FO-1 CALIBRATION OF THE SPECIMEN MASS MEASUREMENT DEVICES (SMMD)

ET	1	2	3	V	*6	CREWMAN	*6
0:00					8	/// SMMD CALIB M074-1 PERF # OF 3 CS-W LOG AND RECORD TEMP/GMT RECORD TIME DISPLAY FOR 0, 50, 150, 400, 500, 750 850, 900, 0 GM (5 TIMES FOR EACH WEIGHT) LOG AND RECORD TEMP/GMT	*6 SCHEDULE CRITERIA ● Calibrate each of the two SMMD 3 times (early, middle, late) in the mission. ● Do not schedule during momentum dumps. ● Do not schedule concurrently with M071-1, -4, -5. CONSTRAINTS ● Calibration will only be performed when the ambient temperature is between 65°F and 80°F. ● The SMMD will not be operated during spacecraft maneuvers. RETURN WEIGHT ● Astronauts log book TBD lbs.
0:15					8	/// SMMD CALIB M074-1 PERF # OF 3 CS-H LOG AND RECORD TEMP/GMT RECORD TIME DISPLAY FOR 0, 50, 150, 400, 500, 750 850, 900, 0 GM(5 TIMES FOR EACH WEIGHT) LOG AND RECORD TEMP/GMT	
0:00					8	/// SMMD CALIB M074-1 PERF # OF 3 CS-W LOG AND RECORD TEMP/GMT RECORD TIME DISPLAY FOR 0, 50, 150, 400, 500, 750 850, 900, 0 GM (5 TIMES FOR EACH WEIGHT) LOG AND RECORD TEMP/GMT	
0:15					8	/// SMMD CALIB M074-1 PERF # OF 3 CS-H LOG AND RECORD TEMP/GMT RECORD TIME DISPLAY FOR 0, 50, 150, 400, 500, 750 850, 900, 0 GM(5 TIMES FOR EACH WEIGHT) LOG AND RECORD TEMP/GMT	

MRD 2/1/72
 EOH VOL I 4/9/71
 EOH VOL II 6/7/71
 ODB 10/2/70
 AE 7/31/71

M092 IN-FLIGHT LOWER BODY NEGATIVE PRESSURE (LBNP)
 FO-1 PERFORM LBNP EXPERIMENT USING A LBNP DEVICE
 M093 VECTOCARDIOGRAM
 FO-1 OBTAIN VCG'S ON EACH ASTRONAUT EVERY THIRD DAY DURING THE MISSION

MRD 11/1/71
 EOH VOL I 4/9/71
 EOH VOL II 6/7/71
 ODB 12/2/71

ET	AM	12	3	V	*6	CREWMAN	*6	CREWMAN
0:00					13	// MED.EXP.ACT. 04.05.01-00//		
						M092/M093-1 SUBJECT CS-E		PHOTO PAD
						MED. EXP. ACTIVATION		
						VERIFY EQUIPMENT CONTROLS		
0:25					53	//M092/M093 OP 10.03.02-00//	53	//M093/M093 OP 10.03.02-00//
0:00						M092/M093-1 SUBJECT CS-E PERF # OF 3		M092/M093-1 OBSERVER CS-E
						M092 OPERATION		AID SUBJECT IN EXPERIMENT PREPARATION
						DON VCG HARNESS AND BTMS EAR PROBE		INSTALL BPMS CUFF AND LVMS BANDS SECURE WAIST SEAL
						INSERT LEG AND HIPS INTO LBNPD		OPEN MAIN VACUUM VALVE AND SET EVENT TIMER
						PERFORM LBNP OPERATING SEQUENCE	$\left[\begin{array}{l} \Delta P = 0, 5 \text{ MIN} \\ \Delta P = \underline{TBS}, 5 \text{ MIN} \\ \Delta P = \underline{TBS}, 5 \text{ MIN} \\ \Delta P = \underline{TBS}, 5 \text{ MIN} \\ \Delta P = 0, 5 \text{ MIN} \end{array} \right]$	CALIBRATE ESS DISPLAYS AND TEST THE VCG ELECTRODES MONITOR HEART RATE, SYSTOLIC PRESSURE, AND VISUAL SYMP- TOMS FOR POSSIBLE TERMINA- TION OF EXPERIMENT CALIBRATE ESS DISPLAYS AND TEST THE VCG ELECTRODES
								SCHEDULE CRITERIA (M092) <ul style="list-style-type: none"> Schedule first performance as soon as possible after OWS activation. Perform every third day at approximately the same time (each crewman 8 times min.) Wait at least 1 hour after eating (2 hours is desirable) before exp. performance. Experiment not to be preceded by activities requiring vigorous exercise. Schedule M092 to be followed immediately by M093 or M171 in order to reduce requirements for subject instrumentation. Initial activation will be performed on exp. M092/M171 which will always preclude this experiment. CONSTRAINTS (M092) <ul style="list-style-type: none"> Any food eaten between 1 and 2 hours prior to the experiment should be low in carbohydrates. Air velocity in vicinity of the experiment should be between 15 to 100 ft/min (use M487 velometer). Ambient temperature to be kept within 67° to 78° at 5 PSIA. No spacecraft maneuvering to be performed which would cause accelerations. RETURN WEIGHT (M092) <ul style="list-style-type: none"> None SCHEDULE CRITERIA (M093) <ul style="list-style-type: none"> Schedule first performance as soon as possible after OWS activation. Obtain VCG's from each crewman every third day. Five performances of M171 (if acquired will be substituted for five performances of M093).

M092 IN-FLIGHT LOWER BODY NEGATIVE PRESSURE (LBNP)
 FO-1 PERFORM LBNP EXPERIMENT USING A LBNP DEVICE
 M093 VECTOCARDIOGRAM
 FO-1 OBTAIN VCG'S ON EACH ASTRONAUT EVERY THIRD DAY DURING THE MISSION

ET	12 AM	1	3	V	*5	CREWMAN	*6	CREWMAN	SCHEDULE CRITERIA CONT. (M093)
						EGRESS LBNP		REMOVE AND STOW BPMS CUFF AND LVMS BANDS	<ul style="list-style-type: none"> The experiment must not be performed less than one-half hour after vigorous exercise of M131. Schedule M093 to follow immediately after M092 in order to reduce requirements for subject instrumentation. Allow at least 2 hours between eating and performance of M093.
						M093 OPERATION		SECURE LBNP	
						MON ERGOMETER RESTRAINTS		MOUNT SIB ON ERGOMETER PNL SET ESS & ERGOMETER CONTROLS FOR M093	<p>CONSTRAINTS</p> <ul style="list-style-type: none"> Ambient temperature to be kept within 67° to 78° at 5 PSIA. <p>RETURN WEIGHT (M093)</p> <ul style="list-style-type: none"> None
						PERFORM VCG EXP.		CALIBRATE ESS DISPLAYS AND TEST THE VCG ELECTRODES	
						DISMOUNT ERGOMETER REMOVE AND CLEAN VCG ELECTRODES AND TEMPERATURE PROBE		MONITOR HEART RATE	
						STOW EQUIPMENT PERSONAL HYGIENE		CALIBRATE ESS DISPLAYS AND TEST THE VCG ELECTRODES	
								DEACTIVATE ESS & ERGOMETER AND SECURE EQUIPMENT	
								ASSIST SUBJECT IF REQUIRED	

1:46

M092 IN-FLIGHT LOWER BODY NEGATIVE PRESSURE (LBNP) FO-1 PERFORM LBNP EXPERIMENT USING A LBNP DEVICE M171 METABOLIC ACTIVITY FO-1 PERFORM CALIBRATED EXERCISE (BY ALL THREE CREWMEN) ON A BICYCLE ERGOMETER		MRD 11/1/71 EOH VOL I 11/19/71 EOH VOL II 6/7/71 ODD 12/2/71		
ET	1 2 3 V *5	CREWMAN	*5	CREWMAN
00:00	23	//MED.EXP.ACT. 04.05.03-00//	23	//MED.EXP.ACT. 04.05.03-00//
		M092/M093/M171 ACTIVATION INSTALL ERGOMETER PANEL ESS POWER UP AND NUMERICS CHECKS VCG/BTMS DOWNING AND CHECKOUT MA OFF TO STANDBY VERIFY EQUIPMENT CONTROLS		AID IN MEDICAL EXPERIMENT ACTIVATION SET UP CAMERAS FOR M151 PHOTOGRAPHY REQUIREMENT (07 MIN.) IF REQUIRED
0:45	73	//M092/M171 OP 10.03.03-00//	73	//M092/M171 OP 10.03.03-00//
0:00		M092/M171-1 SUBJECT CS-E PERF # ___ OF 5 M092 OPERATION DON VCG HARNESS AND BRMS EAR PROBE INSERT LEGS AND HIPS INTO LBNPD		M092/M171-1 OBSERVER CS-E AID SUBJECT IN EXPERIMENT PREPARATION INSTALL BPMS CUFF AND LVMS BANDS SECURE WAIST SEAL OPEN MAIN VACUUM VALVE AND SET EVENT TIMER
<p>SCHEDULE CRITERIA (M092)</p> <ul style="list-style-type: none"> Experiments M092/M171 will be performed before any other medical experiment. Schedule first performance as soon as possible after OWS activation. Perform every third day at approximately the same time (each crewman 8 times min.). Wait at least 1 hour after eating (2 hours is desirable) before exp. performance. Experiment not to be preceded by activities requiring vigorous exercise. Schedule M092 to be followed immediately by M093 or M171 in order to reduce requirements for subject instrumentation. Camera set up for M151 photography required for 8 performances (2 crewmen, 4 times each). Activities to be photographed are VCG harness donning, LBNPD entering. <p>CONSTRAINTS (M092)</p> <ul style="list-style-type: none"> Any food eaten between 1 and 2 hours prior to the experiment should be low in carbohydrates. Air velocity in vicinity of the experiment should be between 15 to 100 ft/min (use M487 velometer). Ambient temperature to be kept within 67° to 78° at 5 PSIA. No spacecraft maneuvering to be performed which would cause acceleration greater than 10-4 g's. <p>RETURN WEIGHT (M092)</p> <ul style="list-style-type: none"> None 				

M092 IN-FLIGHT LOWER BODY NEGATIVE PRESSURE (LBNP)
 FO-1 PERFORM LBNP EXPERIMENT USING A LBNP DEVICE
 M171 METABOLIC ACTIVITY
 FO-1 PERFORM CALIBRATED EXERCISE (BY ALL THREE CREWMEN) ON A BICYCLE ERGOMETER

MRD 11/1/71
 EOH VOL I 11/19/71
 EOH VOL II 6/7/71
 ODB 12/2/71

ET	AM	1	2	3	V	*6	CREWMAN	*6	CREWMAN	SCHEDULE CRITERIA (M171)	CONSTRAINTS (M171)	RETURN WEIGHT (M171)
							<p>PERFORM LBNP OPERATING SEQUENCE</p> <p>$\Delta P = 0, 5 \text{ MIN}$ $\Delta P = \text{TBS}, 5 \text{ MIN}$ $\Delta P = \text{TBS}, 5 \text{ MIN}$ $\Delta P = \text{TBS}, 5 \text{ MIN}$ $\Delta P = 0, 5 \text{ MIN}$</p> <p>EGRESS LBNP</p> <p>M171 OPERATION</p> <p>DON ERGOMETER RESTRAINTS</p> <p>MOUNT ERGOMETER AND ADJUST RESTRAINTS</p> <p>DON NOSE CLIP AND MOUTH PIECE</p> <p>PERFORM CALIBRATED EXERCISE</p> <p>DISMOUNT ERGOMETER</p> <p>REMOVE AND CLEAN BREATHING APPARATUS, VCG ELECTRODES, TEMP PROBE, AND BPMS CUFF</p>	*6	CREWMAN	<p>CALIBRATE ESS DISPLAYS AND TEST THE VCG ELECTRODES OPERATE PRESSURE CONTROL KNOB</p> <p>MONITOR HEART RATE, SYSTOLIC PRESSURE, AND VISUAL SYMPTOMS FOR POSSIBLE TERMINATION OF EXPERIMENT</p> <p>CALIBRATE ESS DISPLAYS AND TEST THE VCG ELECTRODES</p> <p>REMOVE AND STOW LVMS BANDS</p> <p>SECURE LBNP</p> <p>ACTIVATE THE MA & PERFORM CALIBRATION</p> <p>MOUNT SIB ON ERGOMETER PNL</p> <p>SET ESS & ERGOMETER CONTROLS FOR M171</p> <p>SET EVENT TIMER</p> <p>CALIBRATE ESS DISPLAYS AND TEST THE VCG ELECTRODES</p> <p>DEACTIVATE MA, ESS, & ERGOMETER</p> <p>AID IN EQUIPMENT SHUTDOWN</p>	<p>SCHEDULE CRITERIA (M171)</p> <ul style="list-style-type: none"> Experiment will be performed 5 times by each crewman during the mission. ALLOW at least 2 hours between eating and performance of M171. The experiment must not be performed less than 2 hours after vigorous exercise. Experiment should be performed the same time each day for each crewman. Schedule M171 to follow immediately after M092 in order to reduce requirements for subject instrumentation. M151 photography required 8 performances (2 crewmen, 4 times each) activities to be photographed are metabolic apparatus donning, ergometer restraint donning, ergometer mounting, ergometer operation. When M171 has been performed for the last time in the mission, reconfigure stowage as the electrode kit, utility drawer, and ergometer restraint for follow-on mission. <p>CONSTRAINTS (M171)</p> <ul style="list-style-type: none"> Air velocity in vicinity of the experiment should be between 15 to 100 ft/min (use M487 velometer). Ambinet temperature to be kept within 67° to 78° at 5 PSIA. Spacecraft pitch, yaw and roll rates will be limited to 6 degrees per minute during all experiment operations. <p>RETURN WEIGHT (M171)</p> <ul style="list-style-type: none"> None. 	None.

M092 IN-FLIGHT LOWER BODY NEGATIVE PRESSURE (LBNP) F0-1 PERFORM LBNP EXPERIMENT USING A LBNP DEVICE M171 METABOLIC ACTIVITY F0-1 PERFORM CALIBRATED EXERCISE (BY ALL THREE CREWMEN) ON A BICYCLE ERGOMETER		CREWMAN		CREWMAN	
ET	1 2 3 V *6	*6	*6		
2:25		PERSONAL HYGIENE		ASSIST SUBJECT IF REQUIRED	
0:00	15	//RECONFIGURATION STOWAGE //	15	//RECONFIGURATION STOWAGE //	
		M092/M171 STOW EQUIPMENT	M092/M171	AID IN STOWING EQUIPMENT	
0:30					

M110 HEMATOLOGY AND IMMUNOLOGY

ET	AM	12	3	V	*6	CREWMAN	*6	CREWMAN	CREWMAN
0:00					90	// TBS //			
						M110 PREPARE WORK AREA OBTAIN ASP'S FROM STOWAGE	15		
0:15						OBTAIN BLOOD SAMPLES		/// TBS ///	/// TBS ///
						PREPARE FIXED CELL SAMPLES		M110 ASSIST IN OBTAINING BLOOD SAMPLES	M110 ASSIST IN OBTAINING BLOOD SAMPLES
0:45						PERFORM SEPARATION PROCEDURE			

MT10 HEMATOLOGY AND IMMUNOLOGY

ET	AM	123V	*6	CREWMAN	*6		
				STOW BLOOD SAMPLES AND ASP'S			<p>SCHEDULE CRITERIA</p> <ul style="list-style-type: none"> ● Schedule on days 2, 4, 12 +1, 19 +1, 27. ● Schedule over MSFN if possible. ● Schedule prior to breakfast after an all night fast and prior to physical activity.
				CLEAN UP WORK AREA			
1:35			5	/// TBS ///			
0:00				MT10 STOW SAMPLES IN CSM FOR RETURN			
0:10							

M131 HUMAN VESTIBULAR FUNCTION
FO-1 DYNAMIC TESTS (OCULOGYRAL ILLUSION AND MOTION SENSITIVITY)

MRD 2/1/72
EOH VOL I 4/9/71
EOH VOL II 6/7/71
ODD 6/18/71
AE 7/31/71

ET	1 AM	123 V #6	CREWMAN	#6	CREWMAN
0:00		29	M131A NO PH 10.04.01-03 // M131-1 SUBJECT PERF # OF 6	2	// PREPAR 04.05.04-00 // M131-1 OBSERVER CS-E PREPARE CAMERAS
			M131-1 OPERATION ENTER CHAIR & DON OGI GOGGLES ADJUST BITEBOARD PERFORM OCULOGYRAL ILLUSION THRESHOLD (OGI) TEST REMOVE GOGGLES AND BITE- BOARD DON BLINDFOLD PERFORM MOTION SENSITIVITY (MS) TEST	29	M131A NO PH 10.04.01-03 // M131-1 OBSERVER CS-E ADJUST LIGHTING ACTIVATE CONTROL CONSOLE CONDUCT & OBSERVE OGI TEST VOICE RECORD COMMENTS AND ENTER SUBJECT RESPONSES ON CONSOLE PREPARE CONTROL CONSOLE FOR MS TEST CONDUCT & OBSERVE MS TEST VOICE RECORD COMMENTS AND ENTER SUBJECT RESPONSES ON CONSOLE TERMINATE TEST AFTER MAXI- MUM OF 30 SETS OF HEAD MOVEMENTS OR UPON SUBJECT REACHING MALTAISE II LEVEL
0:57					

SCHEDULE CRITERIA

- The dynamic tests will be performed by 2 crewman 6 times each.
- Both crewmen will perform as subjects within a 16-hour period.
- The tests will be performed by each subject no closer than every other day and if possible be equally spaced throughout the mission.
- No tests will be performed within an hour after eating a meal.
- Static tests (FO-2) will not be scheduled on the same day as dynamic tests (FO-1).
- Do not schedule during any OWS maneuver.
- Photograph experiment set up and performance of first OGI test. Photograph each subject accomplishing MS test for first, third, and sixth performances.
- The securing & stowing of equipment need only be performed once if crewmen's performances are scheduled consecutively.
- Initial preparation & final stowage as a one time operation per mission.

CONSTRAINTS

- OWS ambient temperature to be within 67° to 78°F at 5 PSIA.
- Air velocity to be within 15 ft/min to 100 ft/min in vicinity of experiment (use M487 velometer).
- OWS lighting to be adequate enough to recognize color changes in subject's face.
- Experiment shall be interrupted during spacecraft accelerations greater than 0.0003 g's or 0.001 deg/sec² about the X-axis & 0.002 deg/sec² about the Y and Z-axis.

M131 HUMAN VESTIBULAR FUNCTION
FO-2 SPATIAL LOCALIZATION STATIC TESTS

MRD 2/1/72
EOH VOL I 4/9/71
EOH VOL II 6/7/71
ODB 6/18/71
AE 7/31/71

ET	AM	12	3	V	*5	CREWMAN	*6	CREWMAN	SCHEDULE CRITERIA
0:00					23	M131-2 SUBJECT PERF # OF 6 ENTER CHAIR & DON OGI GOGGLES PERFORM VISUAL SPATIAL LOCALIZATION STATIC TESTS	2	// PREPAR 04.004.05.04-00// M131-2 OBSERVER CS-E PREPARE CAMERAS //M131 B NO PH 10.04.02-02 // M131-2 OBSERVER CS-E	<ul style="list-style-type: none"> The static tests will be performed by 3 crewmen 3 times each. All three crewmen will perform as subjects within a 16-hour period. The tests will be performed on all crewmen once early in the mission, once at mid-mission, and once late in the mission. No tests will be performed within an hour after eating a meal. Static tests (FO-2) will not be scheduled on the same day as dynamic tests (FO-1). Do not schedule during any OWS maneuver. Photograph only one subject accomplishing his static tests, preferably his second performance. The securing and stowing of equipment need only be performed once if crewmen's performances are scheduled consecutively.
0:45					5	REMOVE OTG, DON BLINDFOLD PERFORM NON-VISUAL SPATIAL LOCALIZATION STATIC TESTS	5	VOICE RECORD TEST COMMENTS	<p><u>CONSTRAINTS</u></p> <ul style="list-style-type: none"> OWS ambient temperature to be within 67° to 78°F at 5 PSIA. Air velocity to be within 15 ft/min to 100 ft/min in vicinity of experiment (use M487) velometer. <p><u>RETURN WEIGHT</u></p> <ul style="list-style-type: none"> See FO-1. OTG TBS pounds.
0:00						M131 STOWAGE SECURE & STOW EQUIPMENT		// M131 STOWAGE 14.05.03-00//	M131 PHOTO PAD
0:10								SECURE AND STOW EQUIPMENT	

M133 SLEEP MONITORING		MRD		
FO-1 OBTAIN EEG, ECG, AND HEAD MOVEMENT DATA DURING SLEEP		EOH VOL. I	11/1/71	
		EOH VOL. II	11/19/71	
		ODB	6/7/71	
		AE	12/8/71	
			7/31/71	
ET	AM	CREWMAN	*6	CREWMAN
0:00	12	//M133 PREP 04.05.05-00 //		
		M133 EQUIPMENT PREPARATION	CS-E	
		REMOVE PANEL ASSY FROM STORAGE, INSTALL IN SLEEP STATION 1		
		CONNECT CABLES VERIFY CONTROLS EXP PWR-UP		
0:24	6	//SLEEP PREPAR 10.09.01-00//		
0:00		M133 PERF # ___ OF IS	CS-S	
		ENTER SLEEP RESTRAINT, DON & C/O EQUIPMENT		
0:12		8 HR SLEEP PERIOD		
0:00	8	//POST SLEEP ACTIVITIES //		
		M133 LOG COMMENTS	CS-S	
		DOFF CAP DISCONNECT AND STOW EQUIPMENT		
0:15		REMOVE ELECTROLYTE FROM HAIR		

SCHEDULE CRITERIA

- The initial 24 minute preparation need only be scheduled prior to the first sleep period only.
- Schedule during the regular 8 hour sleep period.
- Schedule for same crewman on the following days: 3, 4, 5, 7, 9, 11, 13, 15, 17, 19, 21, 23, 24, 25, 26 (15 total) SL-2 only.
- Schedule the 1 hour equipment stowage after the final equipment performance.
- Schedule for same crewman on the following days 3, 4, 5, 8, 11, 14, 17, 20, 23, 26, 29, 32, 35, 38, 41, 44, 47, 50, 52, 53, 54, (21 total) SL-3 only.

RETURN WEIGHT

- Tape and canister 2.5 lbs.
- Log book TBD lbs.

NOTES

If any of the 6 indicators are not lighted, gently rock corresponding electrode to establish scalp contact.

M133 SLEEP MONITORING
FO-1 OBTAIN EEG, EOG, AND HEAD MOVEMENT DATA DURING SLEEP

ET	AM	123 V #6	CREWMAN	*6	CREWMAN	NOTES
0:00		30 //	M133 POST OP //			<ul style="list-style-type: none"> On SL2 remove both reels on outer recorder only and stow in tape return canister. Reinstall new supply & takeup reels in recorder. On SL3 cut tape at supply reel end place takeup reel in tape return canister for both recorders.
			M133 CS-E, S POST OPER ACTIVITIES AT END OF MISSION EXT PWR-DOWN			
			DISC PWR & TLM CABLES			
			REMOVE PANEL ASSY			
			REPLACE TAPE RECORDER REELS			
			STOW SIA AND HIGH POWER CABLES			
			REPLACE M133 PANEL ASSY IN LAUNCH STORAGE BRACKET			
			STOW TAPE RETURN CANISTER			
1:00						

M151	TIME & MOTION STUDY	FO-1 DURING M092/M171 FO-2 DURING T027/S073 FO-3 DURING T027 FO-4 DURING PGA DOWN/DOFF FO-5 DURING M509 FO-6 DURING MEAL PREP/MEASUREMENT
ET	123V *6	*6
	CREWMAN	
	SEE PHOTOGRAPHY, PAGE 4-42	

ATM DATA TO BE SUPPLIED

EREP:		MRD		1/17/72	
FO-1 AGRICULTURE/FORESTRY		EOH		1/3/72	
FO-2 ENVIRONMENTAL		ERD			
FO-3 GEOGRAPHY		GDB			
AM					
ET	123V*6	CREWMAN	*6	CREWMAN	METEOROLOGY FO-7 OCEANOGRAPHY FO-8 SENSOR TECHNOLOGY FO-9
0:00	07 // EREP ACTIVA 04.02.01-00 // EREP ACTIVATION ST,M TRANSFER MAG. TAPE ACTIVATE STS PANEL CB'S				
0:15	15 // EREP REACTI 04.02.02-00 // EREP ST, M REACTIVATION, SL-3 & SL-4 TRANSFER MAG. TAPE				
0:30	9C // EREP CHECKO 07.01.01-00 // EREP CHECKOUT M OPEN EREP C&D PANEL COVER & VERIFY SWITCH POSITIONS PER C/L				
0:00	90Y // EREP CHECKO 07.01.01-00 // EREP CHECKOUT M VERIFY VTS C&D PANEL SWITCH POSITIONS PER C/L				
<p><u>SCHEDULE CRITERIA</u></p> <ul style="list-style-type: none"> Update pads as follows: EREP MNVR PAD - within 2 hrs. of data take. SETUP PAD - within 2 hrs. of data take. OPERATE PAD - within 2 hrs. of data take. V/TS PAD - night before, 12 hrs. min. PRE-ADVISORY - 1 to 3 hrs. before data take. Aperture Doors and optics covers must be closed except during taka take. Free Liquid Discharges into the OWS waste tank & controllable OA venting must be completed 15 min. prior to exp. operation & avoided during exp. performance of S190, S191 & S192. 70 to 80% of test sites to be in U.S.A. Test sites to be scheduled on basis of priority, weather & proximity of ground tracks of pre-selected EREP passes. For each data pass, voice record site ID, meteorology notations, unusual items, & GMT at start and finish. ATM recorder required. <p><u>CONSTRAINTS</u></p> <ul style="list-style-type: none"> EREP passes are constrained to the following: <ul style="list-style-type: none"> a) 60° max. orbital arc in Z-LV(E) centered about any orbital location. A max. of 4 single orbits in any 24 hr. period. Two consecutive orbits in Z-LV(E) to be followed by 5 orbits in SI. $\beta \pm 50^\circ$. 					

ET		1 AM		12 3 V *6		CREWMAN		CREWMAN		*6		CREWMAN	
EREP : FO-1 AGRICULTURE/FORESTRY FO-2 ENVIRONMENTAL FO-3 GEOGRAPHY		FO-4 GEOLOGY FO-5 HYDROLOGY FO-6 INTERPRETATION TECHNIQUES		FO-7 METEOROLOGY FO-8 OCEANOGRAPHY FO-9 SENSOR TECHNOLOGY		CHECK FOR BROKEN OR DAMAGED CONTROLS, CABLES AND CONNECTORS REMOVE S192 COVER & VERIFY SW POSITIONS CHECK FOR DAMAGED HARDWARE INSPECT PRIMARY & SECONDARY COOLER/DEWAR ASSY CHECK EREP COOLANT VALVE FOR DAMAGE ROTATE S190 TO MID POSITION, REMOVE SHIELDS & INSPECT FOR DAMAGE VERIFY SW POSITIONS		CHECK VTS FOR BROKEN OR DAMAGED CONTROLS, CABLES AND CONNECTORS OPEN TAPE RECORDER COVERS CHECK FOR BROKEN OR DAMAGED HARDWARE LOAD MAGNETIC TAPE IN RECORDERS CHECK OUT RECORDER OPERATION		SCHEDULE CRITERIA(CONTD.-) b) 60° max orbital arc in Z-LV(E) contained within a 120° arc centered about orbital noon. A max. of 4 single Z-LV(E) orbits in any 24 hr. period. Two consecutive orbits in Z-LV(E) to be followed by 4 orbits in SI. $\beta + 50^\circ$. c) 120° max. orbital arc in Z-LV(E) centered about any orbital location, to be followed by 4 orbits in SI $\beta <$ between -50° & $+30^\circ$ • A total of 60 EREP passes in Z-LV(E) & 5 passes in SI are required during Skylab. • OA drift rates not to exceed 0.05°/sec in any axis. • Film is to be stowed in the OWS film vault after each days' photography. when time between data passes exceeds 12 hours or when the next pass requires a new reel. RETURN WEIGHT (SL-2) • S190A film 34.2 pounds. • S191 film 1.0 pounds. • EREP magnetic tape TBD pounds.		NOTES • At the end of the SL-4 mission, the S190A camera filters are to be returned to Earth. • The first of two consecutive Z-LV(E) orbits will require minimum post-pass securing time while the second consecut- ive Z-LV orbit will require less pre- pass preparation. • Missions SL-3 & SL-4 will require 45 to 60 minutes each of additional crew time	

EREP:		AGRICULTURE/FORESTRY		GEOLOGY		METEOROLOGY	
FO-1		FO-4		FO-7		FO-8	
FO-2		FO-5		HYDROLOGY		OCEANOGRAPHY	
FO-3		FO-6		INTERPRETATION		FO-9	
GEOGRAPHY		TECHNIQUES		SENSOR TECHNOLOGY			
AM	ET	1	2	3	V	*6	CREWMAN
						*6	CREWMAN
							POWER UP C&D PANEL
							CHECK S192 DOOR LATCH RELEASE
							ALIGN S192 DETECTOR
							INITIATE S190 CAMERA CHECKOUT
							MONITOR C&D PANEL
							POWER DOWN C&D PANEL
	3:00						
<p>NOTES (CONTD)</p> <ul style="list-style-type: none"> to load the film magazines onboard the OWS with new film cassettes delivered by CM. The Crew will monitor MRD constraints on OA attitude and drift rate limits for TBD number of passes. OWS maneuvers to Z-LV & back to SI can be pre-programmed to occur at the desired time by ground uplink, however, the crew is considered prime in initiating all Z-LV/SI maneuvers. At least 30 minutes cooldown for the spectrometer detector is required prior to S192 data take. At least 20 minutes cooldown/warmup for the scanner is required prior to S192 data take. At least 15 minutes warmup of each S193 sensor is required prior to each S193 data take. At least 30 minutes warmup time is required for the radiometer prior to each S194 data take. Present planning call for 15/25/20 Z-LV(E) passes on SL-2/3/4 respectively. Three data passes are planned on SL-3 & two data passes are planned on SL-4 while in SI attitude. Film 18/30/24 cassettes for SL 2/3/4 respectively. There are 9 EREP disciplines having a total of 291 FO's. 383 test sites are under consideration. Refer to the MRD for FO requirements. 							
<p>VERIFY S191 SHORTING PLUG INSTALLED</p> <p>CHECKOUT S191 PREPARE VTS</p> <p>ALIGN VTS</p> <p>CHECKOUT S191 DOOR OPERATION OPEN (140 SECONDS) CLOSE (108 SECONDS)</p> <p>VTS PANEL POWER OFF</p> <p>ASSIST WITH S190 FILM LOADING & CHANGE-OUT OF FILM</p> <p>STOW S190 FILM</p>							

ET		AM	CREWMAN	*6	CREWMAN	*6	CREWMAN	FO-1. AGRICULTURE/FORESTRY	FO-4. GEOLOGY	FO-7. METEOROLOGY	
0:00		123V	30	//	EREPPREP	30	//	FO-2. ENVIRONMENTAL	FO-5. HYDROLOGY	FO-8. OCEANOGRAPHY	
1:00			15	//	EREPPREP	15	//	FO-3. GEOGRAPHY	FO-6. INTERPRETATION TECHNIQUES	FO-9. SENSOR TECHNOLOGY	
0:00			30	//	EREPPREP	30	//				
					INSTALL M512 FOOT RESTRAINT PLATFORM						
					REVIEW UPDATE PAD POWER UP C&D VERIFY ATM READY FOR EREP EREPPREP COOLANT FLOW (42M) EQUIPMENT WARMUP: <ul style="list-style-type: none"> S191 COOLER ON (30M) S194 POWER ON (30M) S192 POWER STBY (20M) S193 POWER ON (15M) S190A PREPS <ul style="list-style-type: none"> LOAD CAMERA & FILTERS S190 POWER ON CHECK SHUTTER OPERATION POSITION CAMERAS ADVANCE FILM 2 FRAMES VERIFY C&D PANEL SWITCH POSITIONS PERFORM C&D MONITOR Z-LV MNVR AT _____						
					FINAL PANEL SETUP PER UPDATE PAD						
1:00			15	//	EREPPREP	15	//				
0:00					PERFORM Z-LV MNVR IF ATM PANEL IS UNMANNED POSITION VTS FOR FIRST TARGET						
					REMOVE S190 & S191 FILM FROM STORAGE						
					INSTALL S191 DAC FILM COVER WINDOWS & SUBDUCE LIGHTING PREPARE VTS <ul style="list-style-type: none"> S191 POWER ON ADVANCE FILM LAMP TEST ALIGN COLLIMATOR UNSTOW CCU'S & SETUP REVIEW SITE MAPS & PAD						
					PERFORM Z-LV MNVR IF ATM PANEL IS UNMANNED POSITION VTS FOR FIRST TARGET						
					EREPPREP						
					PERF # _____ OF _____						

EREP: FO-1 AGRICULTURE/FORESTRY FO-2 ENVIRONMENTAL FO-3 GEOGRAPHY	FO-4 GEOLOGY FO-5 HYDROLOGY FO-6 INTERPRETATION TECHNIQUES	FO-7 METEOROLOGY FO-8 OCEANOGRAPHY FO-9 SENSOR TECHNOLOGY	
AM ET 1123 V #6	CREWMAN EREP TRANSFER CM, M TRANSFER FILM, TAPE & FILTERS TO CSM		
0:15			

D008 RADIATION IN SPACECRAFT

FO-1 PERFORM TWO ACTIVE DOSIMETER SURVEYS IN THE CM WHILE IN THE SAA

MRD 4/30/71
 EOH VOL. I 11/19/71
 EOH VOL. II 6/7/71
 ODB 1/25/72
 AE 1/3/72

ET	AM	1	2	3	V	*6	*6
0:00						8	//D008 ONE MIN 11.01.01-00// D008-1 CS-CM UNSTOW DOSIMETER SENSOR AT T1 PERFORM 9 BODY & 5 CM. POSITION RADIATION SURVEY (14 PLACES, 1 MIN EACH) STOW DOSIMETER SENSOR
0:15							
							//D008 PAD UP 11.01.03-00 //
							PAD UPDATE
							<p><u>SCHEDULE CRITERIA FO-1</u></p> <ul style="list-style-type: none"> Center test around SAA. Schedule crew performances during two orbits with max. penetration thru the SAA. Perform twice during the mission. Log GET time for each position change of sensor within 5 sec. Update times to crew one rev prior to scheduled experiment. Ground control will manage DSE during dosimeter calibration. The SAA is defined as the area between 60°W 10n/15°E lon and 18°S lat/33°S lat between 200-500 KM alt. Ground to control stowed active dosimeter sensor output on DSE for 5 consecutive SAA passes/day for 14 days. Data can be taken over entire mission span or during 14 consecutive days. <p><u>CONSTRAINTS FO-1</u></p> <ul style="list-style-type: none"> Do not unstow active dosimeter sensor until 15 min. TLM have been recorded for calibration. The 15 minute active survey to be time centered on the SAA zone. <p><u>RETURN WEIGHT</u></p> <ul style="list-style-type: none"> Dosimeters 7 pounds (part of CM). <p><u>NOTES FO-1 & FO-2</u></p> <ul style="list-style-type: none"> Experiment includes one active dosimeter & five passive dosimeters. The active dosimeter has a power/TLM connector. The passive dosimeters have no CM connections and require no crew activity. If S015 radiation level exceeds 0.01 milli-rad/hr., a radiation survey will be required. (Premission requirement)

D008 RADIATION IN SPACECRAFT

FO-2 PERFORM TWO ACTIVE DOSIMETER SURVEYS IN THE CM IN THE MOST NORTHERN LATITUDES OF THE ORBIT

ET	AM	1	2	3	V	*6	*8
0:00						15	//D008 TWO MI 11.01.02-00 //
							D008-2
							PERF # OF 2 UNSTOW DOSIMETER SENSOR AT T1
							PERFORM 9 BODY AND 4 CM POSITION RADIATION SURVEY, (13 PLACES, 2 MIN EACH)
0:30							STOW DOSIMETER SENSOR
							// D008 PAD UP 11.01.03-00//
							PAD UPDATE
							<p>NOTES CONT. FO-1 & FO-2</p> <ul style="list-style-type: none"> The ambient radiation within the CM from all sources before, during, & after the mission must not exceed 0.1 millirad/hr. for the active & .01 millirad/hr for the passive. <p>SCHEDULE CRITERIA FO-2</p> <ul style="list-style-type: none"> Center test in the northernmost latitude at the closest approach to the geomagnetic N. Pole. Perform twice during the mission. Log GET times for each position change of sensor within ±5 sec. Update times to crew one rev prior to scheduled experiment. Repeat for confirmed solar flare at northernmost lat. of orbit Ground control will manage DSE during dosimeter calibration. Ground to control stowed active dosimeter sensor output on DSE one pass each day for 14 days while in the northernmost latitude. Data can be taken over entire mission or during 14 consecutive days. <p>CONSTRAINTS FO-2</p> <ul style="list-style-type: none"> Do not unstow active dosimeter sensor until 7.5 min TLM have been recorded for calibration. The 30 minute active survey is to be time centered on the northernmost lat.

M479 ZERO GRAVITY FLAMMABILITY
 FO-1 UNDISTURBED BURNING THROUGH BURNOUT OF SPECIMENS
 1 THRU 12 (SIX MATERIAL TYPES)

MRD 7/1/71
 EOH VOL I 4/9/71
 EOH VOL II 6/7/71
 ODB 12/3/71
 AE 1/3/72

ET	AM	1	2	3	V	*6	CREWMAN	*6
0:00						10	// M479 EXP PR 11.03.01-00 // M479-1 PREP CS-M PERFORM EXPERIMENT FACILITY INTEGRITY CHECK	
0:19						4	//M479 OP 1 11.03.02-00 // M479-1 CS-M PERF # OF 12 FILM AND COMMENT ON BURNING OF SPECIMEN	
0:00						2	// M479 FILM S 11.03.09-00 // M479-1 FILM STORAGE CS-M	
0:08								
0:00								
0:03								
PHOTO PAD								
<p>SCHEDULE CRITERIA</p> <ul style="list-style-type: none"> • All phases of M512 to be completed prior to M479 performance. • If M479 is scheduled on a different mission to M512 then the Experiment Integrity check must be accomplished once prior to the first performance of M479. • If the experiment is terminated before the completion of the tests, then the film must be removed and stored. Upon resumption of the tests the exp. facility must first be verified. If FO-2 directly follows FO-1 then the stowage of the film need not be performed. • All performances of M479 to be completed prior to the end of SL-3 because of film degradation. <p>CONSTRAINTS</p> <ul style="list-style-type: none"> • S/C maneuvers to be restricted to attitude hold during experiment performance. • Each test will be performed using nominal workshop atmosphere. <p>RETURN WEIGHT</p> <ul style="list-style-type: none"> • Film <u>10.5</u> pounds for all FO's 								

M479 ZERO GRAVITY FLAMMABILITY
 FO-2 UNDISTURBED BURNING WITH A VACUUM QUENCH OF SPECIMENS
 13 THRU 18 (SIX MATERIAL TYPES)

MRD 7/1/71
 EOH VOL I 4/9/71
 EOH VOL II 6/7/71
 ODB 12/3/71
 AE 1/3/72

ET	A.M.	1	2	3	V	*6	CREWMAN	*6
0:00						4	// M479 OP 13 11.03.03-00 // M479-2 PERF # OF 6 FILM AND COMMENT ON BURNING AND QUENCHING	
0:08						2	// M479 FILM S 11.03.09-00// M479-2 FILM STOWAGE CS-M	
0:00								
0:03								
PHOTO PAD								
<p>SCHEDULE CRITERIA</p> <ul style="list-style-type: none"> All phases of M512 to be completed prior to M479 performance. If M479 is scheduled on a different mission to M512 then the Experiment Integrity check must be accomplished once prior to the first performance of M479. If the experiment is terminated before the completion of the tests, then the film must be removed and stored. Upon resumption of the tests the exp. facility must first be verified. If FO-2 directly follows FO-1 then the stowage of the film need not be performed. All performances of M479 to be completed prior to the end of SL-3 because of film degradation. <p>CONSTRAINTS</p> <ul style="list-style-type: none"> S/C maneuvers to be restricted to attitude hold during experiment performance. Each test will be performed using nominal workshop atmosphere. <p>RETURN WEIGHT</p> <ul style="list-style-type: none"> Film <u>10.5</u> pounds for all FO's. 								

M479 ZERO GRAVITY FLAMMABILITY
 FO-3 UNDISTURBED BURNING WITH A WATER SPRAY QUENCH OF SPECIMENS
 19 THRU 24 (SIX MATERIAL TYPES)

MRD 7/1/71
 EOH VOL I 4/9/71
 EOH VOL II 6/7/71
 ODB 12/3/71
 AE 1/3/72

ET	AM	12	3	V	*6	CREWMAN	*6
0:00					7	// M479 OP 19 11.03.04-00 // M479-3 PERF # OF 6 PREPARE WATER QUENCH FILM AND COMMENT ON BURNING AND QUENCH OF SPECIMEN CS-M	<p><u>SCHEDULE CRITERIA</u></p> <ul style="list-style-type: none"> All phases of M512 to be completed prior to M479 performance. If M479 is scheduled on a different mission to M512 then the Experiment Integrity check must be accomplished once prior to the first performance of M479. If the experiment is terminated before the completion of the tests, then the film must be removed and stored. Upon resumption of the tests the exp. facility must first be verified. If FO-2 directly follows FO-1 then the storage of the film need not be performed. All performances of M479 to be completed prior to the end of SL-3 because of film degradation. <p><u>CONSTRAINTS</u></p> <ul style="list-style-type: none"> S/C maneuvers to be restricted to attitude hold during experiment performance. Each test will be performed using nominal workshop atmosphere. <p><u>RETURN WEIGHT</u></p> <ul style="list-style-type: none"> Film 10.5 pounds for all FO's.
0:14					2	// M479 TERM W 11.03.08-00 // M479-3 TERMINATE WATER QUENCH CS-M	
0:00					2	//M479 FILM S 11.03.09-00 //	
0:04						M479-3 FILM STOWAGE CS-M	
0:00							
0:03							
PHOTO PAD							

M479 ZERO GRAVITY FLAMMABILITY
 FO-4 UNDISTURBED BURNING OF PARTIALLY SUPPORTED SPECIMENS
 25 THRU 30 (SIX MATERIAL TYPES)

MRD 7/1/71
 EOH VOL I 4/9/71
 EOH VOL II 6/7/71
 ODB 12/3/71
 AE 1/3/72

ET	AM	1	2	3	V	*6	CREWMAN	*6
0:00						4	// M479 OP 25 11.03.05-00// M479-4 PERF # OF 6 FILM AND COMMENT ON BURNING OF SPECIMEN	
0:08						2	// M479 FILM S 11.03.09-00 // M479 -4 FILM STORAGE CS-M	
0:00								
0:03								
PHOTO PAD								
<p><u>SCHEDULE CRITERIA</u></p> <ul style="list-style-type: none"> All phases of M512 to be completed prior to M479 performance. If M479 is scheduled on a different mission to M512 then the Experiment Integrity check must be accomplished once prior to the first performance of M479. If the experiment is terminated before the completion of the tests, then the film must be removed and stored. Upon resumption of the tests the exp. facility must first be verified. If FO-2 directly follows FO-1 then the stowage of the film need not be performed. All performances of M479 to be completed prior to the end of SL-3 because of film degradation. <p><u>CONSTRAINTS</u></p> <ul style="list-style-type: none"> S/C maneuvers to be restricted to attitude hold during experiment performance. Each test will be performed using nominal workshop atmosphere. <p><u>RETURN WEIGHT</u></p> <ul style="list-style-type: none"> Film 10.5 pounds for all FO's. 								

MRD 7/1/71
 EOH VOL I 4/9/71
 EOH VOL II 6/7/71
 ODB 12/3/71
 AE 1/3/72

M479 ZERO GRAVITY FLAMMABILITY
 BURNING OF TWO IDENTICAL TEST SPECIMENS SEPARATED BY
 DIFFERENT GAP DISTANCES SPECIMENS 31 THRU 37

ET	AM	12	3	V	*8	CREWMAN	*6
0:00					4	// M479 OP 31 11.03.06-00 // M479-5 CS-M PERF # OF 7 FILM AND COMMENT ON BURNING OF SPECIMEN	<p>SCHEDULE CRITERIA</p> <ul style="list-style-type: none"> All phases of M512 to be completed prior to M479 performance. If M479 is scheduled on a different mission to M512 then the Experiment Integrity check must be accomplished once prior to the first performance of M479. If the experiment is terminated before the completion of the tests, then the film must be removed and stored. Upon resumption of the tests the exp. facility must first be verified. If FO-2 directly follows FO-1 then the stowage of the film need not be performed. All performances of M479 to be completed prior to the end of SL-3 because of film degradation. <p>CONSTRAINTS</p> <ul style="list-style-type: none"> S/C maneuvers to be restricted to attitude hold during experiment performance. Each test will be performed using nominal workshop atmosphere. <p>RETURN WEIGHT</p> <ul style="list-style-type: none"> Film 10.5 pounds for all FO's.
0:08					2	// M479 FILM S 11.03.09-00 // M479-5 FILM STOWAGE CS-M	
0:00					3	// M479 STOW E 14.06.01-00 // M479-5 CS-M STOW EXPERIMENT HARDWARE	
0:03					5	// M479 STOW F // M479-5 STOW CS-M STOW FILM IN CM FOR RETURN	
0:00							
0:06							
0:10							

PHOTO PAD

<p>M487 HABILITABILITY/CREW QUARTERS</p> <p>FO-1 OWS ENVIRONMENT</p> <p>FO-2 INTERNAL ARCHITECTURE</p> <p>FO-3 MOBILITY AIDS</p>	<p>FO-4 FOOD & WATER</p> <p>FO-5 PERSONAL ACCOUTREMENTS</p> <p>FO-6 PERSONAL HYGIENE</p>	<p>FO-7 HOUSEKEEPING</p> <p>FO-8 COMMUNICATIONS</p> <p>FO-9 OFF DUTY ACTIVITIES</p>	<p>ET</p> <p>AM</p> <p>1 2 3 V *6</p> <p>*6</p>	<p><u>SCHEDULE CRITERIA CONT.</u></p> <ul style="list-style-type: none"> FO-6, personal hygiene - photograph and voice record comments pertaining to bathing hand and face washing, drying, brushing teeth, shaving and cutting hair. FO-7, housekeeping - photograph and voice comment on normal housekeeping activities such as refuse disposal, stowage, vacuum cleaning, servicing waste management devices, disinfecting, etc. FO-8, communications - voice comment on the use and set-up of onboard communications devices. FO-9, off duty activities - voice comment on all aspects of physical exercises and recreational games. <p><u>CONSTRAINTS</u></p> <ul style="list-style-type: none"> None. <p><u>RETURN WEIGHT</u></p> <ul style="list-style-type: none"> Film 5.25 pounds.
--	--	---	---	--

M509 ASTRONAUT MANEUVERING EQUIPMENT
 FO-1 PERFORM FAMILIARIZATION AND MISSION MANEUVER TESTS
 (SHIRTSLEEVES) TO EVALUATE ALL FOUR OPERATIVE AMRV MODES

AM		CREWMAN	*6	NOTES
ET	123V*6			
0:00	1	//M509 INITIA 09.01.01-00 //	*6	<p>NOTES</p> <ul style="list-style-type: none"> ● Although the battery requires 20 hours to charge, it may be left on the charger for a longer period without being damaged. ● There are two batteries and three PSS tanks to charge prior to first experiment performance.
0:02		M509-1 INITIATE BATTERY CHG		
		WAIT 20 HOURS FOR BATTERY TO CHARGE		
	1	//M509 TERMIN 09.01.02-00 //		
0:00		M509-1 TERMINATE CHARGE		
0:02	5	//M509 BOTTLE 09.01.03-00 //		
		M509-1 CS-A UNSTOW RECHARGE STATION OBTAIN PSS BOTTLE FROM PSS RACK AND INSTALL AT RECHARGE STATION		
0:10		ALLOW 20 MIN FOR PSS BOTTLE TO PRESSURIZE		
		ALLOW <u>1BD</u> MIN FOR BOTTLE TO COOL		
	5	//M509 TOPOFF 09.01.04-00 //		
0:00		M509-1 CS-A TOP OFF PSS BOTTLE RETURN BOTTLE TO RACK STOW RECHARGE STATION		
0:10				

M509 ASTRONAUT MANEUVERING EQUIPMENT
 FO-1 PERFORM FAMILIARIZATION AND MISSION MANEUVER TESTS
 (SHIRTSLEEVES) TO EVALUATE ALL FOUR OPERATIVE AMRV MODES

ET	AM	1	2	3	V	*6	CREWMAN	*6	CREWMAN
0:00		73	//	M509-1	PERF 1	//	M509-1	73	//M509 OBSV 1
				M509-1	SUBJECT	CS-F			M509-1 OBSERVER
					PERF # 1 OF 1				PHOTO PAD
					SET UP CAMERAS AND LIGHTS				
					PREPARE ASMU:				
					<ul style="list-style-type: none"> • INSTALL PSS BOTTLE • INSTALL BATTERY • SPIN UP CMG'S (APPROX. 20 MIN TO ATTAIN FULL SPEED) 				
					CLEAR EXPERIMENT AREA OF OBSTACLES				ASSIST SUBJECT TO PREPARE AND DON ASMU
					DON ASMU				
					DON HARD HAT AND EAR PROTECTORS				DON EAR PROTECTORS
					CONFIGURE EQUIPMENT FOR OPERATION				CUE SUBJECT ON TEST PROCEDURES
					RELEASE DOCKING LATCH				START CAMERAS AND TAPE RECORDERS

M509 ASTRONAUT MANEUVERING EQUIPMENT
 FO-1 PERFORM FAMILIARIZATION AND MISSION MANEUVER TESTS
 (SHIRTSLEEVES) TO EVALUATE ALL FOUR OPERATIVE AMRV MODES

ET	AM	123	CREWMAN	*8	CREWMAN
			PERFORM MANEUVERS IN CMG RATE GYRO, DIRECT AND HHMU CONTROL MODES SWITCH CMG POWER OFF AFTER CMG MODE MANEUVER		CHANGE PSS TANK AFTER 30 MIN OPERATION
			ON COMPLETION OF MANEUVERS RETURN TO DONNING STATION AND DOCK SECURE ASMU: • ASMU POWER OFF • DOFF ASMU • PSS VALVE CLOSED • BLEED LINES BY FIRING THRUSTERS • CONNECT ASMU BATTERY TO CHARGER • INSTALL PSS BOTTLE IN AM RECHARGE STATION		ASSIST SUBJECT TO DOCK RECONFIGURE OWS DOFF EAR PROTECTORS STOW CAMERAS, FILM, FLOOD LIGHTS
2:25					

M509 ASTRONAUT MANEUVERING EQUIPMENT
 FO-1 PERFORM FAMILIARIZATION AND MISSION MANEUVER TESTS
 (SHIRTSLEEVES) TO EVALUATE ALL FOUR OPERATIVE AMRV MODES

ET	AM	1	2	3	V	*6	CREWMAN	*6	CREWMAN
0:00						5	ALLOW 20 MIN FOR PSS BOTTLE TO PRESSURIZE		
							ALLOW TBD MIN FOR BOTTLE TO COOL		
							//M509 TOPOFF 09.01.04-00 //		
0:10							M509-1 TOPOFF PSS BOTTLE RETURN BOTTLE TO RACK		
0:00						5	//M509 BOTTLE 09.01.03-00 //		
							M509-1 OBTAIN PSS BOTTLE FROM RACK AND INSTALL AT RECHARGE STATION		
0:10							ALLOW 20 HOURS FOR BATTERY TO RECHARGE		
0:00						1	//M509 TERMIN 09.01.02-00 //		
0:02							M509-1 TERMINATE CHARGE		

M509 ASTRONAUT MANEUVERING EQUIPMENT
 FO-2 PERFORM FAMILIARIZATION AND MISSION MANEUVER TESTS
 (SHIRTSLEEVES) AND COMPLETE SOME EXPLORATORY MANEUVERS
 TO EVALUATE ALL FOUR OPERATING AMRV MODES

ET	AM			CREWMAN	*6	CREWMAN
	1	2	3			
0:00		93	//	93//M509 PERF 2	//	M509-2 OBSV 2
				M509-2 SUBJECT PERF # 1 OF 1		M509-2 OBSERVER CS-F
				SET UP CAMERAS & LIGHTS		PHOTO PAD
				PREPARE ASMU: <ul style="list-style-type: none"> • INSTALL PSS TANK • SPIN UP CMG'S (APPROX. 20 MIN TO ATTAIN FULL SPEED) 		START ONE CAMERA TO PHOTO-GRAPH SUBJECT DOWNING ASMU (M151 REQUIREMENT)
				CLEAR EXPERIMENT AREA OF OBSTACLES		ASSIST SUBJECT TO PREPARE AND DON ASMU
				DON ASMU, HARDHAT & EAR PROTECTORS		DON EAR PROTECTORS
				CONFIGURE EQUIPMENT FOR OPERATION		CUE SUBJECT ON TEST PROCEDURES

M509 ASTRONAUT MANEUVERING EQUIPMENT
 FO-2 PERFORM FAMILIARIZATION AND MISSION MANEUVER TESTS
 (SHIRTSLEEVES) AND COMPLETE SOME EXPLORATORY MANEUVERS
 TO EVALUATE ALL FOUR OPERATING AMRV MODES

ET	I AM	1234	*6	CREWMAN	*6	CREWMAN
				RELEASE DOCKING LATCH		START CAMERAS AND TAPE RECORDERS
				PERFORM MANEUVERS IN CMG RATE GYRO, DIRECT AND HHMU CONTROL MODES		TAKE STILL PHOTOGRAPHY OF EXPERIMENT MANEUVERS
				SWITCH CMG POWER OFF AFTER CMG MODE		CHANGE PSS TANK AFTER 30 MIN OF OPERATION
				ON COMPLETION OF MANEUVERS RETURN TO DONNING STATION AND DOCK		CHANGE BATTERY AFTER 50 MIN OF OPERATION CHANGE SECOND PSS TANK AFTER 30 MIN OF OPERATION ASSIST SUBJECT TO DOCK

M509 ASTRONAUT MANEUVERING EQUIPMENT
 FO-2 PERFORM FAMILIARIZATION AND MISSION MANEUVER TESTS
 (SHIRTSLEEVES) AND COMPLETE SOME EXPLORATORY MANEUVERS
 TO EVALUATE ALL FOUR OPERATING AMRV MODES

ET	AM	CREWMAN	*6	CREWMAN
	1231			
2:45		SECURE ASMU: o ASMU POWER OFF o DOFF ASMU o PSS VALVE CLOSED o BLEED LINES BY FIRING THRUSTERS o CONNECT ASMU BATTERY TO CHARGER o INSTALL PSS TANK IN AM RECHARGE STATION	*6	RECONFIGURE OWS DOFF EAR PROTECTORS REMOVE & STOW CAMERAS, FILM, FLOOD LIGHTS
		ALLOW 20 MIN FOR PSS BOTTLE TO PRESSURIZE		
		ALLOW TBD MIN FOR PSS BOTTLE TO COOL		
		5 //M509 TOPOFF 09.01.04-00 //		
0:00		M509-2 TOPOFF PSS BOTTLE RETURN BOTTLE TO RACK		
0:10				

M509 ASTRONAUT MANEUVERING EQUIPMENT
 FO-2 PERFORM FAMILIARIZATION AND MISSION MANEUVER TESTS
 (SHIRTSLEEVES) AND COMPLETE SOME EXPLORATORY MANEUVERS
 TO EVALUATE ALL FOUR OPERATING AMRV MODES

ET	AM	12	3	V	*6	CREWMAN	*6
0:00					5	//M509 BOTTLE 09.01.03-00 // M509-2 CS-A OBTAIN PSS BOTTLE FROM RACK AND INSTALL AT RECHARGE STATION	
0:10						ALLOW 20 HRS FOR BATTERY TO RECHARGE	
0:00					1	//M509 TERMIN 09.01.02-00 //	
0:02						M509-2 TERMINATE CHARGE	
0:00					1	//M509 INITIA 09.01.01-00 //	
0:02						M509-2 INITIATE CHARGE	

M509 ASTRONAUT MANEUVERING EQUIPMENT
 FO-3 PERFORM MANEUVERING TESTS TO EVALUATE ALL
 FOUR OPERATING AMRV MODES WEARING A PRESSURIZED SUIT

ET	123 V *6	CREWMAN	*6	CREWMAN
0:00	T07//M509 PERF 3	M509-3 SUBJECT	//T07//M509 OBSV 3	M509-3 OBSERVER
	PERF #1 OF 1 PREPARE ASMU: <ul style="list-style-type: none"> • INSTALL PSS TANK • SPIN UP CMG'S (APPROX. 20 MIN TO ATTAIN FULL SPEED) 	CS-F		CS-F
	CLEAR EXPERIMENT AREA OF OBSTACLES			SET UP CAMERAS & LIGHTS
	UNSTOW ALSA & LSU			START ONE CAMERA TO PHOTOGRAPH SUBJECT DOWNING ASMU (M151 REQUIREMENT)
	DON PGA AND CONNECT LSU			ASSIST SUBJECT TO PREPARE AND DON ASMU
	DON ASMU & EAR PROTECTORS			
	PRESSURIZE SUIT			

M509 ASTRONAUT MANEUVERING EQUIPMENT
 FO-3 PERFORM MANEUVERING TESTS TO EVALUATE ALL
 FOUR OPERATING AMRV MODES WEARING A PRESSURIZED SUIT

AM		1	2	3	V	*S	CREWMAN	*S	CREWMAN
ET							ALLOW TBD MIN FOR PSS BOTTLE TO COOL		
0:00				5		//M509 TOPOFF 09.01.04-00 //	M509-3 CS-A		
0:10							TOPOFF PSS BOTTLE RETURN BOTTLE TO RACK		
0:00				5		//M509 BOTTLE 09.01.03-00 //	M509-3 CS-A		
0:10							OBTAIN PSS BOTTLE FROM RACK AND INSTALL AT RECHARGE STATION		
0:00							ALLOW 20 HRS FOR BATTERY TO RECHARGE		
0:02				1		//M509 TERMIN 09.01.02-00 //	M509-3 TERMINATE CHARGE		
0:00							//M509 INITIA 09.01.01-00 //		
0:02							M509-3 INITIATE CHARGE		

M509 ASTRONAUT MANEUVERING EQUIPMENT
 FO-4 REPEAT SOME OF THE MISSION MANEUVERS AND COMPLETE THE
 EXPLORATORY MANEUVERING TESTS (SHIRTSLEEVE)

ET	I AM 1 2 3 V *S	CREWMAN	*S	CREWMAN
0:00	78	M509-4 SUBJECT PERF # 1 OF 1 SET UP CAMERAS & LIGHTS PREPARE ASMU: • INSTALL PSS TANK • SPIN UP CMG'S (APPROX. 20 MIN TO ATTAIN FULL SPEED) CLEAR EXPERIMENT AREA OF OBSTACLES	// 78	M509-4 OBSERVER PHOTO PAD START ONE CAMERA TO PHOTO- GRAPH SUBJECT DOWNING ASMU (M151 REQUIREMENT)
		DON ASMU, HARDHAT & EAR PROTECTORS CONFIGURE EQUIPMENT FOR OPERATION		ASSIST SUBJECT TO PREPARE AND DON ASMU DON EAR PROTECTORS CUE SUBJECT ON TEST PROCEDURES

M509 ASTRONAUT MANEUVERING EQUIPMENT
 FO-4 REPEAT SOME OF THE MISSION MANEUVERS AND COMPLETE THE
 EXPLORATORY MANEUVERING TESTS (SHIRTSLEEVE)

ET	AM	1	2	3	V	*8	CREWMAN	*6	CREWMAN
							ALLOW 20 MIN FOR PSS BOTTLE TO PRESSURIZE		
							ALLOW TBD MIN FOR PSS BOTTLE TO COOL		
0:00						5	//M509 TOPOFF 09.01.04-00 // M509-4 TOPOFF PSS BOTTLE RETURN BOTTLE TO RACK		
0:10									
0:00						5	//M509 BOTTLE 09.01.03-00 // M509-4 OBTAIN PSS BOTTLE FROM RACK AND INSTALL AT RECHARGE STATION		
0:10									

M516 CREW ACTIVITIES/MAINTENANCE

ET	AM	1	2	3	V	*6	*6		
CREWMAN									
SEE PHOTOGRAPHY PAGE 4-42									

M551 METALS MELTING		MRD		
F0-1 EXAMINE MOLTEN METAL FLOW CHARACTERISTICS IN ZERO GRAVITY		1/1/71		
		EOH VOL I 4/9/71		
		EOH VOL II 6/7/71		
		ODB 12/3/71		
		AE 1/3/72		
ET	AM	*S	*S	CREWMAN
0:00	12	// M551 OP 11.04.03-00 //		
		M551-1 PERF # OF 3 EXPERIMENT FACILITY VERIFICATION	CS-M	
		INSTALL CAMERA AND FILM		
		MOUNT SPECIMEN # OF 3 TO DRIVE MOTOR AND INSTALL IN CHAMBER		
		CONDUCT MELTING SEQUENCE		
0:25		WAIT A MINIMUM OF 2.5 HOURS FOR SPECIMEN TO COOL		
0:00	2	// M551 REMOVA 11.09.04.00//		
0:04		M551 - 1 REMOVE SPECIMEN	CS-M	
0:00	2	//M551 TERMIN 11.04.05-00//		
0:04		M551 - 1 STOW CLEAN CHAMBER STOW SAMPLE	CS-M	
<p>SCHEDULE CRITERIA</p> <ul style="list-style-type: none"> ● Schedule prior to performance of M479. ● If the 3 performances of M551-1 are scheduled consecutively, the camera installation and equipment stowage need only to be performed once. <p>CONSTRAINTS</p> <ul style="list-style-type: none"> ● None. <p>RETURN WEIGHT</p> <ul style="list-style-type: none"> ● Sample TBD pounds. ● Film TBD pounds. 				

M552 MATERIALS PROCESSING IN SPACE
FO-1 EXOTHERMIC HEATING TASK

MRD 7/1/71
EOH VOL I 4/9/71
EOH VOL II 6/7/71
ODB 12/3/71
AE 1/3/72

ET	AM	1	2	3	V	*6	*6	*6
0:00						7	//M552 OP 11.04.08-00 // M552-1 VERIFY FACILITY INSTALL BRAZING MODULE SELECT SPECIMEN SWITCH TRIGGER SPECIMEN	<p><u>SCHEDULE CRITERIA</u></p> <ul style="list-style-type: none"> Schedule prior to the performance of experiment M479. <p><u>RETURN WEIGHT</u></p> <ul style="list-style-type: none"> Samples TBD pounds.
0:14						≠	WAIT MINIMUM OF 2 HOURS FOR SPECIMEN TO COOL	
0:00						2	//M552 CONTIN // M552-1 SET AND TRIGGER SWITCH	
0:04						3	//M552 TERMIN 11.04.09-00// M552-1 STOW REMOVE AND STOW BRAZING MODULE	
0:00								
0:06								

M553 MATERIALS PROCESSING IN SPACE FO-1 SPHERE FORMING TASK		MRD EOH VOL I 4/9/71 EOH VOL II 6/7/71 ODB 12/3/71 AE 1/3/72	
ET	AM	V	*6
0:00	23	//	M553 PREP & 11.09.06-00// M553-1 CS-M PERF # OF 2 EXP FACILITY VERIFICATION INSTALL CAMERA AND LOAD FILM INSERT SPHERE CATCHER MOUNT SPECIMEN # OF 2 TO DRIVE MOTOR AND INSTALL INITIATE VOICE RECORDING CONFIGURE CONTROL PANEL FOCUS ELECTRON BEAM ON TARGET THEN INDEX TO FIRST SAMPLE CONDUCT MELTING SEQUENCE TO FORM THE 14 SPHERES RECONFIGURE CONTROL PANEL WAIT A MINIMUM OF 15 MIN FOR SPHERES TO COOL 5 // M553 REMOVA // M553 - I STOW CS-M REMOVE AND STOW SPHERE CATCHER AND CAMERA AND FILM
0:45			
0:00			
0:10			
		PHOTO PAD	
		<p><u>SCHEDULE CRITERIA</u></p> <ul style="list-style-type: none"> • Schedule prior to the performance of experiment M479. • If the 2 performances of M553 are scheduled consecutively then the camera installation and equipment stowage need only be performed once. <p><u>CONSTRAINTS</u></p> <ul style="list-style-type: none"> • None. <p><u>RETURN WEIGHT</u></p> <ul style="list-style-type: none"> • Samples TBD pounds. • Film TBD pounds. 	

M554 MATERIALS PROCESSING IN SPACE
FO-1 COMPOSITE CASTING TASK

MRD 7/1/71
EOH VOL I 4/9/71
EOH VOL II 6/7/71
ODB 12/3/71
AE 1/3/72

ET	AM	1	2	3	V	*6	CREWMAN	*6
0:00						4	// M554 PREP & 11.04.10-00 // M554 -1 VERIFY FACILITY INSTALL CASTING MODULE INITIATE HEATING CYCLE	<p>SCHEDULE CRITERIA</p> <ul style="list-style-type: none"> Schedule prior to the performance of experiment M479. During the 3 hr. heating cycle and the 3 hr. cooling cycle the temperature of the casting module should be checked intermittently. <p>CONSTRAINTS</p> <ul style="list-style-type: none"> None. <p>RETURN WEIGHT</p> <ul style="list-style-type: none"> Samples TBD pounds. <p>NOTES</p> <ul style="list-style-type: none"> Spacecraft accelerations are to be avoided if possible. However, post-flight telemetry data of accelerations during attitude maneuvers will be required.
0:08							WAIT MIN. OF 3 HRS FOR TEMP TO REACH 950°C	
0:00						1	//M554 INITIATE 11.04.11-00// M554 INITIATE COOLING CYCLE	
0:02							WAIT MIN OF 3 HRS FOR TEMP TO DROP TO 500°C	
0:00						1	//M554 TERMIN 11.04.12-00 // M554 TERMINATE COOL CYCLE	
0:02						4	//M554 STOW 11.04.13-00 // M554-1 STOW REMOVE AND STOW CASTING	
0:00								
0:08								

M555 MATERIALS PROCESSING IN SPACE
FO-1 SINGLE CRYSTAL GROWTH

ET	AM	1	2	3	V	*6	*6
0:00						11	//M555 PREP, I 11.04.14-00// M555-1 CS-M VERIFY FACILITY INSTALL CRYSTAL GROWTH PACKAGE INITIATE HEATING
0:22							RECORD TEMP EVERY 30 MIN FOR 3 HOURS
0:00						5	RECORD TEMP EVERY 12 HOURS FOR 112 HOURS
0:10							//M555 TERMIN 11.04.16-00// M555-1 CS-M TERMINATE HEATING CYCLE PRESSURIZE CHAMBER RECORD TEMP EVERY 30 MIN FOR 3 HOURS WAIT 9 HOURS (TOTAL) FOR COMPLETE COOLING
							<p><u>SCHEDULE CRITERIA</u></p> <ul style="list-style-type: none"> ● Schedule prior to the performance of experiment M479. <p><u>CONSTRAINTS</u></p> <ul style="list-style-type: none"> ● None <p><u>RETURN WEIGHT</u></p> <ul style="list-style-type: none"> ● Samples TBD pounds. <p><u>NOTES</u></p> <ul style="list-style-type: none"> ● Spacecraft accelerations are to be avoided if possible. However, postflight telemetry data of accelerations during attitude maneuvers will be required.

M555 MATERIALS PROCESSING IN SPACE
 FO-1 SINGLE CRYSTAL GROWTH

		AM				
ET	1	2	3	V	*6	*6
0:00					6	// STOW MATERI 14.06.02-00// M555-1 STOW CS-M STOW CRYSTAL GROWTH PACKAGE AND EQUIP
0:12					5	// M555 FILM S //
0:00						M555 -1 FILM STOW CS-CM STOW SAMPLES AND FILM IN CM FOR RETURN
0:10						

S009 NUCLEAR EMULSION FO-1 INSTALL, EXPOSE AND STOW HTE NUCLEAR EMULSION PACKAGE		MRD 4/30/71 EOH VOL I 11/19/71 EOH VOL II 6/7/71 ODB 12/17/71 AE 1/3/72	
ET	AM	CREWMAN	*6
0:00	123 V *6	PREPAR 08.01.01-00// S009-T PREP CS-M PREPARE EXP HOUSING OBTAIN DETECTOR PKG FROM VAULT AND INSTALL	*6
0:11		ACTIVA 08.01.02-00// S009-T ACTIVATE PKG CS-M	2
0:00 0:02		BETA A 08.01.02-00// S009-T BETA ANG & TIME SET	2
0:00 0:02		TERM & 08.01.04-00// S009 TERM & CS-CM TERMINATE EXPOSURE AND REMOVE DETECTOR PKG STOW FOR RETURN	2
0:12			2
SCHEDULE CRITERIA			<ul style="list-style-type: none"> The detector package should be installed in the MDA within 5 days after launch. The detector package should be exposed a minimum of 250 hrs. Update pads will be required throughout the exposure period for changes in the angle adjustments are made in increments of 7.5° and the period timer adjustment can only be made while crossing 30°N latitude going south. It is desirable to make both adjustments at the same time. This will require the new pivot angle setting to be within 3.75° of the actual angle during a southbound crossing of 30°N latitude.
CONSTRAINTS			<ul style="list-style-type: none"> The field-of-view of the detector package must not be directed toward the earth atmosphere. No radiation sources will be allowed in the vicinity of the detector package at any time. The detector package must be in closed configuration whenever the spacecraft is north of 30°N latitude or south of 25.5°S latitude.
RETURN WEIGHT			<ul style="list-style-type: none"> Detector package 28.5 lb.

S015 ZERO GRAVITY SINGLE HUMAN CELLS
 FO-1 OPERATE THE CYTOCHEMICAL EXPERIMENT SUBSYSTEM
 (BIOPACK 1)

ET	1 AM	12	3	V	*6	*6
0:00					5	//S015 BIO 1A 11.09.02-01 // S015-1 CS-CM VERIFY INITIAL SW. POSI- PARTIALLY LABEL BIOPACK 1 BY CONDUCTING SWITCHING SEQUENCE UP TO POSITION 3
0:11					5	WAIT 60 ±3 MIN. FROM WHEN THE BIOPACK 1 SWITCH WAS PLACED IN POSITION 3
0:00					5	//S015 BIO 1B 11.09.02-02 // S015-1 CS-CM RINSE AND COMPLETE LABELING OF BIOPACK 1 BY CONDUCT- ING SWITCHING SEQUENCE UP TO POSITION 5
0:11					5	WAIT 60 ±3 MIN. FROM WHEN THE BIOPACK 1 SWITCH WAS PLACED IN POSITION 5
0:00					7	//S015 BIO 1C 11.09.02-03// S015-1 CS-CM RINSE AND FIX SPECIMENS OF BIOPACK 1 BY CONDUCTING SWITCHING SEQUENCE UP TO POSITION 8 SWITCH BIOPACK 1 OFF
0:13						

SCHEDULE CRITERIA

- Schedule biopack 1 performance at any convenient time on the fourth day after launch.
- Schedule biopack 2 performance at any convenient time on the tenth day after launch.

CONSTRAINTS

- After each step has been initiated on the label, rinse, fix cycle the biopack lamp will come on for about 2-5 minutes. If the lamp does not come on, or if the lamp is on for less than 30 secs. or for more than 8 minutes the flight controllers should be notified so that contingency procedures can be adopted. When the lamp goes out, the biopack switch should be moved to the next position within 10 min. unless a 60 ±3 min. wait period has been scheduled.
- Temperatures beyond the 50°F to 95°F range must be reported to the PI within 24 hours of the occurrence in order to simulate the same temperature for the ground based "control experiment".

RETURN WEIGHT

- The flight hardware package weighs 23.5 pounds and is never removed from the CM

S015 ZERO GRAVITY SINGLE HUMAN CELL

F0-1 OPERATE THE CYTOCHEMICAL EXPERIMENT SUBSYSTEM
(BIOPACK 2)

ET	1	2	3	V	*6	*6
0:00					5	//S015 BIO 2A 11.09.03-01 // S015-1 CS-CM VERIFY INITIAL SW.POSITIONS PARTIALLY LABEL BIOPACK 2 BY CONDUCTING SWITCHING SEQUENCE UP TO POSITION 3
0:11						WAIT 60 ± 3 MIN. FROM WHEN THE BIOPACK 2 SWITCH WAS PLACED IN POSITION 3
0:00					5	//S015 BIO 2B 11.09.03-02// S015-1 CS-CM RINSE AND COMPLETE LABELING OF BIOPACK 2 BY CONDUCT- ING SWITCHING SEQUENCE UP TO POSITION 5
0:11						WAIT 60 + 3 MIN. FROM WHEN THE BIOPACK 2 SWITCH WAS PLACED IN POSITION 5
0:00					7	//S015 BIO 2C 11.09.03-03// S015-1 CS-CM RINSE AND FIX SPECIMENS OF BIOPACK 2 BY CONDUCTING SWITCHING SEQUENCE UP TO POSITION 8 SWITCH BIOPACK 2 OFF
0:13						

S019 ULTRAVIOLET STELLAR ASTRONOMY
FO-1 THROUGH 12 OBTAIN UV PHOTOGRAPHS OF DESIGNATED
STARFIELDS DURING DARKSIDE PASSES

ET	AM			CREWMAN	*6	CREWMAN
	1	2	3			
0:45				RETRACT MIRROR SYSTEM AND CLOSE SAL DOOR BEFORE SUNRISE		
0:00		3		//S019 REMOVE 08.03.08-00//	1	//S019 ENABLE
0:07				S019-1 A-SOL-SAL REMOVE SPECTROGRAPH FROM AMS		S019-1 ENABLE DMG DUMP
0:00		13		//S019 STOW E 08.03.10-00//		
				S019-1 A-SOL-SAL EVACUATE AND STOW MIRROR SYSTEM		
				EVACUATE AND STOW OPTICAL CANISTER		
0:25				STOW FILM CANISTER SECURE SAL		
0:00		5		//S019 LOG AN 08.03.11-00//		
				S019-1 STOW CS-CM		
0:10				STOW LOG AND FILM IN CM FOR RETURN		

CONSTRAINTS (CONTINUED)

- Spacecraft attitude must be known to within 2.5 degrees in the stellar inertial reference system.
- NOTES:
- Spacecraft rate above 10 arc-sec/sec are not acceptable.
 - If the experiment is removed from the airlock for more than 12 hours, all experiment hardware must be evacuated.
 - Of the 150 frames available, a minimum of 115 UV photographs will satisfy the experiment objective. However, to use all the frames, it is expected that an average of 12 exposures can be made on each 12 darkside passes.
 - 3 to 4 starfields will be photographed on each darkside pass using from 2 to 4 exposures per starfield.

RETURN WEIGHT

- Film canister 14.9 lb.

S019 UV STELLAR ASTRONOMY
CANDIDATE STARFIELDS

- A. MORE THAN 3 O/B STARS FROM BSC AND MORE THAN 1 O/B STAR BRIGHTER THAN 5.1 VISUAL AND MORE THAN 9 STARS ALTOGETHER FROM BSC.
- B. MORE THAN 3 O/B STARS FROM BSC AND EITHER MORE THAN 1 O/B STAR BRIGHTER THAN 5.1 VISUAL OR MORE THAN 9 STARS ALTOGETHER FROM BSC.
- C. MORE THAN 1 O/B STAR BRIGHTER THAN 5.1 VISUAL.
- D. MORE THAN 4 O/B STARS FROM BSC.

STARFIELD ID NUMBER	RT ASCENSION-1950		DECLINATION-1950	GUIDE STAR	STARFIELD CRITERIA	NO. OF STARS IN BSC	PRIORITY	EXPOSURE SETTINGS (SEC)			
	h	m						30	90	270	MANUAL
1	00	16	+62.5	TBD	D	8	TBD	TBD	TBD	TBD	
2	00	18	-70.5	TBD		3	TBD	TBD	TBD	TBD	
3	00	25	+52.5	TBD	B	7	TBD	TBD	TBD	TBD	
4	00	55	+60.0	TBD	B	11	TBD	TBD	TBD	TBD	
5	01	36	-57.5	TBD		4	TBD	TBD	TBD	TBD	
6	01	53	+55.0	TBD		5	TBD	TBD	TBD	TBD	
7	01	55	+64.0	TBD		6	TBD	TBD	TBD	TBD	
8	02	42	+27.0	TBD	C	6	TBD	TBD	TBD	TBD	
9	02	48	+16.5	TBD		8	TBD	TBD	TBD	TBD	
10	03	24	+48.0	TBD	A	16	TBD	TBD	TBD	TBD	
11	03	27	+10.5	TBD		7	TBD	TBD	TBD	TBD	
12	03	37	+35.5	TBD		5	TBD	TBD	TBD	TBD	
13	03	42	+47.0	TBD	B	5	TBD	TBD	TBD	TBD	
14	03	44	+32.0	TBD	A	17	TBD	TBD	TBD	TBD	
15	03	46	+32.5	TBD	A	11	TBD	TBD	TBD	TBD	
16	03	47	+ 6.0	TBD		4	TBD	TBD	TBD	TBD	
17	03	50	+35.0	TBD	D	8	TBD	TBD	TBD	TBD	
18	03	51	-24.0	TBD		4	TBD	TBD	TBD	TBD	
19	04	01	+ 5.0	TBD		5	TBD	TBD	TBD	TBD	
20	04	10	+50.5	TBD		4	TBD	TBD	TBD	TBD	
21	04	10	+47.5	TBD	C	3	TBD	TBD	TBD	TBD	
22	04	15	+ 9.0	TBD		10	TBD	TBD	TBD	TBD	
23	04	21	+21.5	TBD		11	TBD	TBD	TBD	TBD	
24	04	21	+14.5	TBD		19	TBD	TBD	TBD	TBD	
25	04	22	+17.5	TBD		17	TBD	TBD	TBD	TBD	
26	04	32	+16.0	TBD		15	TBD	TBD	TBD	TBD	
27	04	37	+11.5	TBD		5	TBD	TBD	TBD	TBD	
28	04	37	- 3.5	TBD	B	6	TBD	TBD	TBD	TBD	
29	04	48	+ 7.0	TBD		3	TBD	TBD	TBD	TBD	
30	04	54	+ 1.5	TBD		8	TBD	TBD	TBD	TBD	
31	04	54	- 6.5	TBD		5	TBD	TBD	TBD	TBD	
32	04	58	+42.0	TBD		7	TBD	TBD	TBD	TBD	
33	05	12	- 8.0	TBD	B	5	TBD	TBD	TBD	TBD	
34	05	15	-13.0	TBD	B	8	TBD	TBD	TBD	TBD	
35	05	22	+ 2.5	TBD	A	10	TBD	TBD	TBD	TBD	

S019 UV STELLAR ASTRONOMY
CANDIDATE STARFIELDS

- A. MORE THAN 3 O/B STARS FROM BSC AND MORE THAN 1 O/B STAR BRIGHTER THAN 5.1 VISUAL AND MORE THAN 9 STARS ALTOGETHER FROM BSC.
- B. MORE THAN 3 O/B STARS FROM BSC AND EITHER MORE THAN 1 O/B STAR BRIGHTER THAN 5.1 VISUAL OR MORE THAN 9 STARS ALTOGETHER FROM BSC.
- C. MORE THAN 1 O/B STAR BRIGHTER THAN 5.1 VISUAL.
- D. MORE THAN 4 O/B STARS FROM BSC.

STARFIELD ID NUMBER	RT ASCENSION-1950		DECLINATION-1950	GUIDE STAR	STARFIELD CRITERIA	NO.OF STARS IN BSC	PRIORITY	EXPOSURE SETTINGS (SEC)		
	h	m						30	90	270
36	05	23	+29.5	TBD		6	TBD	TBD	TBD	TBD
37	05	24	+ 6.5	TBD	B	5	TBD	TBD	TBD	TBD
38	05	24	- 1.5	TBD	A	18	TBD	TBD	TBD	TBD
39	05	25	+16.0	TBD	B	10	TBD	TBD	TBD	TBD
40	05	30	+21.0	TBD	C	4	TBD	TBD	TBD	TBD
41	05	33	+25.0	TBD	D	8	TBD	TBD	TBD	TBD
42	05	33	+ 9.0	TBD	B	6	TBD	TBD	TBD	TBD
43	05	33	- 6.0	TBD	A	22	TBD	TBD	TBD	TBD
44	05	34	+17.0	TBD		6	TBD	TBD	TBD	TBD
45	05	34	+ 3.0	TBD		6	TBD	TBD	TBD	TBD
46	05	34	- 1.5	TBD	A	18	TBD	TBD	TBD	TBD
47	05	38	-34.0	TBD		9	TBD	TBD	TBD	TBD
48	05	45	+14.0	TBD		6	TBD	TBD	TBD	TBD
49	05	48	- 9.0	TBD		4	TBD	TBD	TBD	TBD
50	05	53	-35.0	TBD	C	6	TBD	TBD	TBD	TBD
51	06	06	+19.0	TBD		6	TBD	TBD	TBD	TBD
52	06	08	- 5.5	TBD		9	TBD	TBD	TBD	TBD
53	06	10	+15.0	TBD	B	9	TBD	TBD	TBD	TBD
54	06	16	- 8.0	TBD		7	TBD	TBD	TBD	TBD
55	06	18	-13.0	TBD		4	TBD	TBD	TBD	TBD
56	06	18	-19.0	TBD		4	TBD	TBD	TBD	TBD
57	06	22	-53.0	TBD		8	TBD	TBD	TBD	TBD
58	06	25	-31.5	TBD	C	6	TBD	TBD	TBD	TBD
59	06	28	- 6.0	TBD	B	10	TBD	TBD	TBD	TBD
60	06	30	+68.5	TBD	B	4	TBD	TBD	TBD	TBD
61	06	35	+ 5.5	TBD		10	TBD	TBD	TBD	TBD
62	06	35	+ 0.0	TBD		4	TBD	TBD	TBD	TBD
63	06	39	-37.0	TBD	B	12	TBD	TBD	TBD	TBD
64	06	44	-16.0	TBD		4	TBD	TBD	TBD	TBD
65	06	46	-31.5	TBD	D	8	TBD	TBD	TBD	TBD
66	06	52	-20.0	TBD	B	10	TBD	TBD	TBD	TBD
67	06	52	-24.0	TBD	B	11	TBD	TBD	TBD	TBD
68	06	58	-28.0	TBD		7	TBD	TBD	TBD	TBD
69	07	00	-16.0	TBD	C	4	TBD	TBD	TBD	TBD
70	07	03	-25.0	TBD	B	10	TBD	TBD	TBD	TBD

S019 UV STELLAR ASTRONOMY
CANDIDATE STARFIELDS

- A. MORE THAN 3 O/B STARS FROM BSC AND MORE THAN 1 O/B STAR BRIGHTER THAN 5.1 VISUAL AND MORE THAN 9 STARS ALTOGETHER FROM BSC.
- B. MORE THAN 3 O/B STARS FROM BSC AND EITHER MORE THAN 1 O/B STAR BRIGHTER THAN 5.1 VISUAL OR MORE THAN 9 STARS ALTOGETHER FROM BSC.
- C. MORE THAN 1 O/B STAR BRIGHTER THAN 5.1 VISUAL.
- D. MORE THAN 4 O/B STARS FROM BSC.

STARFIELD ID NUMBER	RT ASCENSION-1950		DECLINATION-1950	GUIDE STAR	STARFIELD CRITERIA	NO. OF STARS IN BSC	PRIORITY	EXPOSURE SETTINGS (SEC)		
	h	m						30	90	270
71	07	07	-11.0	TBD	D	9	TBD	TBD	TBD	TBD
72	07	10	-40.0	TBD		7	TBD	TBD	TBD	TBD
73	07	12	-26.0	TBD	A	18	TBD	TBD	TBD	TBD
74	07	16	-36.5	TBD	B	8	TBD	TBD	TBD	TBD
75	07	17	-24.0	TBD	A	11	TBD	TBD	TBD	TBD
76	07	19	-17.5	TBD		7	TBD	TBD	TBD	TBD
77	07	19	-31.0	TBD	B	14	TBD	TBD	TBD	TBD
78	07	25	-22.5	TBD		7	TBD	TBD	TBD	TBD
79	07	33	-14.5	TBD		6	TBD	TBD	TBD	TBD
80	07	35	-23.5	TBD		9	TBD	TBD	TBD	TBD
81	07	35	-36.0	TBD	A	20	TBD	TBD	TBD	TBD
82	07	37	-27.0	TBD	A	11	TBD	TBD	TBD	TBD
83	07	43	-38.0	TBD	A	23	TBD	TBD	TBD	TBD
84	07	46	-25.5	TBD		7	TBD	TBD	TBD	TBD
85	07	51	-43.5	TBD	B	13	TBD	TBD	TBD	TBD
86	07	52	-47.5	TBD	A	13	TBD	TBD	TBD	TBD
87	07	58	-39.5	TBD	B	6	TBD	TBD	TBD	TBD
88	07	58	-53.0	TBD	D	8	TBD	TBD	TBD	TBD
89	08	01	-49.0	TBD	A	15	TBD	TBD	TBD	TBD
90	08	04	-63.0	TBD		7	TBD	TBD	TBD	TBD
91	08	12	-47.5	TBD	A	12	TBD	TBD	TBD	TBD
92	08	14	-36.5	TBD	B	12	TBD	TBD	TBD	TBD
93	08	20	-70.0	TBD		5	TBD	TBD	TBD	TBD
94	08	26	-47.5	TBD		6	TBD	TBD	TBD	TBD
95	08	28	-51.5	TBD	B	12	TBD	TBD	TBD	TBD
96	08	29	-43.0	TBD	D	6	TBD	TBD	TBD	TBD
97	08	29	-59.0	TBD		8	TBD	TBD	TBD	TBD
98	08	41	-48.5	TBD	B	10	TBD	TBD	TBD	TBD
99	08	44	-46.0	TBD	B	13	TBD	TBD	TBD	TBD
100	08	45	-53.5	TBD	A	12	TBD	TBD	TBD	TBD
101	08	51	-58.0	TBD	A	12	TBD	TBD	TBD	TBD
102	08	54	-59.5	TBD	A	10	TBD	TBD	TBD	TBD
103	09	09	-70.0	TBD		6	TBD	TBD	TBD	TBD
104	09	11	- 8.5	TBD		5	TBD	TBD	TBD	TBD
105	09	11	-44.0	TBD	D	8	TBD	TBD	TBD	TBD

- A. MORE THAN 3 O/B STARS FROM BSC AND MORE THAN 1 O/B STAR BRIGHTER THAN 5.1 VISUAL AND MORE THAN 9 STARS ALTOGETHER FROM BSC.
- B. MORE THAN 3 O/B STARS FROM BSC AND EITHER MORE THAN 1 O/B STAR BRIGHTER THAN 5.1 VISUAL OR MORE THAN 9 STARS ALTOGETHER FROM BSC.
- C. MORE THAN 1 O/B STAR BRIGHTER THAN 5.1 VISUAL.
- D. MORE THAN 4 O/B STARS FROM BSC.

S019 UV STELLAR ASTRONOMY
CANDIDATE STARFIELDS

STARFIELD ID NUMBER	RT ASCENSION-1950		DECLINATION-1950	GUIDE STAR	STARFIELD CRITERIA	NO. OF STARS IN BSC	PRIORITY	EXPOSURE SETTINGS (SEC)		
	h	m						30	90	270
106	09	13	-58.5	TBD		10	TBD	TBD	TBD	TBD
107	09	24	-62.0	TBD	C	8	TBD	TBD	TBD	TBD
108	09	25	-52.0	TBD	D	8	TBD	TBD	TBD	TBD
109	09	49	-45.0	TBD		6	TBD	TBD	TBD	TBD
110	10	01	-53.0	TBD	A	11	TBD	TBD	TBD	TBD
111	10	35	-65.0	TBD	A	13	TBD	TBD	TBD	TBD
112	10	39	-58.5	TBD	B	22	TBD	TBD	TBD	TBD
113	11	10	-71.5	TBD	D	6	TBD	TBD	TBD	TBD
114	11	17	-64.5	TBD		7	TBD	TBD	TBD	TBD
115	11	25	-54.0	TBD	C	3	TBD	TBD	TBD	TBD
116	11	33	-60.5	TBD		13	TBD	TBD	TBD	TBD
117	11	47	-63.0	TBD	A	17	TBD	TBD	TBD	TBD
118	12	03	-78.5	TBD	C	5	TBD	TBD	TBD	TBD
119	12	07	-51.0	TBD	B	6	TBD	TBD	TBD	TBD
120	12	15	-63.5	TBD	A	17	TBD	TBD	TBD	TBD
121	12	19	-58.0	TBD		10	TBD	TBD	TBD	TBD
122	12	29	-50.0	TBD	C	7	TBD	TBD	TBD	TBD
123	12	39	-68.0	TBD	B	9	TBD	TBD	TBD	TBD
124	12	41	-48.0	TBD		8	TBD	TBD	TBD	TBD
125	12	45	-59.5	TBD	B	9	TBD	TBD	TBD	TBD
126	12	48	-56.5	TBD	B	7	TBD	TBD	TBD	TBD
127	13	02	-49.0	TBD	C	6	TBD	TBD	TBD	TBD
128	13	13	-65.5	TBD		9	TBD	TBD	TBD	TBD
129	13	14	-59.5	TBD	C	7	TBD	TBD	TBD	TBD
130	13	28	+55.5	TBD		5	TBD	TBD	TBD	TBD
131	13	42	+54.0	TBD		6	TBD	TBD	TBD	TBD
132	13	43	-53.0	TBD	D	8	TBD	TBD	TBD	TBD
133	13	47	-32.5	TBD	C	6	TBD	TBD	TBD	TBD
134	13	48	-41.5	TBD	B	5	TBD	TBD	TBD	TBD
135	13	52	-46.0	TBD	B	6	TBD	TBD	TBD	TBD
136	14	06	-60.0	TBD		7	TBD	TBD	TBD	TBD
137	14	16	-56.5	TBD		4	TBD	TBD	TBD	TBD
138	14	21	-45.5	TBD	C	7	TBD	TBD	TBD	TBD
139	14	32	-41.5	TBD		5	TBD	TBD	TBD	TBD
140	14	33	-48.5	TBD	C	7	TBD	TBD	TBD	TBD

S019 UV STELLAR ASTRONOMY
CANDIDATE STARFIELDS

- A. MORE THAN 3 O/B STARS FROM BSC AND MORE THAN 1 O/B STAR BRIGHTER THAN 5.1 VISUAL AND MORE THAN 9 STARS ALTOGETHER FROM BSC.
- B. MORE THAN 3 O/B STARS FROM BSC AND EITHER MORE THAN 1 O/B STAR BRIGHTER THAN 5.1 VISUAL OR MORE THAN 9 STARS ALTOGETHER FROM BSC.
- C. MORE THAN 1 O/B STAR BRIGHTER THAN 5.1 VISUAL.
- D. MORE THAN 4 O/B STARS FROM BSC.

STARFIELD ID NUMBER	RT ASCENSION-1950		DECLINATION-1950 °	GUIDE STAR	STARFIELD CRITERIA	NO. OF STARS IN BSC	PRIORITY	EXPOSURE SETTINGS (SEC)		
	h	m						30	90	270
141	14	44	-37.0	TBD		7	TBD	TBD	TBD	TBD
142	14	44	-63.0	TBD		5	TBD	TBD	TBD	TBD
143	14	45	-26.0	TBD		8	TBD	TBD	TBD	TBD
144	14	53	-42.5	TBD	C	4	TBD	TBD	TBD	TBD
145	15	07	+73.0	TBD		3	TBD	TBD	TBD	TBD
146	15	09	-47.5	TBD	C	8	TBD	TBD	TBD	TBD
147	15	13	-44.5	TBD	B	6	TBD	TBD	TBD	TBD
148	15	14	-59.5	TBD		8	TBD	TBD	TBD	TBD
149	15	21	-37.5	TBD		9	TBD	TBD	TBD	TBD
150	15	25	-40.5	TBD	A	11	TBD	TBD	TBD	TBD
151	15	31	-73.0	TBD		3	TBD	TBD	TBD	TBD
152	15	46	-34.0	TBD		5	TBD	TBD	TBD	TBD
153	15	51	-25.5	TBD	B	9	TBD	TBD	TBD	TBD
154	15	55	-21.5	TBD	B	5	TBD	TBD	TBD	TBD
155	16	01	-37.5	TBD	B	9	TBD	TBD	TBD	TBD
156	16	06	-20.0	TBD	C	4	TBD	TBD	TBD	TBD
157	16	08	-24.5	TBD	D	6	TBD	TBD	TBD	TBD
158	16	10	+46.0	TBD		3	TBD	TBD	TBD	TBD
159	16	11	-28.0	TBD		4	TBD	TBD	TBD	TBD
160	16	19	-48.0	TBD		6	TBD	TBD	TBD	TBD
161	16	22	-25.0	TBD	C	6	TBD	TBD	TBD	TBD
162	16	31	-43.0	TBD		4	TBD	TBD	TBD	TBD
163	16	47	-41.5	TBD	B	15	TBD	TBD	TBD	TBD
164	16	48	-59.0	TBD	D	8	TBD	TBD	TBD	TBD
165	16	54	-37.5	TBD	C	7	TBD	TBD	TBD	TBD
166	16	56	-33.5	TBD		6	TBD	TBD	TBD	TBD
167	17	12	-33.0	TBD		6	TBD	TBD	TBD	TBD
168	17	22	-61.5	TBD		5	TBD	TBD	TBD	TBD
169	17	24	-45.5	TBD		8	TBD	TBD	TBD	TBD
170	17	32	-38.0	TBD	C	9	TBD	TBD	TBD	TBD
171	17	35	+53.5	TBD		5	TBD	TBD	TBD	TBD
172	17	39	-32.5	TBD		3	TBD	TBD	TBD	TBD
173	17	42	-39.0	TBD		9	TBD	TBD	TBD	TBD
174	17	50	+52.0	TBD		4	TBD	TBD	TBD	TBD
175	17	56	+ 1.5	TBD		7	TBD	TBD	TBD	TBD

- A. MORE THAN 3 O/B STARS FROM BSC AND MORE THAN 1 O/B STAR BRIGHTER THAN 5.1 VISUAL AND MORE THAN 9 STARS ALTOGETHER FROM BSC.
- B. MORE THAN 3 O/B STARS FROM BSC AND EITHER MORE THAN 1 O/B STAR BRIGHTER THAN 5.1 VISUAL OR MORE THAN 9 STARS ALTOGETHER FROM BSC.
- C. MORE THAN 1 O/B STAR BRIGHTER THAN 5.1 VISUAL.
- D. MORE THAN 4 O/B STARS FROM BSC.

S019 UV STELLAR ASTRONOMY
CANDIDATE STARFIELDS

STARFIELD ID NUMBER	RT ASCENSION-1950		DECLINATION-1950	GUIDE STAR	STARFIELD CRITERIA	NO. OF STARS IN BSC	PRIORITY	EXPOSURE SETTINGS (SEC)		
	h	m						30	90	270
176	18	01	+ 3.0	TBD	C	8	TBD	TBD	TBD	TBD
177	18	03	+21.0	TBD		9	TBD	TBD	TBD	TBD
178	18	10	-20.5	TBD	D	9	TBD	TBD	TBD	TBD
179	18	24	-34.0	TBD		6	TBD	TBD	TBD	TBD
180	18	25	-45.0	TBD	B	7	TBD	TBD	TBD	TBD
181	18	35	+32.5	TBD	D	7	TBD	TBD	TBD	TBD
182	18	38	-62.0	TBD	C	4	TBD	TBD	TBD	TBD
183	18	47	-26.5	TBD	C	3	TBD	TBD	TBD	TBD
184	18	52	+32.5	TBD	B	9	TBD	TBD	TBD	TBD
185	19	01	-21.0	TBD		11	TBD	TBD	TBD	TBD
186	19	14	+21.5	TBD		6	TBD	TBD	TBD	TBD
187	19	45	+46.0	TBD		7	TBD	TBD	TBD	TBD
188	19	48	+ 9.5	TBD		13	TBD	TBD	TBD	TBD
189	19	52	+39.0	TBD	D	9	TBD	TBD	TBD	TBD
190	19	52	+37.0	TBD	B	10	TBD	TBD	TBD	TBD
191	19	55	+31.5	TBD		6	TBD	TBD	TBD	TBD
192	20	14	+36.0	TBD	A	10	TBD	TBD	TBD	TBD
193	20	15	-13.5	TBD		8	TBD	TBD	TBD	TBD
194	20	20	+38.5	TBD	B	14	TBD	TBD	TBD	TBD
195	20	33	+12.5	TBD		10	TBD	TBD	TBD	TBD
196	20	39	+14.5	TBD		9	TBD	TBD	TBD	TBD
197	20	46	+45.0	TBD	A	11	TBD	TBD	TBD	TBD
198	21	00	+45.5	TBD	A	12	TBD	TBD	TBD	TBD
199	21	17	+38.0	TBD		9	TBD	TBD	TBD	TBD
200	21	18	+35.5	TBD		8	TBD	TBD	TBD	TBD
201	21	30	+61.5	TBD		6	TBD	TBD	TBD	TBD
202	21	37	+51.0	TBD	B	8	TBD	TBD	TBD	TBD
203	21	41	-16.0	TBD		5	TBD	TBD	TBD	TBD
204	21	48	+57.0	TBD		6	TBD	TBD	TBD	TBD
205	21	57	+63.0	TBD	B	14	TBD	TBD	TBD	TBD
206	22	02	- 1.5	TBD		4	TBD	TBD	TBD	TBD
207	22	03	+57.5	TBD	B	12	TBD	TBD	TBD	TBD
208	22	05	+51.5	TBD		4	TBD	TBD	TBD	TBD
209	22	05	-28.0	TBD		7	TBD	TBD	TBD	TBD
210	22	07	-13.0	TBD		4	TBD	TBD	TBD	TBD

- A. MORE THAN 3 O/B STARS FROM BSC AND MORE THAN 1 O/B STAR BRIGHTER THAN 5.1 VISUAL AND MORE THAN 9 STARS ALTOGETHER FROM BSC.
- B. MORE THAN 3 O/B STARS FROM BSC AND EITHER MORE THAN 1 O/B STAR BRIGHTER THAN 5.1 VISUAL OR MORE THAN 9 STARS ALTOGETHER FROM BSC.
- C. MORE THAN 1 O/B STAR BRIGHTER THAN 5.1 VISUAL.
- D. MORE THAN 4 O/B STARS FROM BSC.

S019 UV STELLAR ASTRONOMY
CANDIDATE STARFIELDS

STARFIELD ID NUMBER	RT ASCENSION-1950		DECLINATION-1950	GUIDE STAR	STARFIELD CRITERIA	NO. OF STARS IN BSC	PRIORITY	EXPOSURE SETTINGS (SEC)		
	h	m						30	90	270
211	22	22	+57.0	TBD		9	TBD	TBD	TBD	TBD
212	22	24	+47.5	TBD	C	5	TBD	TBD	TBD	TBD
213	22	26	- 0.5	TBD	C	5	TBD	TBD	TBD	TBD
214	22	27	+50.5	TBD		7	TBD	TBD	TBD	TBD
215	22	33	+38.5	TBD	D	9	TBD	TBD	TBD	TBD
216	22	55	+41.0	TBD	D	8	TBD	TBD	TBD	TBD
217	23	15	- 9.5	TBD		4	TBD	TBD	TBD	TBD
218	23	33	+43.0	TBD	B	4	TBD	TBD	TBD	TBD
219	23	57	+57.0	TBD		8	TBD	TBD	TBD	TBD
220	23	59	- 4.5	TBD		4	TBD	TBD	TBD	TBD

S020 X-RAY/UV SOLAR PHOTOGRAPHY
 FO-1 QUIET SUN
 FO-2 ACTIVE SUN

MRD 11/1/71
 EOH 12/1/71
 ERD 10/30/70
 ODB

ET	AM	12	3	V	*6	CREWMAN	*6	CREWMAN
0:00					10	//S020 EXP IN 08.04.01-00 // S020-X PREP S-SAL INSTALL SPECTROGRAPH ASSEMBLY IN THE S-SAL DEPRESSURIZE/LEAK TEST SAL REMOVE FMSC, OWS FILM VAULT VENT SAL & FMSC TO OWS LOAD FILM & FILTER DEPRESSURIZE SAL VVOH TO VENT		
0:20					07	//S020 EXP AL 08.04.02-00 // S020-X ALIGN S-SAL ANNOUNCE ALIGNMENT AT SUNRISE +5 MIN, OPEN SAL OUTER DOOR PERFORM OA ATTITUDE POINTING PROCEDURE RECORD OFFSET BIAS		
0:00					07	//S020 EXP AL 08.04.02-00 // S020-X ALIGN ATM C&D EXECUTE OA X & Y ATTITUDE MANEUVERS AS REQUESTED RECORD BIAS INPUTS φXI φYI φZI		
0:15					15	//S020 EXP OP 08.04.03-00 // S020-X S-SAL PERF # OF 10 EXPOSURE AT • EXPOSE FILM VOICE RECORD • START OF EXPOSURE, GMT • FRAME NUMBER • INTENDED DURATION		
0:00								

SCHEDULE CRITERIA

- The crew timeline will vary depending on how the S020 FO's or individual parts of FO1 and FO2 are sequenced.
- Either FO may be scheduled first and the individual parts of FO1 can be scheduled in any order with FO2 scheduled between any of the ten recommended FO1 exposures.
- For premission planning the best days during the mission for a quiet sun or an active sun are not known in advance nor can this be predicted with any degree of certainty. It is presently planned to schedule blocks of time for flare watch & reschedule during real time as flare predictions are received.
- Two series of exposures are required for FO1. The first series of 5 exposures is to be scheduled early in the mission & the second series of 5 exposures is to be scheduled late in the mission. The exposures can be conducted in any order.
- A minimum of 48 hours is to be assigned to flare watch activity (during an active sun period) with 8 hrs. minimum duration for each solar flare watch. This will require three to six work days on standby status for one crewman.
- Two film magazines are available for each mission. Each film magazine contains ten frames or strips of film. Assuming 3 exposures per solar flare, a total of 9 exposures would be expended on 3 flares and 10 exposures are required for FO1.

S020 X-RAY/UV SOLAR PHOTOGRAPHY
 FO-1 QUIET SUN
 FO-2 ACTIVE SUN

ET	1 AM	123 V	*6	*8
0:30 to 1:10				AT END EXPOSURE, RECORD GMT REPEAT FOR ALL EXPOSURES IN SCHEDULED PERFORMANCE AT SUNSET - 5 MIN, CLOSE SAL OUTER DOOR
0:00			04	// S020-X FILM CHANGE S-SAL CHANGE FILM CANISTER STOW EXPOSED FILM IN FMSC
0:08			10	//S020 EXP ST 08.04.04-00 //
0:00				S020-X STOW S-SAL PRESSURIZE SAL REMOVE SA FROM SAL REMOVE FILM & FILTER & STOW IN FMSC
0:20				DEPRESSURIZE FMSC, TBS MIN SECURE SAL STOW SA AND FMSC
0:00			03	//S020 FILM T //
0:05				S020 FILM TRANS F TRANSFER FMSC TO CSM

SCHEDULE CRITERIA (Continued)

- The exposures required for each FO1 photo series are 60, 30, 15, 8 & 5 min. The 60 min. exposure can be completed during two revs (consec. revs are required) if necessary. With high β angles, one rev. may be sufficient. The 15, 8 & 5 minute exposures can all be taken during one revolution. Each frame exposure will be considered a performance.

FRAME#	PERFORM	DURATION
1	Early	60 min.
2		30 "
3		15 "
4		8 "
5		5 "
6	Late	60 "
7		30 "
8		15 "
9		8 "
10		5 "

- During the flare watch period, the operator may be engaged in other activities but must be able to return and operate the SA within 1BD seconds after a flare alarm occurs.
- The solar SAL is required. Do not schedule concurrently with T027, T027/S073, T025 & S063-2.
- 0A attitude is near SI. Do not schedule concurrently with EREP. ATM recorder data required.

S020 X-RAY/UV SOLAR PHOTOGRAPHY
 FO-1 QUIET SUN PHOTOGRAPHY
 FO-2 ACTIVE SUN PHOTOGRAPHY

ET	AM	12	3	V	*6	CREWMAN	*6	CREWMAN
<p><u>SCHEDULE CRITERIA (Continued)</u></p> <ul style="list-style-type: none"> • All performances are to be conducted during daylight from 5 min. after sunrise to 5 min. before sunset. • An update pad is to be supplied by MCC-H one rev prior to conducting the experiment. • ATM experiments S054 and S082 A&B are to be coordinated & operated with S020-2 during a solar flare. <p><u>CONSTRAINTS</u></p> <ul style="list-style-type: none"> • Voice record vacuum valve position, film advance, offset bias, bias input & other pertinent comments. • The spectrograph principal axis must be within 7 deg. of the sun line for initial coarse pointing. • Experiment pointing accuracy to be within $\pm 0.25^\circ$ of the center of the solar disc for F01 or flare center for F02. • Do not operate concurrently with M509, T020 and T013. • SAL outer door is to be closed for each dark side pass. Do not perform scheduled waste dumps, venting or thruster operation with SAL outer door open. • Film to be stowed in depressurized FMSC when not in use. Minimize film exposure to the OWS atmosphere. Exposure to the oxygen atmosphere is limited to 30 minutes per setup. <p><u>RETURN WEIGHT</u></p> <ul style="list-style-type: none"> • FMSC <u>9.5</u> pounds 								

S020 X-RAY/UV SOLAR PHOTOGRAPHY
 FO-1 QUIET SUN PHOTOGRAPHY
 FO-2 ACTIVE SUN PHOTOGRAPHY

AM		CREWMAN		*S		CREWMAN	
ET	123V	*S		*S			
<p><u>NOTES (Cont.)</u></p> <ul style="list-style-type: none"> ● It is presently believed that an alignment of 1.75° away from SI is the maximum allowable to conduct ATM concurrently with S020. ● The number of exposures & duration of each exposure for each solar flare will be determined in real time. Typically, a solar flare exposure cycle will consist of a one minute exposure for the flare rise followed by several exposures of varied duration during the decay period. ● If S082 & S020 cannot both be pointed as required concurrently, it is expected that ATM will have priority during solar flares. ● Film temperature must be maintained between 35° & 75° F. ● The FMSC stowage evacuation time must be determined. ● If the second film magazine is not utilized in active sun photography because of a lack of solar flares, a decision will be made as to exposure of the film without flare activity. ● S020 is an alternate to T027 for M151 photographic requirements. 							

S063 UV AIRGLOW HORIZON PHOTOGRAPHY
FO-1 PHOTOGRAPH OZONE ATMOSPHERE AT UV WAVELENGTHS
AND EARTH ATMOSPHERE AT VISIBLE WAVELENGTHS

MRD 11/1/71
EOH 1/3/72
ERD 9/70
ODB

ET	AM	123	V	#6	CREWMAN	#6	CREWMAN	SCHEDULE CRITERIA																												
0:00				03	// S063 PRE-PRP 04.03.01-00// S063-1 PRE PREP REMOVE LAUNCH CONSTRAINTS FROM EQUIPMENT			<p>SCHEDULE CRITERIA</p> <ul style="list-style-type: none"> ● ATM recorder data required. ● Concurrent use of anti-solar airlock for UV photography & wardrobe window for visible photography of earths' ozone atmosphere. ● Do not schedule concurrently with S149, S019, T027/S073 and S183. ● A number of passes to obtain 150 UV camera exposures in Z-LV attitude is required. The visible camera will be operated by remote control at the same time. ● Concurrent operation with EREP is desired if crew & SAL time is available. ● Schedule UV photographs as follows: <table border="1"> <thead> <tr> <th>NO.</th> <th>SUN</th> <th>DURATION</th> <th>FILTER</th> </tr> </thead> <tbody> <tr> <td>15</td> <td>60°-40°</td> <td>20S</td> <td>2800 A</td> </tr> <tr> <td>15</td> <td>60°-40°</td> <td>20S</td> <td>3200 A</td> </tr> <tr> <td>30</td> <td>40°-15°</td> <td>8S</td> <td>2800 A</td> </tr> <tr> <td>30</td> <td>40°-15°</td> <td>8S</td> <td>3200 A</td> </tr> <tr> <td>30</td> <td>15°-0°</td> <td>2S</td> <td>2800 A</td> </tr> <tr> <td>30</td> <td>15°-0°</td> <td>2S</td> <td>3200 A</td> </tr> </tbody> </table> <ul style="list-style-type: none"> ● Do not schedule manual overboard vents or RCS operation during exp. performance. ● Do not schedule operation of T013, T020 or M509 when S063 equipment is installed in the anti solar sal. ● Schedule flight crew operations so that no crewman is exposed to UV radiation from the Sal windows for more than 27 minutes in any 24 hr. period. ● Schedule a tape dump after each photo session. <p>CONSTRAINTS</p> <ul style="list-style-type: none"> ● The optical sighting device must be kept pointed to within 0.5° on a chosen tracking feature. 	NO.	SUN	DURATION	FILTER	15	60°-40°	20S	2800 A	15	60°-40°	20S	3200 A	30	40°-15°	8S	2800 A	30	40°-15°	8S	3200 A	30	15°-0°	2S	2800 A	30	15°-0°	2S	3200 A
NO.	SUN	DURATION	FILTER																																	
15	60°-40°	20S	2800 A																																	
15	60°-40°	20S	3200 A																																	
30	40°-15°	8S	2800 A																																	
30	40°-15°	8S	3200 A																																	
30	15°-0°	2S	2800 A																																	
30	15°-0°	2S	3200 A																																	
0:05				07	// S063 PREP 0 08.05.01-00 // S063-1 PREP INSTALL VISIBLE CAMERA MOUNT IN WARDRM WINDOW INSTALL EA I WINDOW & UV MOUNT IN A-SOL-SAL ATTACH OPTICAL SIGHT DEPRESSURIZE SAL		S063-1 UPDATE PAD																													
0:14				05	// S063 FILM 0 08.05.02-00 // S063-1 FILM INSTALL F,W LOAD COLOR & UV FILM IN CAMERAS INSTALL VISIBLE & UV CAMERAS IN WARDROOM & A-SOL-SAL																															
0:10				07	// S063 OZONE 08.05.03-00 // S063-1 PERF # OF SUBDUE LIGHTING AT TRACK TARGET, EXPOSE FILM REPEAT FOR AT OWS LIGHTS ON	03	// S063-1 Z-LV 18.09.06-00 // S063-1 Z-LV ATM C&D MNVR TO Z-LV ATTITUDE AT																													
0:00				03	// S063-1 S1 18.09.02-00 // S063-1 S1 ATM C&D MNVR TO S1 ATTITUDE AT																															
0:15					CONTINUED																															

S063 UV AIRGLOW HORIZON PHOTOGRAPHY
 FO-1 PHOTOGRAPH OZONE ATMOSPHERE AT UV WAVELENGTHS
 AND EARTH ATMOSPHERE AT VISIBLE WAVELENGTHS

ET	AM	12	3	V	*6	CREWMAN	*8	CREWMAN									
0:00					05	// S063 STOW 0 08.05.04-00 // S063-1 STOW FILM F,W REMOVE CAMERAS FROM A-SOL SAL & WARDROOM WINDOW & STOW IN OWS FILM VAULT		<p><u>CONSTRAINTS CONT.</u></p> <ul style="list-style-type: none"> ● Maintain IMC via tracking during the exposure. ● Maximum pitch, yaw or roll rates during the exposure period will not exceed 0.1°/sec. ● Thermal limits on cameras & film are as follows: <table border="0"> <tr> <td>ITEM</td> <td>STORED</td> <td>OPERATED</td> </tr> <tr> <td>Exp. Assy.</td> <td>0-100°F</td> <td>65-90°F</td> </tr> <tr> <td>Film</td> <td>85°F</td> <td>65-90°F</td> </tr> </table> <ul style="list-style-type: none"> ● Constraint on film negative density (increase not to exceed 0.3 density units above baseline fot) cannot be monitored by crew. Crew to minimize film time outside the OWS film vault. ● If experiment operation times are separated by more than 4 hours, the film and camera will be stowed in the OWS film vault. ● Logging requirements shall include orbit number, location of experiments, beginning film exposure number, last exposure number and equip. condition. ● Voice record camera ID, filter code, series number, camera orientation, exposure initiation in GMT to within 1.0 sec., exposure number, and exposure duration. <p><u>RETURN WEIGHT</u></p> <ul style="list-style-type: none"> ● Film 5.7 pounds total for both FO's. <p><u>NOTES</u></p> <ul style="list-style-type: none"> ● Film/filters/batteries will be carried on SL 3 & 4. 	ITEM	STORED	OPERATED	Exp. Assy.	0-100°F	65-90°F	Film	85°F	65-90°F
ITEM	STORED	OPERATED															
Exp. Assy.	0-100°F	65-90°F															
Film	85°F	65-90°F															
0:10					06	// S063 EA1 EQ 08.05.05-00 // S063-1 STOW EQUIP F,W REPRESSURIZE SAL REMOVE EA1 WINDOW & CAMERA MOUNTS, STOW											
0:00																	
0:11																	

S063 UV AIRGLOW HORIZON PHOTOGRAPHY
 FO-1 PHOTOGRAPH OZONE ATMOSPHERE AT UV WAVELENGTHS
 AND EARTH ATMOSPHERE AT VISIBLE WAVELENGTHS

AM	123	V	*6
ET			*6
<p>NOTES CONT.</p> <ul style="list-style-type: none"> • The color exposures taken from the ward-room window will have a standard exposure duration of 1/250 second. • The experiment is battery powered and requires no OWS interface. • If the camera has unexposed film between data takes, the entire camera will be stowed in the OWS vault rather than just the film. • The visible camera will use a single haze cutting filter. 			

S063 UV AIRGLOW HORIZON PHOTOGRAPHY
 FO-2 PHOTOGRAPH EARTHS TWILIGHT AIRGLOW
 AT VISIBLE & UV WAVELENGTHS

ET	AM	V	*6	CREWMAN	*6	CREWMAN
0:00			04	// S063 EA II E 08.05.09-00// S063-2 STOW EQUIP SOL-SAL REMOVE EA II CAMERA MOUNT AND WINDOW FROM THE SOL-SAL		
0:08						

SCHEDULE CRITERIA CONT.

- Schedule twilight airglow photography when the spacecraft is located at solar depression angles between 26.5 deg/ and 37.5 deg.
- Schedule the photography sessions at β angles between 35 and 62 degrees.
- Exposures with any of the four filters can be made whenever the portion of the twilight horizon that is being photographed is not in either the northern or southern auroral zone.
- The photo sequences may be scheduled during CMG dump periods.
- Do not schedule manual overboard vents or RCS operation during exp. performance.
- Do not schedule operation of T013, T020, of M509 when S063 equipment is installed in the solar SAL.
- Schedule flight crew operations so that no crewman is exposed to UV radiation from the SAL windows for more than 27 minutes in any 24 hr. period.
- Schedule a tape dump after each photo session.

CONSTRAINTS

- The camera must be kept pointed at a chosen airglow layer within 0.5°.
- Maintain IMC via tracking during the exposure.
- Maximum pitch, yaw or roll rates during the exposure period will not exceed 0.1°/sec.

S063 UV AIRGLOW HORIZON PHOTOGRAPHY
 FO-2 PHOTOGRAPH EARTHS TWILIGHT AIRGLOW
 AT VISIBLE & UV WAVELENGTHS

AM		CREWMAN		*S	CREWMAN									
ET	123V *S													
<p>CONSTRAINTS CONT.</p> <ul style="list-style-type: none"> Thermal limits on cameras and film: <table border="0" style="margin-left: 20px;"> <tr> <td style="padding-right: 20px;">ITEM</td> <td>STORED</td> <td>OPERATED</td> </tr> <tr> <td>Exp. Assy.</td> <td>0-100°F</td> <td>65-90°F</td> </tr> <tr> <td>Film</td> <td>85° F Max.</td> <td>65-90°F</td> </tr> </table> Constraint on film negative density cannot be monitored by crew. Crew to minimize unstowed film time. If experiment operation times are separated by more than 4 hours, the film and camera will be stowed in the OWS film vault. Logging requirements shall include orbit number, location of experiment, beginning film exposure number, last exposure number and equip. condition. Voice record camera ID, filter code, series number, camera orientation, exposure initiation in GMT to within 1.0 sec., exposure number and exposure duration. The internal lights that may cause light to enter the camera and all external lights will be extinguished. <p>RETURN WEIGHT</p> <ul style="list-style-type: none"> Film - See FO-1. <p>NOTES</p> <ul style="list-style-type: none"> Film/filters/batteries will be carried on SL 3 & 4. The experiment is battery powered and requires no OWS power interface. If the camera has unexposed film between data takes, the entire camera will be stowed in the OWS vault rather than just the film. 						ITEM	STORED	OPERATED	Exp. Assy.	0-100°F	65-90°F	Film	85° F Max.	65-90°F
ITEM	STORED	OPERATED												
Exp. Assy.	0-100°F	65-90°F												
Film	85° F Max.	65-90°F												

S063 UV AIRGLOW HORIZON PHOTOGRAPHY
 FO-2 PHOTOGRAPH EARTHS TWILIGHT AIRGLOW
 AT VISIBLE & UV WAVELENGTHS

ET	AM	1	2	3	V	*6	*6	NOTES CONT.
								<ul style="list-style-type: none"> • The airglow, together with the earth limb & background starfield will be photographed from the OWS night position in the solar direction. • The exposure sequence at evening twilight will generally be 30, 60 & 120 sec. while the sequence at spacecraft dawn will be reversed, i.e., 120, 60 & 30 sec. • With working times greater than 5 consecutive minutes, it is desirable to obtain a series of alternate 6300A & 5577A exposures with exposure times increasing from twilight & decreasing toward dawn. • With working times less than 5 consecutive minutes, only the red or green filters will be used. • Alternating the UV filters in one orbit is not necessary.

ET	AM	12	13	V	*6	CREWMAN	*6	CREWMAN	MRD	2/25/72
0:00					30	<p>FO-1 SYSTEM MONITOR</p> <p>FO-2 CONTAMINATION</p> <p>FO-3 ECLIPTIC</p> <p>FO-4 VERTICAL CIRCLE</p> <p>FO-5 ALL SKY MAP</p> <p>FO-6 CELESTIAL POLES</p> <p>FO-7 GEGENSCHIEIN</p> <p>FO-8 1 TO ECLIPTIC</p> <p>FO-9 ZODIACAL LIGHT/CONTAMINATION</p> <p>FO-10 ECLIPTIC POLES</p>		CREWMAN	ERD	1/3/72
						<p>//S072 PREP P 11.08.03-00 //</p> <p>S073-X PREP</p> <p>REMOVE PHOTOMETER & FILM FROM OWS STOWAGE</p> <p>DEPLOY PHOTOMETER HEAD, LOAD FILM & RETRACT HEAD</p> <p>REMOVE TRIPOD FROM STOWAGE</p> <p>INSTALL PHOTOMETER SYSTEM INTO THE SOL-SAL</p> <p>ATTACH TRIPOD SUPPORT</p> <p>DEPRESSURIZE SAL, LEAK TEST OPEN SAL OUTER DOOR</p> <p>INSTALL POWER & INST CABLES</p> <p>PERFORM PRE-POWER CHECKS POWER-ON (5 MIN WARMUP)</p> <p>ADVANCE FILM TO FRAMES</p> <p>PRESET PANEL</p>				
1:00					05	<p>//S073 DEPLOY 11.08.04-00 //</p> <p>S073-X DEPLOY</p> <p>S-SAL</p>				
0:00						<p>S073-X PREP</p> <p>REMOVE M151 DAC FROM SET UP DAC AT LOCATION E ASSIST WITH PHOTOMETER SETUP</p> <p>PHOTOGRAPH SETUP, M151-2</p>				
						<p>S073 UPDATE PAD</p>				
						<p>Refer to Tables on pages 4-32M & 4-32N for individual FO requirements & test conditions.</p> <p><u>SCHEDULE CRITERIA</u></p> <ul style="list-style-type: none"> Do not schedule solar SAL scans concurrently with T027, T025, S020, S063-2, S183 & S149. Do not schedule anti solar SAL scans concurrently with S149, S019, S063-1, S183 & T027. Do not schedule any performance concurrently with T020, M509, T013 & M092/M093. Do not schedule concurrently with maneuvers into and out of Z-LV(E) or while in Z-LV(E). Do not schedule major OA maneuvers concurrent with experiment data collection. Refer to chart for momentum dump inhibit requirements. Schedule at least 5 min. warmup from power on to initial data collection. When operating out of the solar SAL, extend & retract the photometer system while the OA is in earth shadow. The crewmen will voice record & time correlate comments relevant to experiment operation. Schedule ATM recorder for pointing data during data scans. Knowledge of the pointing direction of the photometer will be required to within 1/2 deg. Programs 0-a & 1-a can be scheduled prior to determining SAL alignment, however, SAL alignment must be performed prior to scheduling the remaining data scans. 				

S073/T027 GEGENSCHNEIN/ZODIACAL LIGHT & CONTAMINATION MEASUREMENT
 FO-1 SYSTEM MONITOR FO-4 VERTICAL CIRCLE FO-7 GEGENSCHNEIN
 FO-2 CONTAMINATION FO-5 ALL SKY MAP FO-8 1 TO ECLIPTIC
 FO-3 ECLIPTIC FO-6 CELESTIAL POLES FO-9 ZODIACAL LIGHT/CONTAMINATION
 FO-10 ECLIPTIC POLES

ET	123 V *6	CREWMAN	*6	CREWMAN
0:10		EXTEND THE PHOTOMETER ROD LENGTHS		CREWMAN
0:00	05	//S073 OA ASA 11.08.06-01 // S073-1 A-SOL-SAL PERF #1 OF 1, PROGRAM 0-a SYSTEM MONITOR START RECORDER & PROGRAM AT _____ MONITOR FOR ___ MIN (94 MIN. OPERATION)	START ANYTIME IN ORBIT	
1:44 1:46	02	S073-1 RECORDERS OFF M,E		
0:00	05	//S073 OA SSA 11.08.06-02 // S073-1 SOL-SAL PERF # OF 3, PROGRAM 0-a SYSTEM MONITOR START RECORDER & PROGRAM AT _____ MONITOR FOR ___ MIN (94 MIN. OPERATION)	START ANYTIME IN ORBIT	
0:10				
1:44 1:46	02	S073-1 RECORDERS OFF M,E		
0:00	05	//S073-1A IAS 11.08.06-03 // S073-2 A-SOL-SAL PERF # OF 3, PROGRAM 1-a CONTAMINATION START RECORDER & PROGRAM	//S073 MOMEN I S073-2 ATM C&D INHIBIT MOMENTUM DUMP	

SCHEDULE CRITERIA (Cont.)

- An update pad is required at least one rev prior to conducting a data scan.
- Calibrate the photometer, Mode 0, before or after each Mode 1 sequence. Calibration is conducted automatically as a part of modes 2 thru 5.
- M151 photography is required only one time each during setup & removal from each SAL.

CONSTRAINTS

- Power is to be left on when the photometer is deployed out the anti solar SAL.
- The Boom is to be retracted when scan sequences are separated by more than TBD hours.
- The photometer shall not be pointed within 15 degrees of the moon or the sun.
- Max allowable scan rate error in shaft or trun during data collection is + 0.025 deg/sec.
- S1 or any other inertial mode with random rates below 0.05 deg/sec is preferred.
- Constant rates up to 0.1 deg/sec causing secular changes in orientation may be acceptable depending on scan mode & direction of vehicle movement.
- Angular accelerations which result in rates less than 0.05 deg/sec are acceptable.

S073/T027 GEGENSCHNEIN/ZODIACAL LIGHT & CONTAMINATION MEASUREMENT		GEGENSCHNEIN	
FO-1 SYSTEM MONITOR		FO-7	
FO-2 CONTAMINATION		FO-8 1 TO ECLIPTIC	
FO-3 ECLIPTIC		FO-9 ZODIACAL LIGHT/CONTAMINATION	
1 AM		FO-10 ECLIPTIC POLES	
ET	1 2 3 V #6	CREWMAN	#8 CREWMAN
0:10		AT MONITOR FOR ___ MIN (20 MIN. OPERATION) (WAIT 50 MINUTES)	START WHILE IN EARTHS SHADOW
1:00		ROTATE SHAFT 180° RESTART PHOTOMETER AT MONITOR FOR ___ MIN (20 MIN. OPERATION)	START 15 MIN. BEFORE OWS NIGHT
1:25	02	S073-2 RECORDERS OFF M,E	//S073 MOMEN E
1:27			S073-2 ATM C&D ENABLE MOMENTUM DUMP
0:00	05 //S073 IA ISS 11.08.06-04 //	SOL-SAL PERF # OF 3, PROGRAM 1-a CONTAMINATION START RECORDER & PROGRAM AT MONITOR FOR ___ MIN (20 MIN. OPERATION) (WAIT 50 MINUTES)	//S073 MOMEN I // S073-2 ATM C&D INHIBIT MOMENTUM DUMP
0:10		ROTATE SHAFT 180° RESTART PHOTOMETER AT MONITOR FOR ___ MIN (20 MIN. OPERATION) (WAIT 50 MINUTES)	START WHILE IN EARTHS SHADOW
1:00		ROTATE SHAFT 180° RESTART PHOTOMETER AT MONITOR FOR ___ MIN (20 MIN. OPERATION) (WAIT 50 MINUTES)	START 15 MIN. BEFORE OWS NIGHT
1:10			

CONSTRAINTS (Cont.)

- Total radiation dosage for each film magazine must not exceed 2 rads.
- Film magazines must not exceed 80°F.
- Extinguish lighting & cover windows that interferes with data collection.
- The photometer temp. should be above the dew point temp. during retrieval procedures.

RETURN WEIGHT

- Film 3.0 pounds.

NOTES

- Unless otherwise noted, after photometer setup and deployment, the crewman need only be present to set the automatic programmer, start the scan and verify proper initial operation.
- Only the crew can turn the AM experiment recorders on and off. The ground can dump the AM experiment recorders over any station. For multiple orbit scans, the recorders will be required to run during the non-data taking periods unless crew time can be scheduled during each orbit for recorder operation.
- The anti-solar SAL will employ a desiccant air filter to dry incoming SAL air.
- There are 18 different operating procedures. The example shows Mode 1-a only. Schedule remaining modes per the attached chart.

S073/T027 GEGENSCHNEID/ZODIACAL LIGHT & CONTAMINATION MEASUREMENT
 FO-1 SYSTEM MONITOR FO-4 VERTICAL CIRCLE FO-7 GEGENSCHNEID
 FO-2 CONTAMINATION FO-5 ALL SKY MAP FO-8 1 TO ECLIPTIC
 FO-3 ECLIPTIC FO-6 CELESTIAL POLES FO-9 ZODIACAL LIGHT/CONTAMINATION
 FO-10 ECLIPTIC POLES

ET	123V *6	CREWMAN	*6	CREWMAN
1:10		MONITOR FOR ___ MIN		
1:30		(20 MIN. OPERATION)		
1:32	02	S073-2 RECORDERS OFF M,E	02	//S073 MOMEN E //
0:00	05	//S073 1B EXP 11.08.06-05 //	02	S073-2 ATM C&D ENABLE MOMENTUM DUMP //S073 MOMEN I //
0:10		S073-7 A-SOL-SAL PERF # OF 3, PROGRAM 1-b GEGENSCHNEID START RECORDERS & PROGRAM AT MONITOR FOR ___ MIN.		S073-7 ATM C&D INHIBIT MOMENTUM DUMP START 24 MIN BEFORE ANTI-SOL POINT
1:00	02	(48 MIN. OPERATION)	02	//S073 MOMEN E //
0:00	05	S073-7 RECORDERS OFF M,E //S073 1C EXP 11.08.06-06 //	03	S073-7 ATM C&D ENABLE MOMENTUM DUMP //S073 1C EXP 11.08.06-06 //
0:10		S073-2 A-SOL-SAL PERF # 1 OF 1, PROGRAM 1-c CONTAMINATION START RECORDER & PROGRAM AT MONITOR FOR ___ MIN		S073-2 ATM C&D INITIATE VENT 10 MIN BEFORE OWS DAYLIGHT START 5 MIN. BEFORE OWS DAYLIGHT
		(20 MIN. OPERATION)		

S073/T027 GEGENSCHNEIN/ZODIACAL LIGHT & CONTAMINATION MEASUREMENT
 FO-1 SYSTEM MONITOR FO-4 VERTICAL CIRCLE FO-7 GEGENSCHNEIN
 FO-2 CONTAMINATION FO-5 ALL SKY MAP FO-8 1 TO ECLIPTIC
 FO-3 ECLIPTIC FO-6 CELESTIAL POLES FO-9 ZODIACAL LIGHT/CONTAMINATION
 FO-10 ECLIPTIC POLES

ET	AM	12	3	V	*6	CREWMAN	*8	CREWMAN
0:10						PERF # OF 3, PROGRAM 2-a ECLIPTIC START RECORDER & PROGRAM AT MONITOR FOR ___ MIN (19 MIN. OPERATION)	02	INHIBIT MOMENTUM DUMP START UPON LEAVING EARTH'S SHADOW //S073 MOMEN E //
0:29					02	S073-3 RECORDERS OFF M,E	02	S073-3 ATM C&D ENABLE MOMENTUM DUMP
0:31					05	//S073 2B EXP 11.08.06-11 //	02	//S073 MOMEN I //
0:00						S073-3 SOL-SAL PERF # OF 3, PROGRAM 2-b ECLIPTIC START RECORDERS & PROGRAM AT MONITOR FOR ___ MIN (19 MIN. OPERATION) (WAIT 40 MINUTES)	02	S073-3 ATM C&D INHIBIT MOMENTUM DUMP START 10 MIN. BEFORE ANTI-SOLAR POINT
0:10						RESTART PHOTOMETER AT MONITOR FOR ___ MIN (19 MIN. OPERATION)	02	//S073 MOMEN E //
0:50							02	S073-3 ATM C&D ENABLE MOMENTUM DUMP
0:55					02	S073-3 RECORDERS OFF M,E	02	//S073 MOMEN I //
1:14					05	//S073 2C EXP 11.08.06-12 //	02	S073-3 ATM C&D ENABLE MOMENTUM DUMP
1:16							02	//S073 MOMEN I //
0:00							02	S073-3 ATM C&D ENABLE MOMENTUM DUMP

S073/T027 GEGENSCHNEIN/ZODIACAL LIGHT & CONTAMINATION MEASUREMENT		S073/T027 GEGENSCHNEIN ZODIACAL LIGHT/CONTAMINATION	
FO-1 SYSTEM MONITOR		FO-7 GEGENSCHNEIN	
FO-2 CONTAMINATION		FO-8 1 TO ECLIPTIC	
FO-3 ECLIPTIC		FO-9 ZODIACAL LIGHT/CONTAMINATION	
AM		FO-10 ECLIPTIC POLES	
ET	123V *6	CREWMAN	*6 CREWMAN
0:10		S073-4 A-SOL-SAL PERF # OF 2, PROGRAM 2-c VERTICAL CIRCLE START RECORDERS & PROGRAM AT MONITOR FOR ___ MIN (30 MIN. OPERATION)	S073-4 INHIBIT MOMENTUM DUMP ATM C&D START 15 MIN. BEFORE TERMINATOR
0:40		ROTATE SHAFT ___ DEGREES RESTART PHOTOMETER AT	START NEAR ANTI SOLAR POINT
0:45		MONITOR FOR ___ MIN (30 MIN. OPERATION)	//S073 MOMEN E //
1:15 1:17	02	S073-4 RECORDERS OFF M,E	S073-4 ENABLE MOMENTUM DUMP ATM C&D
0:00	05	//S073 2D EXP 11.08.06-13 //02	//S073 MOMEN I //
0:10		S073-4 SOL-SAL PERF # OF 2, PROGRAM 2-d VERTICAL CIRCLE START RECORDERS & PROGRAM AT MONITOR FOR ___ MIN (30 MIN. OPERATION)	S073-4 INHIBIT MOMENTUM DUMP ATM C&D START 15 MIN. BEFORE TERMINATOR
0:40		ROTATE SHAFT ___ DEGREES RESTART PHOTOMETER AT	START NEAR ANTI SOLAR POINT
0:45			

073/T027 GEGENSCHNEIN/ZODIACAL LIGHT & CONTAMINATION MEASUREMENT
 FO-1 SYSTEM MONITOR FO-4 VERTICAL CIRCLE FO-7 GEGENSCHNEIN
 FO-2 CONTAMINATION FO-5 ALL SKY MAP FO-8 1 TO ECLIPTIC
 FO-3 ECLIPTIC FO-6 CELESTIAL POLES FO-9 ZODIACAL LIGHT/CONTAMINATION
 FO-10 ECLIPTIC POLES

ET	123V *6	CREWMAN	*6	CREWMAN
1:01				
1:16				
1:18				
0:00				
0:10				
0:27				
0:42				
0:44				

SET TRUNNION TO 30°
 MONITOR FOR 15 MINUTES

RECORDERS OFF

05//S073 3C EXP 11.08.06-16 //

S073-7 OPERATE A-SOL-SAL
 PERF # OF 2, PROGRAM 3-c
 GEGENSCHNEIN
 START RECORDERS & PROGRAM
 AT

MONITOR FOR 15 MINUTES

SET TRUNNION TO 4.2°
 MONITOR FOR 15 MINUTES

RECORDERS OFF

.....
 START 15 MIN. BEFORE
 ANTI SOLAR POINT

S073/T027 GEGENSCHNEIN/ZODIACAL LIGHT & CONTAMINATION MEASUREMENT
 FO-1 SYSTEM MONITOR FO-4 VERTICAL CIRCLE FO-7 GEGENSCHNEIN
 FO-2 CONTAMINATION FO-5 ALL SKY MAP FO-8 1 TO ECLIPTIC
 FO-3 ECLIPTIC FO-6 CELESTIAL POLES FO-9 ZODIACAL LIGHT/CONTAMINATION
 FO-10 ECLIPTIC POLES

ET	AM	12	3	V	#6	CREWMAN	#6	CREWMAN
15:30						MONITOR FOR ___ MIN		
15:32						(WAIT 920 MINUTES)	02	//S073 MOMEN E //
0:00					02	S073-9 RECORDERS OFF M,E		S073-9 ENABLE MOMENTUM DUMP ATM C&D
0:10					05	//S073 5A EXP 11.08.06-20 //		
15:30						S073-5 PERF #1 OF 1, PROGRAM 5-a ALL SKY MAP START RECORDERS & PROGRAM AT		
15:32						MONITOR FOR ___ MIN		
0:00						(WAIT 920 MINUTES)		
0:10					02	S073-5 RECORDERS OFF M,E		
15:30					05	//S073 5B EXP 11.08.06-21 //		
15:32						S073-5 PERF # 1 OF 1, PROGRAM 5-b ALL SKY MAP START RECORDERS AND PROGRAM AT		
0:00						MONITOR FOR ___ MIN		
0:10						(WAIT 920 MINUTES)		
15:30					02	S073-5 RECORDERS OFF M,S		
15:32								

START 16 MIN. PRIOR
 TO ANTI SOLAR POINT

START 15 MIN. PRIOR
 TO SOLAR POINT

S073/T027 GEGENSCHNEIN/ZODIACAL LIGHT & CONTAMINATION MEASUREMENT		CREWMAN		CREWMAN	
ET	123V	*6	*6	*6	*6
0:00		05	//S073 RETRAC 11.08.05-00 //		
			S073-X RETRACT S-SAL		
			RETRACT PHOTOMETER		
			DON GLOVES, RETRACT		
			PHOTOMETER INTO SAL		
			CLOSE OUTER SAL DOOR		
0:10			30//S073 STOW P 11.08.07-00 //B0		
			S073-X STOW S-SAL		
			STOW PHOTOMETER		
			DON GLOVES, RETRACT		
			PHOTOMETER INTO SAL		
			CLOSE SAL OUTER DOOR		
			PRESSURIZE SAL, GN2		
			REMOVE POWER & INST. CABLES		
			REMOVE TRIPOD SUPPORT		
			REMOVE PHOTOMETER FROM		
			SOLAR SAL, ATTACH END		
			COVER		
			ALLOW PHOTOMETER TO		
			WARM ABOVE DEW PT TEMP		
			REMOVE END COVER		
			DEPLOY PHOTOMETER HEAD,		
			REMOVE FILM, RETRACT HEAD		
1:00			STOW FILM IN OWS FILM		
			VAULT		
			STOW PHOTOMETER SYSTEM		
			//S073 TRANS //		
0:00			S073 FILM TRANSFER		
			TRANSFER FILM TO CSM		
0:05					

SL-2 MINIMUM PERFORMANCE REQUIREMENTS
T027/S073

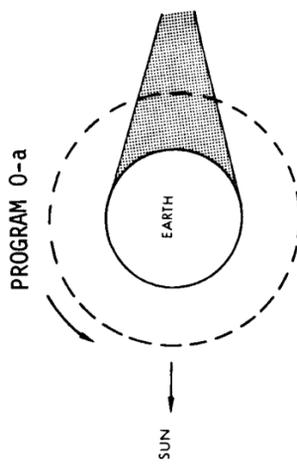
S A L	PRO- GRAM	FO	PROGRAM TITLE	IMD	R O D	SCHEDULING CRITERIA/CONSTRAINTS
S O L	1-a 0-a 2-b 5-b	2 1 3 5	CONTAMINATION SYSTEM MONITOR ECLIPTIC ALL SKY MAP	X X	7 7 7 7	<ul style="list-style-type: none"> ● PERFORM EARLY IN MISSION. TARGET FOR MISSION DAYS 2 THRU 5. ● PROGRAM 1-a TO BE FOLLOWED BY PROGRAM 0-a. PERFORM ON CONSECUTIVE ORBITS AND PRIOR TO ATM, IF POSSIBLE. ● PROGRAM 2-a ANTI SOL-SAL TO FOLLOW PROGRAM 2-b WITHIN 2 DAYS MAXIMUM ● PROGRAMS 1-a, 0-a AND 2-b ARE MANDATORY. ● PROGRAM 5-b IS HIGHLY DESIRABLE.
A	1-a 0-a	2 1	CONTAMINATION SYSTEM MONITOR	X	7/2 2	<ul style="list-style-type: none"> ● PERFORM EARLY IN MISSION. TARGET FOR MISSION DAYS 6 THRU 10. ● PROGRAM 1-a TO BE FOLLOWED BY PROGRAM 0-a. PERFORM ON CONSECUTIVE ORBITS IF POSSIBLE.
S O L	2-a 2-c 1-d	3 4 10	ECLIPTIC VERTICAL CIRCLE ECLIPTIC POLE N	X X X	2 2 2	<ul style="list-style-type: none"> ● SCHEDULE PROGRAM 2-a WITHIN 2 DAYS MAXIMUM OF PROGRAM 2-b ABOVE. ● PROGRAMS 1-a, 0-a, 2-a AND 1-d ARE MANDATORY. ● PROGRAM 2-c IS HIGHLY DESIRABLE.
S O L	0-a 2-d 3-b	1 4 2	SYSTEM MONITOR VERTICAL CIRCLE CONTAMINATION *	X	7 7 7	<ul style="list-style-type: none"> ● PERFORM DURING MID-MISSION. TARGET FOR MISSION DAYS 11 THRU 14. ● PERFORM PROGRAM 3-b FIRST AND AT LEAST 5 DAYS FROM THE PERFORMANCE OF PROGRAM 4-b. ● PROGRAMS 0-a, 2-d and 3-b ARE MANDATORY.
A S O L	1-b 1-e 1-e 4-a	7 6 6 7	GEGENSCHEIN CELESTIAL POLE N. CELESTIAL POLE S. GEGENSCHEIN	X	2 2 2 2	<ul style="list-style-type: none"> ● PERFORM DURING MID-MISSION. TARGET FOR MISSION DAYS 15 THRU 18 ● SCHEDULE PROGRAM 1-b AND 1-e (N) WITHIN + ONE WEEK OF THE NEW MOON. ● COORDINATE PROGRAM 1-b OBSERVATIONS WITH GROUND OBSERVATIONS AT HAWAII. ● SCHEDULE N & S POLE OBSERVATION WITHIN TWO DAYS OF EACH OTHER. ● PROGRAMS 1-b, 1-e (N) AND 4-a ARE MANDATORY. ● PROGRAM 1-e (S) IS HIGHLY DESIRABLE.
S O L	1-a 0-a 4-b	2 1 9	CONTAMINATION SYSTEM MONITOR ZODIACAL LIGHT *	X X	2/7 7 7	<ul style="list-style-type: none"> ● PERFORM LATE IN MISSION. TARGET FOR MISSION DAYS 19 THRU 22. ● PERFORM PROGRAM 4-b AT LEAST 5 DAYS FROM PERFORMANCE OF PROGRAM 3-b. ● PROGRAMS 1-a AND 4-b ARE MANDATORY. ● PROGRAM 0-a IS HIGHLY DESIRABLE.
A S O L	3-a 5-a 3-d 1-a	8 5 2 2	PERP. TO ECLIPTIC * ALL SKY MAP CONTAMINATION CONTAMINATION	X	2 2 7/2 7	<ul style="list-style-type: none"> ● PERFORM LATE IN MISSION. TARGET FOR MISSION DAYS 23 THRU 27. ● PROGRAMS 3-d, 3-a AND 5-a ARE MANDATORY. ● PROGRAM 1-a IS HIGHLY DESIRABLE.

NOTE: THE PROGRAMS DESIGNATED WITH AN ASTERISK CAN BE SCHEDULED AT ANY TIME IN THE MISSION.

Functional Objective Summary Table

FO	PROGRAM	NUMBER OF PERFORMANCES	SAL	NUMBER OF RODS	TIME PER PERFORMANCE	TOTAL OBSERVING TIME	CREW REQUIREMENTS (1)	ENG DUMP INHIBIT	HAWAII COORDINATION	NEW MOON +1 WEEK	REMARKS (2)
1	0a	3	S	7	94	282					Highly desirable for 1 solar SAL performance prior to ATM.
2	1a	1	A	2	94	94					
		3	S	7, 2	40	120					Highly desirable for 1 solar SAL performance prior to ATM.
		3	A	7, 2	40	120		Yes			
3	1c	0	S	-	-	-					To be initiated prior to a TBD overboard venting.
		1	A	2	20	120					Coordinate with Program 4b of FO 9.
4	3b	2	S	7	68	136	Change trumtion angles				
		0	A	-	-	-	Change trumtion angles				
		0	S	-	-	-	Monitor intensity				
		1	A	7, 2	68	68	Change gain and rods				
5	2a	0	S	-	-	-		Yes			
		3	A	2	19	57					
		3	S	7	38	114					
6	2b	0	A	-	-	-		Yes			
		0	S	-	-	-					
7	2c	2	A	2	60	120	Change shaft values	Yes			30 minute dark side passes are highly desirable.
		2	S	7	60	120					
		0	A	-	-	-	Change shaft values	Yes			30 minute dark side passes are highly desirable.
8	5a	0	S	-	-	-					Perform during first photometer deployment.
		1	A	2	320	320					Requires use of orbital counter.
9	5b	1	S	7	290	290					Perform during first photometer deployment following SAL alignment.
		0	A	-	-	-					Requires use of orbital counter.
10	1e	0	S	-	-	-					North Celestial Pole Observations have a higher priority than the South Celestial Pole.
		6	A	2	30	180			North Celestial Pole		One North Celestial Pole Observation
11	1b	0	S	-	-	-					
		3	A	2	48	144		Yes	One Performance		> One Performance
12	3c	0	S	-	-	-					
		2	A	2	34	68	Change trumtion values	Yes			
13	4a	0	S	-	-	-					
		2	A	2	160	320					
14	3a	0	S	-	-	-					Requires use of orbital counter and may be performed during a crew day off or sleep period.
		3	A	2	17	51		Yes			FO 3, Programs 2a and 2b are of higher priority than the third repetition of FO B Program 3a.
15	4b	1	S	7	320	320					Coordinate with Program 3b of FO 2.
		0	A	-	-	-		Yes			Requires use of orbital counter.
16	1d	0	S	-	-	-					
		2	A	2	15	30					Perform early in mission, North Ecliptic Pole is of higher priority.

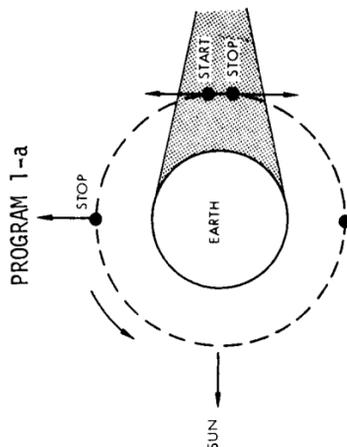
T027/S073 MODE PROGRAM SUMMARY



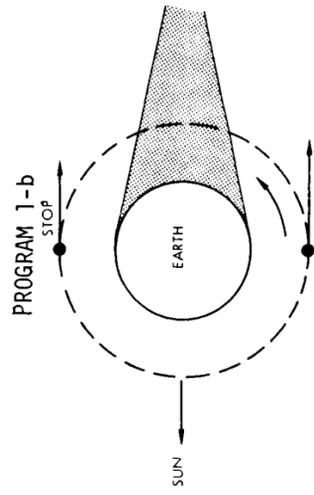
- PROGRAM 0-a**
- SYSTEM MONITOR
 - PHOTOMETER CAPPED
 - SCAN DURING ENTIRE ORBIT
 - SOLAR SAL 3 PERFORMANCES
 - A-SOL-SAL 1 PERFORMANCE

MODE 1

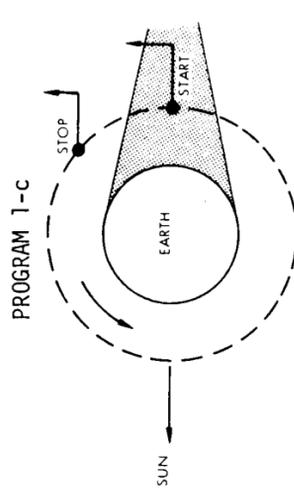
- FIXED POSITION
- OBSERVATIONS ARE MADE WITH THE PHOTOMETER POINTED AT A TARGET. NO PHOTOMETER MOVEMENT IN SHAFT OR TRUNNION.



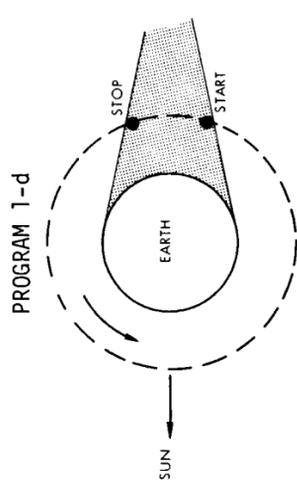
- PROGRAM 1-a**
- CONTAMINATION IN FIXED POSITION
 - PHOTOMETER POINTED 90° TO SUN LINE
 - TWO 20 MINUTE OPERATING PERIODS PER PERFORMANCE
 - SOLAR SAL 3 PERFORMANCES
 - A-SOL-SAL 3 PERFORMANCES
 - CREWMAN REQUIRED TO REPOSITION PHOTOMETER BETWEEN SCANS



- PROGRAM 1-b**
- GEGENSCHWEIN
 - PHOTOMETER POINTED IN ANTI SOLAR DIRECTION
 - 48 MINUTE OPERATING PERIOD
 - A-SOL-SAL 3 PERFORMANCES
 - ONE PERFORMANCE REQUIRED + 1 WEEK OF THE NEW MOON.

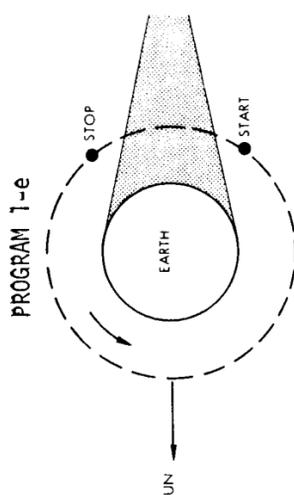


- PROGRAM 1-c**
- CONTAMINATION
 - PHOTOMETER POINTED IN DIRECTION OF CONTAMINATION SOURCE
 - 20 MINUTE OPERATING PERIOD
 - A-SOL-SAL 1 PERFORMANCE
 - START PRIOR TO LEAVING EARTH'S SHADOW & CONTINUE IN SUNLIGHT
 - CAMERA SYSTEM REQUIRED



PROGRAM 1-d

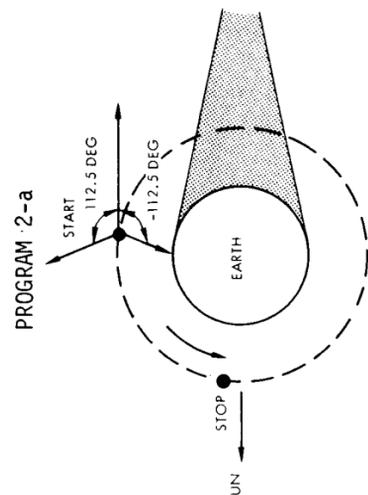
- PROGRAM 1-d Continued**
- F0-10
- ECLIPTIC POLES N. OR S.
 - PHOTOMETER POINTED AT NORTH OR SOUTH ECLIPTIC POLE
 - 15 MINUTE OPERATING PERIOD
 - A-SOL-SAL 2 PERFORMANCES
 - PERFORM WITHIN EARTH'S SHADOW



- PROGRAM 1-e**
- F0-6
- CELESTIAL POLES N. OR S.
 - PHOTOMETER POINTED AT NORTH OR SOUTH ECLIPTIC POLE
 - 30 MINUTE OPERATING PERIOD
 - A-SOL-SAL 6 PERFORMANCES
 - COORDINATE WITH GROUND OBSERVATIONS AT HAWAII

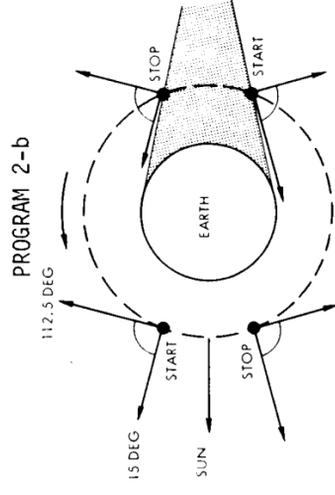
MODE 2

- VERTICAL CIRCLE
- PHOTOMETER SCANS IN TRUNNION BETWEEN SET LIMITS
- SHAFT IS SET AT A FIXED POSITION
- ALL FILTERS ARE USED



PROGRAM 2-a

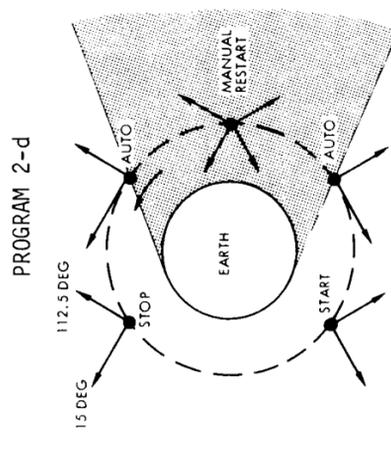
- PROGRAM 2-a Continued**
- F0-3
- ECLIPTIC
 - SCAN IN TRUNNION FROM +112° TO ANTI SOLAR POINT, ROTATE 180°, SCAN TO -112°
 - 19 MINUTE SCAN
 - A-SOL-SAL 3 PERFORMANCES
 - TRUNNION 180° ROTATION IS AUTOMATIC
 - PERFORM IN DARKNESS OUTSIDE THE EARTH'S SHADOW



PROGRAM 2-b

- PROGRAM 2-b**
- F0-3
- ECLIPTIC
 - SCAN IN TRUNNION BETWEEN +112° & +15°
 - TWO 19 MINUTE SCANS PER PERFORMANCE
 - PERFORM IN EARTH'S SHADOW, REPEAT DURING DAYLIGHT PORTION OF ORBIT
 - SOLAR SAL 3 PERFORMANCES
 - CREWMAN REQUIRED TO RESET PROGRAM AFTER INITIAL 19 MIN. OBSERVATION

- PROGRAM 2-c Continued**
- F0-4
- VERTICAL CIRCLE
 - SCAN IN TRUNNION BETWEEN 0° & 112° AS ILLUSTRATED
 - FOUR 15 MINUTE SCANS PER PERFORMANCE
 - START PROGRAM 15 MIN. BEFORE TERMINATOR
 - A-SOL-SAL 2 PERFORMANCES
 - 30 MINUTE MINIMUM DARK PERIOD DESIRABLE
 - CREWMAN REQUIRED TO CHANGE SHAFT SETTING & RESTART PROGRAM AFTER THE FIRST TWO 15 MIN. SCANS



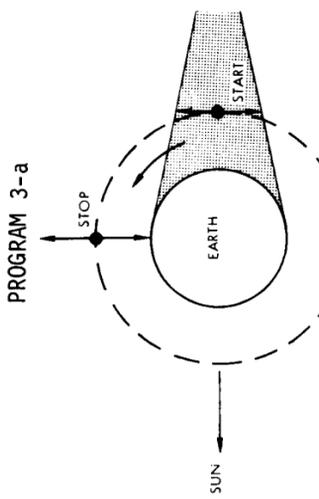
PROGRAM 2-d

- PROGRAM 2-d**
- F0-4
- VERTICAL CIRCLE
 - SCAN IN TRUNNION BETWEEN +15° & +112° AS ILLUSTRATED
 - FOUR 15 MINUTE SCANS PER PERFORMANCE
 - START PROGRAM 15 MIN. BEFORE TERMINATOR
 - SOLAR SAL 2 PERFORMANCES
 - 30 MINUTE MINIMUM DARK PERIOD DESIRABLE
 - CREWMAN REQUIRED TO CHANGE SHAFT SETTINGS & RESTART PROGRAM AFTER THE FIRST TWO 15 MINUTE SCANS

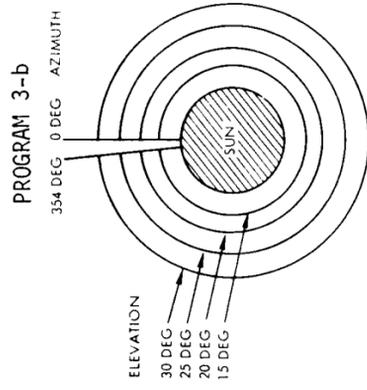
MODE 3

- ALMUCANTER
- PHOTOMETER SCANS IN SHAFT BETWEEN SET LIMITS
- TRUNNION IS SET AT A FIXED POSITION
- ALL 10 FILTERS ARE USED

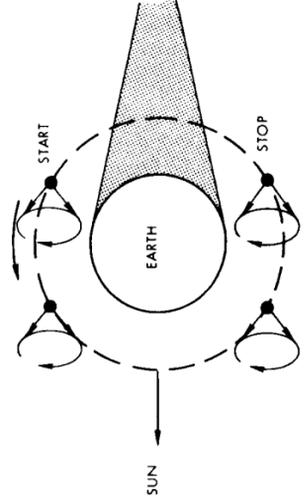
T027 / S073 MODE PROGRAM SUMMARY



- PROGRAM 3-a**
- PERPENDICULAR TO ECLIPTIC
 - TRUNNION PRESET AT 90°
 - SCAN IN SHAFT FROM 0° TO 354°
 - 17 MINUTE SCAN
 - A-SOL-SAL 3 PERFORMANCES
 - PERFORM INSIDE & OUTSIDE THE EARTH'S SHADOW

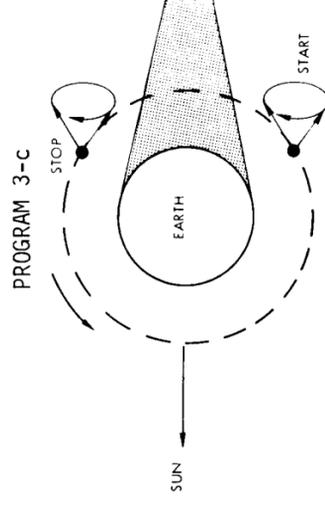


- PROGRAM 3-b**
- 354 DEG 0 DEG AZIMUTH
 - ELEVATION
 - 30 DEG
 - 25 DEG
 - 20 DEG
 - 15 DEG



- PROGRAM 3-d**
- 110 DEG START
 - 112 DEG STOP
 - 110 DEG START
 - 112 DEG STOP
 - CHANGE RODS
 - 110 DEG STOP
 - 112 DEG START
 - STOP/RESTART 2
 - STOP/RESTART 4

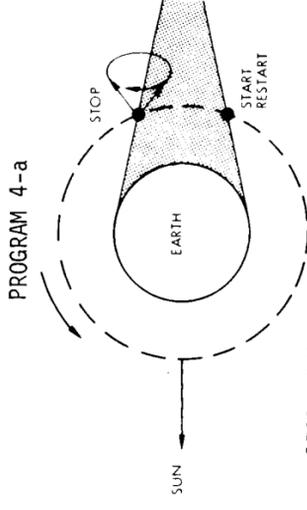
- PROGRAM 3-b Continued**
- CONTAMINATION
 - TRUNNION MANUALLY SET AT 15°, 20°, 25° & 30°
 - SCAN IN SHAFT FROM 0° TO 354° AT EACH TRUNNION SETTING
 - 68 MINUTES PER PERFORMANCE, FOUR 15 MINUTE SCANS
 - SOLAR SAL 2 PERFORMANCES
 - CREWMAN REQUIRED TO CHANGE TRUNNION SETTINGS



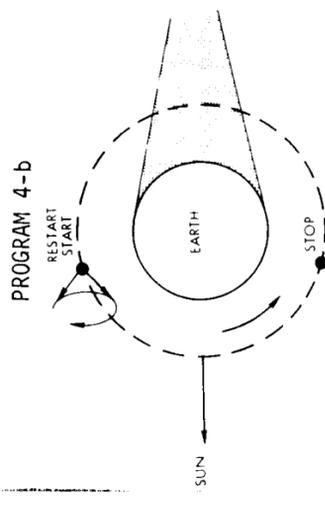
- PROGRAM 3-c**
- GEGENSCHIEIN
 - TRUNNION MANUALLY SET AT 2.8° & 4.2°
 - SCAN IN SHAFT FROM 0° TO 354° AT EACH TRUNNION SETTING
 - 34 MINUTES PER PERFORMANCE, TWO 15 MINUTE SCANS
 - A-SOL-SAL 2 PERFORMANCES
 - CREWMAN REQUIRED TO CHANGE TRUNNION SETTING

- PROGRAM 3-3 Continued**
- CONTAMINATION
 - TRUNNION MANUALLY SET AT 112° & 110°
 - SCAN IN SHAFT FROM 0° TO 354° AT EACH TRUNNION SETTING
 - 34 MINUTES PER PERFORMANCE, TWO 15 MINUTE SCANS
 - ONE PERFORMANCE AT 2 ROD LENGTHS & ONE PERFORMANCE AT 7 ROD LENGTHS
 - A-SOL-SAL ONE PERFORMANCE
 - PERFORM OUTSIDE THE EARTH'S SHADOW
 - CREWMAN REQUIRED TO MONITOR INTENSITY, CHANGE TRUNNION SETTING, PMT GATH & ROD LENGTHS

- MODE 4**
- SKY MAPPING
 - MAP LIMITED REGIONS NEAR SUN OR GEGENSCHIEIN
 - SERIES OF ALMUCANTAR SCANS
 - ALL 10 FILTERS USED

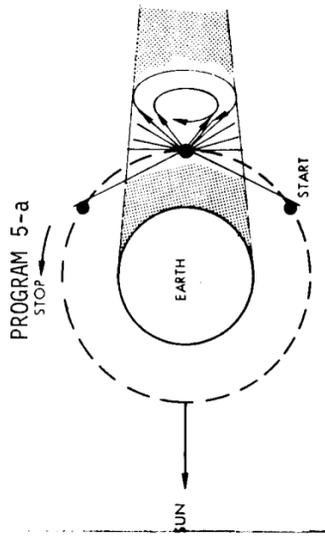


- PROGRAM 4-a**
- GEGENSCHIEIN
 - TRUNNION INITIALLY SET AT 2.8° & AUTOMATICALLY STEPS IN INCREMENTS OF 2.8° TO 28° FOR EACH FILTER
 - SCAN IN SHAFT FROM 0° TO 354° AT EACH TRUNNION SETTING
 - 160 MINUTES PER PERFORMANCE, 16 MINUTES PER FILTER
 - SCAN IN ANTI SOLAR DIRECTION DURING NIGHT SIDE OF ORBIT
 - A-SOL-SAL 2 PERFORMANCES
 - TAKES 10 CONSECUTIVE ORBITS

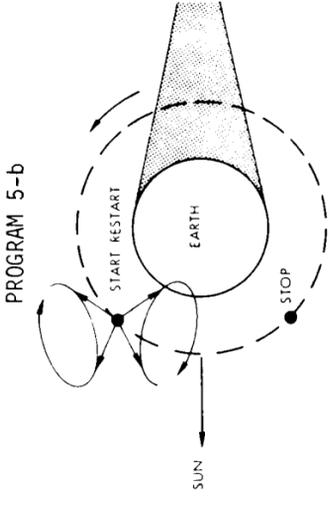


- PROGRAM 4-b**
- INNER ZODIACAL LIGHT/CONTAMINATION
 - TRUNNION INITIALLY SET AT 15° & AUTOMATICALLY STEPS IN INCREMENTS OF 2.8° TO APPROXIMATELY 70°
 - SCAN IN SHAFT FROM 0° TO 354° AT EACH TRUNNION SETTING
 - 320 MINUTES PER PERFORMANCE, 32 MINUTES PER FILTER
 - SCAN IN SOLAR DIRECTION DURING DAY SIDE OF ORBIT
 - SOLAR SAL ONE PERFORMANCE
 - TAKES 10 CONSECUTIVE ORBITS USING THE ORBITAL COUNTER

- MODE 5**
- ALL SKY MAPPING
 - SERIES OF ALMUCANTAR SCANS
 - ALL 10 FILTERS USED



- PROGRAM 5-a Continued**
- ALL SKY MAP
 - TRUNNION INITIALLY SET AT 2.8° & AUTOMATICALLY STEPS IN INCREMENTS OF 5.6° TO APPROXIMATELY 112°
 - SCAN IN SHAFT FROM 0° TO 354° AT EACH TRUNNION SETTING
 - 320 MINUTES PER PERFORMANCE, 32 MINUTES PER FILTER
 - CONCENTRIC SCANS IN THE ANTI-SOLAR DIRECTION
 - A-SOL-SAL ONE PERFORMANCE
 - TAKES 10 CONSECUTIVE ORBITS USING THE ORBITAL COUNTER



- PROGRAM 5-b**
- ALL SKY MAP
 - TRUNNION INITIALLY SET AT 15° & AUTOMATICALLY STEPS IN INCREMENTS OF 5.6° TO APPROXIMATELY 112°
 - SCAN IN SHAFT FROM 0° TO 354° AT EACH TRUNNION SETTING
 - 290 MINUTES PER PERFORMANCE, 29 MINUTES PER FILTER
 - SOLAR-SAL ONE PERFORMANCE
 - SCAN IN THE SOLAR DIRECTION
 - TAKES 10 CONSECUTIVE ORBITS USING THE ORBITAL COUNTER

S149 PARTICLE COLLECTION. DEPLOY, EXPOSE, RETRIEVE & RETURN TO EARTH MICROMETEORITE IMPACT DETECTION CASSETTES
 FO-1 DURING MANNED PORTION OF FLIGHT
 FO-2 DURING UNMANNED PORTION OF FLIGHT

MRD 11/1/71
 EOH 11/1/71
 ERD 1/12/71
 ODB

ET	12	3	V	*6	CREWMAN	*8	CREWMAN
0:00				12	//S149 PREPAR 08.02.01-00// S149-X PREP A-SOL-SAL ATTACH MD/CSU TO T027 EXTENSION MECHANISM INSTALL ASSEMBLY INTO A-SOL-SAL	12	// S149 PREPAR 08.02.01-00// S149-X PREP A-SOL-SAL ASSIST WITH EXTENSION MECHANISM INSTALLATION INTO A-SOL-SAL
0:25					ATTACH TRIPOD SUPPORT OPEN SAL OUTER DOOR DEPLOY 2 RODS OUT CONNECT PWR/INST CABLES PRESET PANEL SWITCHES NOTIFY GROUND EXPERIMENT IS READY FOR ACTIVATION		
0:00				07	//S149 MANNED 08.02.03-00// S149-1 MANNED EXPOSURES PERF # <u> </u> OF <u>1</u> VERIFY PROPER OPERATION (WAIT 72 HRS NOMINAL)		
0:15							S149 UPDATE PAD
0:00				05	//S149 RETRACT 08.02.09-00// S149-X RETRACT A-SOL-SAL MONITOR CASSETTE CLOSURE REMOVE PWR/INST CABLES,STOW RETRACT MD/CSU CLOSE SAL DOOR, PRESSURIZE		
0:10							
0:00				07	//S149 STOWAG 08.02.10-00// (CONTINUED)	07	//S149 STOWAG 08.02.10-00//

SCHEDULE CRITERIA

- Detection system will be exposed to near earth space for 72 hours (min 8 hrs).
- Exp must not be scheduled concurrently with S019, S063, T027/S073, T020, T013, M509, S183, & EREP.
- Schedule during the solar inertial mode only.
- Solar SAL may be used in contingency mode.
- Use wait period shown in CREWMAN column.
- Do not schedule during crew controlled vents.
- Deploy the second cassette at the end of SL-2 mission for unmanned exposure.

CONSTRAINTS

- Close detection cassettes during periods of extraneous contamination, i.e., waste dumps, vents, thruster firing, etc.
- Minimum exposure of 8 hours.

RETURN WEIGHT

- Cassettes & stowage container - 11.5 pounds.

NOTES

- Recorded voice or log entries will be made to describe anomalous conditions, contamination periods such as urine dumps, venting, etc., including time spans and quantities.
- Record attitude deviations of more than +15° from nominal (any axis), (ATM recorder).
- MCC-H will activate & demonstrate experiment remotely with crew backup as required.

S183 ULTRAVIOLET PANORAMA		MRD		4/1/72	
F0-1 THROUGH 12 OBTAIN ULTRAVIOLET PHOTOGRAPHS OF DESIGNATED STARFIELDS DURING DARKSIDE PASSES		EOH VOL I		11/19/71	
		EOH VOL II		6/7/71	
		ODB		2/17/72	
		AE		1/3/72	
ET	123 V #6	CREWMAN	*6	CREWMAN	
0:00	5 // S019 AMS IN 08.03.01-00// S019-1 A-SOL-SAL	INSTALL AMS IN SAL			
0:10	11 // S183 SA INS 08.09.01-00//		1	// S183 ASSIST	
0:00	S183-1 SET UP A-SOL-SAL	ATTACH SPECTROGRAPH ASSEMBLY AND FILM CARROUSEL		S183-1 ASSIST IN EXPERIMENT ASSEMBLY	
		INSTALL MAUER CAMERA			
		VENT SAL TO SPACE			
0:22	23 // S183 OPERAT 08.09.03-00//		1	// S183 INHIBI	
0:00	S183-1 A-SOL-SAL PERF # OF 12 OPEN SAL DOOR, EXTEND MIRROR SYSTEM	STARFIELD TILT ROT EXP		S183-1 INHIBIT CMG DUMP	
		STARFIELD TILT ROT EXP			

SCHEDULE CRITERIA

- Anti-solar SAL required. Do not schedule concurrently with S019, S063, S149, and T027/S073, -2, -3, -5, -7.
- Schedule as close as possible to S019, so results will be comparable (provides similar film and optics degradation).
- Do not schedule concurrently with T013, T020, M509 or any activity requiring excessive crew motion.
- Experiment can only be performed on dark-side passes.
- Waste dumps or outgassing which would affect optics and viewing must be inhibited.
- The mirror system need only be retracted into the SAL during daylight portion of orbit. To minimize film degradation the experiment must be dismantled if observing periods are separated by more than 12 hours. Total exposure time (to OWS ambient radiation) should not exceed 96 hours.
- The mirror system, spectrograph assembly and film canister must be evacuated following experiment disassembly and before restowing.
- Spacecraft should be in solar inertial attitude.

CONSTRAINTS

- All lights which could scatter lights into the spectrograph will be extinguished during exposure periods.

S183 ULTRAVIOLET PANORAMA
FO-1 THROUGH 12 OBTAIN UV PHOTOGRAPHS OF
DESIGNATED STARFIELDS

ET	AM	12	3	V	*6	CREWMAN	*6	CREWMAN	NOTES
0:45						STARFIELD TILT ROT EXP RETRACT MIRROR SYSTEM AND CLOSE SAL DOOR BEFORE SUNRISE			<ul style="list-style-type: none"> The average time for a photographic sequence is assumed to be 15 minutes. Since the exposure times for each starfield vary considerably it may be possible to schedule more than one starfield per night pass. It is anticipated that the 35 photographic plates will be exposed on 12 starfields at the rate of 3 exposures per darkside pass. The question of whether to inhibit momentum dumps or schedule around them during an exposure sequence has not been settled. <p>RETURN WEIGHT</p> <ul style="list-style-type: none"> Film carousel 6.0 lb. Film stowage container 4.0 lb. Film magazine (16mm DAC) 1.0 lb.
0:00					7	// S183 SA REM 08.09.06-00//	1	// S183 ENABLE	
0:13						S183 -1 REMOVE SA AND FILM CARROUSEL FROM AMS		S183 -1 ENABLE CMG DUMP	
0:00					19	// S183 STOW E 08.09.08-00 //		// S183 DISASS	
						S183 -1 REMOVE AMS FROM SAL EVACUATE AND STOW AMS EVACUATE AND STOW SA EVACUATE AND STOW FILM CARROUSEL REMOVE AND STOW MAUER CAMERA		S183 -1 ASSIST IN EXPERIMENT DISASSEMBLY	

S183 ULTRAVIOLET PANORAMA
 F0-1 THROUGH 12 OBTAIN UV PHOTOGRAPHS OF
 DESIGNATED STARFIELDS

ET	AM	1	2	3	V	*6	*6
0:38						SECURE SAL	
0:00						5 // S183 LOG A 08.09.09-00 //	
0:10						S183 -1 STOW STOW LOG AND FILM IN CM FOR RETURN	

S183 ULTRAVIOLET PANORAMA
CANDIDATE STAR FIELDS

*STAR NUMBER IN SMITHSONIAN ASTROPHYSICAL
OBSERVATORY STAR CATALOG

ID NUMBER	GMT	RT ASCENSION-1950			DECLINATION - 1950			CATALOG NUMBER*	PRIORITY	MAGNITUDE	EXPOSURE TIMES (SEC)		
		h	m	s	o	i	u				1	2	3
1	TBD	0	30	8.4	+60	39	22	645	1	4.24	TBD	TBD	TBD
2	TBD	0	34	10.3	+53	37	19	727	2	3.72	TBD	TBD	TBD
3	TBD	1	22	31.5	+59	58	34	1715	2	2.80	TBD	TBD	TBD
4	TBD	2	18	51.2	+55	37	05	2836	1	5.22	TBD	TBD	TBD
5	TBD	3	25	00.1	+59	46	05	4113	TBS	4.42	TBD	TBD	TBD
6	TBD	3	32	55.5	+48	1	41	4287	1	4.26	TBD	TBD	TBD
7	TBD	3	44	30.4	+23	57	08	4541	1	2.96	TBD	TBD	TBD
8	TBD	3	46	22.6	+32	56	23	4592	1	5.10	TBD	TBD	TBD
9	TBD	5	03	00.2	+41	10	08	6226	2	3.28	TBD	TBD	TBD
10	TBD	5	10	55.3	-12	59	57	6387	2	4.46	TBD	TBD	TBD
11	TBD	5	16	43.0	+33	54	28	6515	2	5.16	TBD	TBD	TBD
12	TBD	5	22	26.9	+06	18	22	6668	1	1.70	TBD	TBD	TBD
13	TBD	5	29	30.6	-7	20	13	6850	1	4.64	TBD	TBD	TBD
14	TBD	5	32	23.8	+24	00	30	6916	2	5.28	TBD	TBD	TBD
15	TBD	5	33	40.5	-01	13	56	6960	1	1.75	TBD	TBD	TBD
16	TBD	6	11	33.3	+17	55	20	7956	2	5.74	TBD	TBD	TBD
17	TBD	6	26	18.9	-32	32	51	8410	2	4.48	TBD	TBD	TBD
18	TBD	6	30	12.0	+7	22	16	8506	1	4.50	TBD	TBD	TBD
19	TBD	6	51	52.0	-11	58	29	9051	2	4.25	TBD	TBD	TBD
20	TBD	7	16	49.0	-26	23	36	9740	1	5.40	TBD	TBD	TBD
21	TBD	7	32	02.4	-36	13	43	10139	1	5.51	TBD	TBD	TBD
22	TBD	7	56	17.4	-45	26	31	10790	1	5.16	TBD	TBD	TBD
23	TBD	8	45	25.0	-56	35	07	12138	2	4.63	TBD	TBD	TBD
24	TBD	8	48	03.9	-45	07	16	12204	1	5.02	TBD	TBD	TBD
25	TBD	10	30	14.5	-61	25	40	14489	1	3.58	TBD	TBD	TBD
26	TBD	11	33	27.9	-62	44	35	15899	1	3.34	TBD	TBD	TBD
27	TBD	13	19	23.0	-60	43	37	18087	2	4.62	TBD	TBD	TBD
28	TBD	13	55	34.8	-44	33	38	18883	2	4.17	TBD	TBD	TBD
29	TBD	15	05	27.9	-45	05	20	20356	2	4.39	TBD	TBD	TBD
30	TBD	15	55	34.6	-24	41	20	21442	1	5.41	TBD	TBD	TBD
31	TBD	16	16	05.3	-50	02	06	21933	2	4.14	TBD	TBD	TBD
32	TBD	16	48	28.7	-37	57	49	22677	TBS	3.09	TBD	TBD	TBD
33	TBD	17	44	24.7	-27	48	49	24135	1	Var	TBD	TBD	TBD
34	TBD	17	46	27.2	-37	01	46	24188	1	3.25	TBD	TBD	TBD
35	TBD	18	10	46.3	-21	04	26	24856	1	Var	TBD	TBD	TBD
36	TBD	18	51	58.7	+36	54	29	25934	1	5.51	TBD	TBD	TBD
37	TBD	19	03	35.7	-04	57	33	26285	2	3.55	TBD	TBD	TBD
38	TBD	19	14	04.0	+21	18	03	26569	2	4.60	TBD	TBD	TBD

S183 ULTRAVIOLET PANORAMA
 CANDIDATE STAR FIELDS
 *STAR NUMBER IN SMITHSONIAN ASTROPHYSICAL
 OBSERVATORY STAR CATALOG

ID NUMBER	GMT	RT ASCENSION - 1950			DECLINATION - 1950			CATALOG NUMBER*	PRIORITY	MAGNITUDE	EXPOSURE TIMES (SEC)		
		h	m	s	°	'	"				1	2	3
39	TBD	19	37	52.0	+33	51	45	27216	1	6.12	TBD	TBD	TBD
40	TBD	20	15	56.5	+37	52	35	28218	1	Var	TBD	TBD	TBD
41	TBD	20	54	48.8	+44	43	54	29241	1	6.01	TBD	TBD	TBD
42	TBD	21	36	34.7	+61	51	21	30302	1	4.87	TBD	TBD	TBD
43	TBD	22	00	00.4	+52	38	26	30828	1	5.66	TBD	TBD	TBD
44	TBD	22	39	14.0	+39	57	50	31670	1	Var	TBD	TBD	TBD
45	TBD	23	04	29.5	+59	08	57	32197	2	4.93	TBD	TBD	TBD
46	TBD	23	37	56.2	+44	03	25	32886	2	4.33	TBD	TBD	TBD
47	TBD	0	46	51.7	-75	11	43	983	1	4.96	TBD	TBD	TBD
48	TBD	5	13	47.4	-67	14	30	6444	1	4.78	TBD	TBD	TBD
49	TBD	0	47	02.8	+40	48	25	989	1	4.42	TBD	TBD	TBD
50	TBD	1	18	21.1	+28	28	39	1630	1	5.60	TBD	TBD	TBD
51	TBD	4	33		+16			TBS	1	0.86	TBD	TBD	TBD
52	TBD	8	38		+3	45		TBS	1	4.30	TBD	TBD	TBD
53	TBD	13	43	36	+49	49		TBS	TBS	1.86	TBD	TBD	TBD

S190B EARTH TERRAIN CAMERA

MRD 1/3/72
 EOH 2/18/72
 ERD
 ODB

AM	ET	1	2	3	V	*6	CREWMAN	*6	CREWMAN	SCHEDULE CRITERIA
	0:00					15	//S190B PREP 08.0802-00 // S190B-X PREP A-SOL-SAL INSTALL ETC WINDOW IN ANTI SOLAR SAL REMOVE CAMERA FROM STOWAGE CONTAINER, INSTALL FILTER INSTALL CAMERA IN SAL WINDOW	15	//S190B PREP 08.08.02-00 // S190B-X PREP A-SOL-SAL ASSIST WITH ETC CAMERA INSTALLATION INTO ANTI-SOLAR SAL	<ul style="list-style-type: none"> An update pad is required at least one rev prior to the data taking pass. Attitude data is required on the ATM recorder. Schedule S190B photography in conjunction with S190A. For each data pass voice recorded camera and site data. Data taking periods will vary between 10 to 30 minutes.
	0:30						SET CAMERA CONTROLS & SWITCHES PER C/L			<p><u>CONSTRAINTS</u></p> <ul style="list-style-type: none"> Film temperature - 50 to 90°F. Humidity - 30 to 95% Radiation - less than .37 rads/ds. The camera lens optical axis must be pointed to within $\pm 0.5^\circ$ of the nominal -Z axis. Photography in the summer hemisphere is constrained to periods when the sun elevation angle is greater than 30° and greater than 20° in the winter hemisphere. Spacecraft attitude control must be within $\pm 20^\circ$ about each axis. Allowable S/C rates must not exceed 0.05°/sec in any axis. Film is to be stowed in the OWS film vault if time between data passes exceed 3 hours. A total of _____ data passes in Z-LV(E) and _____ passes in SI are required during SkyLab. <p><u>RETURN WEIGHT</u></p> <ul style="list-style-type: none"> S190B film _____ pounds.
	0:00					02	CONNECT POWER CABLES			
	0:00						//S190B MAGAZ 08.08.03-00 //			
	0:05						S190B-X LOAD A-SOL-SAL INSTALL FILM MAGAZINE IN ETC			
	0:00					15	//S190B CAMER 08.08.04-00 // S190B-X A-SOL-SAL PERF # _____ OF ADVANCE FILM 6 FRAMES VERIFY CAMERA OPERATION SET PER UPDATE PAD <ul style="list-style-type: none"> SHUTTER SPEED MR/SEC FRAMES/MIN VENT SAL, OPEN DOOR CONNECT PLATEN VACUUM SUBDUED REQUIRED LIGHTING			

S190B EARTH TERRAIN CAMERA

ET	AM	12	3	V	*6	CREWMAN	*6	CREWMAN	NOTES
0:30						OPERATE CAMERA, CHANGE SPEED & IMC CONTROL AS REQ			<ul style="list-style-type: none"> Each film canister contains 200 feet of film, approximately 450 frames. Film allocation is as follows: SL-2 4 canisters, SL-3 6 canisters and SL-4 6 canisters. The ETC controls are as follows: Power - on/off Shutter speed - 1/100 sec 1/200 sec 1/500 sec IMC - 0 to 25 milliradians/sec film speed - 0 to 25 frames/min film advance - 2 feet/second Eight filters, neutral density and spectral shaping, are utilized. Surface coverage at 235 miles altitude will be approximately 59 sq. miles. Data collection periods will vary from 10 to 30 minutes depending on the length of Z-LV(E) coverage.
0:50					SECURE CAMERA				
0:00				02	//S190B DOWN 08.08.05-00 //				
0:05					S190B-X UNLOAD A-SOL-SAL DOWNLOAD MAGAZINE REMOVE FILM & STOW				
0:00				10	//S190B STOWA 08.08.06-00 //		//S190B STOWA 08.08.06-00 //		
0:20					S190B-X STOWAGE A-SOL-SAL DISCONNECT PWR CABLES & STOW REMOVE CAMERA FROM WINDOW & STOW IN CONTAINER		S190B-X STOWAGE A-SOL-SAL ASSIST WITH ETC CAMERA REMOVAL FROM ANTI-SOLAR SAL & CAMERA STOWAGE		
0:00				05	CLOSE DOOR, PRESSURIZE SAL REMOVE ETC WINDOW INSTALL SAL WINDOW STOW ETC WINDOW				
0:10					//S190B TRANS 08.08.08-00 //				
					S190B-X TRANSFER CM				
					TRANSFER EXPOSED FILM CASSETTES TO CM STOWAGE				

T002 MANUAL NAVIGATION SIGHTINGS		MRD		11/1/71				
FO-1		EOH VOL I		11/19/71				
ON TWO KNOWN STARS USING A HAND-HELD SEXTANT		EOH VOL II		6/7/71				
		ODB		11/8/71				
		AE		1/3/72				
ET	I AM	12	3	V	*6	*6	CREWMAN	SCHEDULE CRITERIA
0:00					25	// T002 SEXT M 09.03.02-00 //		<ul style="list-style-type: none"> Do not schedule concurrently with S063-1 since both experiments use the wardrobe window. Experiment to be performed on dark side of orbit and external lights near the wardrobe window will be off. Wardroom interior lights will be dimmed during experiment. Waste dumps will not be performed during experiment performance. The sighting periods should be distributed equally in time throughout the mission. If the experiment is not scheduled on SL-2 then one performance of FO-1 would be desirable on SL-2 to test the equipment. One ATM recorder required.
						T002-1		
						UNSTOW EXPERIMENT HARDWARE		
						SUBDUED LIGHTING		
						PERFORM REFERENCE CALIBRATION OF SEXTANT		
						VOICE RECORD GMT, SEXTANT		
						TEMP & TARGET PAIR AT		
						START OF SESSION		
						ACQUIRE:		
						STAR 1 _____		
						STAR 2 _____		
						VOICE "MARK" WHEN IMAGES		
						ALIGNED AND ANNOUNCE		
						MEASURED ANGLE		
						REPEAT SIGHTING FOR A TOTAL		
						OF 10 TIMES MINIMUM		
						STOW EXPERIMENT HARDWARE		
0:50							UPDATE PAD	<p><u>CONSTRAINTS</u></p> <ul style="list-style-type: none"> Maximum allowable spacecraft rates will be nominal solar inertial attitude rates. <p><u>RETURN WEIGHT</u></p> <ul style="list-style-type: none"> N/A

T002 MANUAL NAVIGATION SIGHTINGS
 FO-2 PERFORM 12 PERIODS OF MIDCOURSE TYPE NAVIGATION SIGHTINGS ON
 A KNOWN STAR AND THE LUNAR LIMB USING A HAND-HELD SEXTANT

ET	AM	1	2	3	V	*6	*8
0:00						23	// T002 ST LU 09.03.02-02 //
							T002-2 UNSTOW EXPERIMENT HARDWARE SUBDUCE LIGHTING PERFORM REFERENCE CALIBRATION OF SEXTANT VOICE RECORD GMT, SEXTANT TEMP & TARGET AT START OF SESSION ACQUIRE: STAR AND LUNAR LIMB VOICE "MARK" WHEN IMAGES ALIGNED AND ANNOUNCE ANGLE AND FILTER USED REPEAT SIGHTINGS FOR A MIN. TOTAL OF 10 TIMES
							CS-W STOW EXPERIMENT HARDWARE
0:45							STOW EXPERIMENT HARDWARE
							// T002 PAD UP 09.03.01-00 //
							T002 PAD UP UPDATE PAD
							<u>SCHEDULE CRITERIA</u> <ul style="list-style-type: none"> Do not schedule concurrently with S063-1 since both experiments use the wardrobe window. Experiment to be performed on dark side of orbit and external lights near the wardrobe window will be off. Wardroom interior lights will be dimmed during experiment performance. The sighting periods should be distributed equally in time throughout the mission. Waste dumps will not be performed during experiment performance. One ATM recorder required. <u>CONSTRAINTS</u> <ul style="list-style-type: none"> Maximum allowable spacecraft rates will be nominal solar inertial attitude rates <u>RETURN WEIGHT</u> <ul style="list-style-type: none"> N/A.

T002 MANUAL NAVIGATION SIGHTINGS
 FO-3 PERFORM SIX PERIODS OF MIDCOURSE TYPE NAVIGATION SIGHTINGS
 ON TWO PORTIONS OF THE LUNAR LIMB USING A HAND-HELD SEXTANT

ET	AM	CREWMAN	*S	CREWMAN	SCHEDULE CRITERIA
0:00	23 //	T002 LU LU 09.03.02-03 //	*S		<p><u>SCHEDULE CRITERIA</u></p> <ul style="list-style-type: none"> Do not schedule concurrently with S063-1 since both experiments use the wardroom window. Experiment to be performed on dark side of orbit and external lights near the wardroom window will be off. Wardroom interior lights will be dimmed during experiment. The sighting periods should be distributed equally in time throughout the mission. Waste dumps will not be performed during experiment performance. The lunar limb must be visible through the wardroom window. One ATM recorder required. <p><u>CONSTRAINTS</u></p> <ul style="list-style-type: none"> Maximum allowable spacecraft rates will be nominal solar inertial attitude rates. <p><u>RETURN WEIGHT</u></p> <ul style="list-style-type: none"> N/A.
0:45		<p>T002-3 UNSTOW EXPERIMENT HARDWARE SUBDUE LIGHTING PERFORM REFERENCE CALIBRATION OF SEXTANT VOICE RECORD GMT, SEXTANT TEMP & TARGET AT START OF SESSION</p> <p>ACQUIRE THE TWO PORTIONS OF THE LUNAR LIMB WITH SEXTANT</p> <p>VOICE "MARK" WHEN IMAGES ALIGNED AND ANNOUNCE MEASURED ANGLE</p> <p>REPEAT SIGHTING FOR A MIN. TOTAL OF 10 TIMES</p> <p>VOICE GMT AND SEXTANT TEMP STOW EXPERIMENT HARDWARE</p>			

T002 MANUAL NAVIGATION SIGHTINGS
 FO-4 PERFORM FIVE PERIODS OF ORBIT TYPE NAVIGATION SIGHTINGS
 ON THE EARTH'S HORIZON USING A HAND-HELD STADIMETER

ET	AM	12	3	V	*6	CREWMAN	*6	CREWMAN
0:00					23 // T002 STAD 0 09.03.03-00// T002-4			
					UNSTOW EXPERIMENT HARDWARE SUBDUE LIGHTING			
					VOICE RECORD GMT AT START OF SESSION AND FILTER USED IF IN DAYLIGHT			
					PERFORM STADIMETER HORIZON SIGHTING			
					VOICE "MARK" WHEN IMAGES ALIGNED AND ANNOUNCE READING			
					CONDUCT AS MANY SIGHTINGS AS CONVENIENT DURING SESSION			
0:45					STOW EXPERIMENT HARDWARE			
<p><u>SCHEDULE CRITERIA</u></p> <ul style="list-style-type: none"> Do not schedule concurrently with S063-1 since both experiments use the wardrobe window. Waste dumps will not be performed during experiment performance. The earth's horizon must be visible from the wardrobe window. If the experiment is not scheduled on SL-2 then one performance of FO-4 would be desirable on SL-2 to test the equipment. Wardroom interior lights will be dimmed during experiment. One ATM recorder required. <p><u>CONSTRAINTS</u></p> <ul style="list-style-type: none"> Maximum allowable spacecraft rates will be nominal solar inertial attitude rates. <p><u>RETURN WEIGHT</u></p> <ul style="list-style-type: none"> N/A. 								

T002 MANUAL NAVIGATION SIGHTINGS
 FO-5 PERFORM 2 PERIODS OF ORBIT TYPE NAVIGATION SIGHTINGS ON THE
 EARTH'S HORIZON AND A KNOWN STAR USING A HAND-HELD SEXTANT

ET	L A M	1	2	3	V	#6	CREWMAN	#6	CREWMAN
0:00						25	//T002 SEXT 0 09.03.04-00 // T002-5 CS-W UNSTOW EXPERIMENT HARDWARE SUBDUE LIGHTING PERFORM REFERENCE CALIBRATION OF SEXTANT VOICE RECORD GMT, SEXTANT TEMP & TARGET AT START OF SESSION ACQUIRE: STAR AND HORIZON VOICE "MARK" WHEN IMAGES ALIGNED. ANNOUNCE ANGLE AND CHRONOGRAPH TIME OF SIGHTING REPEAT SIGHTING AS OFTEN AS POSSIBLE DURING NIGHT PASS VOICE GMT AND SEXTANT TEMP STOW EXPERIMENT HARDWARE		
									CREWMAN
									// T002 PAD UP 09.03.01-00// PAD UPDATE
0:50									

SCHEDULE CRITERIA

- Do not schedule concurrently with S063-1 since both experiments use the wardrobe window.
- Experiment to be performed on dark side of orbit and external lights near the wardrobe window will be off.
- Wardroom interior lights will be dimmed during experiment performance.
- Waste dumps will not be performed during experiment performance.
- The night earth horizon must be moonlit and visible from the wardrobe window.
- One ATM recorder required.

CONSTRAINTS

- Maximum allowable spacecraft rates will be nominal solar inertial attitude rates.

RETURN WEIGHT

- N/A.

T002 MANUAL NAVIGATION SIGHTINGS
 FO-6 PERFORM THREE PERIODS OF BOTH ORBIT AND MIDCOURSE TYPE NAVIGATION SIGHTINGS
 CONSISTING OF SEXTANT MEASUREMENTS ON THE EARTH'S HORIZON AND TWO KNOWN
 STARS AND STADIMETER MEASUREMENTS

ET	12	3	V	*6	CREWMAN	*6	CREWMAN
0:00				33	// T002 ST OPE 09.03.05-00//		
					T002 -6 UNSTOW EXPERIMENT HARDWARE SUBDUE LIGHTING		
					PERFORM 3 STADIMETER SIGHT- INGS, 15 MINUTES APART ON THE EARTH'S HORIZON		
					VOICE "MARK" & START STOP- WATCH WHEN IMAGES ALIGNED		
					ANNOUNCE READING		
					WHEN CONVENIENT, STOP STOPWATCH AND RECORD TIME FROM CHRONOGRAPH		
					STOW EXPERIMENT HARDWARE		
1:05							

SCHEDULE CRITERIA

- First stadimeter sighting upon entering day portion of orbit.
- Do not schedule concurrently with S063-1 since both experiments use the wardroom window.
- Sextant sightings to be performed on dark side of orbit and external lights near the wardroom window will be off.
- Wardroom interior lights will be dimmed during sextant sightings.
- Waste dumps will not be performed during experiment performance.
- For the sextant sightings the earth's horizon should be moonlit and visible from the wardroom window.
- One ATM recorder required.

CONSTRAINTS

- Maximum allowable spacecraft rates will be nominal solar inertial attitude rates.

RETURN WEIGHT

- N/A.

T003 IN-FLIGHT AEROSOL ANALYSIS		MRD		11/1/71	
FO-1 INITIAL MEASUREMENTS AFTER OWS ACTIVATION		EOH VOL I		12/28/71	
		EOH VOL II		6/7/71	
		ODB		1/6/72	
		AE		1/3/72	
ET	1 2 3 V	*6	*6		
0:00		15	//	T003	FO-1
				T003-1	AS LISTED
				RECORD AA MEASUREMENTS AT: <ul style="list-style-type: none"> • CS-E, NEAR CEILING • CS-E, NEAR AIR DIFFUSER • CS-W • CS-H • CS-F • CS-D, CENTER OF AM/OWS HATCH • CS-CM, CENTER COUCH 	
0:30					
<p>SCHEDULE CRITERIA</p> <ul style="list-style-type: none"> • Schedule the complete performance as soon as possible after OWS activation, but not later than 5 days. • Schedule just the measurement at CS-E near the ceiling every 8 ±2 hours. • Schedule the removal and stowage of the filter unit only after the last reading has been taken in the mission. <p>CONSTRAINTS</p> <ul style="list-style-type: none"> • During the operation the crewman must keep his movements to a minimum. • During measurements, the air inlet of the analyzer will be pointed perpendicular to the spacecraft longitudinal axis, and away from the couch pad in the CM. <p>RETURN WEIGHT</p> <ul style="list-style-type: none"> • Filter/impact unit <u>0.25</u> pounds. • Filter/impact unit container <u>0.25</u> pounds. • Data cards <u>TBD</u> pounds. 					

T003 IN-FLIGHT AEROSOL ANALYSIS

F0-2 OBTAIN MEASUREMENTS EVERY 10 DAYS AFTER F0-1 PERFORMANCE

ET	AM	1	2	3	V	*6	*6
0:00						4	//T003 MEAS 2 10.07.02-00//
0:08							<p>T003-2 AS LISTED</p> <p>MEASURE AT: CS-E CS-D CS-CM</p>
<p><u>SCHEDULE CRITERIA</u></p> <ul style="list-style-type: none"> ● Schedule performance every 10 days after initial measurements (F0-1). ● Schedule immediately after the regularly scheduled 8 ± 2 hours reading at CS-E (near the ceiling). <p><u>CONSTRAINTS</u></p> <ul style="list-style-type: none"> ● During the operation the crewman must keep his movements to a minimum. ● During measurements, the air inlet of the analyzer will be pointed perpendicular to the spacecraft longitudinal axis, and away from the couch pad in the CM. <p><u>RETURN WEIGHT</u></p> <ul style="list-style-type: none"> ● See F0-1 page 4-30. 							

T003 IN-FLIGHT AEROSOL ANALYSIS
 FO-3 OBTAIN MEASUREMENT EVERY 10 DAYS AFTER FO-1 PERFORMANCE

ET	AM		CREWMAN	*δ	*δ
	1	2			
0:00			T003 // T003 FO-3		
			T003-3 AS LISTED PERFORM AA MEASUREMENT		
			<ul style="list-style-type: none"> ● CS-W ● CS-H ● CS-F 		
0:24					

SCHEDULE CRITERIA

- Schedule performances every 10 days after initial measurements (FO-1).
- Schedule analyzer measurements before and after a meal during each performance (CS-W).
- Schedule analyzer measurements before a period of sanitary facility use and after a weighing of the wet fecal sample bag (M071) during each performance (CS-H).
- Schedule analyzer measurements at CS-F immediately after change and/or suit donning when convenient, during each performance.

CONSTRAINTS

- During the operation the crewman must keep his movements to a minimum.
- During measurements, the air inlet of the analyzer will be pointed perpendicular to the spacecraft longitudinal axis, and away from the couch pad in the CM.

RETURN WEIGHT

- See FO-1 page 4-30.

T003 IN-FLIGHT AEROSOL ANALYSIS
 FO-4 OBTAIN UP TO 20 MEASUREMENTS AT LOCATIONS OF SUSPECT
 PARTICULATE GENERATION

AM		ET	1	2	3	V	*5	CREWMAN	*6
0:00							4	// T003 MEAS 8 10.07.06-00// T003-4 AS LISTED	<p>SCHEDULE CRITERIA</p> <ul style="list-style-type: none"> Performances will be at astronaut's discretion at locations which may be a source of particulate generation. <p>CONSTRAINTS</p> <ul style="list-style-type: none"> During the operation the crewman must keep his movements to a minimum. During measurements, the air inlet of the analyzer will be pointed perpendicular to the spacecraft longitudinal axis, and away from the couch pad in the CM. <p>RETURN WEIGHT</p> <ul style="list-style-type: none"> See FO-1 page 4-30.
0:09							2	RECORD AA MEASUREMENT AT: <ul style="list-style-type: none"> CREW SELECTED LOCATION 	
0:00								// T003 DEACTI 14.05.04-00//	
0:04								T003-4 STOW CS-E REMOVE AND STOW FILTER	

T013 CREW VEHICLE DISTURBANCES

FO-1 PERFORM VARIOUS BODY MOVEMENTS IN OWS USING A LIMB MOTION SENSOR (LIMS) AND FORCE MEASURING UNITS

MRD 4/1/72
 EOH VOL. I 11/19/71
 EOH VOL. II 6/7/71
 ODB 1/26/72
 AE 1/3/72

ET	AM	1	2	3	V	*6	*6		
0:00						5	//T013 PRE OPE 04.06.02-00// T013-1 CS-F EQUIPMENT CHECKOUT		
0:10						24	// T013 OPERAT // 24 T013-1 SUBJECT CS-F PERF # 1 OF 1 SET UP DACS DON LIMS ASSEMBLY UNCAGE FMU SAND CALIBRATE		
0:00							//T013 ASSIST // T013-1 OBSERVER CS-F CONNECT LIMS DATA CABLE ACTIVATE VOICE AND DATA RECORDING SYSTEMS ASSIST SUBJECT START CAMERAS VERIFY GROUND TRACKING CONTACT ANNOUNCE EACH STEP IN EXPERIMENT PERFORMANCE		
0:47							CAGE FMU'S DOFF AND STOW LIMS ASSY STOW CAMERAS AND FILM		
							PHOTO PAD		

SCHEDULE CRITERIA

- Schedule during daylight portion of orbit with the acquisition sun sensors of the ATM/APCS active.
- The console simulation part of the experiment has to be conducted over a ground tracking station for ATM PCS real-time data transmission. At least 5 min. contact required.
- Spacecraft should be in Solar Inertial attitude.
- All equipment that may obstruct the experiment performance must be cleared from the surrounding area. This requirement prohibits scheduling experiments using the A-SOL SAL during T013 performance.
- Do not schedule experiment performance during a CMG desaturation maneuver.
- One ATM recorder is required.

CONSTRAINTS

- Observer and third crewman to remain motionless while the operator is performing the experiment tasks.

RETURN WEIGHT

- Film - 3.0 pounds.

NOTES

- None

T020 FOOT CONTROLLED MANEUVERING UNIT
 FO-1 PERFORM VARIOUS MANEUVERS (MODE 1) WHILE FLYING THE FOOT
 CONTROLLED MANEUVERING UNIT (FCMU) IN SHIRTSLEEVES

ET	A.M.	CREWMAN	*8	CREWMAN
	1230	CREWMAN	*8	CREWMAN
		RUN NO 2: PERFORM PRESCRIBED MANEUVERS		
		VOICE RECORD COMMENTS ON RUN		
		RUN NO 3: PERFORM PRESCRIBED MANEUVERS VOICE RECORD COMMENTS ON RUN		PHOTO PAD (3)
		DOCK TO MOUNTING FIXTURE AND BLEED THRUSTERS DISMOUNT FCMU AND REMOVE EAR PROTECTORS, SHOE PLATES AND HARD HAT		ASSIST SUBJECT TO DOCK AND CLOSE PSS VALVE INSTALL SHOE PLATES ON FOOT CONTROLLERS
		REMOVE PSS TANK FROM BACK- PACK AND GIVE INITIAL CHARGE AT AM RECHARGE STATION. REPLACE IN PSS STORAGE RACK AND ALLOW TBS HRS TO COOL		
		REMOVE BATTERY FROM BACK- PACK AND STOW IN M509 DOWNING STATION		
		STOW HANDLEBARS, BACKPACK AND SHOULDER RESTRAINTS REMOVE AND STOW PHOTOGRAPH- IC EQUIPMENT		
		WAIT TBS HRS FOR PSS TANK TO COOL		

2:25

T020 FOOT CONTROLLED MANEUVERING UNIT
 FO-1 PERFORM VARIOUS MANEUVERS (MODE) WHILE FLYING THE FOOT
 CONTROLLED MANEUVERING UNIT (FCMU) IN SHIRTSLEEVES

ET	1	2	3	V	*6	CREWMAN	*6	CREWMAN
0:00					5	// T020 TOPOFF	//	
0:10						T020-1 GIVE PSS TANK TOPPING OFF CHARGE AT AM STATION THEN REPLACE IN STORAGE RACK		

T020 FOOT CONTROLLED MANEUVERING UNIT
 FO-2 PERFORM VARIOUS MANEUVERS (MODE II) WHILE FLYING THE FOOT
 CONTROLLED MANEUVERING UNIT (FCMU) SUITED

MRD 4/1/72
 EOH VOL I 11/19/71
 EOH VOL II 6/7/71
 ODB 1/17/72
 AE 1/3/72

ET	AM	1	2	3	V	*6	CREWMAN	*6	CREWMAN	SCHEDULE CRITERIA
0:00						75	// T020 FO 2 OP	// 75	// T020 FO 2 AS	<p>The following experiments must not be performed concurrently with T020 because of the experiment flight envelope - S019, S020, S063, S073, S149, S183, M172, M509, T003, T013, T025, and T027.</p> <p>Performances of M509 or T020 must be scheduled at least 5 days apart to maintain OWS atmosphere.</p> <p>Two crewman will each complete one performance of FO-1 (MODE I) and FO-2 (MODE II).</p> <p><u>CONSTRAINTS</u></p> <ul style="list-style-type: none"> Maximum allowable spacecraft rate about any axis (during rim) will be 6 degrees/minutes. Maximum acceleration (during rim) will be 1.5×10^{-3} g along any axis and 1×10^{-3} degrees/second-squared. <p><u>RETURN WEIGHT</u></p> <ul style="list-style-type: none"> Film TBD pounds. <p><u>NOTES</u></p> <ul style="list-style-type: none"> ATI crewman should wear ear protectors during performance of experiment.
							T020 -2 SUBJECT PERF # OF 2		T020-2 OBSERVER	<p>CS-F</p> <p>CLEAR EXPERIMENT AREA AND SET UP PHOTOGRAPHIC EQUIP</p> <p>ADJUST FCMU AND CONNECT PSS AND POWER UMBILICAL TO BACKPACK</p> <p>INSTALL DAC IN FCMU</p> <p>INSTALL BATTERY IN BACKPACK</p> <p>REMOVE PSS FROM PSS RACK AND INSTALL IN BACKPACK</p> <p>ASSIST SUBJECT TO DON SUIT AND SHOE PLATES</p> <p>DON EAR PROTECTORS</p> <p>VERIFY PROPER THRUSTER FIRING RESPONSE AND CAMERA OPERATION</p> <p>POSITION SUBJECT IN CENTER OF OWS AND ADJUST FCMU LENGTH</p> <p>PHOTOGRAPH MANEUVERS</p>
							UNSTOW EVA EQUIPMENT			
							DON PGA, AND CONNECT LSU TO PCU			
							DON SHOE PLATES AND MOUNT FCMU			
							LOCK SHOE PLATES IN POSITION AND ADJUST HARNESS STRAPS			
							PRESSURIZE SUIT			
							COMMAND THRUSTER FIRINGS AND CAMERA OPERATION TO VERIFY RESPONSE			
							RELEASE FCMU AND MOVE UP OUT OF MOUNTING FIXTURE			
							RUN NO 4: PERFORM PRESCRIBED MANEUVERS			

T020 FOOT CONTROLLED MANEUVERING UNIT
 FO-2 PERFORM VARIOUS MANEUVERS (MODE II) WHILE FLYING THE FOOT
 CONTROLLED MANEUVERING UNIT (FCMU) SUITED

ET	12	3	V	*6	CREWMAN	*6	CREWMAN
					VOICE RECORD COMMENTS ON RUN		OBSERVE SUBJECT AND ASSIST WHEN REQUIRED
					RUN NO 5: PERFORM PRESCRIBED MANEUVERS		
					VOICE RECORD COMMENTS ON RUN		
					DOCK TO MOUNTING FIXTURE AND BLEED THRUSTERS		ASSIST SUBJECT TO DOCK AND CLOSE PSS VALVE
					DEPRESSURIZE PGA		DOFF EAR PROTECTORS INSTALL SHOE PLATES ON FOOT CONTROLLERS
					DISMOUNT FCMU AND REMOVE SHOE PLATES		REMOVE PSS TANK FROM BACK-PACK AND GIVE INITIAL CHARGE AT AM RECHARGE STATION. REPLACE IN PSS STORAGE RACK AND ALLOW TBS HRS TO COOL
					DISCONNECT LSU FROM PCU AND DOFF PGA		REMOVE BATTERY FROM BACK-PACK AND STOW IN M509 DONNING STATION
					STOW EVA EQUIPMENT AND COMMENCE SUIT DRYING		STOW HANDLEBARS, BACKPACK AND SHOULDER RESTRAINTS REMOVE AND STOW PHOTOGRAPHIC EQUIPMENT

2:30

T020 FOOT CONTROLLED MANEUVERING UNIT
 FO-2 PERFORM VARIOUS MANEUVERS (MODE II) WHILE FLYING THE FOOT
 CONTROLLED MANEUVERING UNIT (FCMU) SUITED

ET	AM	1	2	3	V	*6	*6	CREWMAN	*6	CREWMAN
0:00							5		5	WAIT TBS HRS FOR PSS TANK TO COOL
0:10									//	T020 TOPOFF
0:00							5	// T020 FILMST		T020-2 CS-F GIVE PSS TANK TOPPING OFF CHARGE AT AM STATION THEN REPLACE IN STORAGE RACK
0:10								T020-2 STOW	CS-CM	PHOTO PAD (3)
								STOW FILM AND LOG IN CM FOR RETURN		

T025 PARTICULATE CORONAMETER		CREWMAN		CREWMAN		CONSTRAINTS
ET	AM	123V	*6	*6		
						<ul style="list-style-type: none"> ● Inhibit controlled vents & waste dumps prior to conducting the first photo sequence. ● Position the OA to align the photometer to within $\pm 0.5^\circ$ of the center of the solar disc. ● OA drift rates not to exceed 0.012 deg/sec ● Roll rates about the major exp. pointing axis to remain less than 0.012 deg/sec. ● Max. film temp. 85°F. ● Max. film radiation TBD. ● Film to be stowed in film vault during non-operating periods. ● Camera settings: <ul style="list-style-type: none"> ● Film magazine may be kept on camera if two photo taking orbits are scheduled within _____ hrs of each other. ● The occulting disc is to be retracted into the SAL during non-operating periods. <p>RETURN WEIGHT</p> <ul style="list-style-type: none"> ● Film 3.5 pounds.
1:05						
0:00						
0:05						
0:05						
0:00						
0:10						

T025 PARTICULATE CORONAMETER
 FO-1 PHOTOGRAPH PARTICULATE MATTER SURROUNDING THE GA
 FO-2 PHOTOGRAPH PARTICULATE MATTER IN THE UPPER ATMOSPHERE
 FO-3 F-CORONA PHOTOGRAPHY

ET	AM	1	2	3	V	*6	CREWMAN	*6	CREWMAN
0:00						08	// T025 STOW E 11.07.07-00// T025-X STOW EQUIP REMOVE CAMERA, FILTERS & FILM. STOW REMOVE BOOM ASSY & STOW REMOVE CORONAGRAPH FROM SAL & STOW		
0:16							SECURE SAL		
0:00						02	//T026 TRANSFER //		
0:05							T025 TRANS FILM CSM TRANSFER FILM TO CSM		
									T025 PHOTO PAD

T027 CONTAMINATION MEASUREMENT
 FO-1 DEPLOY, EXPOSE, RETRIEVE & RETURN SAMPLE ARRAY

ET	A.M.	12	3	V	*6	CREWMAN	*6
1:00						STOW SAS IN CONTAINER EVACUATE (5 MIN) SECURE SAL AREA	
0:00						05// T027 TRANSFER //	
0:10						T027-1 TRANSFER EQUIPMENT REMOVE SAS FROM OWS STOWAGE CONTAINER & STOW IN CM	

SKYLAB PHOTOGRAPHY (M151, M487, M516, OPS)

EXP	FO	ACTIVITY			SOURCE		PERFORMANCES			FILMING TIME PER PERF (MIN)	FRAME RATE (FPS)	FILM FOOTAGE (ONE PERF)	FILM TYPE	CAMERA TYPE	FILM FOOTAGE (ALL PERFS)
		PRIME	ALT	CONT	MEALS (M509)	MEALS (M509)	MEALS (M509)	MEALS (M509)	MEALS (M509)						
M151	1	ENCLOSURE; DOWNING OF METABOLIC BELT; TRANSLATION TO AND FROM AND INGRESS AND EGRESS OF CONFINED ENCLOSURE; MOUNTING, APPLYING RESTRAINTS, AND OPERATING ENGOMETER.	M092/M171	1. M092	TBD	8 PERFORMANCES OF M092 OF AT LEAST 4 TIMES EACH AND 2 CREWMEN 4 TIMES EACH AND M171 METABOLIC ACTIVITY BY ONE OF THE ABOVE CREWMEN.	M092-29	M171-15	6		M092-261	M171-135	16 MM	M092 2086 M171 540	
	2	HARDWARE DEPLOY AND RETRIEVAL OF LARGE	1027 OR	1. 5149	TBD	(1) DEPLOY AT SOLAR SAL (1) DEPLOY AT ANTI-SOLAR SAL (1) RETRIEVE SOLAR SAL (1) RETRIEVE ANTI-SOLAR SAL	43.0	25.0	6		387	225	16 MM	784 450	
	3	DEPLOY AND RETRIEVAL OF MEDIUM SIZE HARDWARE.	1027 (SAMPLE ARRAY)	1. 5020 2. 5025 3. 5019	TBD	(1) DEPLOY (ANY SAL) (1) RETRIEVE (ANY SAL)	20.0	360	6		180	360	16 MM	405 180	
	4	DONNING AND DOFFING OF THE PRESSURE GARMENT ASSEMBLY.	PRE-EVA & POST-EVA	1. M509 (PRE- AND POST-SUITED)	TBD	ONE PERFORMANCE OF 2 CREWMEN DONNING AND DOFFING PGA.	43.0	405	6		405	405	16 MM	405 405	
	5	PERIODIC MAINTENANCE OF HARDWARE WHICH REQUIRES REMOVAL & INSTALLATION OF ASSEMBLIES AND DOWNING OF SUCH HARDWARE WHERE APPLICABLE.	M509	1. 1020 (PREP)	TBD	3 PERFORMANCES OF AT LEAST	18.0	162	6		162	486	16 MM	486	
	6	FOOD PREPARATION AND FOOD RESIDUE MASS MEASUREMENT	MEALS (OWSI) MEALS (CSM)	MEALS (M509) MEALS (CSM)	MEALS (M509) MEALS (CSM)	4 PERFORMANCES OF AT LEAST 2 CREWMEN WORKING SIMULTANEOUSLY	13.0	117	6		117	468	16 MM	468	
	M487	1	DINING (FOOD AND WATER, MOBILITY AND RESTRAINTS, ARCHITECTURE, HOUSEKEEPING)			PERFORM 3 TIMES DURING THE MISSION ON DAYS 12, 24, AND TBD.	6	54	6		54	162	16 MM	162	
		2	SLEEP PREP. (DOFFING GARMENTS AND INGRESS SLEEP RESTRAINTS)			PERFORM 3 TIMES DURING THE MISSION ON DAYS 12, 24, AND TBD.	2	18	6		18	54	16 MM	54	
		3	EGRESS SLEEP RESTRAINTS AND DON GARMENTS			PERFORM 3 TIMES DURING THE MISSION ON DAYS 13, 25, AND TBD.	2	18	6		18	54	16 MM	54	
		4	CLEAN OWS AIR MIXING SCREEN			PERFORM 3 TIMES DURING THE MISSION ON DAYS 7, 14, AND 21	3	27	6		27	81	16 MM	81	
5		STILL PHOTOGRAPHY OF ALL CREW AREAS			PERFORM 3 TIMES DURING THE MISSION ON DAYS 14, 26, AND TBD.	1	3	2		1	3	16 MM	3		
M516		1	RECONFIGURE FOOD STORAGE CONTAINERS TO ON ORBIT LOCATION			PERFORM TWICE ON MISSION DAY 2 AT TWO DIFFERENT LOCATIONS	3	27	6		27	81	16 MM	81	
		2	1027 HEAD REMOVAL, ASSEMBLE AND MOUNT 5149, INSTALL IN SAL			PERFORM ONCE ON MISSION DAY 13	7	63	6		63	189	16 MM	189	
		3	CONFIGURE WHOLE BODY SHOWER			TBD	185	185	6		185	555	16 MM	555	
	4	M509 REMOVAL OF SHOCK MOUNTS, BATTI RESTRAINTS DUST COVERS, INSTALLATION OF ASMU IN OPS, POS, IN PADDLES			PERFORM ON MISSION DAY 5	15	135	6		135	405	16 MM	405		
	5A	REMOVE FILM CASSETTES FROM 5504 MAGAZINE			PERFORM ON MISSION DAY 26 (POST-EVA)	8	72	6		72	216	16 MM	216		
	5B	REMOVE 5506 SHOE FROM M143 STORAGE LOCKER			PERFORM ON MISSION DAY 26	2	18	6		18	54	16 MM	54		
	6	UNSCHEDULED MAINTENANCE			TBD	2	18	6		18	54	16 MM	54		
	1	SWS PRIOR TO AND DURING CSM DOCKING			PHOTO DOCKING TARGETS, DOCKING LIGHTS, FLOATING DEBRIS DURING DAYLIGHT	5	45 FT (16 MM)	6		45	135	16 MM	135		
OPS	2	SWS AFTER UNDOCKING			PHOTO EREP (M509), OWS AFT SKIRT, EVA AREA (AM), EVA/EXP, PREA (ATM) DURING DAYLIGHT	NA	NA	NA		NA	30 FRAMES	5068	20 MM	30 FR	
	20.8														

20.10 ENVIRONMENTAL MICROBIOLOGY

FO-1 OBTAIN INFLIGHT MICROBIOLOGICAL SAMPLES FROM THE SWS HARDWARE

ET	AM	12	3	V	#6	CREWMAN	#6
0:00						<p>/// TBD ///</p> <p>20.10-1 CS-AS LISTED</p> <p>TAKE SWAB SAMPLE AT <u>TBD</u> SITE</p> <p>PLACE SWAB SAMPLE IN TRANSPORT MEDIA TUBE AND REPLACE CAP</p> <p>REPEAT SAMPLE TAKING AT 15 DESIGNATED SITES</p> <p>STOW TRANSPORT MEDIA TUBES IN FOOD CHILLER</p>	
0:30							
<p>SCHEDULE CRITERIA</p> <ul style="list-style-type: none"> ● Schedule 16 days prior to the end of the mission and again at the end of the mission. ● Schedule sample taking shortly after high crew activity periods on the days specified. ● Schedule the transport media tubes to be placed in the CM resupply and return system of the Inflight Medical Support System (IMSS) within 6 hours prior to separation. <p>RETURN WEIGHT</p> <ul style="list-style-type: none"> ● Sample media tubes <u>TBD</u> lbs. 							

20.10 ENVIRONMENTAL MICROBIOLOGY
 FO-2 OBTAIN INFLIGHT MICROBIOLOGICAL SAMPLES FROM
 THE CREW

ET	AM	12	3	V	*6	CREWMAN	*6
0:00						///	
0:12						TBD CS-TBD TAKE 4 BODY SWAB SAMPLES AND PLACE IN TRANSPORT MEDIA TUBES STOW TRANSPORT MEDIA TUBES IN FOOD CHILLER	
<p><u>SCHEDULE CRITERIA</u></p> <ul style="list-style-type: none"> ● Schedule 16 days prior to the end of the mission. ● Schedule shortly after crew sleep period. ● Four body samples are to be taken from each crewman. ● Schedule the transport media tubes to be placed in the CM resupply and return system of the Inflight Medical Support System (IMSS) within 6 hours prior to separation. <p><u>RETURN WEIGHT</u></p> <ul style="list-style-type: none"> ● Sample media tubes TBD lbs. 							

20.10 ENVIRONMENTAL MICROBIOLOGY

F0-3 OBTAIN INFLIGHT MICROBIOLOGICAL SAMPLES OF THE SMS AIR

ET	AM	12	3	V	*6	CREWMAN	*6
0:00						/// TBD 20.10-3 ATTACH AIR SAMPLE RECEPTACLE TO VACUUM SOURCE AT TBD OWS SITE OBTAIN AIR SAMPLE AT 3 TBD FLOW RATES CHANGE AIR SAMPLE RECEPTACLE AT MDA SITE OBTAIN AIR SAMPLE AT 3 TBD FLOW RATES STOW AIR SAMPLE RECEPTACLES IN FOOD CHILLER	<p><u>SCHEDULE CRITERIA</u></p> <ul style="list-style-type: none"> o Schedule performance at close of mission. o Do not schedule performance when the OWS vacuum cleaner is being used for general cleaning or for M512 and M479 chamber cleaning. o Schedule the air samples to be placed in the CM resupply and return system of the Inflight Medical Support System (IMSS) within 6 hours prior to separation. <p><u>RETURN WEIGHT</u></p> <ul style="list-style-type: none"> o Two air sample receptacles <u>TBD</u> lbs.
0:30							